

EXPERIMENTAL HEAT TRANSFER COEFFICIENTS  
FOR THE COOLING OF OIL  
IN HORIZONTAL INTERNAL FORCED CONVECTIVE TRANSITIONAL FLOW

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A dissertation submitted to the Faculty of Engineering,  
University of the Witwatersrand, Johannesburg  
for the Degree of Master of Science in Chemical Engineering.

Johannesburg, 1981

DECLARATION

I declare that this dissertation is my own, unaided work. It is being submitted for the degree of Master of Science in Chemical Engineering at the University of the Witwatersrand, Johannesburg. It has not been submitted before for any degree or examination at any other University.

SIGNED this 3<sup>rd</sup> day of June in the year 1981.

  
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S Y N O P S I S

The heat transfer coefficients for the cooling of a Newtonian oil were determined for the ranges  $550 < Re < 2800$  and  $132 < Pr < 642$ . The Reynolds number alone proved to be an insufficient criterion for the division of transfer regimes, in line with the view of Eckert. In each of three of the transfer regimes recognised heat transfer was correlated by a Hausen type equation

$$Nu = A(Re^{0.8} - B)(1.8Pr^{1/3} - 0.8) \left[ 1 + (D/L)^{2/3} \right] \left[ \frac{\mu}{\mu_s} \right]^{0.14}$$

The values found for A and B were 0,0184 and 213,9 for the mixed turbulent regime, 0,0277 and 272,5 for the upper transitional regime, 0,0176 and 106,1 for the middle transitional regime, while in the mixed laminar regime the correlating equation was  $Nu = 0,0002 Re^{1.42} Pr^{1/3} \left[ \frac{\mu}{\mu_s} \right]^{0.14}$

In all the regimes the Stanton number increases with Reynolds number, while in true turbulent or laminar flow it decreases. In the lower transitional regime, characterised by low wall Reynolds numbers, the Stanton number fluctuated with Reynolds number and no good correlation was found.

KEYWORDS      Transfer, heat, transitional, flow, pipe, Nu, Re, Pr

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## 1 INTRODUCTION

The flow of fluids is commonly divided into only two main flow patterns, the laminar flow and the turbulent flow regions. In most situations this subdivision is adequate in that the flow is either fully turbulent or fully laminar. However, an intermediate flow situation exists where the fluid flows in a so-called transitional fashion. In this transitional region a further distinction is made as to whether the flow is changing from laminar to turbulent or if the flow is changing from turbulent to laminar; the latter is called reversion, reverse transition or relaminarisation.

### 1.1 OCCURRENCE OF RELAMINARISATION

There are a number of instances where the relaminarisation of fluid flows may occur. These are<sup>1,2</sup>:

*Reversion by dissipation.* This is described as the relative increase in dissipation that occurs when the Reynolds number in a flow decreases. This can occur with the gradual enlargement of a pipe or channel if the angle of divergence is sufficiently small so that flow separation does not occur. It has also been experimentally observed that the decaying turbulent flow is strongly anisotropic under these conditions and that although the skin friction reaches the laminar value, the rest of the flow is still strongly turbulent. Other instances where this reversion may take place are when branching occurs or after a restriction in a pipe, for example an orifice, or in any flow where there is a Reynolds number decrease with time.

*Stably stratified flows.* This is the suppression of turbulence in the presence of a stabilising density gradient and can be observed to occur frequently in the atmosphere. In these instances it is the presence of a lighter fluid on top that causes the rising fluid to work against gravity and so turbulent energy may be converted into gravitational potential energy. This energy absorption leads to reversion of the flow.

*Highly accelerated flows.* Reversion has been observed to occur to a turbulent boundary layer subjected to a large favourable pressure gradient. This phenomenon is mostly limited to external flows.

*Curved flows.* Reversion has been observed during the radial Poiseuille flow between two parallel discs. However, the phenomenon does not appear to occur at a fixed Reynolds number and hence is more complex than may first be imagined.

*Rotation.* Reversion has been observed in a channel rotating about a spanwise axis. Under these circumstances it appears that the Coriolis force is providing the force for stabilising the flow.



*Surface mass transfer.* Experiments show that fully developed pipe flow exhibits some laminar features when a uniform circumferential injection is applied to the flow. It is probable that the injection of fluid is in some way affecting the eddies close to the wall and hence the entire flow.

*Magnetohydrodynamic duct flows.* Experiments have shown that with a sufficiently strong field the skin friction changes from the turbulent value to one characteristic of laminar magnetohydrodynamic flow. The experiments further show that under these circumstances the turbulent fluctuations do not disappear although they do not appear to contribute to momentum transport.

*Thermal effects.* There are two instances where thermal effects may cause reversion

- a. The heating of a gas in internal flow where with increasing temperature the gas density decreases and the velocity increases, therefore causing reversion due to acceleration. It is also possible that it is the increase in kinematic viscosity that is causing the reversion. This cause is seldom encountered in the process industry.
- b. The cooling of a liquid in internal flow where it is likely that the increase in viscosity is causing the reversion, that is, reversion by dissipation. This case was selected for further study.

## 1.2 SPECIFIC PROBLEM SELECTED

The change in flow pattern necessarily changes the heat transfer rate which in turn affects the sizing of equipment to effect a given heat exchange duty. Since the heat transfer rate under laminar flow conditions is much less than the rate under turbulent conditions it is desirable to use turbulent flow conditions in equipment design. However, this cannot always be done, either due to limitations on available pumping power or because of the physical properties of the fluid. Situations may therefore exist for which equipment must be designed with transitional flow occurring

Data and correlations for the heat transfer rate for laminar and turbulent conditions for internal fluid flow are numerous and acceptably accurate, however, a survey has shown<sup>(2)</sup> that there is a dearth of information concerning the transitional region.

The situation of turbulent to laminar transition is more often encountered in industry (cooling of liquids) than the laminar to turbulent transition and it has therefore been decided to investigate the former situation, more specifically the effect of turbulent to laminar transition on the heat transfer of a cooling Newtonian liquid.

\* Reference (2) forms an addendum to this report and serves as a survey of the pertinent literature. It is suggested that the addendum be read at this stage. The addendum starts on page 115.

The boundary conditions chosen for the experiments are:

- i. Internal forced flow of a Newtonian fluid in a horizontal, circular tube
- ii. A uniform temperature of the fluid at the start of the test section
- iii. A uniform wall temperature on the outside of the tube
- iv. A variable length test section to observe the effect of the length to diameter ratio.
- v. Steady state condition.

These conditions have been chosen so as to simulate the situation occurring in a shell and tube exchanger where the tube side heat transfer coefficient limits the design.

## 2 EXPERIMENTAL AND PROCESSING OF RESULTS

The flow diagram of the experimental system is given in Figure 1 (Section 6.15) and may be summarised as follows:



There are two loops in the process, the one for the oil flowing on the inside of the test pipe and the other for the cooling water flowing on the outside or annulus. At no time are the fluids in direct contact. There are nine individual test sections, each identical, and fitted together to form one long heat exchanger. Detailed drawings of the test sections and the connecting pieces are given in Figures 2 to 15 (Section 6.15).

In the experiments Regal Oil B has been used as the inside fluid. A description of each process loop is given in Section 6.1.

The methods of flow measurement, temperature measurement and determination of physical properties are given in Sections 6.2, 6.3 and 6.4 respectively.

The quantities measured during the experimentation are the

- a. bulk oil inlet temperature
- b. bulk oil outlet temperature
- c. oil mass flow rate
- d. cooling water temperature.

The range of dimensionless groups that the experiments cover is given in Table 1.

TABLE 1 Range of dimensionless groups covered by the data

Dimensionless group based on arithmetic average properties	Minimum	Maximum
Re	553	2 808
Pr	132,5	642,0
Gr	4 846	59 636
$\frac{\eta_w}{\eta}$	1,04	8,80
L/D	107	321

The experimental apparatus was designed with the aim of measuring the temperature profile with respect to length as the oil flowing in the jacketed pipe is cooled by the water in the jacket. The oil will enter in turbulent flow and as it cools will relaminarise and exit from the test section flowing laminary. The temperature profile obtained will thus indicate how the heat transfer changes as the flow relaminarises.

However, it has proved impossible to measure the bulk temperature of the oil with a platinum resistance thermometer extending over the pipe diameter. The temperature profile in the oil causes the thermometer to measure some average value that is not the mean bulk value as is shown by the results in Section 6.9. It has also not been practicable to use a thermistor and traverse across the diameter to obtain a temperature and velocity profile and thus a bulk temperature, since the time required to do such measurements would require that a constant state condition exists for a long period of time, a requirement that is not usually achieved in relatively small apparatus of the nature used.

... immediately before temperature  
 ... the disturb the flow completely it has only been possible to measure  
 the bulk temperature at the outlet from the test section. The temperature of the oil at the inlet  
 is assumed to be the same as the temperature of the lagged section used and thus this temperature can be measured without a  
 mixing device being used.

Using the experimental results in Section 6.12, the heat transfer coefficient has been  
 calculated from

$$\frac{m_1 \lambda \left\{ \ln \left[ \frac{\theta_{w0}}{\theta_{w1}} + (C_o + b) \ln \left[ \frac{\theta_{w0}}{\theta_{w1}} - \frac{\theta_{w0}}{\theta_{w1}} \right] \right\}}{2\pi \lambda L - m_1 \ln \left\{ b(\theta_{w0} - \theta_{w1}) + (C_o + b) \ln \left[ \frac{\theta_{w0}}{\theta_{w1}} - \frac{\theta_{w0}}{\theta_{w1}} \right] \right\}} \quad (2.1)$$

The calculation has been performed by computer using the programs  
 described in Section 6.6.

#### ERROR ANALYSIS

The effects of both random and systematic errors for the experimentally determined  
 heat transfer coefficient are given in Section 6.7, and an expected error of  $E(h_1)/h_1 \leq 0.4\%$ . The  
 systematic errors are a result of the assumption that

the outside wall temperature is equal to the cooling water temperature. This  
 assumption is used since

- a) the heat flux in each section is unknown, and
- b) the flow rate of cooling water to each section is not known accurately.

It is thus impossible to calculate the outside wall temperature accurately. (Correlations  
 for the outside heat transfer coefficient are typically no more accurate than 10% to 20%.)

The assumption introduces a systematic error into the experimental results which has  
 been shown to be of the order of 2% in the calculation of the heat transfer  
 coefficient and of the order of 5% in the calculation of the Grashof number  
 (See Table 6.7). Since the direction in which the error lies is always the same (under-  
 estimation of the heat transfer coefficient and Grashof number) it can be allowed  
 for in correlating the data.

If however the assumption is not made and the outside wall temperature is  
 calculated using generally accepted correlations for the outside heat transfer coefficient  
 with reasonable assumptions for the heat flux in each section, the error introduced  
 will be random and of unknown magnitude. This is undesirable and hence working  
 with a known magnitude systematic error has been chosen.

The wall temperature is constant throughout the tube. Experimental results (Section 6.10) that there is a temperature rise of  $\theta = 1.1\%$  of the inlet water flowing through each section. This will increase the mean temperature by  $\theta_s \approx 0.3^\circ\text{C}$  which will cause a further systematic error of the order of 0.5% in the calculation of the heat transfer coefficient and of the order of 1% in the Grashof number. This error acts in the same direction as the other systematic error and increases the systematic error in the Grashof number to  $\frac{\Delta Gr}{Gr} \approx 1\%$  and in the Grashof group to  $\frac{\Delta Gr}{Gr} \approx 1\%$ .

3. DISCUSSION

3.1 COMPARISON OF EXPERIMENTAL RESULTS WITH CORRELATIONS FROM THE LITERATURE

Initially the data obtained from the experiments have been plotted using the equation of Hausen<sup>(3)</sup>.

$$Nu = 0.0235(Re^{0.8} - 230)(1.8Pr^{0.4} + 0.8) \left(1 + \frac{L}{D}\right)^{0.14} \quad (3.1)$$

with the result as plotted in Figure 21. From this it has been noted that it is not possible to correlate the data adequately using this form of correlation and relying on the Reynolds number alone to distinguish among the transfer regimes. The criteria established by Metz and Eckert<sup>(4)</sup> have then been applied to the data as indicated in Figure 22, to separate the data into various transfer regimes. This method effectively uses the Grashof, Prandtl and  $L/D$  ratios to characterize the effect of heat transfer on the transition mechanism. However, these subdivisions have proved inadequate and no reasonable correlation has been obtained for the data neither with existing equations in the literature nor with modifications to such equations. Figure 23 shows the familiar Colburn form that is most often used as an example for correlation.

It has therefore been necessary to redefine the limits of the transfer regimes in such a manner that correlation equations can be found for the data in each regime.

### 3.2 REDEFINITION OF TRANSFER REGIMES

Sieder and Tate (1936)<sup>(5)</sup> noted that the Stanton number decreases with increasing Reynolds number in both the laminar and the turbulent regimes. However, in the inbetween region the Stanton number appears to increase with increasing Reynolds number. This criterion has been applied to the data to subdivide the flow regions as given in Figure 24.

It has further been found that a Reynolds number effectively based on the wall viscosity as given by  $Re_{\frac{\mu_w}{\mu_b}}$  is more effective in correlating the regions than a bulk Reynolds number. This may intuitively be expected since the heat transfer is predominantly concerned with the region close to the wall and hence properties based on a wall temperature may be more effective in convection correlations.

The mixed turbulent and mixed laminar regions have been so termed because the Stanton number is still increasing with increasing Reynolds number in these regions, however, they are distinct from the transitional regions.

The transitional region further appears to divide into three distinct subregions: the upper, middle and lower transitional regions. This indicates that the cooling rate is not adequately characterised by the  $L/D$  ratio and the Grashof number, and that a more convenient length scale may exist on which to base dimensionless groups. It is possible that such a length scale may come either from a residence time concept from the view of allowing time and distance for free convection effects to set up, or from an entry length consideration.

### 3.3 CORRELATING EQUATIONS FOR EACH TRANSFER REGIME

With the transfer regions divided as given in Figure 24, correlating equations have been obtained for each regime.

#### 3.3.1 Mixed turbulent regime

In this region the equation of Hausen has been found to be able to be conveniently modified to fit the data. As with correlations tried for all the regions the effects of the Prandtl group, the  $L/D$  ratio and the  $\mu_w/\mu_b$  ratio have been tested without any conclusive evidence being found for cause to change the form presented by Hausen.

The modified equation that has been found to be most suited to the mixed turbulent data is given in Figure 25 as

$$Nu = 0,0184(Re^{0,8} - 213,9)(1,8Pr^{1/3} - 0,8)[1 + (D/L)^{2/3}]^{1/4} \left(\frac{\eta}{\eta_w}\right)^{0,14} \quad (32)$$

The exponent in the Reynolds group has been retained as 0,8 since this is generally accepted as being correct for turbulent flow. Equations tested with other values of the exponent did not prompt a change of the exponent from 0,8.

The correction factor of Gregorig<sup>(6)</sup>,  $\left[\frac{Pr}{Pr_w}\right]^n$ , has been tested as a replacement of the  $\left(\frac{\eta}{\eta_w}\right)^{0,14}$  group but shows no marked improvement in the correlation. Because of the complexity of calculating the Gregorig factor it has been decided to retain the  $\left(\frac{\eta}{\eta_w}\right)^{0,14}$  factor.

Points on the boundaries of the region may be noted as correlating with subsequent correlations. The average data scatter for this correlation is of the order of 6%.

### 3.3.2 Upper transitional regime

As with the mixed turbulent region the Hausen form of the equation has been found to be the form most suitable for correlating the data. This is given in Figure 26 as

$$Nu = 0,0277(Re^{0,8} - 272,5)(1,8Pr^{1/3} - 0,8)[1 + (D/L)^{2/3}]^{1/4} \left(\frac{\eta}{\eta_w}\right)^{0,14} \quad (33)$$

The average data scatter for the correlation is of the order of 10% to 15%.

### 3.3.3 Middle transitional regime

As with the mixed turbulent and the upper transitional regions the Hausen form of the equation has been found to correlate the data most suitably. The equation as given in Figure 27

$$Nu = 0,0176(Re^{0,8} - 106,1)(1,8Pr^{1/3} - 0,8)[1 + (D/L)^{2/3}]^{1/4} \left(\frac{\eta}{\eta_w}\right)^{0,14} \quad (34)$$

The data point 59, has been assumed to be an experimental error since it will not fit in with any form of the equation nor in any other region.

### 3.3.4 Mixed laminar regime

The equation as given in Figure 28 has been found to be most suitable for correlating results in this region.

$$Nu = 0.002 Re^{1.42} Pr^{0.4} \quad (3.5)$$

The exponent of the Reynolds group, 1.42, raises doubt as to whether this region may in fact be termed laminar since it is characteristic of the laminar region when the exponent is  $1/3$ . It is thus possible that there are two regions rather than one in this area of Figure 24. However, the data set is too small to be able to separate the regions. It has thus been decided to accept the correlation.

### 3.3.5 Lower transitional regime

No suitable equation has been found to correlate the data in this region. The Stanton number appears to be random with respect to the Reynolds number as given in Figure 29 and it has thus been impossible even to use the data for scale up. The only group that appears to correlate the data at all is

$$\left[ \frac{Re_T}{Re_w} \right] (L/D)^{0.2} Pr^{0.8} \quad (3.6)$$

as given in Figure 30. It is doubtful whether this is in any way adequate, however.

Since the data in this region are predominantly for short length tubes it is possible that entry effects are causing the data scatter. These data are also characteristic of low mass flow rates used in order to obtain reasonable temperature drops during experimentation and it is possible that the nature of the flow is completely different to that of the other data.

Chronologically these data were obtained shortly before the temperature probe at the oil inlet was found to be faulty and thus experimental error could also be the cause of the data scatter.

## 3.4 COMPARISON OF EQUATION WITH THE HAUSEN EQUATION

In order to show the effect of subdividing the flow as given in Figure 24, a plot has been prepared of the data against the Hausen equation as given in Figure 31.

This shows the value of subdividing the flow into various regions in order to find an accurate correlating equation. However, it also indicates that the subdivisions as chosen here may



not be accurate since there is no definite trend in the equations obtained. It is therefore necessary to obtain additional data so that the regime may be defined more clearly.

#### 4 CONCLUSIONS AND RECOMMENDATIONS

Conclusions resulting from the experimental programme and the subsequent processing of the experimental results are as follows.

##### 1. *Correlations obtained for each transfer regime.*

It has not been possible to determine the flow regime during internal pipe flow when heat transfer is taking place on the basis of the Reynolds group alone. Other factors influencing the transfer have been taken into account using the Graetz, Prandtl and  $L/D$  ratio groups.

With the transfer divided into regimes as given in Figure 24, it has been possible to find equations that correlate each regime except for the lower transitional regime.

$$Nu = 0,184(Re^{0,8} - 213,9)(1,8Pr^{1/3} - 0,8)[1 + (D/L)^{2/3}](\frac{\eta}{\eta_w})^{0,14}$$

Upper transitional: [Equation (3.3)]

$$Nu = 0,0277(Re^{0,8} - 213,9)(1,8Pr^{1/3} - 0,8)[1 + (D/L)^{2/3}](\frac{\eta}{\eta_w})^{0,14}$$

Middle transitional: [Equation (3.4)]

$$Nu = 0,0176(Re^{0,8} - 106,1)(1,8Pr^{1/3} - 0,8)[1 + (D/L)^{2/3}](\frac{\eta}{\eta_w})^{0,14}$$

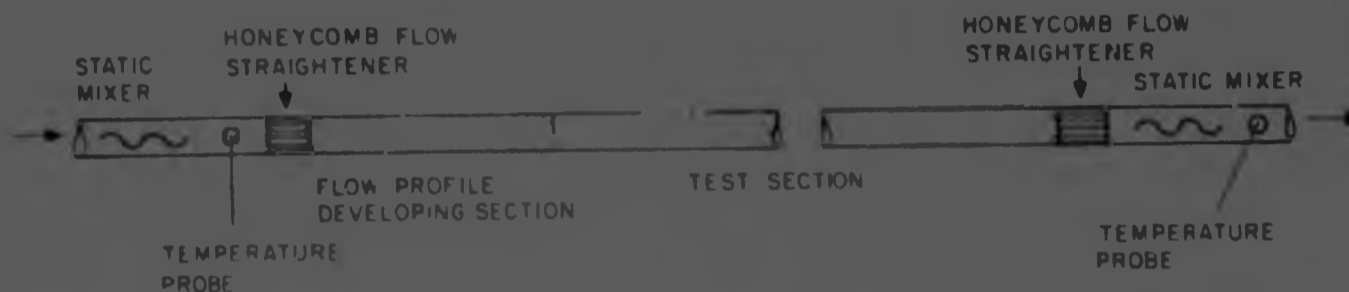
Mixed laminar: [Equation (3.5)]

$$Nu = 0,0002Re^{1,42}Pr^{1/3}(\frac{\eta}{\eta_w})^{0,14}$$

##### 2. *Recommendations for further work.*

Experience gained during this experimental programme has indicated that our understanding of transfer in the transitional region requires considerable additional data. A programme encompassing the following conditions and equipment is proposed

- i. four different tube lengths ranging from 2 to 10 metres,
- ii. straight lengths of tube without joints or internal protrusions to disturb the flow,
- iii. three different tube diameters ranging from 19 to 38 millimetres,
- iv. horizontal, intermediate and vertical inclination of the test section,
- v. a test section constructed as indicated below



- vi. that the cooling medium be water supplied to a multisectioned jacket around the test section,
- vii. a turbine flow meter or other accurate type of flow meter in the test fluid loop,
- viii. that the cooling water flow rate and bulk inlet and outlet temperatures are monitored to give an overall heat balance,
- ix. a fully automated data logging with immediate evaluation of the data and graphical display and using this facility to obtain a complete distribution of data points on a Metais Eckert type plot,
- x. use of two or three different test fluids,
- xi. steady state operation only,
- xii. Reynolds numbers in the range  $500 < Re < 10\,000$ ,
- xiii. turbulent to laminar transition initially.

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## 6 APPENDIX

### 6.1 EXPERIMENTAL

The two process loops as given in Figure 1 are as follows

*Oil side.* Starting at the header tank the oil flows across electrical heaters to a rotary vane pump. This is a Variflo pump from Blackmer Pump Co. that is adjustable to obtain various flow rates. After the pump, the hot oil passes through a heavily insulated calming section. This section is to eliminate swirl introduced by the pump and to allow the fluid to reach thermal equilibrium.

The first temperature probe is at the end of the developing section, as detailed in Figures 9 to 15. These probes and connecting pieces have all been custom made to suit the dimensions of the pipe at each particular joint to prevent any projection or edge triggering turbulence in the flow. This is also why the temperature probes are retractable. To check the system for possible projections, water has been run down the inside pipe at a very high Reynolds number and the pressure drop has been monitored at each connecting piece using a multiple arm manometer as shown in Figure 16. Any surface irregularities are observed as apparent errors in the pressure drop and in this way each section has been corrected for alignment.

The first temperature probe is extended into the fluid at all times to measure the inlet oil temperature. The flat temperature profile at this point ensures that the temperature measured by the resistance thermometer is in fact the mean bulk temperature.

Undisturbed, the oil then flows for a variable number of sections with all the thermometers in the path withdrawn. The length of the test section flow has been varied during the experiments to obtain data on the effect of the length to diameter ratio on the heat transfer.

At the end of the test section the temperature of the fluid is measured after it has passed through a mixing device as detailed in Figure 17. This mixing device is necessary because of the strong temperature gradient in the fluid. Results without a mixing device (Section 6.9) show that the temperature interpreted by the thermometer can be up to 2 °C from the true bulk temperature.

After the final section of heat exchanger, the oil is routed through a rotameter and back to the holding tank.

*Water side.* The water is taken from a holding tank that is kept at constant temperature by continually introducing fresh water from the mains supply and allowing a continual overflow. The water is pumped by a Matheson and Bremner Multiflo pump into two headers, in one of which

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The first temperature probe is extended into the fluid at all times to measure the inlet oil temperature. The flat temperature profile at this point ensures that the temperature measured by the resistance thermometer is in fact the mean bulk temperature.

Undisturbed, the oil then flows for a variable number of sections with all the thermometers in the path withdrawn. The length of the test section flow has been varied during the experiments to obtain data on the effect of the length to diameter ratio on the heat transfer.

At the end of the test section the temperature of the fluid is measured after it has passed through a mixing device as detailed in Figure 17. This mixing device is necessary because of the strong temperature gradient in the fluid. Results without a mixing device (Section 6.9) show that the temperature interpreted by the thermometer can be up to 2% from the true bulk temperature.

After the final section of heat exchanger, the oil is routed through a rotameter and back to the holding tank.

*Water side.* The water is taken from a holding tank that is kept at constant temperature by continually introducing fresh water from the mains supply and allowing a continual overflow. The water is pumped by a Matheson and Bremner Multiflo pump into two headers, in one of which

is a temperature probe as detailed in Figures 18 to 20. This probe gives the bulk temperature of the water supplied to each test section since the temperature of the water in each header is necessarily the same. The flow rate to each header has been determined using orifice plates with an orifice diameter of 20 mm (Section 6.8).

With the headers arranged in this way it is generally accepted that the flow will distribute evenly provided that each section is identical. The temperature of the water exiting from each section has been measured using platinum resistance thermometers and it has been found that the temperature rise is less than 1 °C (Section 6.10). The water is then returned to the holding tank.

## 6.2 FLOW MEASUREMENT

### 6.2.1 Flow rate of water in the annulus

The flow rate of the cooling water to each header has been measured on one occasion and thereafter conditions have been kept constant and the flow rate unaltered for all the experiments.

Orifice plates have been used in each of the lines and the pressure drop across each orifice has been measured using a differential pressure gauge (Section 6.8).

It has been found that the flow rate is essentially constant for small variations in the water temperature and that the flow appears to divide evenly among the nine test sections. The flow rate to each section has been taken as 0.98 kg/s which is the arithmetic average flow rate.

### 6.2.2 Flow rate of oil inside test section

The flow rate of the oil flowing on the inside of the test pipe is determined using a calibrated Fischer and Porter precision bore rotameter. The rotameter has been calibrated with water at 5 °C and using a weighing tank system. The calibration points so obtained have been fitted using a linear least squares regression technique to give the equation for new flow as

$$m_1 = -0.015603 + 0.007762N_R \quad (6.2.1)$$

with a regression coefficient  $r = 0.9989$ .  $N_R$  is the rotameter reading, and the equation is limited to  $15 < N_R / \% < 100$  for an estimated error in the flow rate of  $\frac{E(m_1)}{m_1} \leq 3\%$ .

Correction has been made for different fluids using the equation recommended by Fischer and Porter.

$$m_2^c = \frac{m_1^2 (p_1 - p_2) w_2}{w_1 - p_1 p_2} \quad (5.23)$$

No correction factor is necessary for changes in viscosity.

### 6.3 TEMPERATURE MEASUREMENT

Calibrated 100 ohm platinum resistance thermometers are used to measure temperature. This type of thermometer has a relatively fast response time  $\tau$  where  $\tau < 1$  second, a negligible offset drift with time, and is accepted as the International Practical Temperature Scale standard for temperatures  $-180 < t(^{\circ}\text{C}) < +300$ .

All the resistance thermometers used have been calibrated in an ice bath against a standard 100 ohm platinum resistance thermometer that has been calibrated by the Precise Physical Measurements Group of the CSIR. All the thermometers are thus referred to the same standard state and, therefore, temperature differences are known with great accuracy. The absolute temperatures are known only with respect to the standard thermometer used for the calibration and this is known to have an accuracy  $E(\theta) = 0,10^{\circ}\text{C}$ .

The bridge used for determining the resistance of each thermometer is a Leeds and Northrup 3078 portable precision temperature bridge utilising a four lead, alternating current system of resolution  $0,025^{\circ}\text{C}$ . The reading given by the bridge is in ohms and has to be subsequently processed to give temperatures. The various thermometers at the various points in the system have been selected for measurement using a simple manual switching box.

The temperature is determined from the resistance reading using the method proposed by Leeds and Northrup in the user manual for the bridge. This method is to first obtain a corrected value based on the calibrated value

$$R_t = R_0 \frac{R_t}{R_0} \quad (6.31)$$

where  $0$  refers to the calibration point

Using the corrected value for the resistance, the temperature is then found by linear interpolation in the standard tables for 100 ohm platinum resistance thermometers based on the International Practical Temperature Scale of 1968. The expected error in the temperature is estimated as  $E(\theta) < 0,05^{\circ}\text{C}$ .

$$\frac{R_{100} - R_0}{R_0 - R_{273.15}} = \frac{T - 273.15}{100 - 273.15}$$

(6.2)

TEMPERATURE MEASUREMENT

Calibrated 100 ohm platinum resistance thermometers are used to measure temperature. This type of thermometer has a relatively fast response time  $\tau$  where  $\tau < 1$  second, negligible self-heating with time, and is accepted as the International Practical Temperature Scale standard for temperatures 180 K to 600 K.

All thermometers used must have been calibrated in an ice bath against a standard reference platinum resistance thermometer that has been calibrated by the Precise Physical Measurements Group at the C-17. All the thermometers are thus referred to the same standard and, therefore, temperature differences are known with great accuracy. The absolute temperatures are known only with respect to the standard thermometer used for the calibration and this is known to have an accuracy  $\pm 0.10$  K.

To measure the temperature, the resistance of each thermometer is a Leeds and Northrup portable precision temperature bridge utilizing a four lead alternating current system of resolution 0.025  $^{\circ}$ C. The reading given by the bridge is in ohms and has to be subsequently corrected to give temperature. The various thermometers at the various points in the system can be selected by connecting with a simple manual switching box.

The temperature is determined from the resistance reading using the method proposed by Leeds and Northrup by the use of tables for the purpose. This method is to first obtain a corrected value based on the calibrated value.

$$R = R_0 \left[ 1 + \alpha (T - T_0) \right] \tag{6.3.1}$$

where  $R$  refers to the coil at the point

and, the corrected value for the resistance, the temperature is then found by linear interpolation in the standard table for 100 ohm platinum resistance thermometers based on the International Practical Temperature Scale of 1968. The expected error in the temperature is estimated as  $\pm 0.05$  K.



6.4 PHYSICAL PROPERTIES

No correction is included in the physical properties for effects of pressure since all the experiments have been conducted at ambient pressure using liquid only and the effect of pressure on liquid properties is negligible for small changes in pressure.

The physical properties are thus needed only for the liquid phase and only as a function of temperature. Values for the oil used (R<sub>1</sub> or B) and for water are tabulated in Section 6.14.

Since the physical properties are available only at discrete values of temperature and not in functional form, it is necessary to use interpolatory techniques to find values at arbitrary temperatures. The various interpolation techniques used are described below.

*Viscosity.* The liquid viscosity correlates fairly closely with the Andrade equation<sup>10</sup>

$$\eta = A_0 \exp(B/T) \quad (6.4.1)$$

A plot of  $\ln(\eta)$  as a function of  $\frac{1}{T}$  should, therefore, be linear. This has been noticed not to be quite accurate and a better fit has been obtained using a cubic spline<sup>11</sup> to  $\ln(\eta)$  as a function of  $\frac{1}{T}$ .

The interpolated value is thus obtained from the cubic spline fitted to the data points in the immediate vicinity of the interpolation point.

*Thermal conductivity.* The liquid thermal conductivity follows the form<sup>10</sup>

$$\lambda = \lambda_0 [1 - a(\theta - \theta_0)] \quad (6.4.2)$$

Since  $\theta_0$  is an arbitrary temperature,  $\lambda$  may be assumed to be linear with the centigrade temperature. This has been found to be true and a linear interpolation technique is adopted.

*Specific heat.* The specific heat is generally<sup>110</sup> assumed to be linear with temperature. It has been found that over reasonable variations in temperature this is the case and hence a linear interpolation technique is used.

*Density.* The form

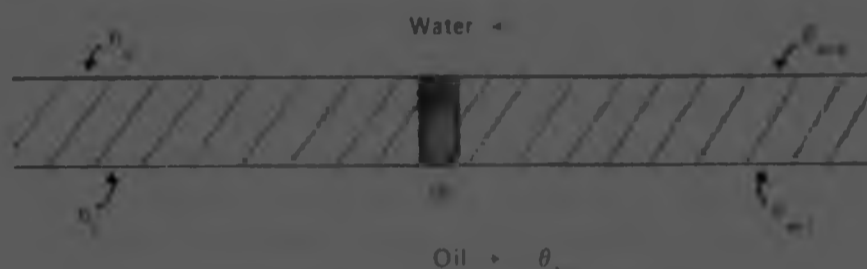
$$\frac{(\rho_k - \rho_v)_2}{(\rho_k - \rho_v)_1} = \left[ \frac{T_1 - T_2}{T_1 - T_4} \right]^{1/3} \quad (6.4.3)$$

is suggested in Perry<sup>110</sup>. This is not usable with the oil properties since the vapour properties and the critical temperature are unknown. The form of the equation suggests that  $\ln(\rho)$  may be

linear; with  $\ln(\theta)$ , however, this is not so and the method of fitting cubic splines to  $p$  versus  $\theta$  is used giving good results. This method is only acceptable for interpolation and not for extrapolation; however, all experiments have been conducted under conditions within the bounds of known.

### 6.5 CALCULATION OF THE HEAT TRANSFER COEFFICIENT

The average heat transfer coefficient over the test section is calculated as follows and the geometry is given in the figure below:



A simple heat balance has been used to give

$$\theta - \theta_{\infty} = dQ \left[ \frac{1}{h_1 2\pi r_1 dl} + \frac{1}{2\pi \lambda_w dl} \right] \quad (6.5.1)$$

where  $dQ$  may be written as

$$dQ = m_1 (C_o + b\theta) d\theta \quad (6.5.2)$$

$$\text{where } C(\theta) = C_o + b\theta \quad (6.5.3)$$

Equation (6.5.2) is substituted in Equation (6.5.1) and integrated from  $\theta = \theta_{b1}$  to  $\theta = \theta_{\infty}$  and from  $l = 0$  to  $l = L$  assuming that:

- i. the thermal conductivity of the wall ( $\lambda_w$ ) is constant. This is reasonable for relatively small variations in temperature,
- ii. the tube dimensions remain constant,
- iii. the outside wall temperature  $\theta_{\infty}$  is constant.

Using separation of variables and solving for  $h_1$  gives:

$$h_1 = \frac{\frac{m_1 \lambda}{r_1} \left[ h_1 \theta_{ax} - \theta_{ax} \right] + (C_p + h_1 r_{w,0}) \ln \left[ \frac{\theta_{b,0} - \theta_{w,0}}{\theta_{b,1} - \theta_{w,0}} \right]}{2H\lambda L - m_1 n \frac{r_2}{r_1} \left[ h_1 \theta_{b,0} - \theta_{b,1} \right] + (C_p + h_1 r_{w,0}) \ln \left[ \frac{\theta_{b,0} - \theta_{w,0}}{\theta_{b,1} - \theta_{w,0}} \right]} \quad (6.54)$$

In this form  $h_1$ , the heat transfer coefficient, is an average value

## 6.6 COMPUTER PROGRAMS

Data accumulated during experimentation is in the form of resistance measurements from the platinum resistance thermometers and of rotameter readings. From this raw data it is necessary to determine flow rates, temperatures, physical properties (dimensionless members) and the heat transfer coefficient. It is not practical to do this by hand for each experiment due to the amount of work necessary to interpolate physical properties. This task is therefore computerised as program 1 in FORTRAN on the CSIR's Control Data Corporation (CDC) Cyber 174. Program 1 reads in rotameter readings and resistances and gives out point values for physical properties and dimensionless groups and gives an average heat transfer coefficient and also arithmetic average dimensionless groups. This data is then in a reasonable format to be used in correlating other data.

In correlating the data a large number of variations need to be considered in the organisation of correlating equations. This is most conveniently done using computer program 2. All the data from program 1 is incorporated as a data input file to program 2. An interactive plotting routine and a Tektronix 4662 digital plotter are coupled to the program to obtain plots of the correlations tried.

Printouts of both programs are given in Section 6.13.

## 6.7 EXPECTED ERRORS

There are two types of errors that may be associated with experimental data, random errors and systematic errors, both forms of which are present in these experimental data. The random errors are associated with instrument accuracy and given in Section 6.7.1 and the systematic errors given in Section 6.7.2 are associated with the assumption of the outside wall temperature of the test sections being equal to the cooling water temperature. A summary of the expected errors is given in Table 2.

TABLE 2 Summary of expected errors

Variable	Random error	Systematic error	Maximum total error
$\bar{\theta}_1/^{\circ}\text{C}$	0,05	0	0,05
$\theta_o/^{\circ}\text{C}$	0,05	0	0,05
$\theta_{wo}/^{\circ}\text{C}$	0,05	-1,3	1,35
$m_1/\%$	3	0	3
$h_1/\%$	3,4	-3	6,4
$Nu/\%$	3,4	-3	6,4
$Re/\%$	3	0	3
$Pr/\%$	0	0	0
$Gr/\%$	0,5	-6	6,5

6.7.1 Random errors

The precision index of a general function,  $F(x)$ , is defined as<sup>(11)</sup>

$$F(x)^2 = \sum_{i=1}^n \left[ \frac{\partial F(x)}{\partial x_i} \right]^2 E(x_i)^2 \quad (6.7.1)$$

Thus if the errors in the variables which make up the function  $F(x)$  are known then the error in  $F(x)$  is also known.

*Heat transfer coefficient.* The heat transfer coefficient is given by Equation (6.5.4). For practical purposes it is assumed that:

- i. the error in the thermal conductivity is zero,
- ii. the error in length measurement is zero,
- iii. the errors in  $C_o$  and  $b$  are zero.

Let

$$A_1 = \frac{m_1 \lambda}{r_1} \left[ b(\theta_{bo} - \theta_{bi}) + (C_o + b\theta_{wo}) \ln \left[ \frac{\theta_{bo} - \theta_{wo}}{\theta_{bi} - \theta_{wo}} \right] \right] \quad (6.7.2)$$

$$\bar{\theta}_1 = \frac{2|A_1|}{m_1 \ln \frac{r_2}{r_1}} \left[ b(\theta_{bo} - \theta_{bi}) + (C_o + b\theta_{wo}) \ln \left[ \frac{\theta_{bo} - \theta_{wo}}{\theta_{bi} - \theta_{wo}} \right] \right] \quad (6.7.3)$$

Then it may easily be shown that

$$\frac{dh}{dm_1} = \frac{2l\lambda^2 bL}{B_1^2 r_1} \left[ (\theta_{bo} - \theta_{bl}) + \left( \frac{C}{b} + \theta_{wo} \right) \ln \left[ \frac{\theta_{bo} - \theta_{wo}}{\theta_{bl} - \theta_{wo}} \right] \right] \quad (6.7.4)$$

$$\frac{dh}{d\theta_{bo}} = \frac{\lambda b m_1}{B_1^2 r_1} \left[ 1 + \left( \frac{C}{b} + \theta_{wo} \right) \right] + \frac{A_1 b \ln \left( \frac{r_2}{r_1} \right) m_1}{B_1^2 r_1} \left[ 1 + \frac{C}{b} + \theta_{wo} \right] \quad (6.7.5)$$

$$\frac{dh}{d\theta_{bl}} = \frac{(-1)\lambda b m_1}{B_1^2 r_1} \left[ 1 + \left( \frac{C}{b} + \theta_{wo} \right) \right] + \frac{A_1 b \ln \left( \frac{r_2}{r_1} \right) m_1}{B_1^2 r_1} \left[ 1 + \frac{C}{b} + \theta_{wo} \right] \quad (6.7.6)$$

$$\frac{dh}{d\theta_{wo}} = \frac{\lambda b m_1}{B_1^2 r_1} \left[ \frac{(\theta_{bo} - \theta_{bl}) \left( \frac{C}{b} + \theta_{wo} \right)}{(\theta_{bo} - \theta_{wo}) (\theta_{bl} - \theta_{wo})} + \ln \left( \frac{\theta_{bo} - \theta_{wo}}{\theta_{bl} - \theta_{wo}} \right) \right] + \frac{m_1 A_1}{B_1^2 r_1} \left[ \frac{(\ln \left( \frac{r_2}{r_1} \right) \left( \frac{C}{b} + \theta_{wo} \right) (\theta_{bo} - \theta_{bl})}{(\theta_{bo} - \theta_{wo}) (\theta_{bl} - \theta_{wo})} + b \ln \left( \frac{r_2}{r_1} \right) \ln \left( \frac{\theta_{bo} - \theta_{wo}}{\theta_{bl} - \theta_{wo}} \right) \right] \quad (6.7.7)$$

The error in h is then

$$E(h)^2 = \left( \frac{dh}{dm_1} \right)^2 E(m_1)^2 + \left( \frac{dh}{d\theta_{bo}} \right)^2 E(\theta_{bo})^2 + \left( \frac{dh}{d\theta_{bl}} \right)^2 E(\theta_{bl})^2 + \left( \frac{dh}{d\theta_{wo}} \right)^2 E(\theta_{wo})^2 \quad (6.7.8)$$

The expected errors in temperature and flow measurement are:

- i. Temperature. The temperatures are measured using calibrated 100 ohm platinum resistance thermometers and a Leeds and Northrup 8078 portable precision temperature bridge. The makers claim a resolution of 0,025 °C under normal operating conditions. However, it is more realistic to double this value. The value for the precision index for temperature,  $E(\theta)$ , is thus 0,05 °C.
- ii. Flow rate. The flow rate is measured using a Fischer and Porter FP-1-35-6-10/80 precision bore rotameter with a GNSVT66 stainless steel float. The rotameter has been calibrated with water at 16 °C using a weighing tank system. The expected error in the range  $15 < N_n \% < 100$  is  $\frac{E(N_n)}{N_n} < 3\%$ .

The correction for a different fluid in the rotameter is performed using the recommended equation of Fischer and Porter

$$N_n^2 = \frac{m_1^2 (\rho_1 - \rho_2) \mu_2}{(\rho_1 - \rho_2) \rho_1} \quad (6.7.9)$$

This correction causes further degradation of accuracy in measurement.

In the experimental apparatus the rotameter is some distance from the last temperature measuring point. The effect of the error in temperature on the density correction factor has been determined as in Table 3.

TABLE 3 Error in flow rate due to temperature

Rotameter reading $N_R$ / %	Fluid temperature $T_1$ / °C	Calculated mass flow rate $m_2$ / kg s <sup>-1</sup>	Error due to temperature / %
88,5	54,16	0,6264	+ 0,09
88,5	56,76	0,6258	0,0
88,5	58,85	0,6254	- 0,06
88,5	59,03	0,6253	- 0,06

It is thus assumed that this error is negligible.

The overall accuracy of flow measurement is thus assumed to be  $\frac{E(m)}{m_1} < 3\%$ .

iii. The average expected error. An arbitrary experimental run selected is

$$\theta_{b1} = 66,96 \text{ } ^\circ\text{C}, \quad \theta_{bo} = 62,50 \text{ } ^\circ\text{C}; \quad \theta_{wo} = 22,56 \text{ } ^\circ\text{C} \quad (\text{run 1})$$

$$m_1 = 0,5055 \text{ kg/s}, \quad L = 8,154 \text{ m} \quad h = 169,9 \text{ W/m}^2\text{K}$$

Fixed constants in the system are

$$\begin{aligned} C &= 1804,32 \text{ J/kgK} \\ b &= 3,56 \text{ J/kgK}^2 \\ \lambda &= 100 \text{ W/mK mean value for brass for } \theta, ^\circ\text{C} = 0 \dots 100 \\ r_1 &= 0,01257 \text{ m} \\ r_h &= 0,01588 \text{ m} \end{aligned}$$

Using these values the following have been calculated

$$E(m_1) = 0,015 \text{ kg/s}$$

$$A_1 = 866 \text{ 259,85 W}^2/\text{m}^2$$

$$B_1 = 5097,83 \text{ WK}$$

$$\begin{aligned} \frac{A_1}{B_1} = h &= 169,927 && \text{W/m}^2\text{K} \\ \frac{dh}{dm_1} &= 337,837 && \text{Ws/m}^2\text{K kg} \\ \frac{dh}{d\theta_{bo}} &= 36,478 && \text{W/m}^2\text{K}^2 \\ \frac{dh}{d\theta_{bl}} &= -39,836 && \text{W/m}^2\text{K}^2 \\ \frac{dh}{d\theta_{wt}} &= 4,6 && \text{W/m}^2\text{K}^2 \end{aligned}$$

Therefore

$$E(h)^2 = 25,68 + 3,327 + 3,967 + 0,041 = 33,051 \text{ W}^2/\text{m}^4\text{K}^2$$

$$\text{and } \frac{E(h)}{h} = 3,4\%$$

Therefore the expected error in the heat transfer coefficient is  $\frac{E(h)}{h} < 3,4\%$ .

Reynolds number. As previously it is assumed that:

- there is no error in length measurement.
- there is no error in physical properties.

Thus

$$\frac{E(Re)}{Re} = 3\% \text{ (the error in } m_1 \text{).}$$

Nusselt number. With the previous assumption this reduces to

$$\frac{E(Nu)}{Nu} = 3,4\%$$

Prandtl number. With the previous assumption, the Pr has no error.

Grazhof number. Using the previous assumptions and assuming no error in  $\beta$  the only error is associated with the temperature.

Using the general form of the error equation

$$\frac{dGr}{d\theta_1} = \frac{\beta g L^3 \rho^2}{\eta^2} \quad (6.7.10)$$

$$\frac{dGr}{d\theta_2} = \frac{-\beta g L^3 \rho^2}{\eta^2} \quad (6.7.11)$$

$$\left[ \frac{E(\text{Gr})}{\text{Gr}} \right]^2 = \frac{E(\theta_1)^2 + E(\theta_2)^2}{\theta_1 - \theta_2} \quad (6.7.12)$$

and

$$\frac{E(\text{Gr})}{\text{Gr}} = 0,5\%, \text{ with } \theta_1 = 25^\circ\text{C}, \theta_2 = 40^\circ\text{C} \text{ (characteristic temperatures).}$$

### 6.7.2 Systematic errors

The only known systematic error that is encountered is the error in the outside wall temperature used in the calculation of the heat transfer coefficient (Nusselt number) and the Grashof number. This error is caused by assuming that the cooling water temperature is the same as the outside wall temperature and that the cooling water temperature is constant over the section. The reasons for these assumptions are given in Section 2. The error introduced has been estimated as follows:

*Calculation of error in the outside wall temperature.* The outside wall to fluid temperature difference is given by

$$\theta_{wo} - \theta_{wa} = \frac{q}{h_o A} \quad (6.7.13)$$

Typical experimental values for the right hand side of Equation (6.7.13) are

$$\begin{aligned} q &\approx 1,5 \text{ KW/section} \\ A &= 0,09 \text{ m}^2/\text{section} \\ h_o &\approx 17 \text{ 000 W/m}^2\text{K} \quad (\text{Section 6.11}) \end{aligned}$$

Using these values

$$\theta_{wo} - \theta_{wa} = 0,98^\circ\text{C}.$$

Also the mean cooling water temperature is expected to be  $0,3^\circ\text{C}$  higher than the inlet temperature (Section 6.10).

Therefore, effectively, using the outside inlet fluid temperature equal to the outside wall temperature introduces an error of  $E(\theta_{wo}) \approx 1,3^\circ\text{C}$ .

*Effect on the Nusselt number of a  $1^\circ\text{C}$  error in the outside wall temperature.* Using two arbitrary different wall temperatures to calculate the Nusselt number and using results from an arbitrary experiment gives for  $\theta_{wo} = 22,56$ ,  $\text{Nu} = 32,8$ , and for  $\theta_{wo} = 21,56$ ,  $\text{Nu} = 32,1$ .



That is, the Nusselt number is underestimated by 7% per degree Celsius of the outside wall temperature.

*Effect on the Grashof number of a 1 °C error in the outside wall temperature.* This is the only other group that was affected by the systematic error. The effect of an error in the outside wall temperature gives for  $\theta_{\text{wall}} = 66,96$ ,  $Gr = 17\,347$ , and for  $\theta_{\text{wall}} = 65,89$ ,  $Gr = 16\,434$ . Therefore the Grashof number is underestimated by approximately 5% per degree Celsius of the outside wall temperature.

#### 68 ORIFICE PLATE RESULTS

These results have been obtained for the flow rate of water to the two inlet headers.

TABLE 4      Header with five sections

Temperature ( $\theta$ )/ °C	$\frac{m}{\text{kg/s}}$
25	4,793
28	4,816
30	4,817
34	4,812
38,5	4,797
43	4,777
44,8	4,777
-----	
$\bar{m} = 4,793/\text{kg s}^{-1}$	
Standard deviation = 0,017/ $\text{kg s}^{-1}$	

TABLE 5 Header with four sections

Temperature ( $\theta$ )/ °C	$\dot{m}$ kg/s
27	4,0895
29	4,057
37	4,084
42,5	4,080
45	3,987
Mean = 4,0595/kg s <sup>-1</sup> Standard deviation = 0,042/kg s <sup>-1</sup>	

The mean flow to each section is thus 0.93 kg/s

#### 6.2 ERROR IN TEMPERATURE MEASUREMENT OF OIL WITHOUT MIXING DIVIDER

The following measurements have been taken under normal operating conditions with

- a a plain thermometer, and
- b a mixing section and thermometer

TABLE 6 Error in temperature measurement

Run	1	2	3	4	5
Plain $\theta_1$ / °C	52,24	49,77	49,85	27,74	25,26
Mixed $\theta_2$ / °C	50,31	47,12	47,20	26,55	24,52
Difference ( $\theta_1 - \theta_2$ ) / °C	1,93	2,65	2,65	1,19	0,76

Run no. 5 has been done under adiabatic conditions and reflects the error in calibration of the thermometers. Taking its calibration error into effect, the average difference in the measured temperature to the true temperature is 1.35 °C

#### 6.10 TEMPERATURE RISE IN WATER ACROSS T'ST SECTIONS

TABLE 7

	Run 1	Run 2	Run 3	Mean rise
Temperature of water $t_1$ /°C	27,25	27,07	26,84	$\theta_o - \theta_1$
Temperature of water $\theta_o$ /°C				
Section 1	28,26	28,08	27,82	1,01
2	27,97	27,77	27,54	0,71
3	27,82	27,64	27,41	0,57
4	27,77	27,59	27,36	0,52
5	28,03	27,87	27,61	0,78
6	27,82	27,64	27,41	0,57
7	27,77	27,61	27,38	0,53
8	27,77	27,59	27,38	0,53
9	27,74	27,56	27,36	0,50
				$m = 0,64$ /°C

#### 6.11 HEAT TRANSFER COEFFICIENT IN THE ANNULUS

The heat transfer coefficient on the outside of the pipe is estimated using the equation of Monrad and Pelton<sup>(10)</sup> for annuli. The equation is

$$Nu = 0,020 Re^{0,8} Pr^{1/3} \left[ \frac{D_2}{D_1} \right]^{0,63} \quad (6.11.1)$$

with  $Re = \frac{4m}{\pi \eta (D_1 + D_2)}$  and  $Nu = \frac{h(D_2 - D_1)}{\lambda}$

In this case the dimensions of the annulus are:

Inside pipe ID = 24,94 mm OD = 31,75 mm  
 Outside pipe ID = 35,74 mm  
 $\frac{D_2}{D_1} = 1,126$ .

The mass flow rate in each section is 0,98 kg/s as given in Section 6.8, and the fluid is water. From this it is possible to determine the heat transfer coefficient as in Table 8.

TABLE 8 Heat transfer coefficient in the annulus

$\frac{D_2}{D_1}$	Re	$\frac{1}{f}$	Nu	$h_o / \text{Wm}^{-2}\text{K}^{-1}$
15	16 274	7,99	99,561	14 846
20	18 451	6,95	105,085	15 881
25	20 773	6,09	110 566	16 931
30	22 197	5,39	120,444	17 960
35	25 749	4,80	121,285	18 998
40	28 399	4,3	126,454	20 029

6.12 EXPERIMENTAL RESULTS

$\frac{\theta_1}{^\circ\text{C}}$	$\frac{\theta_2}{^\circ\text{C}}$	$\frac{\theta_{21}}{^\circ\text{C}}$	$\frac{m}{\text{kg}}$	$\frac{h_1}{\text{W/m}^2\text{K}}$	Nu	St	Re	Pr	$\frac{L}{\text{m}}$	
66.960	62.500	22.560	.498	169.70	32.80	.8240E-04	1891.0	212.0	.5719E+12	8.154
67.820	62.910	22.720	.498	181.90	35.20	.8960E-04	1904.0	208.0	.6068E+12	8.154
59.550	56.740	24.420	.5354	140.00	26.96	.6500E-04	1591.0	262.0	.2905E+12	8.154
59.760	56.890	24.470	.5354	142.00	27.36	.6590E-04	1601.0	260.0	.2955E+12	8.154
62.780	59.320	25.500	.6576	202.00	38.96	.7590E-04	2170.0	238.0	.3767E+12	8.154
62.780	59.340	25.470	.6578	200.20	38.63	.7530E-04	2171.0	238.0	.3770E+12	8.154
70.120	64.870	25.470	.5773	229.50	44.43	.9710E-04	2364.0	195.0	.6860E+12	8.154
67.450	62.760	25.780	.7077	267.00	51.74	.9280E-04	2681.0	209.7	.5498E+12	8.154
67.400	62.700	25.810	.7077	268.10	51.84	.9298E-04	2677.0	210.0	.5469E+12	8.154
70.460	65.160	26.040	.6783	273.80	53.02	.9860E-04	2805.0	194.0	.6951E+12	5.154
70.480	65.180	26.040	.6783	273.60	52.99	.9850E-04	2808.0	193.0	.6967E+12	8.154
71.820	66.280	22.480	.4833	183.70	35.60	.9260E-04	2079.0	186.0	.8368E+12	8.154
71.690	66.250	22.460	.4833	180.40	34.96	.9096E-04	2073.0	187.0	.8112E+12	8.154
78.520	71.900	22.380	.4833	132.90	25.85	.9770E-04	1695.0	158.0	.1367E+13	8.154
69.100	64.660	22.100	.3424	109.50	21.19	.7730E-04	1390.0	199.0	.6988E+12	8.154
68.840	64.320	22.100	.3537	114.40	22.15	.7920E-04	1406.0	200.0	.6815E+12	8.154
65.370	60.880	22.050	.3611	125.00	24.14	.8520E-04	1281.0	223.0	.5035E+12	8.154
65.130	60.620	22.050	.3613	126.40	24.41	.8620E-04	1270.0	224.0	.4920E+12	8.154
53.070	50.650	21.970	.783	196.20	37.61	.6298E-04	1828.0	329.0	.1603E+12	8.154
54.600	52.520	21.970	.6611	139.50	26.80	.5130E-04	1700.0	308.0	.1930E+12	8.154
65.970	62.510	25.450	.4893	177.70	34.16	.8746E-04	1835.0	14.8	.2399E+12	6.341
61.580	59.030	24.470	.4893	180.20	34.71	.7135E-04	2009.0	243.7	.1689E+12	6.341
61.450	58.850	24.420	.6254	184.30	35.54	.7300E-04	1998.0	244.9	.1666E+12	6.341
70.120	65.780	25.550	.3246	136.20	26.39	.1024E-03	1346.0	192.5	.3327E+12	6.341
70.010	65.340	25.550	.3247	141.80	28.61	.1112E-03	1337.0	194.1	.3257E+12	6.341
68.550	64.710	26.750	.5268	204.60	39.60	.9500E-04	2097.0	200.0	.2910E+12	6.341
66.660	64.740	26.250	.5268	208.40	40.34	.9680E-04	2101.0	199.6	.2928E+12	6.341
66.410	63.040	26.400	.6476	279.70	44.40	.8778E-04	2402.0	211.8	.2439E+12	6.341
66.640	63.200	26.400	.6498	236.50	41.10	.8900E-04	2445.0	210.5	.2484E+12	6.341
72.180	67.770	25.960	.3822	157.90	30.62	.1005E-03	1690.0	181.6	.3908E+12	6.341
72.000	67.770	25.960	.3822	151.60	29.40	.9650E-04	1685.0	182.0	.3870E+12	6.341
62.650	59.710	25.680	.4164	143.50	27.69	.8720E-04	1413.0	236.9	.1780E+12	6.341
62.420	59.260	25.650	.4269	154.90	29.88	.8980E-04	1396.0	239.6	.1726E+12	6.341
63.640	60.150	25.580	.3324	129.00	25.08	.9650E-04	1179.0	231.6	.1911E+12	6.341
65.990	60.670	25.600	.2095	120.60	23.30	.1418E-03	748.0	221.9	.2221E+12	6.341
65.390	61.660	25.600	.2889	115.80	22.37	.9867E-04	1038.0	219.9	.2236E+12	6.341
58.980	57.310	25.030	.6438	182.60	35.11	.7050E-04	1912.0	261.9	.4864E+11	4.530
59.000	57.470	25.030	.4437	167.90	32.35	.6480E-04	1918.0	261.0	.4907E+11	4.530
61.040	58.820	24.880	.4171	170.60	32.89	.8860E-04	1512.0	246.6	.5854E+11	4.530
61.580	58.480	25.010	.3109	155.70	30.02	.1241E-03	982.0	246.0	.5931E+11	4.530
65.500	63.250	25.290	.5997	196.00	37.89	.8038E-04	2213.0	213.9	.8810E+11	4.530
71.870	67.300	25.140	.3173	187.90	36.42	.1440E-03	1366.0	183.7	.1406E+12	4.530
71.980	67.220	25.140	.3173	195.30	37.86	.1440E-03	1387.0	183.6	.1409E+12	4.530
77.180	69.910	25.160	.1873	163.50	31.77	.2109E-03	924.0	165.2	.1968E+12	4.530
76.950	69.490	25.160	.1946	175.10	34.01	.2176E-03	950.6	166.6	.1922E+12	4.530
68.760	66.020	26.500	.6349	243.50	47.15	.9370E-04	2587.0	195.0	.1120E+12	4.530

35.805	49,811	24,905	2111	258,79	46,71	29411-11	210,68	176,0	24791-07	4.516
42.857	59,276	34,436	2382	304,20	45,94	21884-01	243,68	175,6	25457-11	4.531
44.777	64,266	34,181	4433	311,28	37,69	22109-02	278,28	184,7	26421-01	4.516
45.196	65,152	35,051	2784	321,35	44,27	22531-04	279,17	184,7	27031-07	4.516 50
46.841	69,430	35,499	4772	327,73	45,14	22848-04	283,28	185,7	28331-01	4.516
48.258	72,771	36,929	3911	337,41	51,75	23081-04	348,21	194,7	28131-07	4.516
54.376	81,445	38,131	2188	373,51	71,24	23451-02	359,6	212,7	28451-01	4.516
59.741	87,273	38,199	2918	374,3	16,77	23401-03	390,2	213,1	21251-02	4.516
67,446	94,186	38,943	2414	375,11	14,28	23741-03	384,2	194,7	24011-07	4.516
68,616	97,376	38,131	2478	375,31	14,14	23879-07	385,1,0	194,1	23731-02	4.516
67,577	94,888	38,943	4443	382,0	25,18	23811-04	384,0	282,1	21211-01	4.516
74,410	100,410	38,410	2484	388,18	28,33	27131-01	382,0	147,5	21811-01	4.516
62,783	94,498	38,311	3667	388,81	27,86	23811-01	374,0	185,1	27441-11	4.516
84,811	111,898	38,617	3425	447,38	47,88	24810-04	373,0	245,7	21841-11	4.516 60
73,299	101,998	38,943	3191	437,43	24,78	24811-03	378,0	193,7	21851-11	4.516
74,888	76,538	38,943	2377	437,11	22,81	24811-01	378,1	185,7	25841-11	4.516
62,254	74,948	38,943	2355	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
61,741	71,911	38,943	3812	391,28	37,57	24741-01	378,1	185,7	22731-07	4.516
69,777	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	27711-11	4.516
69,777	72,741	38,943	3812	391,28	37,57	24741-01	378,1	185,7	23551-07	4.516
69,777	71,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,411	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516 70
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516
77,441	74,311	38,943	3812	391,28	37,57	24741-01	378,1	185,7	24741-11	4.516 80

84.342	79.751	37.896	.4385	191.41	35.77	.1039E-03	2774.3	132.3	.7667E+12	6.342
85.576	81.613	48.705	.4239	202.63	39.59	.1135E-03	2751.9	127.4	.844E+11	6.342
87.791	80.039	47.278	.4132	199.31	38.91	.9804E-03	2723.6	132.7	.5259E+12	6.342
81.433	74.961	47.045	.4118	193.87	37.83	.1056E-03	2706.2	133.0	.1940E+12	6.342
81.194	77.813	45.961	.4531	191.03	37.24	.1008E-03	2637.7	141.8	.5076E+12	6.342
79.280	76.17	45.526	.4605	187.69	36.56	.9874E-04	2571.7	141.3	.4416E+12	6.342
73.153	70.877	44.684	.4445	186.76	32.37	.7333E-04	2279.0	171.1	.2715E+12	6.342
68.08	66.227	44.139	.4448	186.54	32.24	.8063E-04	2041.9	196.6	.1701E+12	6.342
65.940	64.400	43.699	.5197	152.11	29.41	.7183E-04	1970.1	208.6	.1397E+12	6.342
66.999	59.602	35.023	.5445	117.13	24.47	.5616E-04	1571.1	278.2	.7583E+11	6.342 90
74.995	54.06	36.025	.5711	127.46	24.1	.8726E-04	1496.1	297.4	.5094E+11	6.342
73.565	52.785	38.517	.5787	124.40	23.90	.5737E-04	1436.8	312.6	.471E+11	6.342
70.133	49.187	38.517	.6020	115.04	22.06	.4822E-04	1293.9	354.7	.2422E+11	6.342
47.097	46.707	38.362	.6245	112.55	21.55	.4172E-04	1184.6	399.1	.1428E+11	6.342
43.981	47.647	38.362	.6507	107.92	20.63	.4232E-04	1075.2	454.7	.7054E+10	6.342
40.194	47.044	39.293	.6370	165.81	31.64	.1011E-04	898.5	534.0	.7099E+09	6.342
36.116	35.927	31.227	.6470	111.42	21.21	.4725E-04	766.4	642.0	.3001E+10	6.342
42.144	41.858	35.413	.6181	105.73	20.19	.4377E-04	943.1	491.5	.712E+10	6.342
46.733	46.247	31.227	.5883	116.85	22.37	.543E-04	1196.1	415.9	.1583E+11	6.342
52.795	51.537	37.197	.5711	121.84	24.31	.5127E-04	1181.1	377.7	.3734E+11	6.342 100
53.669	50.785	31.227	.5433	123.63	18.75	.5706E-04	1336.1	112.1	.4335E+11	6.342
56.817	55.724	31.227	.5711	124.47	23.55	.5952E-04	1443.7	279.5	.6463E+11	6.342
60.149	58.647	39.025	.5711	124.46	23.91	.6079E-04	1178.6	250.0	.9524E+11	6.342
55.785	56.847	34.300	.4999	115.49	21.14	.4781E-04	1400.0	249.4	.1128E+12	6.342
63.448	61.948	34.458	.4459	117.35	20.66	.4736E-04	596.1	716.7	.1587E+12	6.342
67.507	65.444	34.768	.4787	116.51	22.61	.4731E-04	1735.7	200.6	.2245E+12	6.342
70.693	68.238	31.049	.4774	119.64	23.19	.4750E-04	1771.7	183.9	.7937E+12	6.342
43.716	42.688	33.654	.4205	113.07	21.61	.641E-04	770.3	470.3	.1109E+11	6.342
43.673	43.077	33.835	.4611	114.1	21.11	.6271E-04	816.7	462.3	.1172E+11	6.342
41.057	47.097	34.171	.4283	121.46	23.26	.7185E-04	836.4	398.4	.2375E+11	6.342 110
55.125	53.513	34.481	.3619	117.81	23.04	.6747E-04	945.8	299.9	.6033E+11	6.342
62.641	64.791	42.411	.3613	117.92	22.81	.7980E-04	1313.4	205.2	.1571E+12	6.342

END OF REPORT.

Mixed Carburetor

Run	51.000	57.770	5.010	.1437	167.90	32.34	-.04681-04	1918.0	261.0	.4907E+11	4.52E
38	62.840	60.830	5.990	.6112	151.70	-1.61	.9840E-04	1501.0	231.7	.6818E+11	4.53E
51	72.499	69.336	36.317	.6959	155.01	30.08	.2589E-04	7.72.8	175.2	-.3299E+12	6.34E
76	75.334	72.342	38.938	.4840	173.00	11.62	-.0631E-04	2408.5	162.5	.4120E+12	6.34E
77	77.971	73.990	37.110	.1752	178.05	34.64	-.9019E-04	2490.2	155.0	-.4762E+12	6.34E
78	80.144	76.215	37.404	.4402	185.10	36.06	-.2487E-04	2585.6	145.6	.5761E+12	6.34E
79	82.427	78.180	37.818	.4531	189.60	36.98	.7792E-04	2696.2	138.1	.6730E+12	6.34E
80	84.342	79.751	37.896	.4185	191.42	37.13	.1019E-04	2734.3	112.3	.7667E+12	6.34E
81	86.510	81.611	38.705	.4239	202.63	39.59	-.1125E-04	2781.8	127.4	.4544E+12	6.34E
82	83.791	80.039	37.278	.4385	199.11	18.81	-.1037E-04	2731.6	111.1	.5219E+12	6.34E
83	83.633	79.968	37.015	.370	191.83	37.83	.1078E-04	2706.2	133.0	-.910E+12	6.34E
84	81.192	77.813	36.981	.4531	191.03	31.81	-.1086E-04	2637.7	140.8	.5016E+12	6.34E
85	79.280	77.132	36.566	.4605	187.69	36.26	.8226E-04	2351.3	147.3	.4416E+12	6.34E
86	73.153	70.877	36.684	.4849	166.76	32.37	.8332E-04	2279.0	171.2	-.715E+12	6.34E
87	68.081	66.227	42.139	.5050	166.54	32.21	-.0963E-04	2541.9	196.6	.1705E+12	6.34E
88	65.940	64.550	43.692	.4197	152.11	29.41	.7183E-04	1972.1	208.6	.1397E+12	6.34E
89											



Mixed laminar

Run	3	59.550	56.740	24.820	.5354	140.00	26.96	.6500E-04	1591.0	262.0	.2905E+12	8.154
	4	59.760	56.890	24.470	.5353	142.00	27.36	.6590E-04	1601.0	260.0	.2955E+12	8.154
	5	62.780	59.320	25.500	.6578	202.00	38.96	.7590E-04	2170.0	238.0	.3767E+12	8.154
	6	62.780	59.340	25.470	.6578	200.20	38.63	.7530E-04	2171.0	238.0	.3770E+12	8.154
	19	53.070	50.650	21.970	.7834	196.20	37.80	.6798E-04	1828.0	329.0	.1603E+12	8.154
	20	54.600	52.520	21.970	.6811	139.50	26.80	.5130E-04	1700.0	308.0	.1934E+12	8.154
	72	63.383	61.714	37.663	.5273	263.73	27.75	.6721E-04	1830.6	226.4	.1363E+12	6.342
	73	63.383	61.766	37.870	.5345	142.18	27.45	.6599E-04	1857.3	226.2	.1155E+12	6.342
	74	54.267	53.253	34.068	.5795	119.45	22.96	.5165E-04	1457.3	306.0	.5724E+11	6.342
	75	51.901	51.017	34.068	.5945	170.47	23.32	.5099E-04	1364.1	333.5	.2226E+11	6.342
	90	56.997	55.802	35.023	.5645	127.13	24.67	.5616E-04	1571.2	278.2	.7583E+11	6.342
	91	54.995	54.085	38.025	.5771	127.46	24.20	.5576E-04	1496.2	297.4	.5094E+11	6.342
	92	53.565	52.785	38.517	.5836	124.40	23.90	.5377E-04	1436.8	312.6	.4071E+11	6.342
	93	50.133	49.587	38.517	.6070	115.04	22.06	.4877E-04	1193.9	354.7	.2422E+11	6.342
	94	47.097	44.707	38.362	.6245	112.55	21.57	.4572E-04	1194.6	399.1	.1428E+11	6.342
	95	43.880	43.647	38.232	.6507	107.92	20.63	.4237E-04	1075.2	454.7	.7054E+10	6.342
	96	40.096	40.044	39.293	.6370	165.87	31.64	.6690E-04	888.5	534.0	.7099E+09	6.342
	97	36.136	35.929	31.227	.6670	111.42	21.21	.4325E-04	766.4	642.0	.3001E+10	6.342
	98	42.144	41.958	37.463	.6183	105.73	20.19	.4377E-04	943.3	490.5	.7128E+10	6.342
	99	46.733	46.240	36.680	.5883	116.85	22.37	.5043E-04	1096.1	405.9	.1593E+11	6.342
	100	52.395	51.537	37.197	.5472	126.84	24.35	.5877E-04	1281.6	327.2	.1734E+11	6.342
	101	53.669	52.785	37.689	.5433	123.63	23.75	.5706E-04	1338.1	312.1	.4335E+11	6.342
	102	56.817	55.724	37.870	.5211	124.47	23.95	.5958E-04	1443.2	279.5	.6463E+11	6.342
	103	60.149	58.847	38.075	.5061	124.06	23.51	.6079E-04	1578.6	250.0	.9524E+11	6.342
	104	60.305	58.847	34.300	.4609	109.69	21.14	.5788E-04	1470.0	249.4	.1128E+12	6.342
	105	63.748	61.948	34.455	.4550	117.35	22.66	.6356E-04	1596.1	221.3	.1587E+12	6.342

Table 6-continued

867.450	17.750	5.780	7.077	267.60	51.74	.9280E-04	2281.0	209.7	15498E+12	8.154
937.400	17.750	25.810	7.077	268.10	51.84	.7290E-04	2677.0	280.0	5488E+12	8.154
1070.460	17.750	24.040	6.783	273.80	53.02	.9840E-04	2805.0	194.0	6951E+12	8.154
1197.400	17.750	26.040	6.783	273.60	52.99	.9910E-04	2804.0	194.0	6967E+12	8.154
2261.450	17.750	24.420	6.253	380.20	34.75	.719E-04	2009.0	243.7	1689E+12	6.341
2764.510	17.750	24.400	6.254	184.30	33.34	.7300E-04	1998.0	246.9	1868E+12	6.341
2966.740	17.750	24.400	6.426	229.70	44.40	.8770E-04	2402.0	211.8	2419E+12	6.341
3135.900	17.750	25.030	6.458	246.50	51.73	.8700E-04	2445.0	210.5	2784E+12	6.341
3165.500	17.750	25.030	6.438	182.60	35.17	.7520E-04	1912.0	261.9	14864E+11	4.530
4668.760	17.750	26.500	5.991	196.00	37.89	.8050E-04	2213.0	213.9	8810E+11	4.530
5063.180	17.750	25.910	6.349	243.50	47.15	.9170E-04	2587.0	195.0	3120E+12	4.530
5167.800	17.750	25.790	7.082	231.30	44.63	.8070E-04	2344.0	236.7	6410E+11	4.530
7121.181	17.750	32.663	6.792	251.70	41.63	.7830E-04	2101.0	231.7	6818E+11	4.530
7263.381	17.750	37.870	6.273	143.73	27.72	.6730E-04	1830.6	226.4	1167E+12	6.342
7672.599	17.750	35.317	5.345	142.18	27.45	.6570E-04	1857.3	226.2	1355E+12	6.342
7775.534	17.750	36.938	6.259	155.01	30.08	.7500E-04	2272.8	175.2	3299E+12	6.342
8868.081	17.750	44.139	4.840	173.00	33.62	.8630E-04	2408.5	162.5	4120E+12	6.342
8965.940	17.750	43.699	5.050	166.54	32.24	.8060E-04	2041.9	194.6	1701E+12	6.342
8840.096	17.750	38.293	5.197	152.11	29.41	.7183E-04	1970.1	208.6	1397E+12	6.342
10563.758	17.750	35.355	6.520	165.87	33.68	.6690E-04	1908.5	534.0	7099E+09	6.342
10667.207	17.750	36.765	4.550	117.35	22.66	.6356E-04	1596.1	224.3	1587E+11	6.342
10770.691	17.750	35.049	4.387	116.85	22.61	.6250E-04	1735.7	200.6	2253E+12	6.342
			4.076	119.64	23.19	.7150E-04	1771.8	103.9	2937E+12	6.342

Lower Transitional

Rum

1	60.920	62.500	20.540	5035	169.70	32.80	8240E	04	1891.0	212.0	.5739E+12	8.154
2	67.820	62.910	22.720	4982	181.90	15.20	-8960E	04	1904.0	-08.0	.6068E+12	8.154
7	70.170	64.870	25.410	73	29.50	4.43	.710F	-04	2364.0	195.0	.6860F+12	8.154
12	71.820	66.280	27.900	4833	183.7	35.60	-9260F	04	2379.0	186.0	.8168E+12	8.154
13	71.830	66.280	27.900	1833	180.40	34.96	.9096I	04	2073.0	182.0	.8312I+12	8.154
14	69.100	64.630	22.560	1311	109.50	11.19	.710E	-04	1490.0	197.0	.6988E+12	8.154
15	68.840	64.320	22.100	3537	114.44	22.15	.7920E	04	1408.0	-00.0	.6812E+12	8.104
17	65.770	67.880	21.720	4612	125.00	24.14	.8570E	04	1281.0	123.0	.5072E+12	8.154
18	65.230	60.870	21.210	1613	124.80	21.34	.7180E	-04	1770.0	171.0	.4910E+12	8.154
21	55.970	62.570	18.850	4893	177.7	34.78	.8740I	04	1835.0	214.8	.2199E+12	6.3
26	68.550	64.710	24.250	5268	204.60	39.48	.7200I	04	2097.0	120.0	.2910I+12	6.148
27	68.660	64.750	24.280	2168	208.50	18.31	.8600E	04	2101.0	199.6	.2979E+12	6.341
30	72.230	67.710	27.960	342	127.10	30.62	.1000I	03	1690.0	181.6	.3908I+12	6.341
31	72.000	67.770	27.960	5522	131.60	29.40	.9600I	04	1685.0	182.0	.3870E+12	6.341
32	61.650	67.110	25.680	4264	143.50	17.69	-83.0I	04	1413.0	276.9	.1780E+12	6.341
33	67.520	69.240	25.650	4777	134.90	29.88	.8980E	04	1396.0	219.6	.1726E+12	6.341
38	69.000	67.470	25.030	443	167.90	12.35	-6.80I	04	1918.0	261.0	.7107E+12	4.530
39	61.040	63.820	23.880	4221	170.60	32.89	.8860I	-04	1512.0	246.6	.5854E+12	4.530
47	72.080	68.810	26.580	5335	228.70	54.36	-1040I	02	2392.0	179.0	.1450E+12	1.530
48	72.860	68.570	26.610	5762	224.00	43.46	.1030E	03	2415.0	175.0	.1545E+12	4.510
49	66.750	64.240	26.190	4692	171.00	31.02	-6930I	-04	1798.0	206.7	.9540E+12	4.530
62	65.550	62.770	24.620	3971	147.60	28.53	.9138E	04	1461.0	212.7	.6860E+12	1.530
57	67.530	64.900	26.040	1041	151.80	29.38	.9202I	-04	1586.0	202.3	.1021F+12	1.530
58	66.780	64.690	25.320	6063	299.80	57.80	.1708E	03	2341.0	205.1	.2136E+12	2.7200
60	64.870	63.300	25.630	6426	247.70	17.88	.9480I	04	2350.0	215.7	.1834E+12	2.7200
88	68.081	66.227	44.139	5050	166.54	32.24	.8063E	04	201.9	196.6	.1701E+12	6.342
89	65.940	64.600	43.699	5197	152.11	29.41	.7183I	-04	1970.1	208.6	.1397E+12	6.342

Mixed Trans 10 80

Run	78.520	79.700	22.380	137.7	132.10	25.24	9.170	1.050	7.820	1.340E+03	8.134
14	79.100	65.780	25.550	134.6	136.20	26.39	1024E-07	13.80	7.920	1.870E+02	8.385
24	79.000	65.340	25.550	134.7	147.00	28.41	1112E-07	17.70	8.140	3.050E+02	8.385
25	63.143	60.350	25.510	133.4	129.70	25.04	9.53E-07	17.00	7.770	1.880E+02	8.141
34	65.990	60.670	25.650	133.5	126.40	23.28	1.418E-07	7.480	7.147	1.220E+02	6.381
35	65.300	61.460	25.590	133.0	115.80	22.77	9.867E-04	17.10	16.08	1.250E+02	6.381
36	64.580	58.480	25.010	132.8	155.70	30.52	1.241E-03	98.7	26.21	1.030E+02	6.381
40	64.870	67.300	25.140	131.8	187.70	36.10	1.440E-03	170.0	19.17	1.006E+02	6.270
42	64.700	67.220	25.240	131.2	195.10	37.86	1.198E-03	150.0	18.16	1.009E+02	6.270
43	64.200	66.010	25.260	130.7	183.20	31.77	2.094E-07	170.0	16.72	1.861E+02	6.270
44	64.200	66.010	25.260	130.7	183.20	31.77	2.094E-07	170.0	16.72	1.861E+02	6.270
45	64.200	66.010	25.260	130.7	183.20	31.77	2.094E-07	170.0	16.72	1.861E+02	6.270
53	64.200	66.010	25.260	130.7	183.20	31.77	2.094E-07	170.0	16.72	1.861E+02	6.270
54	64.200	66.010	25.260	130.7	183.20	31.77	2.094E-07	170.0	16.72	1.861E+02	6.270
55	62.840	64.380	25.800	131.3	172.10	31.3	1.074E-03	100.0	18.17	1.403E+02	6.270
56	64.000	65.330	25.830	131.4	177.40	32.94	1.627E-03	110.0	19.03	1.795E+02	6.270
58	64.830	70.600	25.420	130.4	257.10	50.14	2.133E-03	121.0	17.09	1.797E+02	6.270
59	64.780	64.600	25.520	130.3	228.00	51.80	1.208E-03	134.0	20.21	2.347E+02	6.270
61	64.260	60.580	25.960	130.2	175.40	28.23	1.688E-03	90.0	17.12	3.492E+01	6.270
62	64.310	61.100	25.500	130.0	177.10	22.92	5.01E-03	105.0	18.25	4.922E+01	6.270
63	64.310	61.100	25.500	130.0	177.10	22.92	5.01E-03	105.0	18.25	4.922E+01	6.270
64	64.310	61.100	25.500	130.0	177.10	22.92	5.01E-03	105.0	18.25	4.922E+01	6.270
65	64.310	61.100	25.500	130.0	177.10	22.92	5.01E-03	105.0	18.25	4.922E+01	6.270
66	64.310	61.100	25.500	130.0	177.10	22.92	5.01E-03	105.0	18.25	4.922E+01	6.270
67	64.310	61.100	25.500	130.0	177.10	22.92	5.01E-03	105.0	18.25	4.922E+01	6.270
68	64.310	61.100	25.500	130.0	177.10	22.92	5.01E-03	105.0	18.25	4.922E+01	6.270
69	63.320	79.300	27.640	131.8	174.00	26.01	1.002E-03	183.0	13.48	7.927E+01	6.270
70	62.840	81.430	27.510	131.8	174.00	26.01	1.002E-03	183.0	13.48	7.927E+01	6.270
71	62.840	81.430	27.510	131.8	174.00	26.01	1.002E-03	183.0	13.48	7.927E+01	6.270

513 COMPUTER PROGRAM

514 Program 1

PROGRAM MAIN (TEMP, DENS, TAPZ, THERM, TAI, CO, DDT, DT,  
STAP, SOUT, TAP10, TAP15, TAP18, TAP20, TAP25, TAP30, TAP35,  
TAP40) FOR USE WITH THE BLENK NAME

DIMENSION RT(10), R(10), RE(10), C(10), HWE(10), SW(10),  
A(10), VE(10), VIB(10)  
RT IS THE REGISTERED THERMOMETER READING  
R IS THE CALCULATED REYNOLDS NUMBER  
RE IS THE REYNOLDS NUMBER  
C IS THE STAGNANT INTERPOLATED TEMPERATURE  
HWE IS THE SURFACE STAGNANT INTERPOLATED DENSITY  
SW IS THE SURFACE INTERPOLATED THERMAL CONDUCTIVITY  
A IS THE SURFACE INTERPOLATED SPECIFIC HEAT  
VE IS THE SURFACE INTERPOLATED VISCOSITY  
VIB IS THE VELOCITY AT EACH POINT  
M IS THE NUMBER OF TEMPERATURE POINTS  
C  
WRITE(\*,\*) BLENK NAME PROGRAM  
WRITE(\*,\*) ENTER NO. OF POINTS IN ARRAY  
WRITE(\*,\*)  
READ(\*,\*) M  
WRITE(\*,\*)  
CONTINUE  
IF (M.EQ.1) THEN  
IF (MATER.EQ.1) THEN  
IF (MATER.EQ.2) THEN

\*\*\*\*\*  
OBTAIN THE EXPERIMENTAL DATA  
APPLY TEMPERATURES DOWN THE EXCHANGER  
DO 100 I=1, M  
WRITE(\*,\*)  
400 FORMAT(' GIVE THE DISTANCE THERMOMETER READING (170)')  
WRITE(\*,\*) 'ENTER REGISTERED THERMOMETER READING'  
C  
READ(\*,\*) RT(I), TAP10(I)  
IF (I.EQ.2) THEN  
CONTINUE  
READ THE TEMPERATURE READING ON THE VIB SIDE  
WRITE(\*,\*)  
500 FORMAT(' GIVE THE TEMPERATURE READING IN PERCENT')  
READ(\*,\*)  
IF (I.EQ.2) THEN  
CONTINUE

```

C FROM THIS DATA CALCULATE ALL THE TEMPERATURES
C AND PHYSICAL PROPERTIES, AND PRANDTL AND REYNOLDS NUMBERS
CALL TEMP(RT,T,N)
IF(IWATER.EQ.1) GO TO 25
CALL DENC(T,DEN,N)
CALL COND(T,CND,N)
CALL SPECIF(T,CP,N)
CALL VISC(T,VIS,N)
25 CONTINUE
CALL DENCW(T,DEN,N)
CALL CONDW(T,CND,N)
CALL SPECIFW(T,CP,N)
CALL VISCW(T,VIS,N)
27 CALL ROTA(RR,UMASS,VEL,DEN,N)
CALL REYN(DEN,D.O.539,VIS,UMASS,VEL,RE,N)
CALL PRANDTL(CP,VIS,CND,PR,N)

C FROM THE ABOVE ALL THE REQUIRED DATA HAS BEEN CALCULATED AND STORED
C IN THE RELEVANT ARRAYS OF DIMENSION 10 .THE DIAMETER OF THE TUBE IS
C 25.39 MM IN THE REYNOLDS CALC.
-----
C THIS IS NOW PRINTED OUT
C
C CALL WRT(RR,UMASS,RT,T,DEN,CND,CP,VIS,RE,PR," " N)
-----
C CALCULATE THE HEAT BALANCE FROM THE DATA
C
C IF(IWATER.EQ.1)CALL HEATLW(T,UMASS,CP,FE,FR,CND,N)
C IF(IWATER.EQ.0)CALL HEATL(T,UMASS,CP,FE,FR,CND,DEN,VIS,N)
-----
C REQUESTS FOR PLOTTER
C
C GOTO 71
17 WRITE(6,506)
506 FORMAT(/15," DO YOU WANT TO PLOT (Y/N)")
READ(5,507)(PLOT
507 FORMAT(A10)
IF(EOF(5))22,15
15 IF(IFLOT.EQ.1HN)GO TO 22
WRITE(6,508)
508 FORMAT(15,"WHICH TWO VARIABLES DO YOU WISH TO PLOT X-Y"
1,/, " OR TYPE HELP TO LIST")
READ(5,507)I,PLOT
IF(EOF(5))20,16
16 IF(IFLOT.EQ.1HH)WRITE(6,509)
IF(IFLOT.EQ.1HH) GO TO 15
IF(IFLOT.EQ."RES-TEMP")CALL RPLOT(RT,T,N)
IF(IFLOT.EQ."RES-DEN")CALL RPLOT(RT,DEN,N)
IF(IFLOT.EQ."RES-COND")CALL RPLOT(RT,CND,N)
IF(IFLOT.EQ."RES-CP")CALL RPLOT(RT,CP,N)

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IF (IPLOT.EQ."RES-VIS")CALL RILOT(RT,VIS,N)
IF (IPLOT.EQ."RES-REY")CALL RFLOT(RT,RE,N)
IF (IFLOT.EQ."RES-FR")CALL RFLOT(RES,FR,N)
IF (IPLOT.EQ."TEMP-RES")CALL RILOT(T,RT,N)
IF (IPLOT.EQ."TEMP-DEN")CALL RILOT(T,DEN,N)
IF (IPLOT.EQ."TEMP-COND")CALL RFLOT(T,COND,N)
IF (IPLOT.EQ."TEMP-CF")CALL RFLOT(T,CF,N)
IF (IPLOT.EQ."TEMP-VIS")CALL RILOT(T,VIS,N)
IF (IFLOT.EQ."TEMP-REY")CALL RILOT(T,RE,N)
IF (IFLOT.EQ."TEMP-FR")CALL RILOT(T,FR,N)
GO TO 17
509 FORMAT(/," RESISTANCE RES, TEMPERATURE TEMP, DENSITY DEN"
A,/" CONDUCTIVITY COND, SPECIFIC HEAT CF, VISCOSITY VIS,"
E,/" REYNOLDS REY, FRANDTL FR "
C,/" E.G. TEMP-DEN TEMP-FR")
C
C
C 22 GO TO 71
C 20 STOP
C END
C SUBROUTINE DENC(T,DEN,N)
C E.I. UNITS ARE USED
C
C SUBROUTINE TO INTERPOLATE FOR
C DENSITY FROM A GIVEN TEMPERATURE
C USING CUBIC SPLINES
C DATA IS FOR REGAL OIL I
C
C T CONTAINS THE TEMPERATURE(S)
C DEN CONTAINS THE INTERPOLATED DENSITY(S)
C N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
C DTABLE CONTAINS THE LOOK UP TABLE
C
C DIMENSION T(N),DEN(N),DTABLE(9,9),C(4,9)
C DATA (DTABLE(1,2),I=1,9)/
C A 872.5,862.1,848.,837.8,825.,799.8,
C B 772.5,750.1,727.1/
C DATA (DTABLE(1,1),I=1,9)/
C A 20.,40.,60.,80.,100.,
C B 150.,200.,250.,300 /
C
C CALCULATE THE CONSTANTS FOR THE SPLINE FIT
C CALL SPLCON(DTABLE,C,9)
C
C SET UP LOOP FOR NUMBER OF DATA POINTS
C
C DO 100 K 1,N
C
C FIND THE POSITION OF T IN THE LOOK UP TABLE
C
C DO 1 J=2,9
C J I
C COUNTER VALUE AT EXIT
C IF (T(K).LT.DTABLE(1,1)) GO TO 3
C IF (T(K).LE.DTABLE(1,1)) GO TO 2
C IF NOT LOOP
C 1 CONTINUE

```

THE VALUE IS OUT OF THE TOP OF THE TABLE  
CONTINUE

THE VALUE IS NOT FOUND WITHIN THE BOUNDS OF THE TABLE

PRINT WARNING AND INTERPOLATE USING LAST 2 VALUES IN THE TABLE  
WRITE(6,99=1(K)

FORMAT(115,"THE TEMPERATURE VALUE ",F10.2," WAS NOT FOUND"  
IN THE BOUNDS OF THE TOP OF TABLE. THE DENSITY HAS BEEN"  
NEARLY EXTRAPOLATED FROM EXISTING VALUES")

CARRY ON INTO THE INTERPOLATION SECTION

INTERPOLATE LINEARLY FOR T

$DEN(K) = (T(K) - DTABLE(J,1)) / (DTABLE(J,1) - DTABLE(J-1,1))$   
 $DEN(K) - DEN(K) * (DTABLE(J,2) - DTABLE(J-1,2)) + DTABLE(J-1,2)$

DEN(K) IS NOW INTERPOLATED  
CALCULATE THE NEXT VALUE  
GO TO 100

CONTINUE

DEN(K) IS NOW INTERPOLATED USING A  
CUBIC SPLINE ROUTINE

ROUTINE SPLCON CALCULATES THE INTERPOLATION CONSTANTS

$DEN(K) = (DTABLE(I,1) - T(K)) * C(1,J) + (DTABLE(J,1) - T(K)) * C(2,J)$   
 $+ (DTABLE(J,1) - T(K)) * C(3,J) + (DTABLE(J,1) - T(K)) * C(4,J)$   
 $+ DTABLE(J,2)$   
 $DEN(K) = DEN(K) + (T(K) - DTABLE(J,1)) * C(1,J)$   
 $+ (T(K) - DTABLE(J,1)) * C(2,J)$   
 $+ (T(K) - DTABLE(J,1)) * C(3,J)$   
 $+ (T(K) - DTABLE(J,1)) * C(4,J)$

DEN(K) HAS BEEN INTERPOLATED BY SPLINES  
CALCULATE THE NEXT VALUE

CONTINUE

FINISHED

RETURN WITH INTERPOLATED VALUE

RETURN

END

ROUTINE SPLCON(TABLE,C,N)

ROUTINE TO CALCULATE THE CONSTANTS FOR THE  
SPLINE III

III: FINNINGTON, R.H.

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ANALYSIS. MAC MILLAN 1965



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DIMENSION TABLE(M,3),C(4,M),D(10),F(10),E(10),
A A(10,3),B(10),Z(10),X(10),Y(10)
DO 100 I=1,M
X(I)=TABLE(I,1)
Y(I)=TABLE(I,2)
100 CONTINUE
MM=M-1
DO 2 I=1,MM
D(K)=X(H+1)-X(K)
F(K)=D(K)/6.
2 E(K)=(Y(K+1)-Y(K))/D(K)
DO 3 K=2,MM
3 E(K)=E(K)-E(K-1)
A(1,2)=1.-D(1)/D(2)
A(1,3)=D(1)/D(2)
A(2,3)=F(2)-F(1)*A(1,3)
A(2,2)=1.*(F(1)+F(2))-F(1)-A(1,2)
A(2,3)=A(2,3)/A(2,2)
E(2)=E(2)/A(2,2)
DO 4 K=3,MM
4 A(K,2)=2.*(F(K-1)+F(K))-F(K-1)-A(1,3)
B(K)=B(K)-F(K-1)*B(K-1)
A(K,3)=F(K)/A(K,2)
6 B(K)=B(K)/A(K,2)
Q=D(M-2)/D(M-1)
A(M,1)=1.+Q+A(M-2,3)
A(M,2)=-Q-A(M,1)-A(M-1,3)
B(M)=B(M-2)/A(M,1)+B(M-1)
Z(M)=B(M)/A(M,2)
MN=M-2
DO 6 I=1,MN
6 Z(K)=E(K)/A(K,3)+Z(K+1)
Z(1)=A(1,2)+Z(2)/A(1,3)+Z(3)
DO 7 H=1,MM
7 Q=1./(6.*D(K))
C(1,K)=Z(K)+Q
C(2,K)=Z(K+1)+Q
C(3,K)=Y(K)/D(K)+Z(K)+F(K)
C(4,K)=Y(K+1)/D(K)+Z(K+1)+F(K)
RETURN
END
SUBROUTINE VIS(T,VIS,N)
S.I. UNITS ARE USED

SUBROUTINE TO INTERPOLATE FOR
VISCOSITY FROM A GIVEN TEMPERATURE
DATA IS FOR REGAL OIL P

T CONTAINS THE TEMPERATURE(S)
VIS CONTAINS THE INTERPOLATED VISCOSITY(S)
N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
TABLE CONTAINS THE LOOK UP TABLE

```

```

DIMENSION T(N),VIS(N),VTABLE(10,7),C(10,7)
DATA (VTABLE(I,2),I=1,10)
A4.626,3.57,2.75,2.135,1.656,.802,.239/
DATA (VTABLE(I,1),I=1,7)
A.00341,.00319,.003,.0028,.00268,
B.00236,.00211/
-----
C CALCULATE THE CONSTANTS FOR THE SPLINE FIT
CALL SPLICON(VTABLE,C,7)
C
C SET UP LOOP FOR NUMBER OF DATA POINTS
C
DO 100 K=1,N
C
C FIND THE POSITION OF T IN THE LOOK-UP TABLE
C
T(K)=1./(273.15+T(K))
DO 1 I=2,9
J=1
C
C COUNTER VALUE AT EXIT
IF(T(K).GT.VTABLE(1,1)) GO TO 3
IF(T(K).GE.VTABLE(1,1), GO TO 2
C IF NOT LOOP
1 CONTINUE
C
-----
C THE VALUE IS OUT OF THE TOP OF THE TABLE
3 CONTINUE
C
C THE VALUE IS NOT FOUND WITHIN THE BOUNDS OF THE TABLE
C
C PRINT WARNING AND INTERPOLATE USING LAST 4 VALUES IN THE TABLE
TK=1./T(K)-273.15
WRITE(6,99) TK
99 FORMAT(/15,"THE TEMPERATURE VALUE ",F10.2," WAS NOT FOUND"
C " IN THE BOUNDS OF THE LOOK-UP TABLE. THE VISCOSITY HAS BEEN"
D,/,15," LINEARLY EXTRAPOLATED FROM EXISTING VALUES")
C
C CARRY ON INTO THE INTERPOLATION SECTION
C
C INTERPOLATE LINEARLY FOR T
C
VIS(K)=(T(K)-VTABLE(J,1))/(VTABLE(J,1)-VTABLE(J-1,1))
VIS(K)=VIS(K)*(VTABLE(J,7)-VTABLE(J-1,7))+VTABLE(J-1,7)
VIS(K)=EXP(VIS(K))*0.001
T(K)=1./T(K)-273.15
C
C THE VISCOSITY IS NOW INTERPOLATED
C CALCULATE THE NEXT VALUE
GO TO 100
-----

```

CONTINUE

THE VISCOSITY IS NOW INTERPOLATED USING A  
CUBIC SPLINE ROUTINE  
SUBROUTINE SPLICON CALCULATES THE INTERPOLATION CONSTANTS

```

VIS(K) = (VTABLE(J,1) - T(K)) * (C(1,J-1) * (VTABLE(J,1) - T(K)) + 2
Z + C(3,J-1))
VIS(K) = VIS(K) + (T(K) - VTABLE(J-1,1)) * (C(2,J-1) * (T(K) - VTABLE(J-1,1)
Z)) * 2 + C(4,J-1)
VIS(K) = EXP(VIS(K)) * 0.001
T(K) = 1./T(K) - 273.15

```

THE VISCOSITY HAS BEEN INTERPOLATED BY SPLINES  
CALCULATE THE NEXT VALUE

130 CONTINUE

ALL FINISHED  
RETURN WITH INTERPOLATED VALUES

```

RETURN
END
SUBROUTINE SPECIF(T,CF,N)
S.I. UNITS ARE USED

```

SUBROUTINE TO INTERPOLATE LINEARLY FOR  
SPECIFIC HEAT FROM A GIVEN TEMPERATURE  
DATA IS FOR REGAL OIL B

T CONTAINS THE TEMPERATURE(S)  
CF CONTAINS THE INTERPOLATED SPECIFIC HEAT(S)  
N IS THE NUMBER OF DATA POINTS TO BE PROCESSED  
CPTABLE CONTAINS THE LOOK UP TABLE

```

DIMENSION T(N),CF(N),CPTABLE(9,9)
DATA (CPTABLE(1,2),1-1,9) /
A 1871.5,1946.02,2018.04,2093.4,2160.35,
B 2343.77,2522.55,2701.58,2890.99 /
DATA (CPTABLE(1,1),1-1,9) /
A 20.,40.,60.,80.,100.,120.,140.,160.,180. /

```

SET UP LOOP FOR NUMBER OF DATA POINTS

```
DO 100 K 1,N
```

FIND THE POSITION OF T IN THE LOOK UP TABLE

```
DO 11 2,9
J 1

```

```

C           COUNTER VALUE AT EXIT
C           II (T(I).LT.CFTABLE(1,1)) GO TO 3
C           IF (T(K).LE.CFTABLE(1,1)) GO TO 1
C IF NOT L001
1          CONTINUE
C
C -----
C THE VALUE IS OUT OF THE TOP OF THE TABLE
3          CONTINUE
C
C THE VALUE IS NOT FOUND WITHIN THE BOUNDS OF THE TABLE
C
C PRINT WARNING AND INTERPOLATE USING LAST 2 VALUES IN THE TABLE
C           WRITE(6,99) T(K)
99          FORMAT(75,"THE TEMPERATURE VALUE ",F10.2," WAS NOT FOUND"
C           " IN THE BOUNDS OF THE LOOK UP TABLE. THE SPECIFIC HEAT HAS BEEN"
C           " LINEARLY INTERPOLATED FROM EXISTING VALUES")
C
C CARRY ON INTO THE INTERPOLATION SECTION
C -----
C
2          CONTINUE
C INTERPOLATE LINEARLY FOR CP
C
C           CP(K) = (T(K) - CFTABLE(J-1,1)) / (CFTABLE(J,1) - CFTABLE(J-1,1))
C           CP(K) = CP(K) * (CFTABLE(J,2) - CFTABLE(J-1,2)) + CFTABLE(J-1,2)
C
C THE SPECIFIC HEAT IS NOW INTERPOLATED
C CALCULATE THE NEXT VALUE
100         CONTINUE
C -----
C
C ALL FINISHED
C RETURN WITH INTERPOLATED VALUES
C
C           RETURN
C           END
C           SUBROUTINE COND(T,CND,N)
C           S.I. UNITS ARE USED
C
C           SUBROUTINE TO INTERPOLATE LINEARLY FOR
C           THERMAL CONDUCTIVITY FROM A GIVEN TEMPERATURE
C           DATA IS FOR REGAL OIL
C
C           T CONTAINS THE TEMPERATURE(S)
C           CND CONTAINS THE INTERPOLATED THERMAL CONDUCTIVITY(S)
C           N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
C           CTABLE CONTAINS THE LOOK UP TABLE
C
C           DIMENSION T(N),CND(N),CTABLE(9,2)
C           DATA (CTABLE(1,1),1,1,9) /
C           A 20.,40.,60.,80.,100.,120.,140.,160.,180.,
C           DATA (CTABLE(1,2),1,1,9) /
C           A 132.36,131.048,129.66,128.19,126.9,123.
C           B 119.42,115.19,112.36/

```

```

C
C SET UP LOOP FOR NUMBER OF DATA POINTS
E
DO 100 K 1,N
C
C FIND THE POSITION OF T IN THE LOOK-UP TABLE
C
DO 1 1 2,9
J I
C          COUNTER VALUE AT EXIT
IF (T(K).LT.CTABLE(1,1)) GO TO 3
IF (T(K).GE.CTABLE(1,1)) GO TO 2
C IF NOT LOOP
1 CONTINUE
C
E -----
C THE VALUE IS OUT OF THE TOP OF THE TABLE
3 CONTINUE
C
C THE VALUE IS NOT FOUND WITHIN THE BOUNDS OF THE TABLE
E
C PRINT WARNING AND INTERPOLATE USING LAST 2 VALUES IN THE TABLE
WRITE(6,99) T(K)
99 FORMAT(/15,"THE TEMPERATURE VALUE ",F10.2," WAS NOT FOUND"
C " IN THE BOUNDS OF THE LOOK UP TABLE"
A " . THE THERMAL CONDUCTIVITY HAS BEEN"
D,/ 15," LINEARLY EXTRAPOLATED FROM EXISTING VALUES")
C
C CARRY ON INTO THE INTERPOLATION SECTION
E -----2-----
2 CONTINUE
C INTERPOLATE LINEARLY FOR K
C
CND(K) (T(K)-CTABLE(J-1,1))/(CTABLE(J,1)-CTABLE(J-1,1))
CND(K) CND(K)*(CTABLE(J,2)-CTABLE(J-1,2))+CTABLE(J-1,2)
CND(K) CND(K)+0.001
C
C THE THERMAL CONDUCTIVITY IS NOW INTERPOLATED
C CALCULATE THE NEXT VALUE
100 CONTINUE
E -----
C
C ALL FINISHED
C RETURN WITH INTERPOLATED VALUES
C
RETURN
END
SUBROUTINE TEMP(RT,I,N)
C S.I. UNITS ARE USED
E
C SUBROUTINE TO INTERPOLATE LINEARLY FOR
C TEMPERATURE FROM A GIVEN RESISTANCE THERMOMETER
C RESISTANCE.
C PRIMARILY INTENDED FOR USE WITH PRT'S.

```

RT CONTAINS THE RESISTANCE(S)  
 T CONTAINS THE INTERPOLATED TEMPERATURE(S)  
 N IS THE NUMBER OF DATA POINTS TO BE PROCESSED  
 RTTABLE CONTAINS THE LOOK UP TABLE

```

DIMENSION RT(N),T(N),RTTABLE(35,2)
DATA (RTTABLE(I,1),I=1,35)/
A 100.000,101.753,103.904,105.891,107.772,
A109.737,111.675,113.611,115.543,117.477,
B119.399,121.322,123.243,125.160,127.075,
C128.986,130.895,132.801,134.703,136.604,
D138.500,140.394,142.285,144.173,146.058,
E147.941,149.820,151.694,
F153.570,155.440,157.308,159.173,161.035,
G162.894,164.750/
DATA (RTTABLE(I,2),I=1,35)/
Z0.,5.,10.,15.,20.,25.,30.,35.,40.,45.,50.,55.,
E60.,65.,70.,75.,80.,85.,90.,95.,100.,
I105.,110.,115.,120.,125.,130.,135.,
G140.,145.,150.,155.,160.,165.,170.

```

```

-----1-----
C
C SET UP LOOP FOR NUMBER OF DATA POINTS
C
C DO 100 K=1,N
C
C FIND THE POSITION OF RT IN THE LOOK UP TABLE
C
C DO 1 I=1,35
C   J=I
C     COUNTER VALUE AT EXIT
C     IF(RT(K).LT.RTTABLE(I,1)) GO TO 3
C     IF(RT(K).GT.RTTABLE(I,1)) GO TO 2
C IF NOT -- LOOP
C   CONTINUE
C
C -----2-----
C THE VALUE IS OUT OF THE TOP OF THE TABLE
C   CONTINUE
C
C THE VALUE IS NOT FOUND WITHIN THE BOUNDS OF THE TABLE
C PRINT WARNING AND INTERPOLATE USING LAST 2 VALUES IN THE TABLE
C   WRITE(1,99) RT(K)
C   FORMAT(1,3,"THE RESISTANCE VALUE ",RT(K)," WAS NOT FOUND"
C     " IN THE BOUNDS OF THE LOOK UP TABLE. THE TEMPERATURE HAS BEEN"
C     " LINEARLY EXTRAPOLATED FROM EXISTING VALUES")
C
C CARRY ON INTO THE INTERPOLATION SECTION
C
C -----2-----

```



```

C
C ROUTINE TO CALCULATE THE REYNOLDS NUMBER FROM EITHER
C THE VELOCITY OR THE MASS FLOWRATE
C DENS IS THE DENSITY
C DIA IS THE PIPE DIAMETER
C VISCOS IS THE VISCOSITY
C VMASS IS THE CALCULATED MASS FLOW
C VEL IS THE CALCULATED VELOCITY
C REYNOLD IS THE CALCULATED REYNOLDS FROM THE ABOVE
C N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
C
C IF THE MASS FLOW IS ZERO THEN THE INDIVIDUAL VELOCITIES ARE SPECIFIED
C DIMENSION DENS(N),VEL(N),VISCOS(N),REYNOLD(N)
C DO 2 I=1,N
C IF(VEL(I).LE.1.E-4) GO TO 1
C -----
C REYNOLD(I)=DENS(I)*VEL(I)*DIA/VISCOS(I)
C GO TO 2
C -----
1 CONTINUE
C CHECK THAT THE MASS FLOW IS ALSO NOT ZERO
C IF(VMASS.LE.1.E-4) GO TO 3
C -----
2 REYNOLD(I)=4*VMASS/(1.1159*VISCOS(I)*DIA)
C GO TO 2
C -----
3 CONTINUE
C WRITE WARNING
C WRITE(6,400)
400 FORMAT(,'TS'," REYNOLDS CAN NOT BE CALCULATED AS BOTH"
A" THE MASSFLOW AND VELOCITY ARE ZERO."',/,'TS',
B" THE RE IS SET TO ZERO.")
C DO 6 J=1,N
6 REYNOLD(J)=0.0
C
2 CONTINUE
C RETURN
C END
C SUBROUTINE ROTA(PERCEN,VMASS,VEL,DEN,N)
C DIMENSION VEL(N),DEN(N)
C
C ROUTINE TO CALCULATE THE MASS FLOW RATE (VELOCITY) FROM THE ROTAMETER READ
C THE VARIATION IN DENSITY OF THE FLUID IS INCLUDED IN THE CORRECTION
C METHOD. THE VISCOSITY IS NOT INFLUENCING IF THE REYNOLDS NUMBER IS
C GREATER THAN 1.E7
C
C CHECK IF READING IS INSIDE CALIBRATION RANGE
C IF(PERCEN.GT.100.) GO TO 1
C IF(PERCEN.LE.10.) GO TO 1

```



```

3 CONTINUE
C CALCULATE THE MASS FLOW FROM THE CORRELATING EQN.
C
  VMASS1 = 0.015603+0.007762 * PERCLN
C CORRECT THIS FOR THE VARIATION IN DENSITY
C
  VMASS = VMASS1 * SQRT((8020. * DEN(N)) * DEN(N) / ((8020. * 999.) * 999.))
C DENSITY OF THE FLOAT IS 8020 KG/M3
C DENSITY OF H2O AT 16 IS 999 KG/M3
C
C CALCULATE THE VELOCITY IN THE PIPE AT EACH POINT
C DIAMETER IS 25.39 MM
  DIA = .02539
  DO 7 JJ=1,N
7 VEL(JJ) = 4. * VMASS / (DEN(JJ) * (3.14159 * DIA * DIA))
C
  GO TO 2
C ----- 1 -----
1 CONTINUE
C PRINT WARNING MESSAGE AND THEN CALCULATE FLOWRATE
C
  WRITE(6,400)
400 FORMAT(/,15," THE ROTAMETER READING IS OUT OF RANGE ",
  B" OF THE CALIBRATION."
  A,/, " THE MASS FLOW MUST BE CONSIDERED SUSPECT")
  GO TO 3
C -----
2 CONTINUE
  RETURN
  END
  SUBROUTINE RPLOT(X,Y,N)
  DIMENSION X(12),Y(12,1),ICHAR(10),RANGE(4),ITITLE(144)
  DIMENSION NINCH(2),MASK(2000)
C READ THE TITLE AND AXIS CHARS
  WRITE(6,8)
8 FORMAT(" TITLE ")
  READ(5,7) (ITITLE(I),I=1,72)
7 FORMAT(80A1)
  WRITE(6,9)
9 FORMAT(" XAXIS ")
  READ(5,7) (ITITLE(I),I=73,108)
  WRITE(6,6)
6 FORMAT(" YAXIS ")
  READ(5,7) (ITITLE(I),I=109,144)
C FIND MAX VALUES VALUES OF X AND Y
  XMAX = 1.E10
  YMAX = 1.E10
  DO 1 I=1,N
  IF(X(I).GE.XMAX) GOTO 1
  XMAX = X(I)
  IF(Y(I).GE.YMAX) GOTO 2
  YMAX = Y(I)
1 CONTINUE

```

```

C FIND MIN VALUES OF X AND Y
  XMIN=1.0E10
  YMIN=1.0E10
  DO 10 I=1,N
    IF(X(I).GE.XMIN) 6 TO 20
    XMIN=X(I)
20  IF(Y(I).GE.YMIN) GOTO 10
    YMIN=Y(I)
10  CONTINUE
    RANGE(1)=XMIN+0.1*(XMIN-XMAX)
    RANGE(2)=XMAX+0.1*(XMAX-XMIN)
    RANGE(3)=YMIN+0.1*(YMIN-YMAX)
    RANGE(4)=YMAX+0.1*(YMAX-YMIN)
    ICHAR(1)="+"
    IER=0
    NINCH(1)=20
    NINCH(2)=20
    CALL PLOTS(NINCH,"RICH",10)
    X(N+1)=RANGE(1)
    X(N+2)=(RANGE(2)+RANGE(1))/2
    Y(N+1)=RANGE(3)
    Y(N+2)=(RANGE(4)+RANGE(3))/2
    TITLE="MAMBA"
    NI=5
    TITLEX="XAXIS"
    NX=5
    TITLEY="YAXIS"
    NY=5
    WRITE(6,999)"N ",N
999  FORMAT(A10,15)
    WRITE(6,989)"X-VALUE ",X
989  FORMAT(A10,(4G16.7,3X))
    WRITE(6,989)"Y-VALUE ",Y
    CALL GENPLT(X,Y,N,3,1,0.0,1,TITLE,NT,TITLEX,NX,TITLEY,NY,
1      25.,15.,1,2.)
    CALL PLOT(0.,0.,999)
    RETURN
  END
  SUBROUTINE DENCW(T,DEN,N)
C .I. UNIT ARE USED
C
C  SUBROUTINE TO INTERPOLATE FOR
C  DENSITY FROM A GIVEN TEMPERATURE
C  USING CUBIC SPLINES
C  DATA IS FOR WATER
C
C  T CONTAINS THE TEMPERATURE(S)
C  DEN CONTAINS THE INTERPOLATED DENSITY(S)
C  N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
C  DTABLE CONTAINS THE LOOK UP TABLE
C
C  DIMENSION T(N),DEN(N),DTABLE(9,2),C(4,9)

```

```

C FIND MIN VALUES OF X AND Y
  XMIN=1.0E10
  YMIN=1.0E10
  DO 10 I=1,N
  IF(X(I).GE.XMIN) GOTO 20
  XMIN=X(I)
20 IF(Y(I).GE.YMIN) GOTO 10
  YMIN=Y(I)
10 CONTINUE
  RANGE(1)=XMIN+0.1*(XMIN-XMAX)
  RANGE(2)=XMAX+0.1*(XMAX-XMIN)
  RANGE(3)=YMIN+0.1*(YMIN-YMAX)
  RANGE(4)=YMAX+0.1*(YMAX-YMIN)
  ICHAR(1)="+"
  IER=0
  NINCH(1)=20
  NINCH(2)=20
  CALL PLOTS(NINCH,"RICH",10)
  X(N+1)=RANGE(1)
  X(N+2)=(RANGE(2)+RANGE(1))/25
  Y(N+1)=RANGE(3)
  Y(N+2)=(RANGE(4)+RANGE(3))/15
  TITLE="MAMBA"
  NT=5
  TITLX="XAXIS"
  NX=5
  TITLY="YAXIS"
  NY=5
  WRITE(6,999)"N ",N
999 FORMAT(A10,I5)
  WRITE(6,989)"X-VALUE",X
989 FORMAT(A10,(4B16.7,3X))
  WRITE(6,989)"Y-VALUE",Y
  CALL GENFIT(X,Y,N,3,1,0.0,1,TITLE,NT,TITLX,NX,TITLY,NY,
1      25.,15.,1,2.)
  CALL PLOT(0.,0.,999)
  RETURN
  END
  SUBROUTINE DENWCT(DEN,N)
C .1. UNITS ARE U/I
C
C SUBROUTINE TO INTERPOLATE FOR
C DENSITY FROM A GIVEN TEMPERATURE
C USING CUBIC SPLINES
C DATA IS FOR WATER
C
C I CONTAINS THE TEMPERATURE(S)
C DEN CONTAINS THE INTERPOLATED DENSITY(S)
C N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
C VTABLE CONTAINS THE LOG OF TABLE
C
  DIMENSION T(N),DEN(N),DTABLE(9,2),C(4,9)

```

```
DATA (DTABLI(1,1),1-1.9)
A 998.203,992.16,984.73,971.01,957.26,942.09,
B 867.304,799.361,717.001
DATA (DTABLI(1,1),1-1.9)
A 20.,40.,60.,80.,100.,
150.,200.,250.,300.,
```

```
INITIALIZE THE CONSTANTS FOR THE TABLE FIT
DATA FITCON(DTABLE 1,9)
```

```
1. THE LOOK UP NUMBER OF DATA POINTS
THE LOOK UP N
```

```
2. THE AMPLITUDE OF T IN THE LOOK UP TABLE
```

```
DO I = 1, N
  DO J = 1, 3
    COUNTED VALUE AT EXIT
    DTABLI(DTABLE(I,J), DTABLI(I,J))
  END DO
END DO
```

```
3. THE VALUE OF T OF THE TOP OF THE TABLE
CONTINUE
```

```
4. THE VALUE OF T WITHIN THE BOUNDS OF THE TABLE
```

```
5. THE WARNING AND INTERPOLATE USING LAST 3 VALUES IN THE TABLE
6. THE MESSAGE "THE TEMPERATURE VALUE '110.'" WAS NOT FOUND
IN THE BOUNDS OF THE LOOK UP TABLE. THE DENSITY HAS BEEN
EXTRAPOLATED FROM EXISTING VALUES"
```

```
7. THE INTERPOLATION SECTION
```

```
DO I = 1, N
  DO J = 1, 3
    DTABLI(DTABLE(I,J), DTABLI(I,J))
  END DO
END DO
```

```
8. THE DENSITY IS NOW INTERPOLATED
9. THE NEXT VALUE
10. TO 100
```

```
CONTINUE
```

```
11. THE DENSITY IS NOW INTERPOLATED USING A
12. THE MESSAGE
13. THE MESSAGE "THE TEMPERATURE VALUE '110.'" WAS NOT FOUND
IN THE BOUNDS OF THE LOOK UP TABLE. THE DENSITY HAS BEEN
EXTRAPOLATED FROM EXISTING VALUES"
```

```

DEN(K) = (DTABLE(J,1) - T(K)) * (1, J - 1) / (DTABLE(J,1) - T(K)) * *2
DEN(K) = DEN(K) + (T(K) - DTABLE(J,1)) * (1, J - 1)
A = (T(K) - DTABLE(J,1)) * *2
A = A + C(4, J - 1)

```

THE DENSITY HAS BEEN INTERPOLATED BY SPLINE  
CALCULATE THE NEXT VALUE

100 CONTINUE

ALL FINISHED  
RETURN WITH INTERPOLATED VALUES

RETURN  
END  
SUBROUTINE VISCW(T,VIS,H)  
S.I. UNITS ARE USED

SUBROUTINE IS INTENDED FOR  
VISCOSITY FROM A GIVEN TEMPERATURE  
DATA IS FOR WATER

T CONTAINS THE TEMPERATURE  
VIS CONTAINS THE INTERPOLATED VISCOSITY  
N IS THE NUMBER OF DATA POINTS TO BE PROCESSED  
VTABLE CONTAINS THE LOOK UP TABLE

```

DIMENSION T(N),VIS(N),VTABLE(2,2),C(4,7)
DATA (VTABLE(2,1),1-1.7)
A 12.816,13.384,13.845,12.719,12.539,17.106,11.8067
DATA (VTABLE(1,1),1-1.7)
A .00341,.00319,.003,.00283,.00268,
E .00236,.00217

```

CALCULATE THE CONSTANTS FOR THE SPLINE FIT  
CALL SPLICOR(VTABLE,C,7)

SET UP LOOP FOR NUMBER OF DATA POINTS

DO 100 I=1,N

FIND THE POSITION OF T IN THE LOOK UP TABLE

```

T(K) = 1.71773.154T(K)
DO 1 I=1,9

```

```

2) COUNTER VALUE AT EXIT
IF(T(K).GT.VTABLE(1,1)) GO TO 1
IF(T(K).GT.VTABLE(1,1)) GO TO 1

```

```

DEN(K) = (DTABLE(J,1) - T(K)) * C(3,J) + (DTABLE(J,2) - T(K)) * C(4,J)
DEN(K) = DEN(K) + (T(K) - DTABLE(J,1)) * C(3,J-1)
        + (T(K) - DTABLE(J-1,1)) * C(4,J-1)

```

THE DENSITY HAS BEEN INTERPOLATED BY SPLINES  
CALCULATE THE NEXT VALUE

100 CONTINUE

ALL FINISHED  
RETURN WITH INTERPOLATED VALUES

RETURN  
END  
SUBROUTINE VISCW(T,VIS,N)  
S.I. UNITS ARE USED

SUBROUTINE TO INTERPOLATE  
VISCOSITY FROM A GIVEN TEMPERATURE  
DATA IS FOR WATER

T CONTAINS THE TEMPERATURE(S)  
VIS CONTAINS THE INTERPOLATED VISCOSITY(S)  
N IS THE NUMBER OF DATA POINTS TO BE PROCESSED  
VTABLE CONTAINS THE LOOK UP TABLE

```

DIMENSION T(N),VIS(N),VTABLE(7,2)/C(3,2)
DATA (VTABLE(1,1),VTABLE(1,2))
A 13.816,13.382,13.245,12.769,12.539,12.306,11.976
DATA (VTABLE(1,1),VTABLE(1,2))
A .00341,.00319,.003,.00283,.00268,
B .00256,.00241

```

CALCULATE THE CONSTANTS FOR THE SPLINE III  
CALL SPLICON(VTABLE,C,7)

SET UP LOOP FOR NUMBER OF DATA POINTS

DO 100 I = 1,N

FIND THE POSITION OF T IN THE LOOK-UP TABLE

```

T(K) = 1. / (273.15 + T(K))
DO 1 I = 1,7

```

IF (T(K) .GT. VTABLE(1,1)) GO TO 2

IF (T(K) .GE. VTABLE(1,1)) GO TO 2



```

SUBROUTINE SPECHEAT(T,CF,N)
C S.I. UNITS ARE USED
C
C SUBROUTINE TO INTERPOLATE LINEARLY FOR
C SPECIFIC HEAT FROM A GIVEN TEMPERATURE
C DATA IS FOR WATER
C
C T CONTAINS THE TEMPERATURE(S)
C CF CONTAINS THE INTERPOLATED SPECIFIC HEAT(S)
C N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
C CFTABLE CONTAINS THE LOOK UP TABLE
C
C DIMENSION T(N),CF(N),CFTABLE(9,2)
C DATA (CFTABLE(I,2),I=1,9)/
C A 4183.,4179.,4175.,4198.,4219.,4310.,4510.,
C B 4870.,5650./
C DATA (CFTABLE(I,1),I=1,9)/
C A 20.,40.,60.,80.,100.,150.,200.,250.,300./
C -----1-----
C
C SET UP LOOP FOR NUMBER OF DATA POINTS
C
C DO 100 K 1,N
C
C FIND THE POSITION OF T IN THE LOOK UP TABLE
C
C DO 1 J 1, 9
C J I
C COUNTER VALUE AT EXIT
C IF(T(K).LT.CFTABLE(1,1)) GO TO 3
C IF(T(K).LE.CFTABLE(1,1)) GO TO 2
C IF NOT - LOOP
C 1 CONTINUE
C
C THE VALUE IS OUT OF THE TOP OF THE TABLE
C 3 CONTINUE
C
C THE VALUE IS NOT FOUND WITHIN THE BOUNDS OF THE TABLE
C
C PRINT WARNING AND INTERPOLATE USING LAST 2 VALUES IN THE TABLE
C WRITE(6,99) T(K)
99 FORMAT(/15,"THE TEMPERATURE VALUE ",T10.2," WAS NOT FOUND"
C " IN THE BOUNDS OF THE LOOK UP TABLE. THE SPECIFIC HEAT HAS BEEN"
C " LINEARLY EXTRAPOLATED FROM EXISTING VALUES")
C
C CARRY ON INTO THE INTERPOLATION SECTION
C
C 2 CONTINUE
C INTERPOLATE LINEARLY FOR CF
C
C CF(K) = (T(K) - CFTABLE(J-1,1)) / (CFTABLE(J,1) - CFTABLE(J-1,1))
C CF(K) = CF(K) * (CFTABLE(J,2) - CFTABLE(J-1,2)) + CFTABLE(J-1,2)

```



```

C
C THE SPECIFIC HEAT IS NOW INTERPOLATED
C CALCULATE THE NEXT VALUE
100 CONTINUE
C
C ALL FINISHED
C RETURN WITH INTERPOLATED VALUES
C
C RETURN
C END
C SUBROUTINE CONDW(T,END,N)
C S.I. UNITS ARE USED
C
C SUBROUTINE TO INTERPOLATE LINEARLY FOR
C THERMAL CONDUCTIVITY FROM A GIVEN TEMPERATURE
C DATA IS FOR WATER
C
C T CONTAINS THE TEMPERATURE(S)
C COND CONTAINS THE INTERPOLATED THERMAL CONDUCTIVITY(S)
C N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
C CTABLE CONTAINS THE LOOK UP TABLE
C
C DIMENSION T(N),COND(N),CTABLE(9,2)
C DATA (CTABLE(1,1),1-1,9)/
C 20.,40.,60.,80.,100.,120.,140.,160.,180./
C DATA (CTABLE(1,2),1-1,9)/
C .603,.637,.653,.670,.687,.697,.665,.616,.547
C
C SET UP LOOP FOR NUMBER OF DATA POINTS
C DO 100 K 1,N
C FIND THE POSITION OF T IN THE LOOK-UP TABLE
C DO 1 J=1,9
C J I
C COUNTER VALUE AT EXIT
C IF(T(K).LT.CTABLE(1,1)) GO TO 3
C IF(T(K).LE.CTABLE(1,1)) GO TO 2
C IF NOT LOOP
C CONTINUE
C
C THE VALUE IS OUT OF THE TOP OF THE TABLE
C CONTINUE
C
C THE VALUE IS NOT FOUND WITHIN THE BOUNDS OF THE TABLE
C
C PRINT WARNING AND INTERPOLATE USING LAST 2 VALUES IN THE TABLE
C WRITE(6,99) T(K)
C FORMAT(/15,"THE TEMPERATURE VALUE ".F10.2," WAS NOT FOUND"
C " IN THE BOUNDS OF THE LOOK UP TABLE"

```



```

DIMENSION RT0(10),PRC(10),CP(10),DT(10),TO(10),CPAVG(10),
A DTL(10),REAV(10),FRAV(10),T(10),HEAT(10),AU(10)
B ,ARU(10),ARUC(10),ARUC(10),FRU(10),CPAVG(10)
B ,RT0(10)

```

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 200

```

-----1-----
SET LOOP TO N 1
K N-1

CALCULATE THE HEAT BALANCE
DO 2 1-1,K
CPAVG(1)=(C(1+1)+C(1))/2
DT(1)=T(1+1)-T(1)
HEAT(1)=UMASS*CPAVG(1)*DT(1)
CONTINUE

```

```

-----2-----
CALCULATE THE LOG MEAN TEMP DIFF
WRITE(6,101)
FORMAT(/," WHAT IS THE INLET HEAT RESISTANCE")
101 A," READING")
READ(5,*)RT0(10)
WRITE(6,102)
104 FORMAT(/," WHAT ARE THE OUTLET HEAT")
A," RESISTANCES (9)")
READ(5,*)RT0(1),RT0(2),RT0(3),RT0(4),RT0(5),RT0(6),RT0(7),RT0(8),RT0(9)
INLET OIL 1 2 3 4 5 6 7 8 9 SHELL OIL

```

```

CALL TEMP(RT0,TO,N)
WRITE(6,106)
WRITE(6,105,101,102,103)
106 FORMAT(/," THE TEMPERATURES ARE (1-9 AND INLET)")
105 FORMAT(/,10(F8.2))
DO 3 1-1,K
T1=TO(10)-T(1+1)
T2=TO(1)-T(1)
DTL(1)=(T1-T2)/(ALOG(T1,T2))
CONTINUE

```

```

-----3-----
CALCULATE THE OVERALL HTC
DO 4 1-1,K

THE INSIDE AREA IS 0.07116 ; RECIPROCAL 14.0534
AU(1)=HEAT(1)*14.0534/DTL(1)
CONTINUE

```

```

-----4-----
CALCULATE THE AVERAGE RE ,FR AND CONDUCTIVITY
DO 5 1-1,K

```

```

REAV(I) = (RF(1+1)+RF(I))/2.
FRAV(I) = (FK(1+1)+FK(I))/2.
CNDAV(I) = (CND(1+1)+CND(I))/2.
CONTINUE

```

```

WRITE THIS ALL OUT

```

```

102 WRITE(6,102)
   HEAT LOSS      DELTA T      DELTA T PER      U"
   FORMAT(10.2,3X, REAVG, FRAVG, AVG CONDUCTIVITY",
   A "          REAVG          FRAVG          I          W/M2 K "
   B /, "          W          "          W/M K"
   C "
   DO 3 1=1,N
   WRITE(6,103)HEAT(I),D(I),U(I),AVG(I),REAV(I),FRAV(I),CNDAV(I)
   CONTINUE
103 FORMAT(3(F10.2,3X),3X,3(F10.2,3X),F10.4)

```

```

C CALCULATE THE INSIDE HTC FROM THE EQUATION OF EDE (1961)

```

```

998 WRITE(6,998)
   FORMAT(/" USING EQUATION OF EDE (1961)" )

```

```

DO 2 1=1,N
   ANU(I) = .026 + (REAV(I) + .8 * FRAV(I)) / D(I)
   AHU(I) = ANU(I) * CNDAV(I) * D(I)
   THE TUBE DIAMETER IS .5 CM
   ARU(I) = 1. / (1. / AHU(I) + 1. / ANU(I))
   CONTINUE

```

```

PRINT OUT THE RESULTS

```

```

109 WRITE(6,109)
   FORMAT(/, "          U          NU          HI
   " , "RESISTANCE OUTSIDE"
   B /, "          W/M2 K          W/M2 K          W/M2 K"
   C "
   DO 2 1=1,N
   WRITE(6,110)AU(I),ANU(I),AHU(I),ARU(I)
   CONTINUE
110 FORMAT(4(F10.2,3X))

```

```

C CALCULATE THE INSIDE HTC FROM THE EQUATION OF HAUSEN (1974)

```

```

999 WRITE(6,999)
   FORMAT(/" USING EQUATION OF HAUSEN (1974)" )

```

```

C XL = .906
   THE TUBE LENGTH IS 906 MM
   DO 2 1=1,N
   ANU(I) = .0235 + (REAV(I) + .8 * FRAV(I)) / D(I)
   AHU(I) = ANU(I) * CNDAV(I) * D(I)
   ARU(I) = 1. / (1. / AHU(I) + 1. / ANU(I))
76 CONTINUE

```

```

C PRINT OUT THE RESULTS
WRITE(6,177)
199 FORMAT(7," " U NU HI "
A,"RESISTANCE OUTSIDE " W/M2 K W/M2 K")
B," " W/M2 K
DO 80 I=1,N
WRITE(6,220)AU(I),ANU(I),AHU(I),ARU(I)
80 CONTINUE
C
220 FORMAT(F10.2,3X,F10.4,3X,F10.2,3X,F10.2,3X)
RETURN
END
SUBROUTINE HEATL(T,UMASS,CF,RE,FR,CND,DEN,VIS,N)

```

ROUTINE TO CALCULATE THE H.T.C.

DIMENSION RE(N),FR(N),CND(N),CF(N),DEN(N),VIS(N),T(N)

READ THE TUBE LENGTH AND THE WALL TEMPERATURE

```

WRITE(6,100)
100 FORMAT(7," GIVE THE WALL RESISTANCE THREE READINGS")
WRITE(2,*) "ENTER KEWATER/WM2K"
READ(5,*)RT0
WRITE(6,101)
101 FORMAT(7," GIVE THE TUBE LENGTH (M)")
READ(5,*)XL
CALL TEMP(RT0,TO,1)
WRITE(6,102)TO
102 FORMAT(7," THE WALL TEMPERATURE IS (14.2)

```

CALCULATE THE AVERAGE RESULTS

$$\begin{aligned}
CF_{AVG} &= (CF(1)+CF(2))/2. \\
CND_{AV} &= (CND(1)+CND(2))/2. \\
RE_{AV} &= (RE(1)+RE(2))/2. \\
FR_{AV} &= (FR(1)+FR(2))/2.
\end{aligned}$$

THE THERMAL CONDUCTIVITY OF BRASS IS TAKEN AT 50 DEG C AS 100.3 W/MH THE TUBE I.D. IS 12.47 MM

```

CALCULATE THE H.T.C. ASSUMING CONSTANT CF
HI=100.34-UMASS*CF_AVG*XL*(2.47+(X11*(2.47+3.14159*
*100.34-UMASS*CF_AVG*XL*(2.47+X11))
CALCULATE THE NU AND ST NUMBERS
AS11=HI/CF_AVG/(UMASS/0.0005)
ANU1=HI*0.025/CND_AV

```

```

C -----
C CALCULATE THE GRASHOF NUMBER
C BEIA IS 636 E-6
C
C GR1=(9.0*636.E-6*(XL**3)*(DENM2**2)**T(15 101.77018013)*21
C GR2=(9.0*636.E-6*(XL**3)*(DENM2**2)**T(15 101.77018013)*21
C GRAV=(GR1+GR2)*2
C
C -----
C WRITE THIS OUT
C WRITE(6,103)
C103 FORMAT(/,' HI ST BU KE FR SK")
C WRITE(6,104)HI1,AST1,ANU1,REAV,FRAV,GRAV
C104 FORMAT(6(E10.4))
C
C -----
C CALCULATE THE H.T.C ASSUMING CP(T)
C CP 0 =1804.52 L=3.26
C X2=(1804.52+3.56*T0)*ALOG((T(2)-T0)/(T(1)-T0))
C X3=3.56*(T(2)-T(1))*L
C HI2=(-1.)*(VMASS*100.34*X3/0.01247)/(2.*3.14159*100.34)
C XL VMASS*ALOG(15.88/12.62)*L
C AST2=HI2/CPAVG/(VMASS/0.0005)
C ANU2=HI2*0.025/CNDAV
C
C -----
C WRITE(6,105)
C WRITE(6,106)HI2,AST2,ANU2,REAV,FRAV,GRAV
C WRITE(6,107)T(1),T(2),VMASS,HI2,ANU2,AST2,REAV,FRAV,GRAV,
C105 READ(64,*)
C L XL
C106 WRITE(64,107) T(1),T(2),VMASS,HI2,ANU2,AST2,REAV,FRAV,GRAV,
C107 FORMAT(3(F6.3,1X),F6.4,1X,F6.2,1X,F6.2,1X,E10.4,1X,F6.1,
C 1 1X,F6.1,1X,E10.4,2X,F3.3)
C
C RETURN
C END

```

## b.132 Program 2

```

PROGRAM DGR(INPUT,IN,OUT,OUTPUT,TAPE5-IN,TAPE6-OUT,TAPE7
A,TAPE63,TAPE9,TAPE10,TAPE61-INPUT,TAPE62-OUTPUT,TAPE3)

```

```

PROGRAM TO PLOT AND CORRELATE THE DATA TAKEN FROM THE GREEN MAMBA
THE EXPERIMENTAL DATA IS ON TAPE7

```

```

DIMENSION TEM(200),TQM(200),W(200),WABS(200),H1(200),
  A(200),AST(200),DE(200),T(K200),GR(200),W(200),YR(200),
DIMENSION ANUC(200),D(200),M(200),X(200),Y(200),Z(200),NR(9)
A,GR(150)
COMMON X,Y,NR

```

```

C -----
C SET THE FLAGS FOR THE VARIOUS CASE
C
C REENTRY POINT
713= CONTINUE
  ICHI=0
  C IF ICHI=1 AND IAVG=1 THEN THE CHI2 TEST IS DONE
  EXTRA=0
  C IF EXTRA=1 THEN THE DATA OF SIEDER/TATE ETC IS INCLUDED
  IHAUSEN=0
  C IF IHAUSEN=1 THEN THE HAUSEN CORRELATION IS USED
  ICOLB=0
  C IF ICOLB=1 THEN THE COLBURN TYPE PLOT IS DONE
  IST=0
  C IF IST=1 THEN THE SIEDER TATE TYPE PLOT IS DONE
  IDEFEW=0
  C IF IDEFEW=1 THE A DEFEW/AUGUST TYPE PLOT IS DONE
  IDOUG=0
  C IF IDOUG=1 THEN MY CORRELATIONS ARE DONE
  IAVG=0
  C IF IAVG=1 THEN THE AVERAGE PERCT. ERROR IS CALCULATED
  IMET=0
  C IF IMET=1 THEN THE METZS ECKERT PLOT IS DONE
  ILENGTH=12
  C IF ILENGTH NOT = 0 THEN NUMBERED PLOT IS DONE FOR
  C THE SPECIFIED EXCHANGER LENGTH
  C SEE STATEMENTS 810-890 FOR CODES (1-9)
  C
  C IF ILENGTH = 12 THEN A PLOT OF ALL THE DATA IS MADE
  C
  C
  IKUZ=0
  C IF IKUZ=1 THEN THE KUZNETSOVA PLOT IS DONE
  INK=0
  C IF INK=1 THEN A NU GR PLOT IS DONE
  C
  ING=0
  C IF ING=1 THEN A NU GR PLOT IS DONE
  C

```

```

IFE 0
  IF IFE 1 THEN A 01  IF PLOT IS DONE

IRG 0
  IF IRG 1 THEN A 01  IF GR PLOT IS DONE

IRE 0
  IF IRE 1 THEN A 01  IF PLOT IS DONE

ISG 0
  IF ISG 1 THEN A 01  IF GR PLO. IS DONE

INF 0
  IF INF 1 THEN THE NU  IF PLOT IS DONE

IGR 0
  IF IGR 1 THEN THE NU - IF GR PLOT IS DONE

```

## READ IN SELECTION

```

732 READ(5,732)IM
    FORMAT(A3)
    IF (IM.EQ.3H5TO)STOP
    IF (IM.EQ.3HCON)GO TO 733
    IF (IM.EQ.3HICH)ICH=1
    IF (IM.EQ.3HI)IEXTRA=1
    IF (IM.EQ.3HIA)IHAVEIN=1
    IF (IM.EQ.3HIC)ICOLR=1
    IF (IM.EQ.3HIST)IST=1
    IF (IM.EQ.3HIDE)IDELEW=1
    IF (IM.EQ.3HIDO)IDOUNG=1
    IF (IM.EQ.3HIAV)IAVG=1
    IF (IM.EQ.3HIME)IMIT=1
    IF (IM.EQ.3HIKU)IKU2=1
    IF (IM.EQ.3HILE)IFAD(5,*)ILENGTH
    IF (IM.EQ.3HINR)INR=1
    IF (IM.EQ.3HJNG)JNG=1
    IF (IM.EQ.3HIF)IF=1
    IF (IM.EQ.3HIRG)IRG=1
    IF (IM.EQ.3HIRI)IRI=1
    IF (IM.EQ.3HISG)ISG=1
    IF (IM.EQ.3HINF)INF=1
    IF (IM.EQ.3HIGR)IGR=1
    GO TO

```

CONTINUE

```

READ IN THE DATA
REWIND 7

```



```

DO 699 I=1,9
699 NR(I) 0
N=0
DO 1 I=1,400
N=N+1
READ(7,*)TIN(I),TOUT(I),TW(I),VMASS(I),HI(I),ANU(I),AST(I),RE(I)
FR(I),GR(I),XL(I)
IF(EOF(7))2,1
1 CONTINUE
2 CONTINUE
-----
N=N-1
KKK=N
C IF IXTRA IS 1 READ IN THE DATA
C OF SIEDER/TATE NORRIS/SIMMS
IF(IXTRA.EQ.0) GO TO 89
KK=N+1
KKK=N+20
DO 88 I=KK,KKK
READ(5,*)VRAT(I),ANU(I),RE(I),FR(I),AL(I)
VRAT(I)=1./VRAT(I)
88 CONTINUE
89 CONTINUE
C
C
-----
C
C DETERMINE THE VISCOSITY RATIO
C
CALL VISRAT(VRAT,TIN,TOUT,VMASS,XL,HI,N)
VRAT IS V W/V B
C
C DETERMINE THE GREGORIG CORRECTION IF IXTRA DATA IS NOT PRESENT
C IF(IXTRA.EQ.0)CALL GRLG(FRAT,TIN,TOUT,VMASS,XL,HI,RE,N)
C
C
-----
C
C CALCULATE THE GRASHOF AND GRAETZ NO. OR FURTHER USE
C THE GRASHOF NUMBER BASED ON LENGTH IS GRL
C
DO 115 I=1,KKK
GZ(I)=.785*RE(I)*FR(I)*0.0254/XL(I)
GRL(I)=GR(I)
IF(IXTRA.EQ.0)GRL(I)=GR(I)*((1.639E-5/(XL(I)+.42))
115 CONTINUE
C
C WRITE THESE RESULTS TO THE OUTPUT FILE
C
C
C
C BRANCH ROUND THE FOLLOWING IF AVERAGE ERROR IS TO BE FOUND
IF(AVG.EQ.1)GO TO 789
C
C BRANCH ROUND IF A NU - RE PLOT IS TO BE DONE
IF(INR.EQ.1)GO TO 789

```

DO 77 I-1, RKR

ROUTINE TO DO CORELLATION

```

IF (IHAUSEN.EQ.1)
A RE(I) = 0.0184*(RE(I)**0.8)
B (1.8*FR(I)**0.3)
C (1.+(0.025./XL(I))*0.65)
D (VRAT(I)**(-0.14))
IF (IDFEW.EQ.1)
A RE(I) = 0.0235*(RE(I)**0.8)
B (FR(I)**0.33)
C (1.+(0.0254/XL(I))*0.65)
D (VRAT(I)**(-0.14))
IF (ICOLD.EQ.1) RE(I) = 24.00*(RE(I))
IF (IST.EQ.1) RE(I) = 1.488*(RE(I)**0.73)
A (FR(I)**0.33)
B (VRAT(I)**(-0.14))
IF (IDOOD.EQ.1)
A RE(I) = 0.0002*(RE(I)**1.42)
IF (IMET.EQ.1) ANU(I) = ALOG(RE(I)/VRAT(I))
IF (IMET.EQ.1) RE(I) = ALOG(ANU(I))
IF (INDZ.EQ.1) ANU(I) = ALOG(ANU(I))
IF (INDZ.EQ.1) RE(I) = ALOG(RE(I))
IF (INDZ.EQ.1) RE(I) = ALOG(RE(I)/VRAT(I))
A ((XL(I)/0.0254)**0.7)
IF (IFE.EQ.1) ANU(I) = AST(I)
IF (IFE.EQ.1) RE(I) = RE(I) - FR(I)
IF (IRG.EQ.1) ANU(I) = AST(I)
IF (IRG.EQ.1) RE(I) = RE(I) - GR(I)
IF (IRE.EQ.1) ANU(I) = AST(I)
IF (ISG.EQ.1) ANU(I) = AST(I)
IF (ISG.EQ.1) RE(I) = GR(I)
IF (INF.EQ.1) RE(I) = RE(I) - (I)
IF (IGR.EQ.1) RE(I) = RE(I) - GR(I)
CONTINUE

```

789

CONTINUE

DETERMINE THE AVERAGE PERCENTAGE ERROR  
 BETWEEN THE EQUATION AND THE DATA

```

IF (IAVG.EQ.0) GO TO 799
E2 = 0.
NN = 0

```

E2 IS THE AVERAGE PERCENTAGE ERROR

```

DO 799 I=1, NKK
IF (IHAUSEN.EQ.1) ANU(I) = 0.0235*(RE(I)**0.8)
A **0.33) - 0.0) + (1.+(0.0254/XL(I))*0.65)
IF (ICHI.EQ.1) GO TO 891

```

```

E1=100. * ((ANUC(I)-ANUC(I-1))/ANUC(I))
IF (RE(I).LE.1600. .AND. XL(I).GE.1.) E2=E2+ABS(E1)
IF (RE(I).LE.1600. .AND. XL(I).GE.1.) NN=NN+1
GO TO 799

```

C

391

```

CONTINUE
E1=((ANUC(I)-ANUC(I-1))/ANUC(I))
IF (RE(I).LE.1600. .AND. XL(I).GE.1.) E2=E2+ABS(E1)
IF (RE(I).LE.1600. .AND. XL(I).GE.1.) NN=NN+1

```

C

799

```

CONTINUE
IF (ICHI.EQ.1) GO TO 892
E2=E2/NN
THE AVERAGE PERCENTAGE ERROR IS FINISHED
STOP

```

C

892

```

CONTINUE
WRITE(6,*) "CH12", E2, "POINTS", NN
STOP

```

C

799

CONTINUE

C

```

IF (ILENGTH.EQ.12) GO TO 65

```

```

DO 10 I=1,N

```

810

```

IF (XL(I).EQ.0.906) CALL SETSUB(RE, ANU, 1, I)

```

820

```

IF (XL(I).EQ.1.012) CALL SETSUB(RE, ANU, 2, I)

```

830

```

IF (XL(I).EQ.2.718) CALL SETSUB(RE, ANU, 3, I)

```

840

```

IF (XL(I).EQ.1.424) CALL SETSUB(RE, ANU, 4, I)

```

850

```

IF (XL(I).EQ.4.236) CALL SETSUB(RE, ANU, 5, I)

```

860

```

IF (XL(I).EQ.7.456) CALL SETSUB(RE, ANU, 6, I)

```

870

```

IF (XL(I).EQ.6.342) CALL SETSUB(RE, ANU, 7, I)

```

880

```

IF (XL(I).EQ.7.254) CALL SETSUB(RE, ANU, 8, I)

```

890

```

IF (XL(I).EQ.8.157) CALL SETSUB(RE, ANU, 9, I)

```

10

CONTINUE

66

CONTINUE

C

C

```

NOW DO THE PLOTTING OF THE DATA

```

C

```

CALL RPLOT(KKK, ILENGTH, RE, ANU)
GO TO 7132

```

END

```

SUBROUTINE SETSUB(RE, ANU, L, I)

```

C

```

SUBROUTINE TO SET UP 2-DIMENSIONAL ARRAY CONTAINING ALL DATA
FOR PLOTTING SUBROUTINE, SORTED ACCORDING TO L/D RATIO.

```

C

```

DIMENSION NR(1), X(100), Y(100), RE(100), ANU(100)

```

```

COMMON X, Y, NR

```

```

NR(L)=NR(L)+1

```

```

X(NR(L), L)=RE(I)

```

```

Y(NR(L), L)=ANU(I)

```

```

RETURN

```

END

```

SUBROUTINE RPLOT (N, ILENGTH, RE, ANU)

```

C

C

```

ROUTINE TO PLOT THE DATA

```

```

COMMON X,Y,NR
DIMENSION X(200,9),Y(200,9),ICHAR(10),ANU(10),NINCH(2),
1 X1(200),X2(200),X3(200),X4(200),X5(200),X6(200),X7(200),
1 X8(200),X9(200),Y1(200),Y2(200),Y3(200),Y4(200),Y5(200),
1 Y6(200),Y7(200),Y8(200),Y9(200),XX(10),YY(10),NR(9),
1 RE(400),ANU(400),A(25),B(25)
EQUIVALENCE (X(1,1),X1(1)),(X(1,2),X2(1)),(X(1,3),X3(1)),
1 (X(1,4),X4(1)),(X(1,5),X5(1)),(X(1,6),X6(1)),
1 (X(1,7),X7(1)),(X(1,8),X8(1)),(X(1,9),X9(1)),
1 (Y(1,1),Y1(1)),(Y(1,2),Y2(1)),(Y(1,3),Y3(1)),
1 (Y(1,4),Y4(1)),(Y(1,5),Y5(1)),(Y(1,6),Y6(1)),
1 (Y(1,7),Y7(1)),(Y(1,8),Y8(1)),(Y(1,9),Y9(1))
C -----
C READ THE TITLE AND AXIS CHARS
TITLE=" "
TITLEX=" "
TITLEY=" "
C -----
C SCALE AXES FOR ALL PLOTS
C
CALL SCALE(RE,15.,N,1)
CALL SCALE(ANU,10.,N,1)
XX(1)-XX(2) RE(N+1)
XX(3)-RE(N+2)
YY(1) YY(2) ANU(N+1)
YY(3)-ANU(N+2)
C SET UP THE CALLING ROUTINES
NT=10
NX=10
NY=10
X1(NR(1)+1)=X2(NR(1)+1)=X3(NR(3)+1)=X4(NR(4)+1)=X5(NR(5)+1)=XX(1)
X6(NR(6)+1)=X7(NR(7)+1)=X8(NR(8)+1)=X9(NR(9)+1)=XX(1)
X1(NR(1)+2)=X2(NR(2)+2)=X3(NR(3)+2)=X4(NR(4)+2)=X5(NR(5)+2)=XX(2)
X6(NR(6)+2)=X7(NR(7)+2)=X8(NR(8)+2)=X9(NR(9)+2)=XX(2)
Y1(NR(1)+1)=Y2(NR(2)+1)=Y3(NR(3)+1)=Y4(NR(4)+1)=Y5(NR(5)+1)=YY(1)
Y6(NR(6)+1)=Y7(NR(7)+1)=Y8(NR(8)+1)=Y9(NR(9)+1)=YY(1)
Y1(NR(1)+2)=Y2(NR(2)+2)=Y3(NR(3)+2)=Y4(NR(4)+2)=Y5(NR(5)+2)=YY(2)
Y6(NR(6)+2)=Y7(NR(7)+2)=Y8(NR(8)+2)=Y9(NR(9)+2)=YY(2)
C -----
C
CALL PLOTS(NINCH,"RICH",10)
IF (ILENGTH.RE.EQ.50.40)
50 IF (ILENGTH.NE.12) CALL NUMPLOT(ILENGTH,XX,YY,RE,ANU,N)
IF (ILENGTH.EQ.12) CALL NUMPLOT(ILENGTH,XX,YY,RE,ANU,N)
GOTO 99
60 CONTINUE
CALL GENPLOT(XX,YY,1,3,1,0,0,0,TITLE,NT,TITLEX,NX,TITLEY,NY,
15.,10.,1,0.)
IF (NR(1).NE.0) CALL GENPLOT(X1,Y1,NR(1),1,1,0,0,0)
IF (NR(2).NE.0) CALL GENPLOT(X2,Y2,NR(2),2,1,0,0,0)
IF (NR(3).NE.0) CALL GENPLOT(X3,Y3,NR(3),3,1,0,0,0)
IF (NR(4).NE.0) CALL GENPLOT(X4,Y4,NR(4),4,1,0,0,0)
IF (NR(5).NE.0) CALL GENPLOT(X5,Y5,NR(5),5,1,0,0,0)
IF (NR(6).NE.0) CALL GENPLOT(X6,Y6,NR(6),6,1,0,0,0)

```

```

IF (NR(7).NE.0) CALL GENPTI(X7,Y7,NR(7),7,1.0,0.0)
IF (NR(8).NE.0) CALL GENPTI(X8,Y8,NR(8),8,-1.0,0.0)
IF (NR(9).NE.0) CALL GENPTI(X9,Y9,NR(9),9,-1.0,0.0)
GOTO 99
99 CALL PLOT(0.,0.,999)
-----
RETURN
END
SUBROUTINE NUMFIT (IL,XX,YY,RE,ANU,N)
-----
C THIS SUBROUTINE PLOTS THE DATAPOINTS OF A DATASET WITH SPECIFIED L/D
C RATIO USING NUMBERS TO IDENTIFY THE POINTS.
C THE DATA IS PRINTED IN THE SAME SEQUENCE IN FILE "OUT".
C NOTE: SYMBOLS ARE NOT SELF CENTERED
C-----
DIMENSION X(200,?),Y(200,?),XX(3),YY(3),NR(?)
C ,ANU(400),RE(400)
COMMON X,Y,NR
20 WRITE (6,30) IL
30 FORMAT (//,19,"----- NUMFIT : L/D RATIO CODE ",IL," -----",
117,"POINT NO SYMBOL VALUE VALUE")
PNUM=0.
M=NR(IL)
CALL PLOT(1.0,1.0,-3)
CALL FACTOR(6)
CALL AXISE(0.0,0.0,108,,-10,15,-0.,,12),XX(3))
CALL AXISE(0.0,0.0,64,6,10.,90.,YY(2),YY(3))
IF(1L.EQ.12)GO TO 100
GO TO 101
100 DO 60 K=1,N
XX(1)=(RE(K)-XX(2))/XX(3)
YY(1)=(ANU(K)-YY(2))/YY(3)
PNUM=PNUM+1.
CALL NUMBER(XX(1),YY(1),.14,PNUM,0.,-1)
60 CONTINUE
GO TO 99
101 CONTINUE
DO 50 L=1,M
XX(1)=(X(L,IL)-XX(2))/XX(3)
YY(1)=(Y(L,IL)-YY(2))/YY(3)
PNUM=PNUM+1.
WRITE (6,40) L,1,X(L,IL),Y(L,IL)
40 FORMAT (,10,I3,T19,I3,T28,E10.,,141,E10.)
CALL NUMBER(XX(1),YY(1),.14,PNUM,0.,17)
50 CONTINUE
99 RETURN
END
SUBROUTINE VISC(T,VIS,N)
C S.I. UNITS ARE USED
C
C SUBROUTINE TO INTERPOLATE FOR
C VISCOSITY FROM A GIVEN TEMPERATURE
C DATA IS FOR REGAL OIL B
C

```

```

T CONTAINS THE TEMPERATURE(S)
VIS CONTAINS THE INTERPOLATED VISCOSITY(S)
N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
VTABLE CONTAINS THE LOOK UP TABLE

DIMENSION T(N),VIS(N),VTABLE(1,2),VTABLE(1,1)
DATA (VTABLE(1,2),VTABLE(1,1))
4.626,3.57,2.753,2.135,1.656,.802,.239/
DATA (VTABLE(1,1),VTABLE(1,2))
1.00341,.00319,.003,.00283,.00268,
8.00236, .10211
-----
CALCULATE THE CONSTANTS FOR THE SILLINE FIT
CALL SPLICON(VTABLE,C,2)

SET UP LOOP FOR NUMBER OF DATA POINTS
DO 100 K=1,N

FIND THE POSITION OF T IN THE LOOK UP TABLE
T(K)=1./(273.15+T(K))
DO 1 I=2,9
  J=I
  C COUNTER VALUE AT EXIT
  IF(T(K).GT.VTABLE(I,1)) J=I-1
  IF(T(K).GE.VTABLE(I,1)) GO TO 2
C IF NOT IN LOOP
  CONTINUE
1
2
C THE VALUE IS OUT OF THE TOP OF THE TABLE
  CONTINUE
3
C THE VALUE IS NOT FOUND WITHIN THE BOUNDS OF THE TABLE
C
C PRINT WARNING AND INTERPOLATE USING LAST 2 VALUES IN THE TABLE
  WRITE(6,99) T(K),T(K)-273.15
  WRITE(6,99) T(K)
99 FORMAT(15,"THE TEMPERATURE VALUE ",F10.2," WAS NOT FOUND"
  " IN THE BOUNDS OF THE LOOK UP TABLE. THE VISCOSITY HAS BEEN"
  " LINEARLY EXTRAPOLATED FROM EXISTING VALUES")
CARRY ON INTO THE INTERPOLATION SECTION
INTERPOLATE LINEARLY FOR T
  VIS(K)=VTABLE(J,2)+VTABLE(J,1)-VTABLE(J,1,1)
  VIS(K)-VIS(K)+VTABLE(J,2)-VTABLE(J,1,2)+VTABLE(J,1,2)
  VIS(K)-EXP(VIS(K))-0.001
  T(K)=1./T(K)-273.15
C THE VISCOSITY IS NOW INTERPOLATED
C CALCULATE THE NEXT VALUE
  GO TO 100

```

```

2      CONTINUE
C
C      THE VISCOSITY IS NOW INTERPOLATED USING A
C      CUBIC SPLINE ROUTINE
C      SUBROUTINE SPLICON CALCULATES THE INTERPOLATION CONSTANTS
L
      VIS(K) = (VTABLE(J,1) - T(K)) + (C(1,J-1) * (VTABLE(J,1) - T(K)) + 2
Z      + C(3,J-1))
      VIS(K) = VIS(K) + (T(K) - VTABLE(J-1,1)) * (C(2,J-1) * (T(K) - VTABLE(J-1,1)
Z      )) + 2 * C(1,J-1)
      VIS(K) = EXP(VIS(K)) * 0.001
      T(K) = 1./T(K) - 273.15
C
C      THE VISCOSITY HAS BEEN INTERPOLATED BY SPLINES
C      CALCULATE THE NEXT VALUE
C
C
100     CONTINUE
C-----1-----
C
C      ALL FINISHED
C      RETURN WITH INTERPOLATED VALUES
C
      RETURN
      END
      SUBROUTINE SPECIF(T,CP,N)
C      S.I. UNITS ARE USED
C
C      SUBROUTINE TO INTERPOLATE LINEARLY FOR
C      SPECIFIC HEAT FROM A GIVEN TEMPERATURE
C      DATA IS FOR REGAL OIL B
C
C      T CONTAINS THE TEMPERATURE(S)
C      CP CONTAINS THE INTERPOLATED SPECIFIC HEAT(S)
C      N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
C      CPTABLE CONTAINS THE LOOK UP TABLE
C
      DIMENSION T(N),CP(N),CPTABLE(9,2)
      DATA (CPTABLE(I,2),I=1,9)/
A 1871.5,1946.02,2018.04,2093.4,2160.39,
B 2343.77,2522.55,2702.58,2890.99/
      DATA (CPTABLE(I,1),I=1,9)/
A 20.,40.,60.,80.,100.,150.,200.,250.,300./
C-----1-----
C
C      SET UP LOOP FOR NUMBER OF DATA POINTS
C
      DO 100 K=1,N
C
C      FIND THE POSITION OF T IN THE LOOK-UP TABLE
C
      DO 1 I=2,9
      J=I

```

```

C           COUNTER VALUE AT EXIT
C           IF(T(K).LT.CPTABLE(1,1)) GO TO 3
C           IF(T(K).LE.CPTABLE(I,1)) GO TO 2
C IF NOT - LOOP
1           CONTINUE
C
C -----
C THE VALUE IS OUT OF THE TOP OF THE TABLE
3           CONTINUE
C
C THE VALUE IS NOT FOUND WITHIN THE BOUNDS OF THE TABLE
C
C PRINT WARNING AND INTERPOLATE USING LAST 2 VALUES IN THE TABLE
C           WRITE(*,99) T(K)
99          FORMAT(/15,"THE TEMPERATURE VALUE ",F10.2," WAS NOT FOUND
C " IN THE BOUND OF THE LOOK UP TABLE. THE SPECIFIC HEAT HAS
C D,/,T5," LINEARLY EXTRAPOLATED FROM EXISTING VALUES")
C
C CARRY ON INTO THE INTERPOLATION SECTION
C -----2-----
C
2           CONTINUE
C INTERPOLATE LINEARLY FOR CP
C
C           CP(K)=(T(K)-CPTABLE(J-1,1))/(CPTABLE(J,1)-CPTABLE(J-1,1))
C           CP(K)=CP(K)*(CPTABLE(J,2)-CPTABLE(J-1,2))+CPTABLE(J-1,2)
C
C THE SPECIFIC HEAT IS NOW INTERPOLATED
C CALCULATE THE NEXT VALUE
100        CONTINUE
C -----1-----
C
C ALL FINISHED
C RETURN WITH INTERPOLATED VALUES
C
C           RETURN
C           END
C           SUBROUTINE VISRAT(VRAT,TIN,TOUT,UMASS,XL,HI,N)
C
C           SUBROUTINE TO CALCULATE THE VISCOSITY RATIO
C           DEFINED AS VIS WALL/ VIS BULK
C           WHERE THE BULK VISCOSITY IS ARITHMETIC MEAN OF
C           THE VISCOSITY AT THE INLET AND OUTLET.
C
C           DIMENSION VRAT(N),TIN(N),TOUT(N),UMASS(N),XL(N)
C           A ,HI(N),T(3),V(3)
C           DO 1 I=1,N
C
C           FIND THE AVERAGE SPECIFIC HEAT OF THE OIL
C           T(1)=TIN(I)
C           T(2)=TOUT(I)
C           CALL SPECIF(T,V,2)
C           CPAV=(V(1)+V(2))/2.

```



```

C      CALCULATE THE INSIDE WALL TEMPERATURE
C
      TW1 VMASS(1) CFAV*(TOUT(I)-TIN(I))/(3.14*0.0254*
A XL(I)*HI(I))+(TOUT(I)+TIN(I))/2.
C
C      CALCULATE THE VISCOSITIES
C
      T(3)=TW1
      CALL ISC(T,V,3)
      VB=(V(1)+V(2))/2.
      VRAT(I)=V(3)/VB
1     CONTINUE
C
C     ALL CALCULATIONS DONE
C     RETURN
C     END
C     SUBROUTINE SFLICON(TABLE,C,M)
C
C     SUBROUTINE TO CALCULATE THE CONSTANTS FOR THE
C     SFLINE I I
C     REF: FENNINGTON ,R.H
C           INTRODUCTORY COMPUTER METHODS AND NUMERICAL
C           ANALYSIS.  MAC MILLAN 1965
C
      DIMENSION TABLE(M,2),C(4,M),D(10),F(10),E(10),
A A(10,3),B(10),Z(10),X(10),Y(10)
      DO 100 I=1,M
      X(I)=TABLE(I,1)
      Y(I)=TABLE(I,2)
100    CONTINUE
      MM=M-1
      DO 2 K=1,MM
      D(K)=X(K+1)-X(K)
      F(K)=D(K)/6.
2     E(K)=(Y(K+1)-Y(K))/D(K)
      DO 3 K=2,MM
3     B(K)=E(K)*F(K-1)
      A(1,2)=1.-D(1)/D(2)
      A(1,3)=D(1)/D(2)
      A(2,3)=F(2)-F(1)*A(1,3)
      A(2,2)=2.*(F(1)+F(2))-F(1)*A(1,2)
      A(2,3)=A(2,3)/A(2,2)
      B(2)=B(2)/A(2,2)
      DO 4 K=3,MM
      A(K,2)=2.*(F(K-1)+F(K))-F(K-1)*A(K-1,3)
      B(K)=B(K)-F(K-1)*B(K-1)
      A(K,3)=F(K)/A(K,2)
4     B(K)=B(K)/A(K,2)
      Q=D(M-2)/D(M-1)
      A(M,1)=1.+Q/A(M-2,3)
      A(M,2)=Q-A(M,1)*A(M-1,3)
      B(M)=B(M-2)-A(M,1)*B(M-1)
      Z(M)=B(M)/A(M,2)
      MN=M-2

```

```

DO 6 I=1,MN
K=M-I
6 Z(K)=B(K)-A(K,3)*Z(K+1)
Z(1)=A(1,2)*Z(2)-A(1,3)*Z(3)
DO 7 K=1,MN
Q=1./(6.*D(K))
C(1,K)=Z(K)*Q
C(2,K)=Z(K+1)*Q
C(3,K)=Y(K)/D(K)-Z(K)*P(K)
7 C(4,K)=Y(K+1)/D(K)-Z(K+1)*P(K)
RETURN
END
SUBROUTINE COND(T,CND,N)
C S.I. UNIST ARE USED
C
C SUBROUTINE TO INTERPOLATE LINEARLY FOR
C THERMAL CONDUCTIVITY FROM A GIVEN TEMPERATURE
C DATA IS FOR REGAL OIL B
C
C T CONTAINS THE TEMPERATURE(S)
C CND CONTAINS THE INTERPOLATED THERMAL CONDUCTIVITY(S)
C N IS THE NUMBER OF DATA POINTS TO BE PROCESSED
C CTABLE CONTAINS THE LOOK-UP TABLE
C
C DIMENSION T(N),CND(N),CTABLE(9,2)
C DATA (CTABLE(I,1),I=1,9)/
C A 20., 0.,60.,80.,100.,150.,200.,250.,300./
C DATA (CTABLE(I,2),I=1,9)/
C A 132.36,131.048,129.66,128.19,126.9,123.
C B ,119.42,115.19,112.36/
-----1-----
C
C SET UP LOOP FOR NUMBER OF DATA POINTS
C
C DO 100 K=1,N
C
C FIND THE POSITION OF T IN THE LOOK UP TABLE
C
C DO 1 I=2,9
C J=I
C COUNTER VALUE AT EXIT
C IF(T(K).LT.CTABLE(I,1)) GO TO 3
C IF(T(K).LE.CTABLE(I,1)) GO TO 2
C IF NOT - LOOP
C 1 CONTINUE
C
C -----2-----
C THE VALUE IS OUT OF THE TOP OF THE TABLE
C 3 CONTINUE
C
C THE VALUE IS NOT FOUND WITHIN THE BOUNDS OF THE TABLE.
C
C PRINT WARNING AND INTERPOLATE USING LAST 2 VALUES IN THE TABLE

```

```

WRITE(6,99) T(K)
99  FORMAT(/15,"THE TEMPERATURE VALUE ",F10.2," WAS NOT FOUND"
C " IN THE BOUNDS OF THE LOOK UP TABLE"
A " THE THERMAL CONDUCTIVITY HAS BEEN"
D,F15," LINEARLY EXTRAPOLATED FROM EXISTING VALUES")
C
C CARRY ON INTO THE INTERPOLATION SECTION
C-----2-----
C
C CONTINUE
C INTERPOLATE LINEARLY FOR K
C
CND(K)=(T(K)-CTABLE(J,1))/(CTABLE(J,1)-CTABLE(J-1,1))
CND(K)=CND(K)*(CTABLE(J,2)-CTABLE(J-1,2))+CTABLE(J-1,2)
CND(K)=CND(K)*0.001
C
C THE THERMAL CONDUCTIVITY IS NOW INTERPOLATED
C CALCULATE THE NEXT VALUE
100 CONTINUE
C-----1-----
C
C ALL FINISHED
C RETURN WITH INTERPOLATED VALUES
C
C RETURN
C END
SUBROUTINE GREG(FRAT,TIN,TOUI,UMASS,XL,HI,RE,N)
C
C ROUTINE TO CALCULATE THE CORRECTION FACTOR OF GREGORIG
C
C DIMENSION RE(N),FRAT(N),TIN(N),TOUI(N),UMASS(N),XL(N)
C ,HI(N),T(3),CF(3),AK(3),V(3)
C DO 1 I=1,N
C
C FIND THE AVERAGE SPECIFIC HEAT OF THE OIL
C T(1)=TIN(I)
C T(2)=TOUI(I)
C CALL SPECIF(T,V,2)
C CFAV=(V(1)+V(2))/2.
C
C CALCULATE THE INSIDE WALL TEMPERATURE
C
C TWI=UMASS(I)*CFAV*(TOUI(I)-TIN(I))/(3.14*0.0254*
C A XL(I)*HI(I)+(TOUI(I)+TIN(I))/2.
C
C T(3)=TWI
C T(1)=(TIN(I)+TOUI(I))/2.
C T(2)=(T(1)+T(3))/2.
C CALL SPECIF(T,CF,3)
C CALL VISI(T,V,3)
C CALL COND(T,AK,3)
C FRI=CF(2)*V(2)/AK(2)
C FRW=CF(3)*V(3)/AK(3)

```

```

FRE=C1(1)*V(1)/AK(1)
XIFK=(PRF-PRW)/,PRB-PRW) 0.5
DENM=(PRB**0.05)*(RI(1)**0.02)*((0.0 XIFR)**0.01)
XNUM=((0.5 XIFK)**0.2)*0.1A+0.0757
XI=XNUM/DENM
FRAT(1)=(PRP/PRW)**XI
CONTINUE

```

ALL CALCULATIONS DONE

RETURN

END

SUBROUTINE REG(X,Y,N)

DIMENSION X(120),Y(120)

XBAR=0.

YBAR=0.

XX=0.

YY=0.

XY=0.

DO 1 I=1 N

XX=XX+X(I)\*X(I)

YY=YY+Y(I)\*Y(I)

XY=XY+X(I)\*Y(I)

XBAR=XBAR+X(I)

YBAR=YBAR+Y(I)

CONTINUE

WRITE(6,\*)N," XX",XX," YY",YY

WRITE(6,\*)" XBAR",XBAR," YBAR",YBAR

IFLOAT(N)

XBAR=XBAR/I

YBAR=YBAR/I

SX=XX-C\*XBAR\*XBAR

SY=YY-C\*YBAR\*YBAR

SXY=XY-C\*XBAR\*YBAR

WRITE(6,\*)" SX",SX," SY",SY," SXY",SXY

R=SXY/SQRT(SX\*SY)

A=SXY/SX

I=YBAR-A\*XI/AI

WRITE(6,\*)" R ",R," A ",A," I ",I," E "

STOP

END

```

PRB = (L1 + V1) / ZAN111
X11 = (V1 + RM) / (K1 + RW) - 0.45
DENM = (PRB + 0.05) + (K1 + RW) + 0.02 + (V1 + 0.01)
XNUM = (0.5 - X11) * 0.21 + 0.04 + 0.07
XF = XNUM / DENM
FRAT(1) = (FR1 + FR2) * XF
CONTINUE

```

1  
2  
3

ALL CALCULATIONS DONE

```

RETURN
END
SUBROUTINE REBIA, Y, NI
DIMENSION X(170), Y(170)
XBAR = 0.
YBAR = 0.
XX = 0.
YY = 0.
XY = 0.
DO I = 1, N
  XX = XX + X(I) * X(I)
  YY = YY + Y(I) * Y(I)
  XY = XY + X(I) * Y(I)
  XBAR = XBAR + X(I)
  YBAR = YBAR + Y(I)
CONTINUE
WRITE (6, 0) N, " XBAR = ", XBAR, " YBAR = ", YBAR
WRITE (6, 0) " XBAR**2 = ", XBAR**2, " YBAR**2 = ", YBAR**2
CALL CQ1(N)
XBAR = XBAR / N
YBAR = YBAR / N
EXBAR = XX - N * XBAR**2
EYBAR = YY - N * YBAR**2
EXY = XY - N * XBAR * YBAR
WK1 = (6.0) * SX * SY * (BY - 0.5) * BX * 0.5
R = EXY / WRT(SX**2 + SY**2)
A = EXY / SX
B = YBAR - A * XBAR
WRITE (6, 0) " R = ", R, " A = ", A, " B = ", B
STOP
END

```

4

6.14 PHYSICAL PROPERTIES6.14.1 Regal oil B (R and O)TABLE 9 Physical properties

$\theta$ °C	C J/kgK	$\lambda$ W/mK $\times 10^3$	$\eta$ kg/ms $\times 10^3$	$\rho$ kg/m <sup>3</sup>	Pr
20	1871,5	132,36	102,1	872,5	1443,64
40	1946,02	131,048	35,53	862,1	527,61
60	2018,04	129,66	15,69	848,0	244,2
80	2093,40	128,19	8,46	837,8	138,13
100	2160,39	126,90	5,24	825,0	89,21
150	2343,77	123,00	2,23	799,8	42,49
200	2522,55	119,42	1,27	772,5	26,82
250	2702,58	115,91		750,1	
300	2890,99	112,36		727,1	

$$C_o = 1804,52 \text{ J/kgK}, \quad b = 3,56 \text{ J/kgK}^2$$

$$\beta = \frac{1}{V} \left[ \frac{\partial V}{\partial T} \right]_p = 636 \times 10^{-6} / ^\circ\text{C}$$

6.14.2 Water

Physical properties for water were taken from "Thermodynamic and Transport Properties of Fluids", Y R Mayhew and G F C Rogers, Oxford, (1973).

6.15 FIGURES

- Figure 1 Schematic layout of the experimental rig  
Figure 2 Assembled section of the heat exchanger  
Figure 3 Item 2 of Figure 2  
Figure 4 Item 3 of Figure 2  
Figure 5 Item 4 of Figure 2  
Figure 6 Item 5 of Figure 2  
Figure 7 Item 6 of Figure 2  
Figure 8 Item 7 of Figure 2  
Figure 9 Connecting piece  
Figure 10 Item 1 of Figure 9  
Figure 11 Item 2 of Figure 9  
Figure 12 Item 3 of Figure 9  
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Figure 16 Manometer  
Figure 17 Static mixer  
Figure 18 Temperature probe for water temperature in the inlet header  
Figure 19 Detail of Figure 18  
Figure 20 Detail of Figure 18  
Figure 21 Data plotted against the Hausen equation  
Figure 22 Metais and Eckert plot of the data  
Figure 23 Data plotted against the Colburn type equation  
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Figure 25 Mixed turbulent equation  
Figure 26 Upper transitional equation  
Figure 27 Middle transitional equation  
Figure 28 Mixed laminar equation  
Figure 29 Lower transitional data as a function of the Stanton and Reynolds groups  
Figure 30 Lower transitional equation  
Figure 31 Hausen equation compared with derived equations

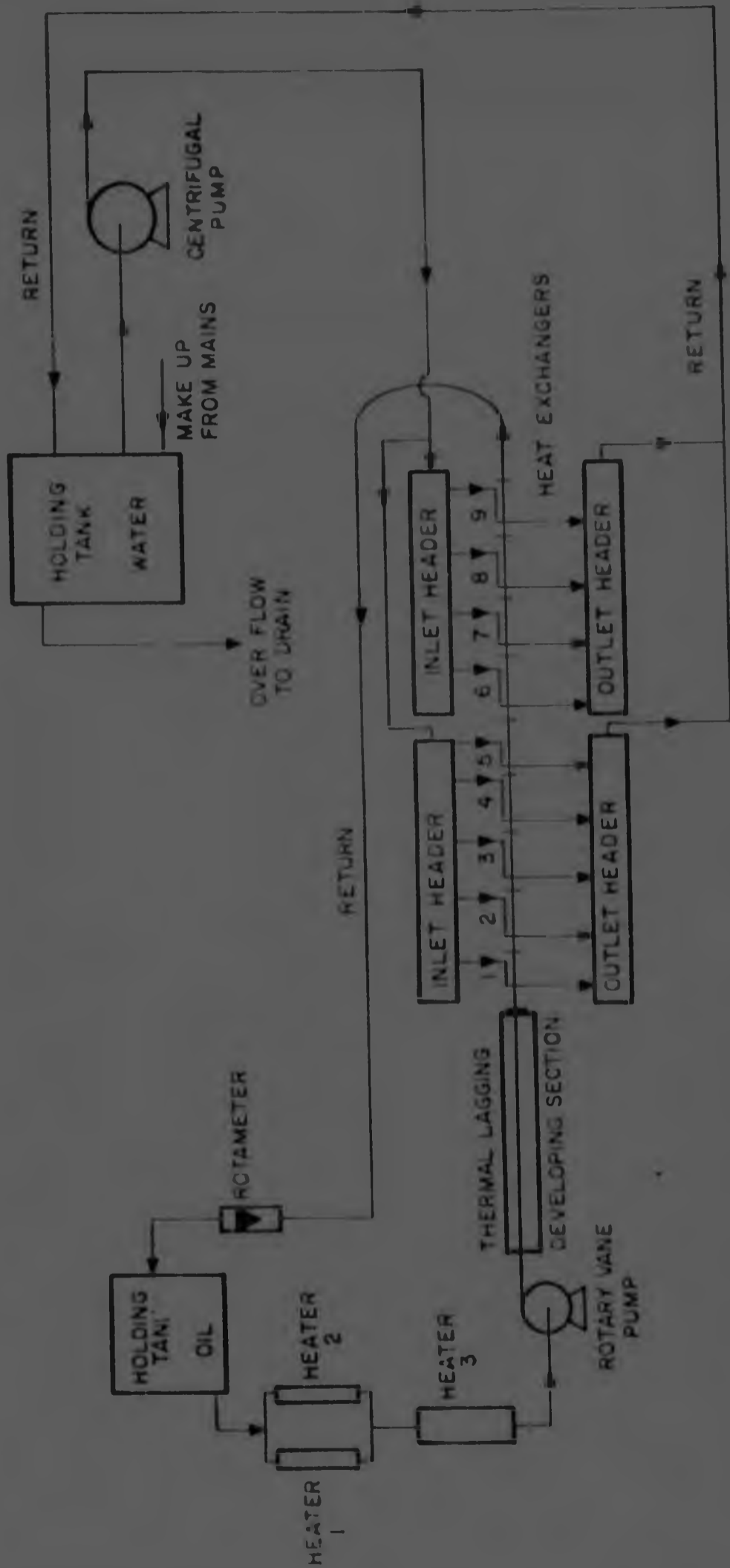


FIGURE 1 Schematic layout of the experimental rig



FIGURE 2 Assembled section of the heat exchanger

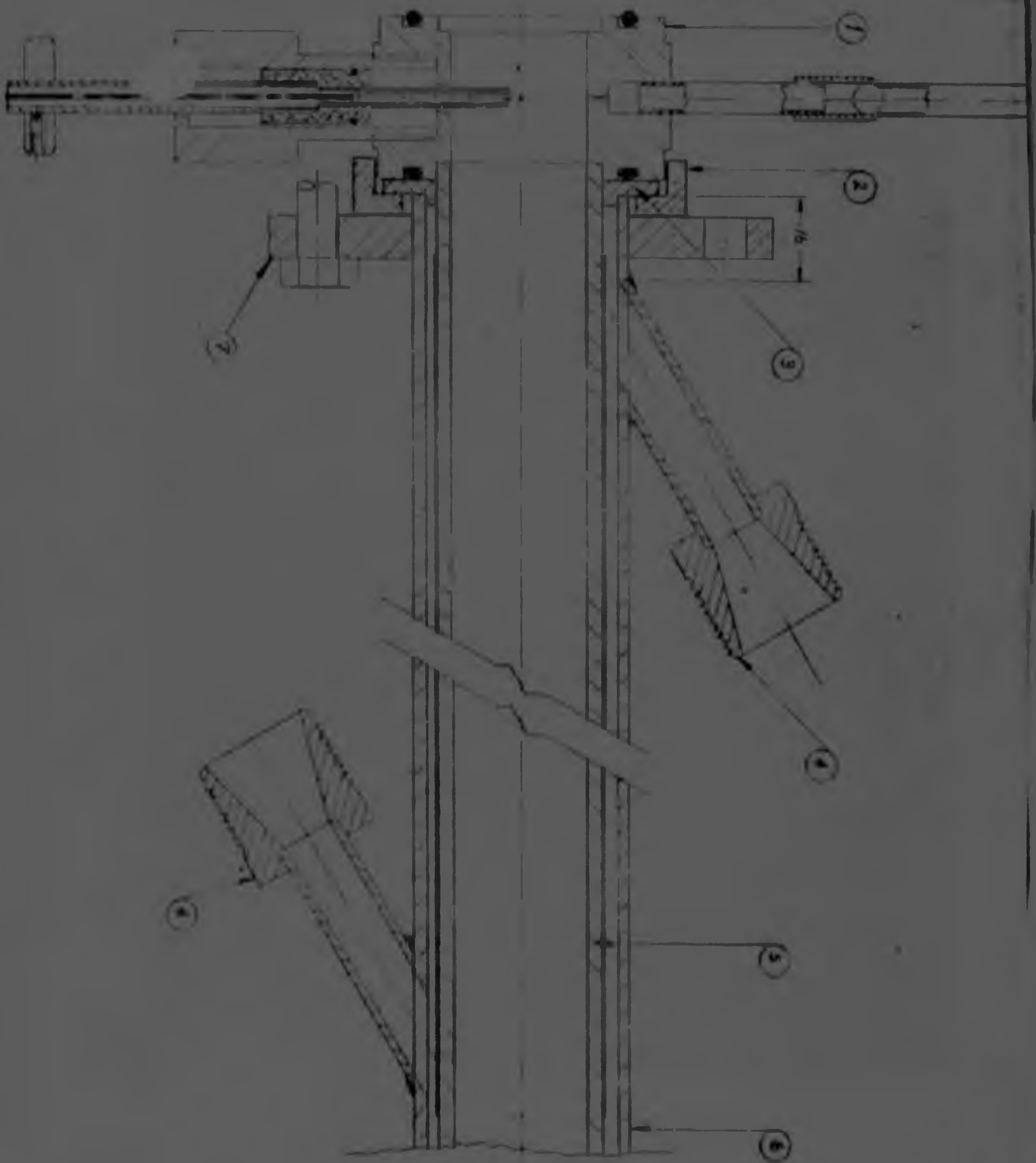


Figure 3 (cont.) of Figure 2

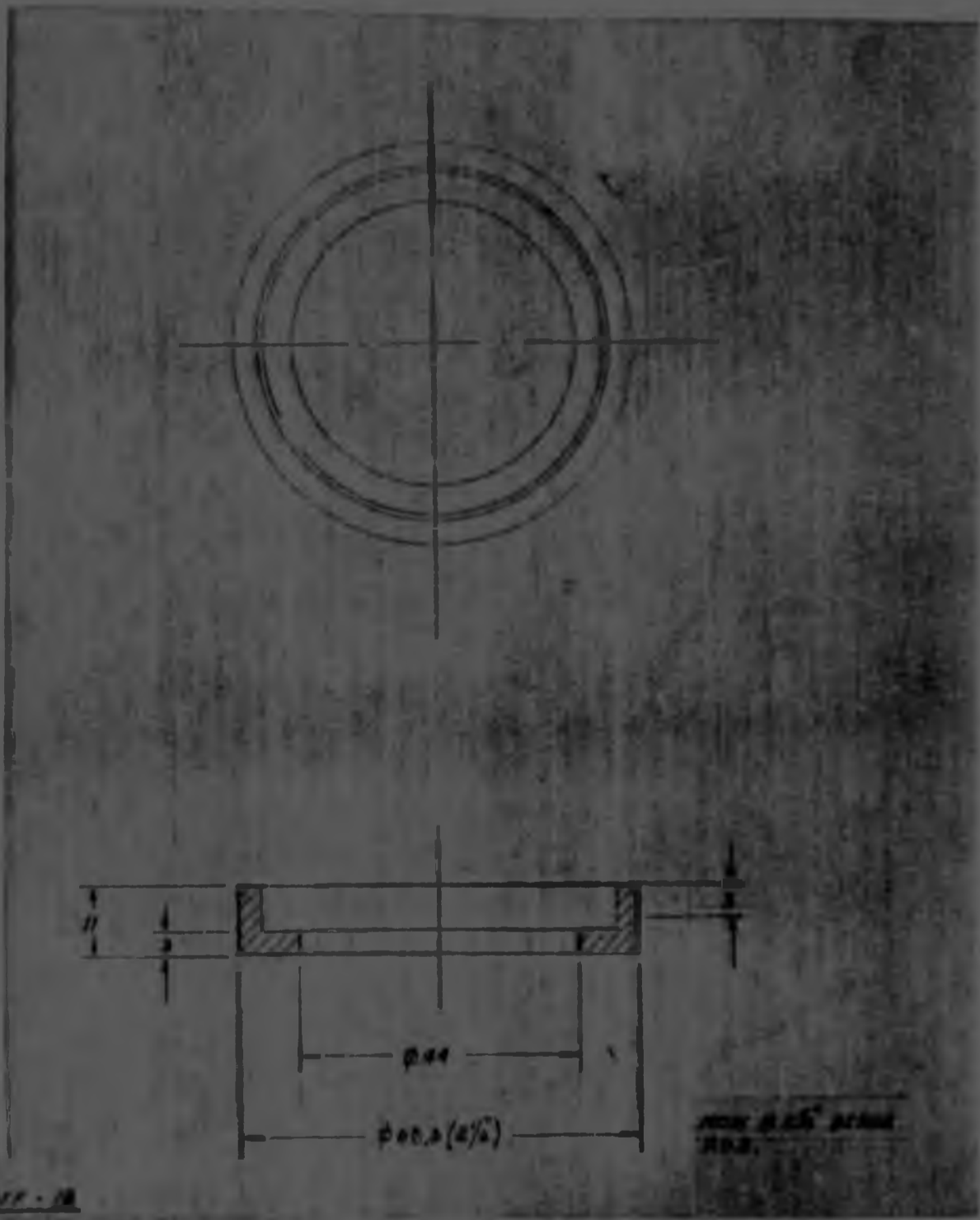
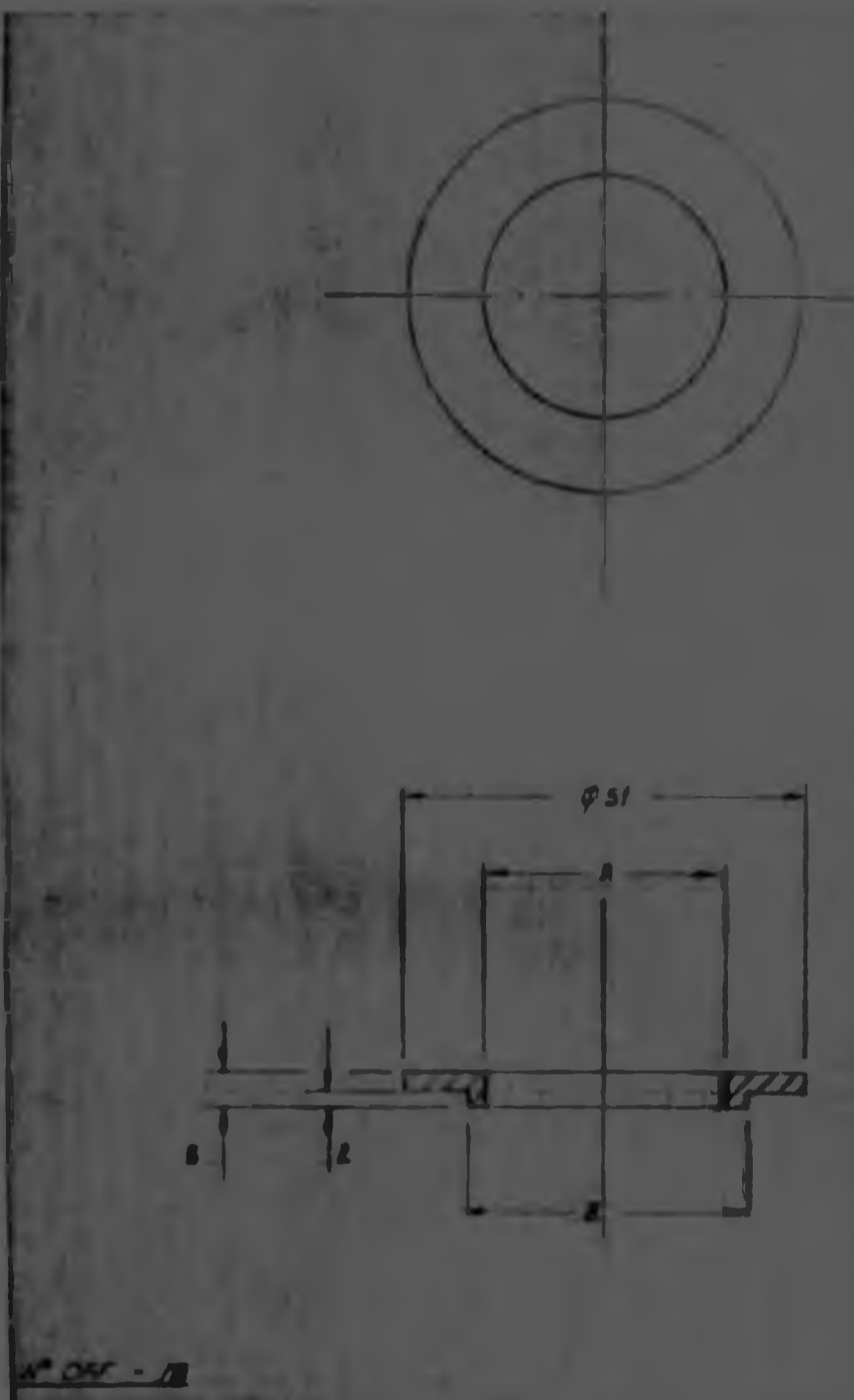


FIGURE 4 Item 3 of Figure 2

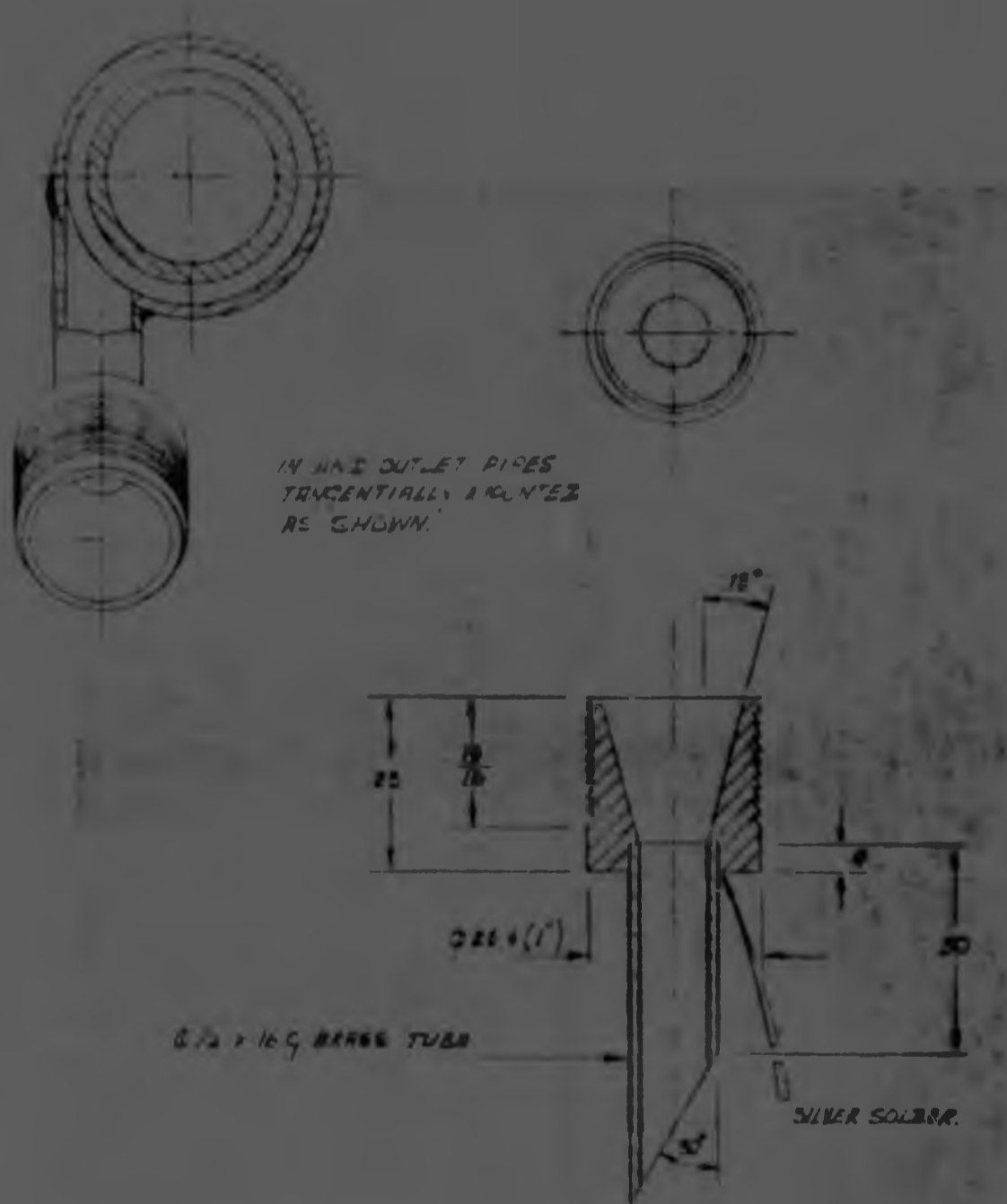


A. TO SUIT O.D. OF  $1\frac{1}{2}$  C.D.  
2 10 G BRASS TUBING.  
ALLOW 0,127 mm  
(0,005") FOR SOLDER JOINT.

B. TO SUIT I.D. OF  $1\frac{1}{2}$  C.D.  
x 12 G STEEL CYCLE FRAME.

AS SHOWN -

FIGURE 5 Item 4 of Figure 2



AC OFF 18

FIGURE 6    Item 5 of Figure 2



1/2" O.D. x 104  
BRASS TUBING

2" DIA. 1/2"

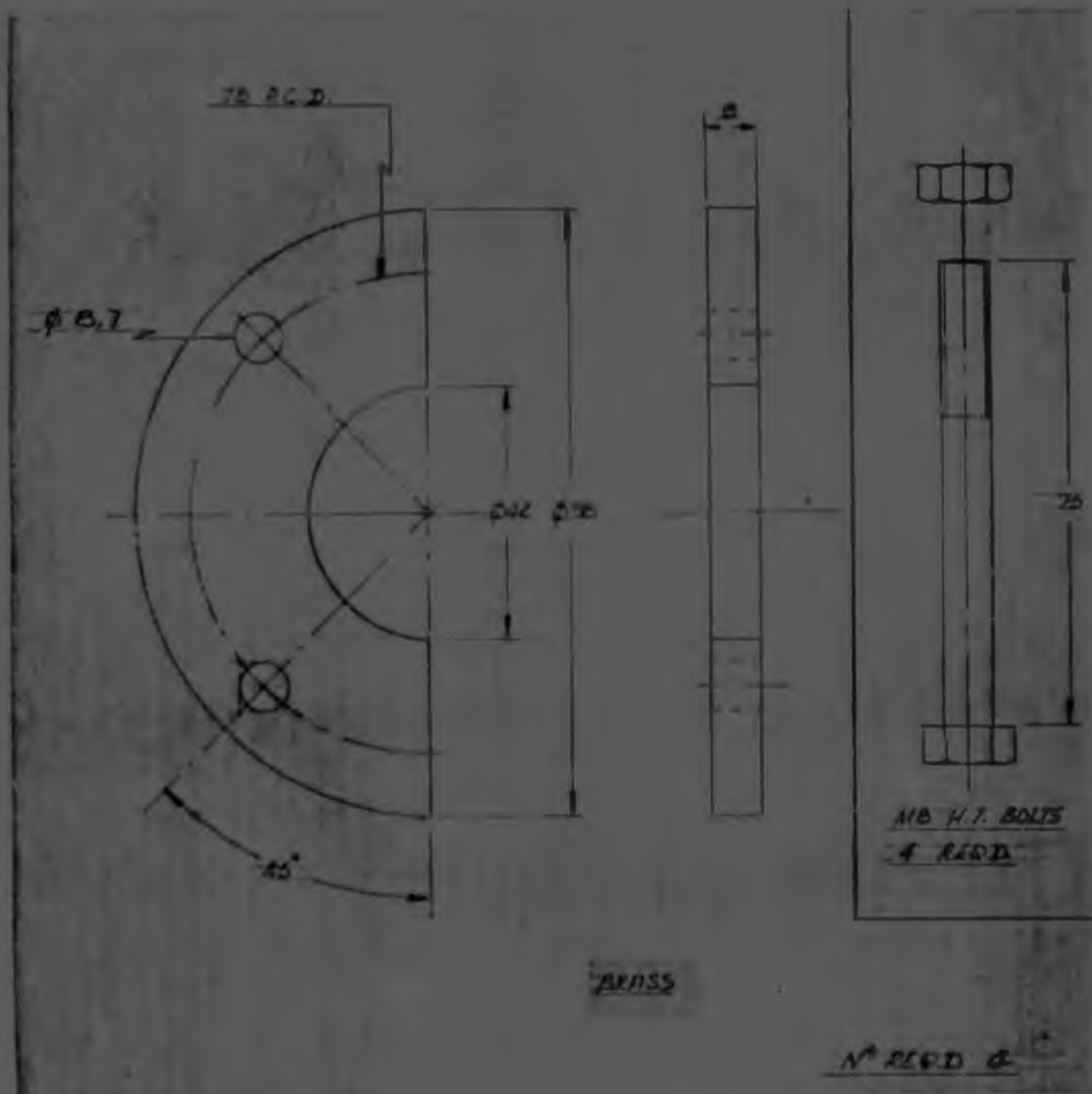
FIGURE 7    Item 6 of Figure 2



190 C.D. x 129 STEEL  
CYCLE FERRI TUB.

10 DIA. 2

FIGURE 8 Item 1 of Figure 2



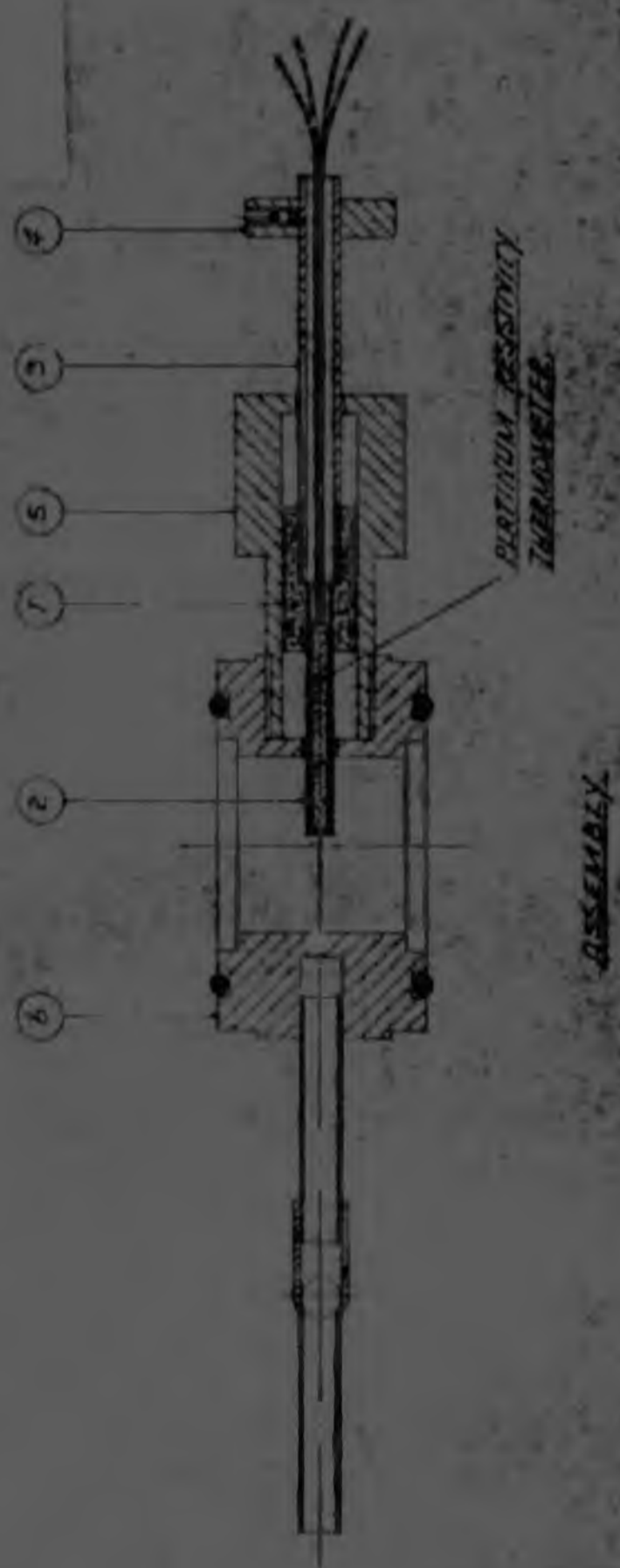


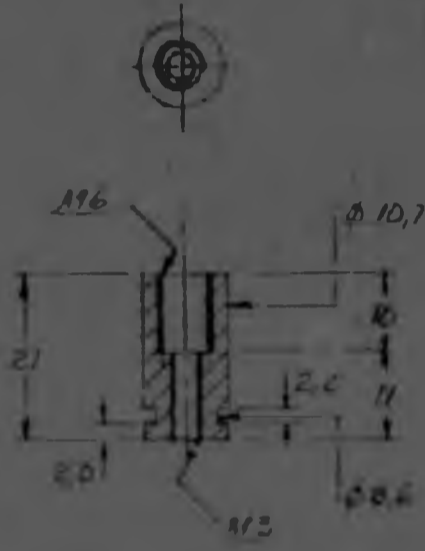
FIGURE 9 Connecting piece



FIGURE 10

Item 1 of Figure 9

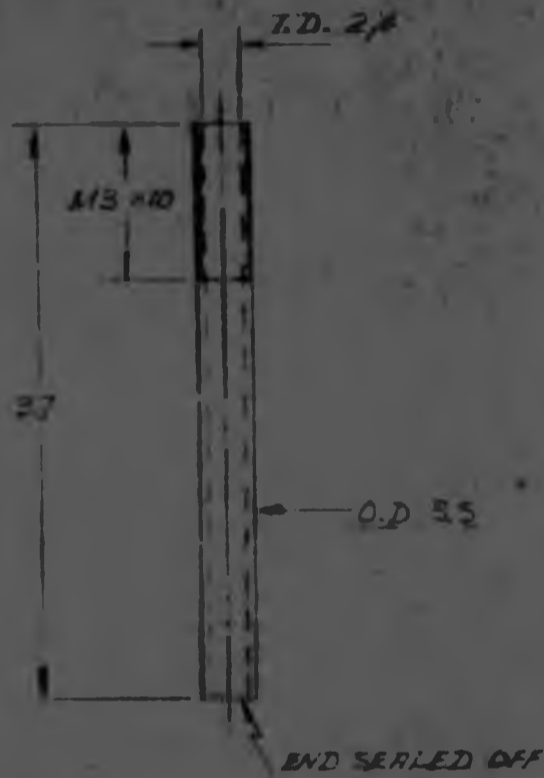
CALCO O RING  
N° RM 0081-16



PART N° 1 OF ITEM 1  
MAT. PHENOLIC RESIN  
PAPER BASE.

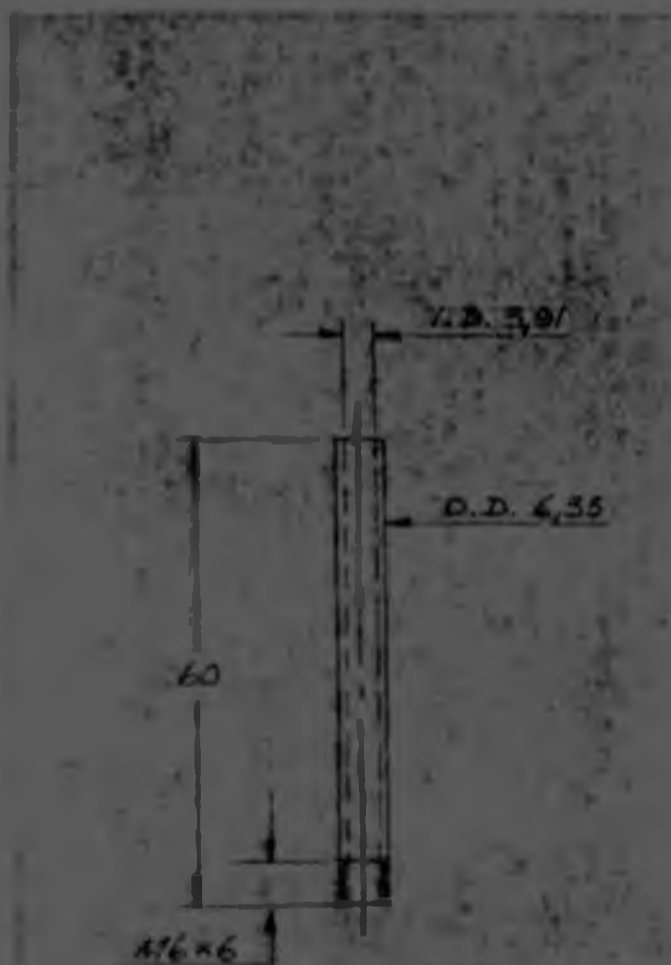
SCALE 1:1

FIGURE 11      Item 2 of Figure 9



PART N° 2 OF ITEM 1  
MHT. COPPER  
SCALE 2:1

FIGURE 12 Item 3 of Figure D



PART N° 3 OF ITEM 1

MAT. BRASS

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FIGURE 13

Item 4 of Figure 9

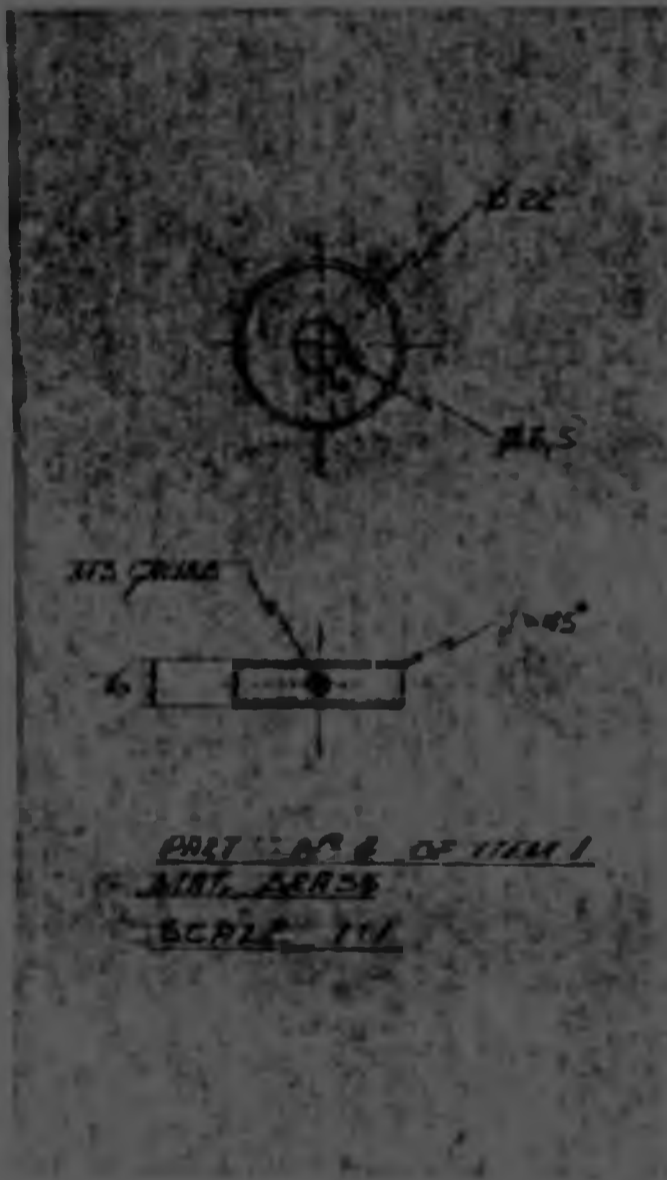
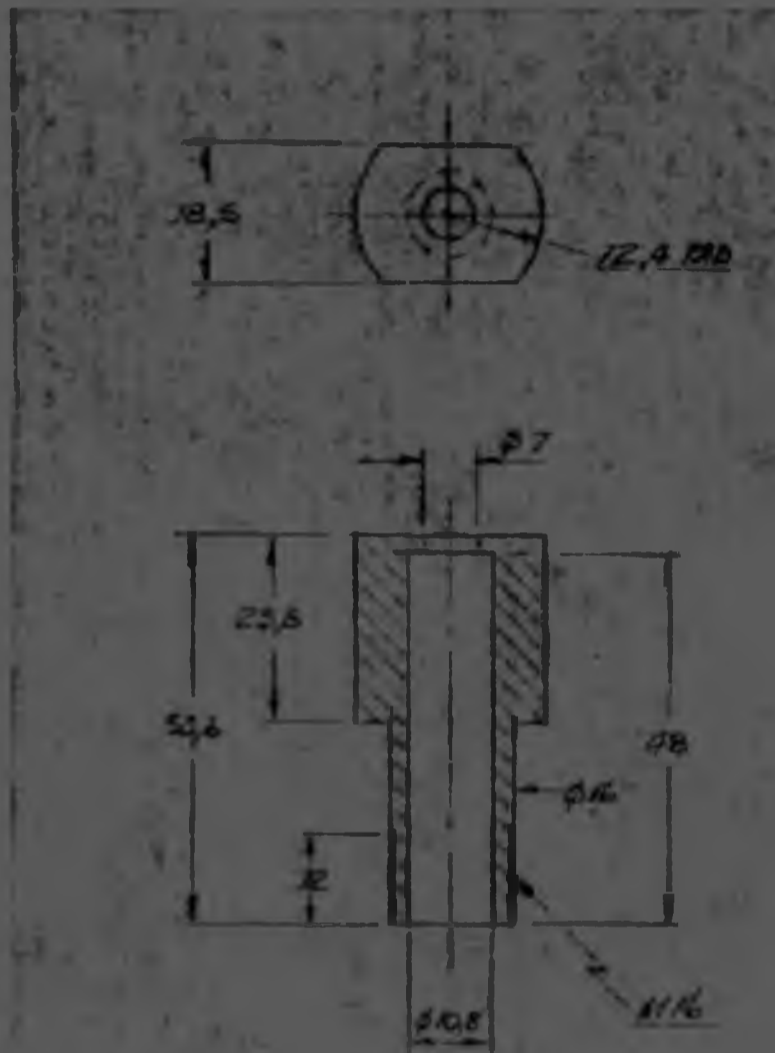
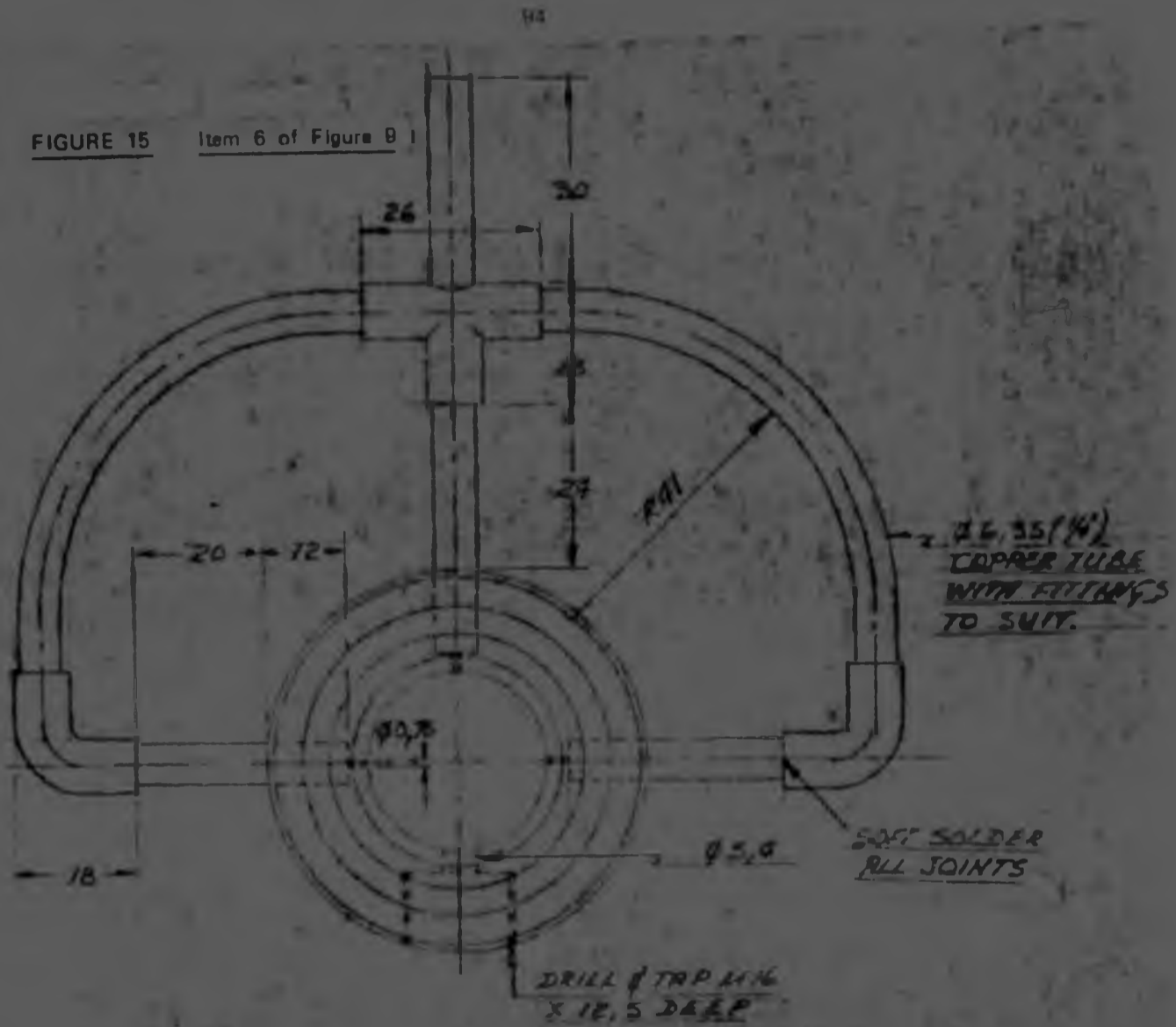


FIGURE 14 Item b of Figure 11



PART N° 5 OF ITEM 1  
MAT BRASS  
SCALE 1/1

FIGURE 15 Item 6 of Figure 9



2x GACO O RINGS N° FM 0395-50  
REDD.

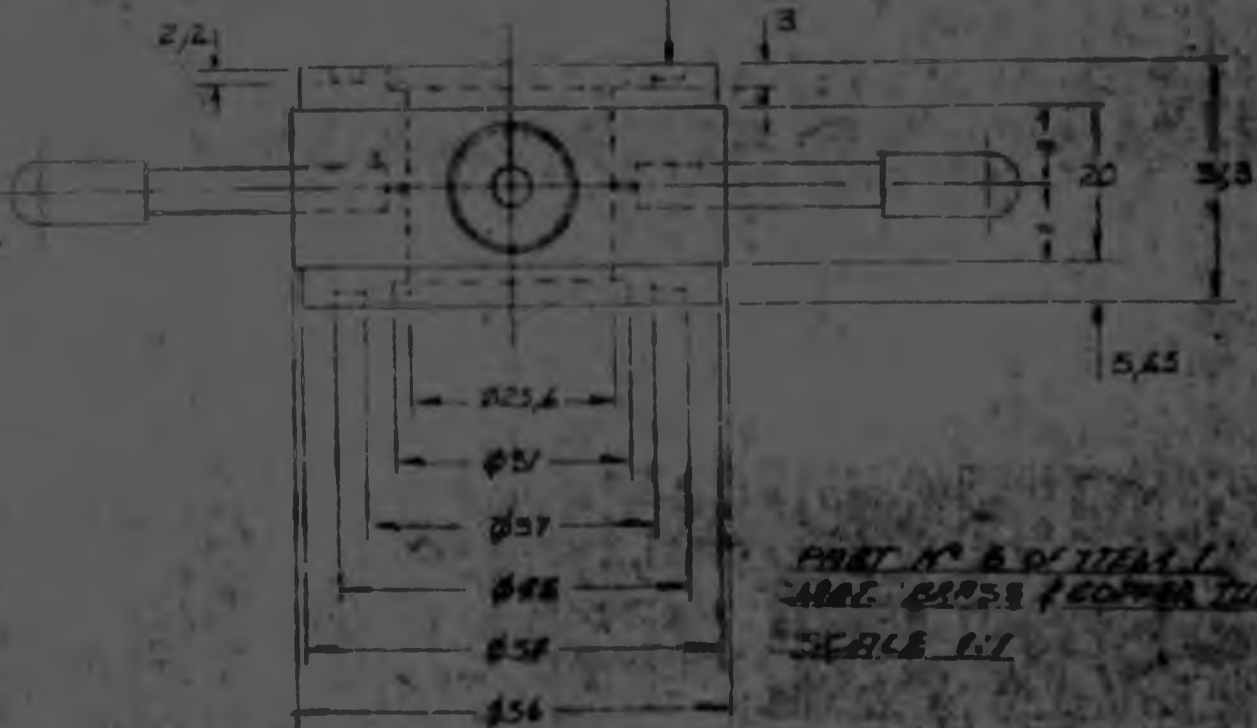


FIGURE 16a Manometer

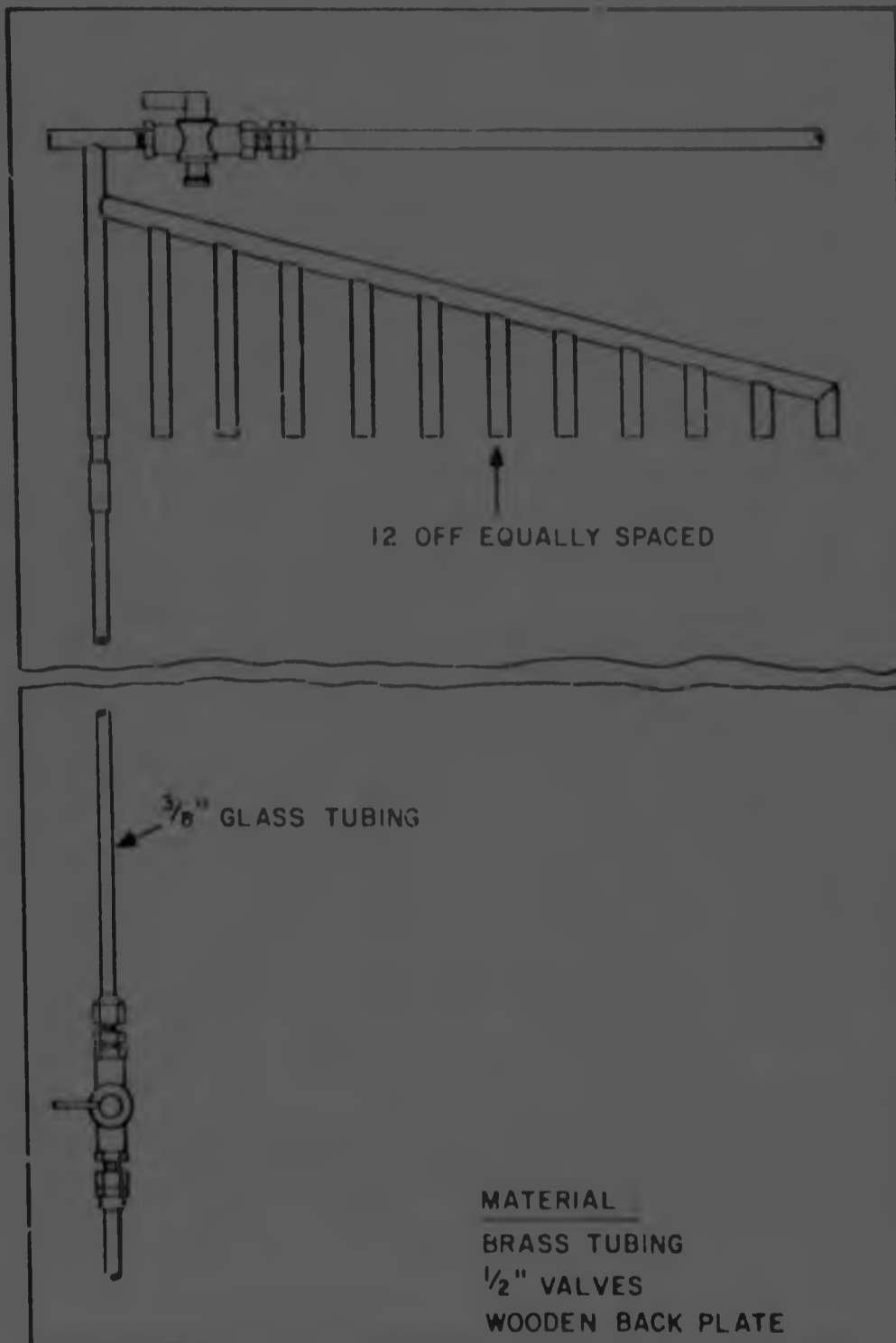


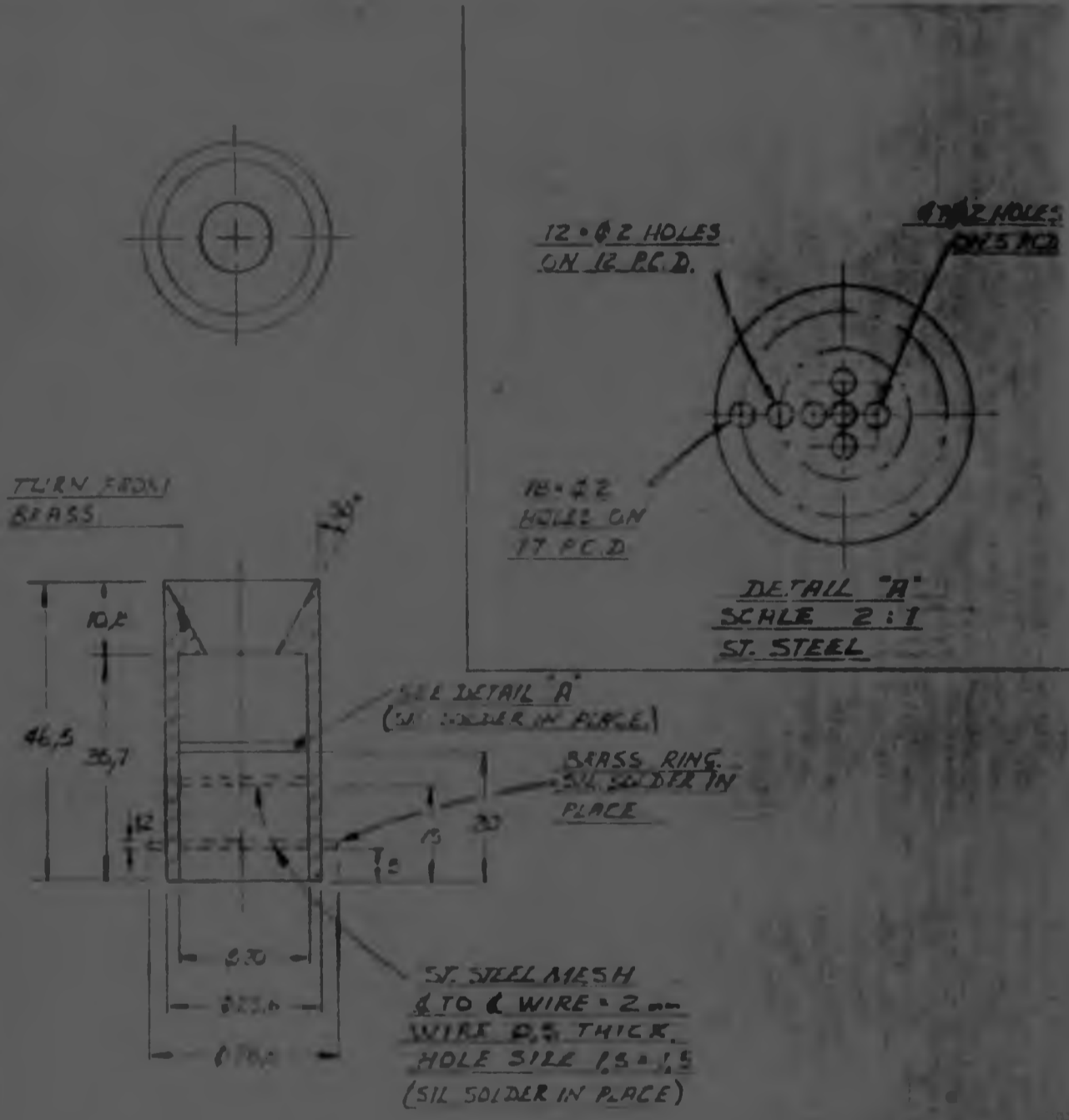
FIGURE 16b

Manometer





FIGURE 17 Static mixer



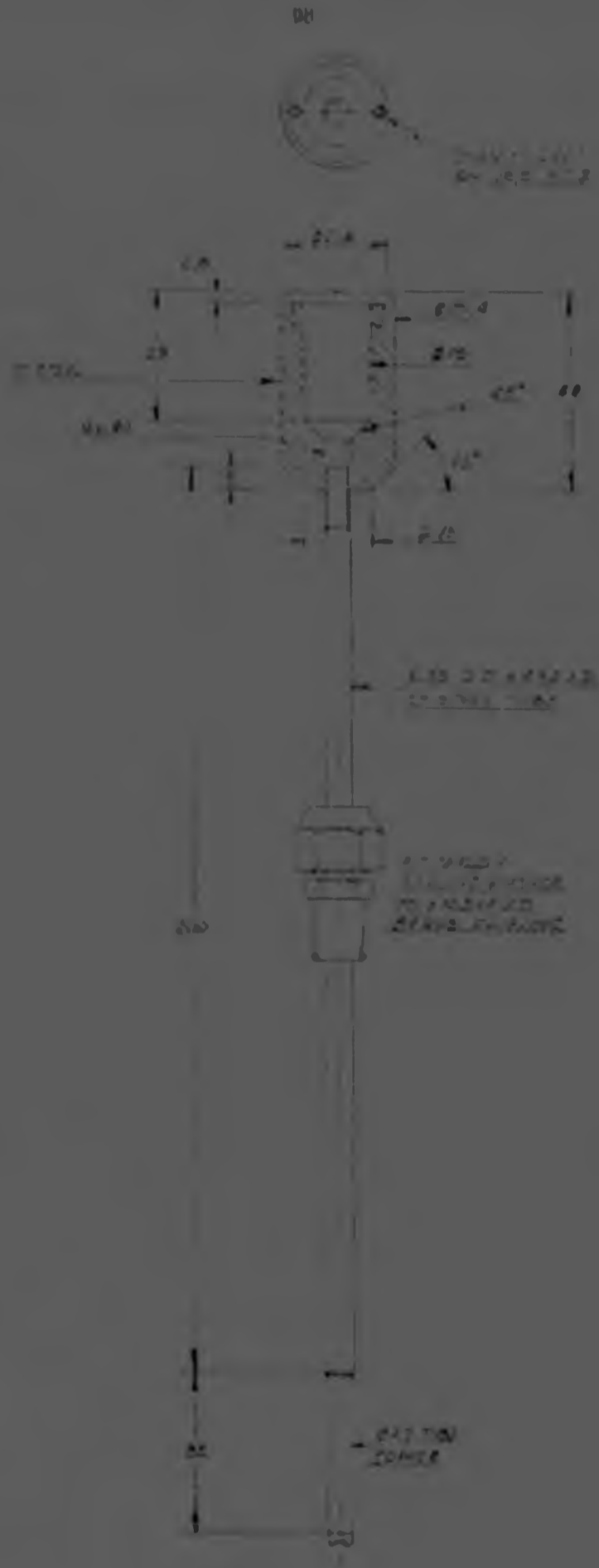


FIGURE 18 Temperature probe for water temperature in the inlet header

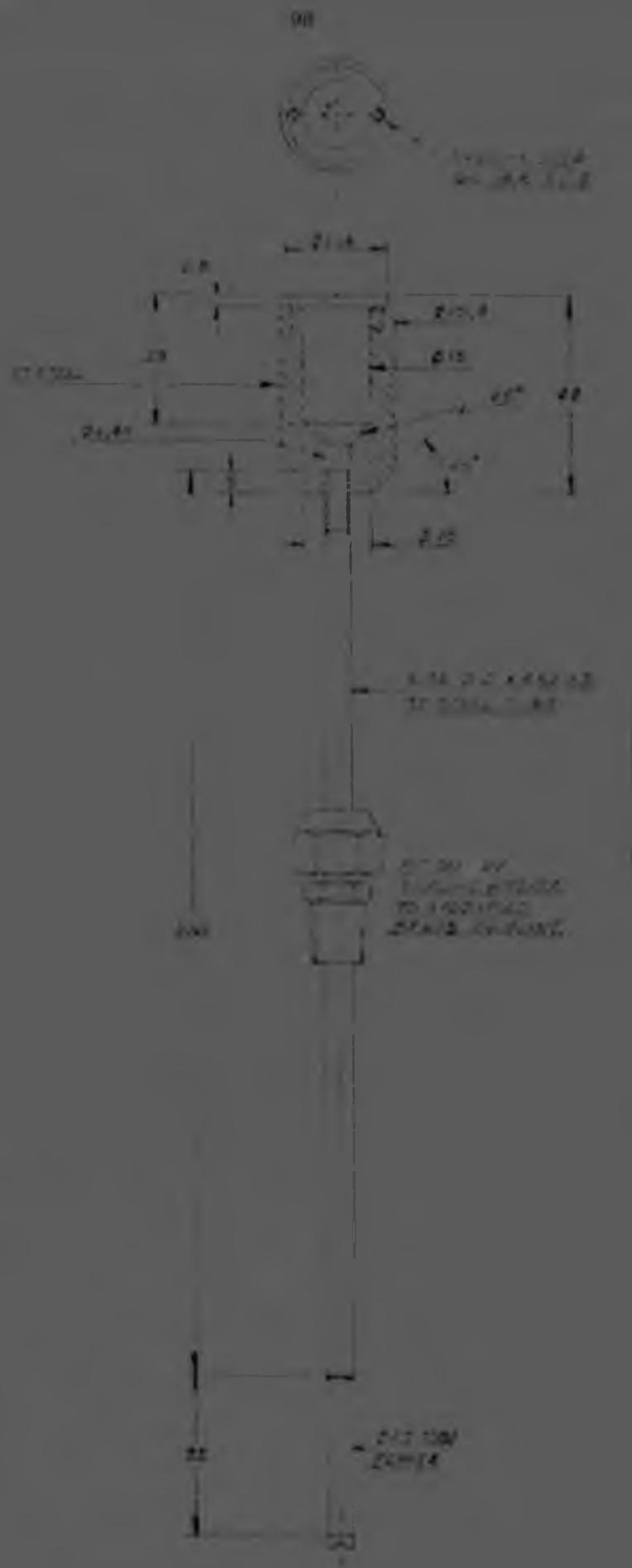


FIGURE 18 Temperature probe for water temperature in the inlet receiver

SCALE 1:1

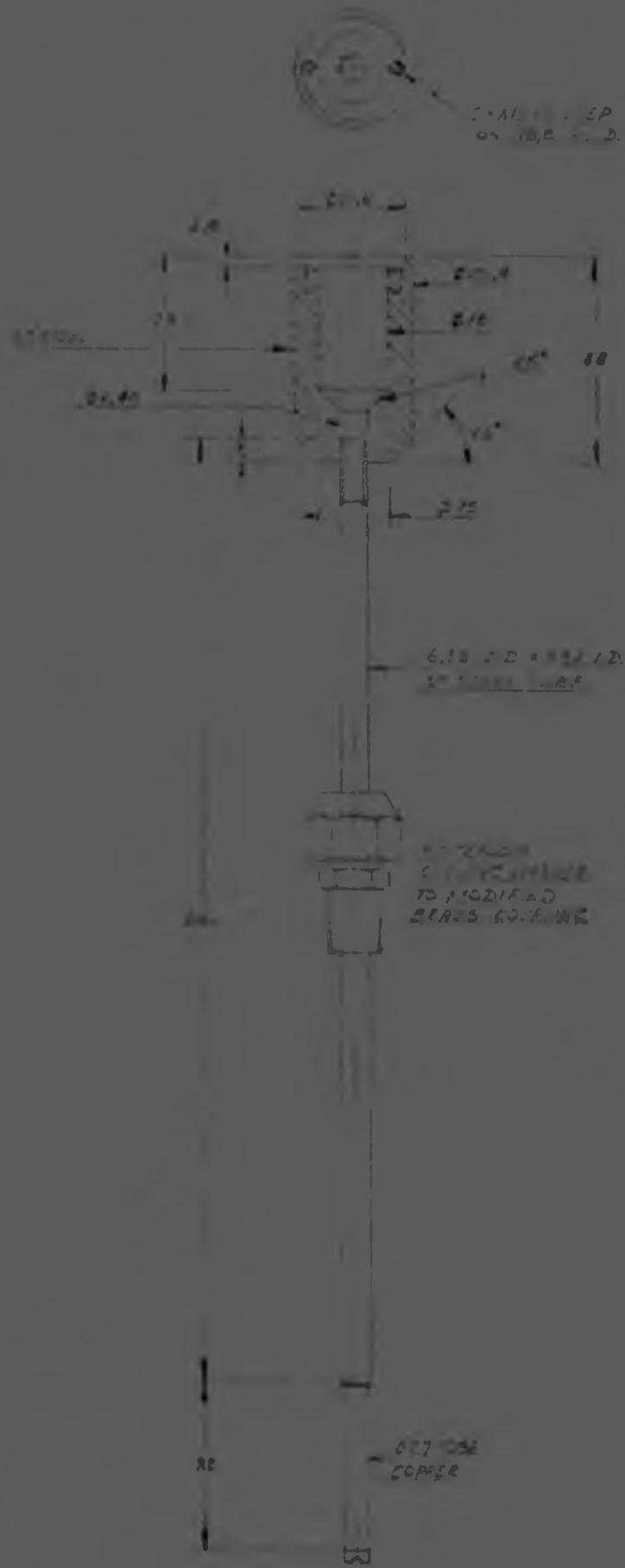


FIGURE 18 Temperature probe for water temperature in the inlet heater

FIGURE 19 Detail of Figure 18

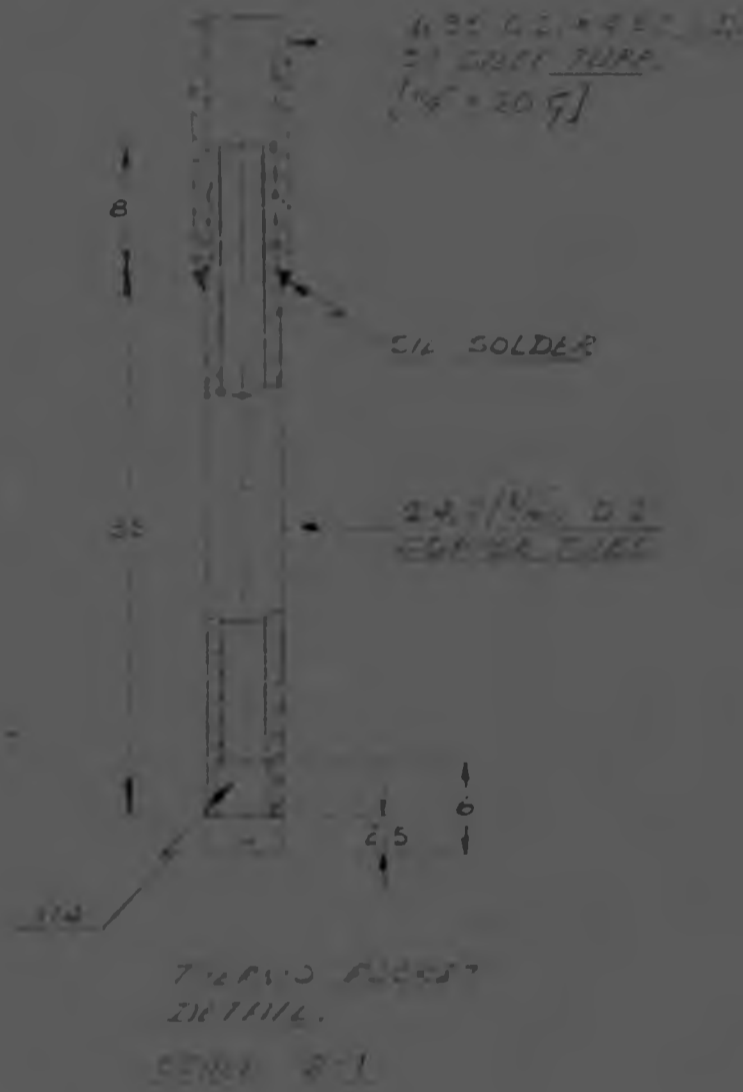
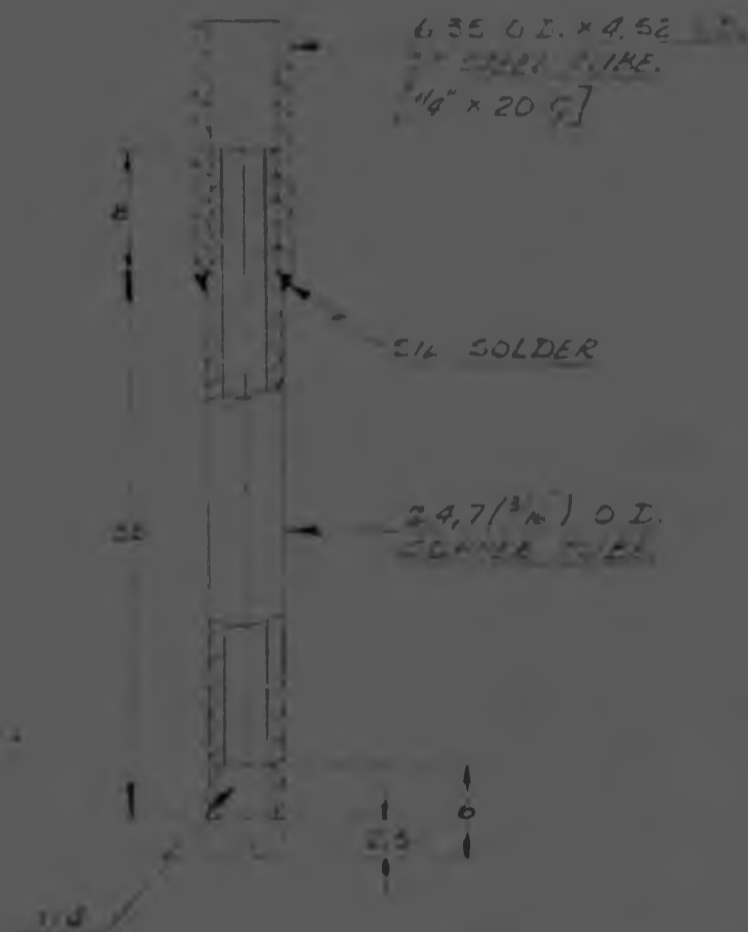


FIGURE 19

Detail of Figure 18



BYRON'S LOCKE  
DETAIL.  
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FIGURE 20

(Detail of Figure 10)



FIGURE 21 Data plotted against the Hauser equation

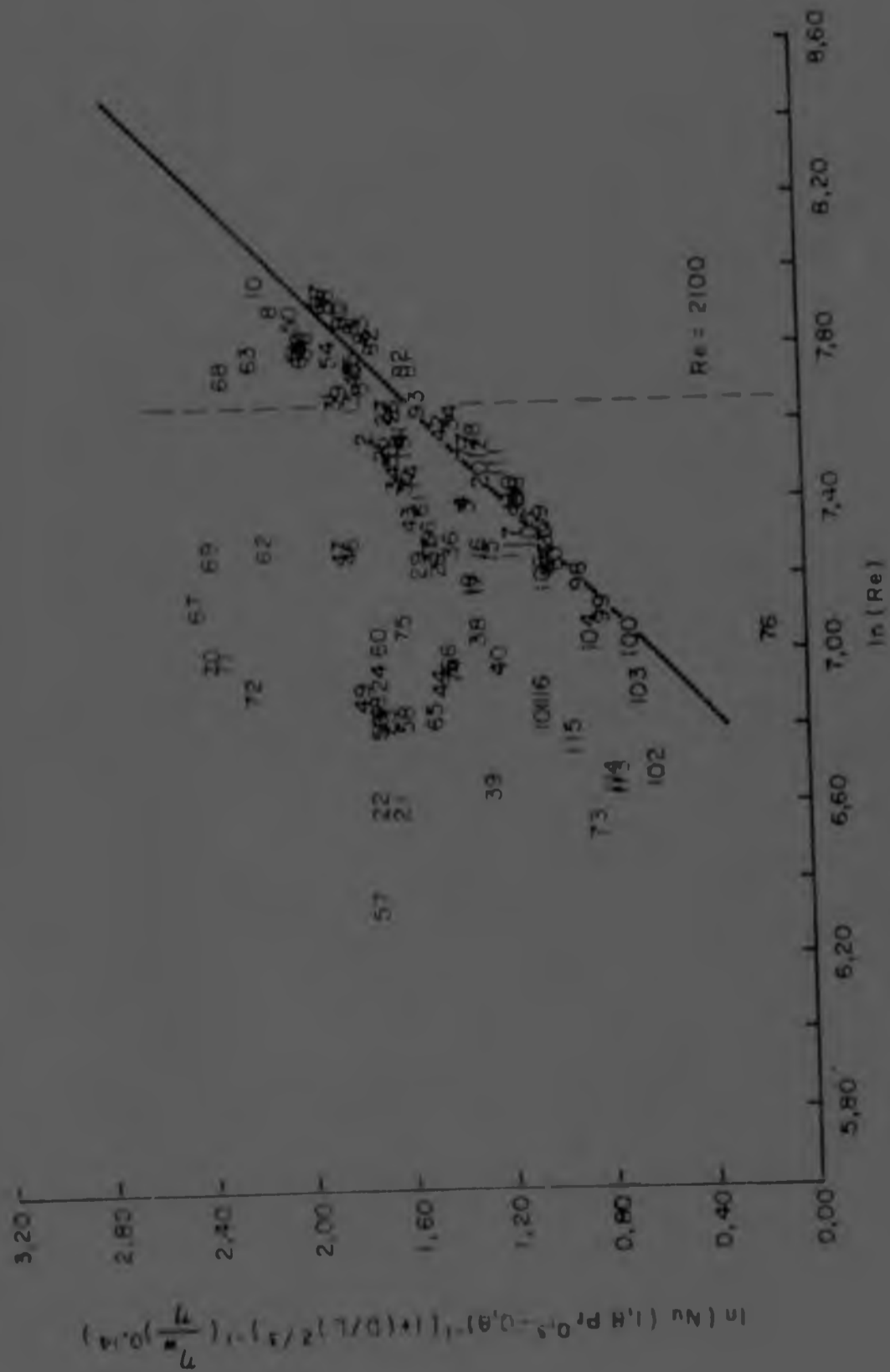




FIGURE 22 Metals and Eckert plot of the data

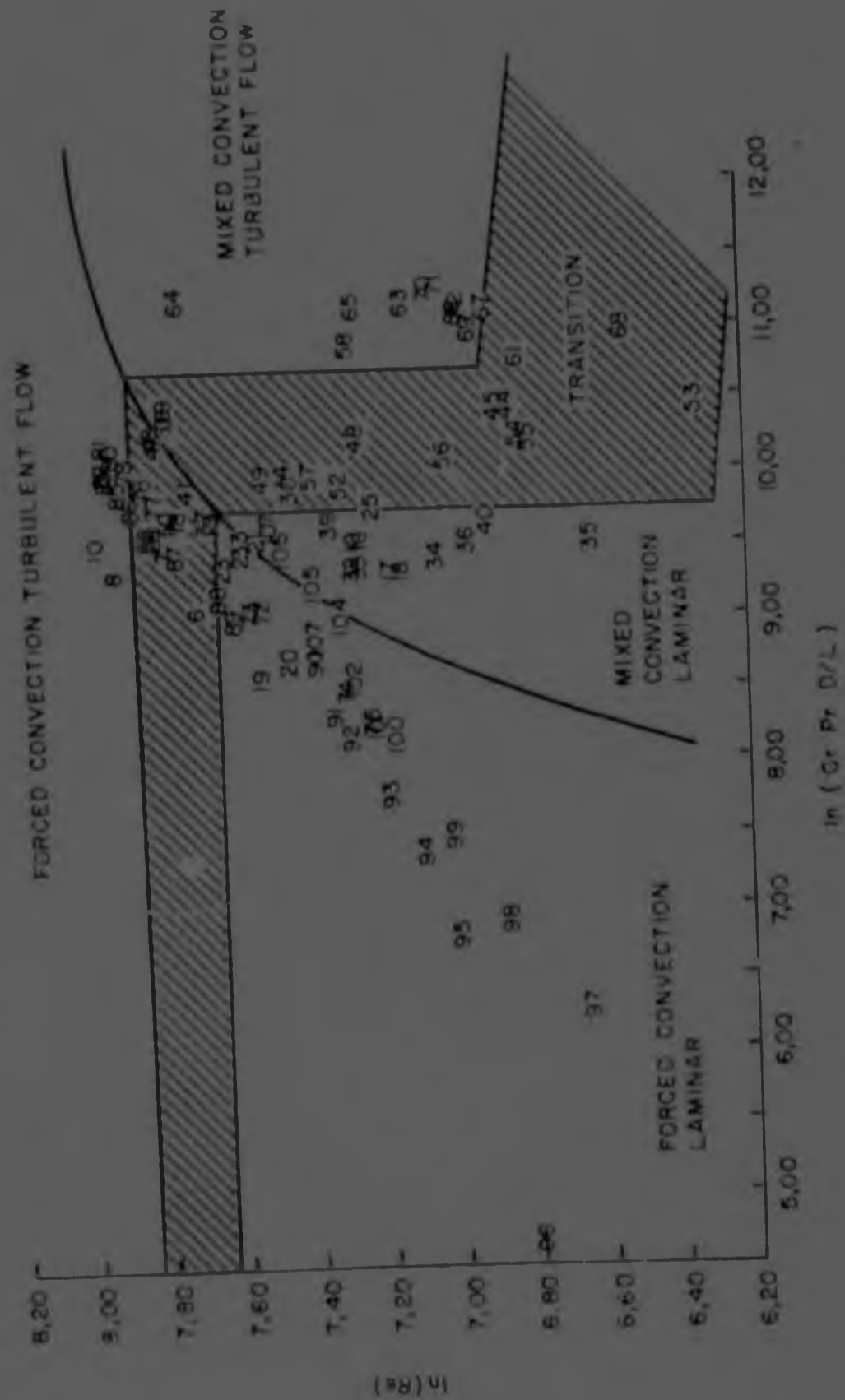


FIGURE 23 Data plotted against the Colburn type equation

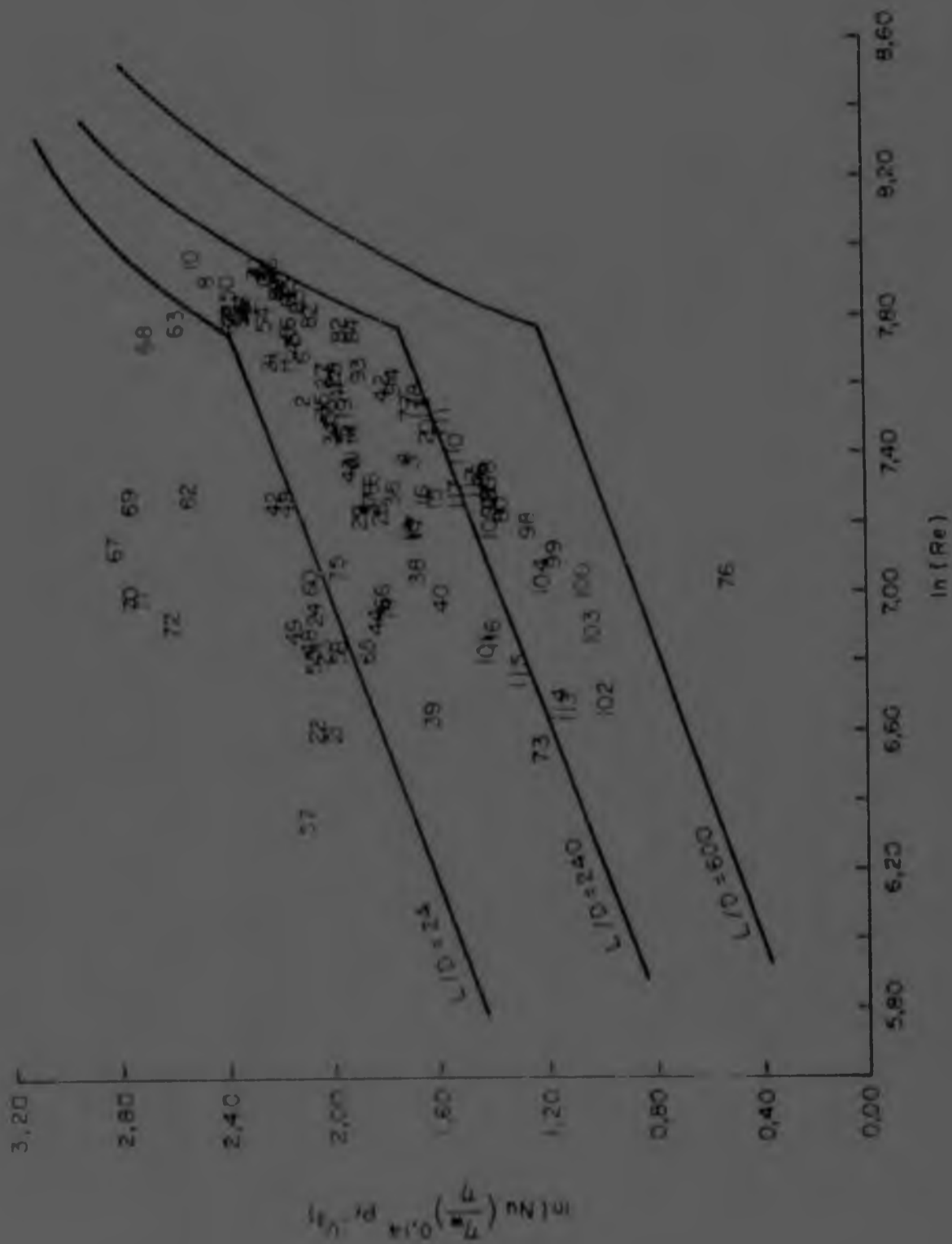


FIGURE 24 Modified Metz and Eckert plot

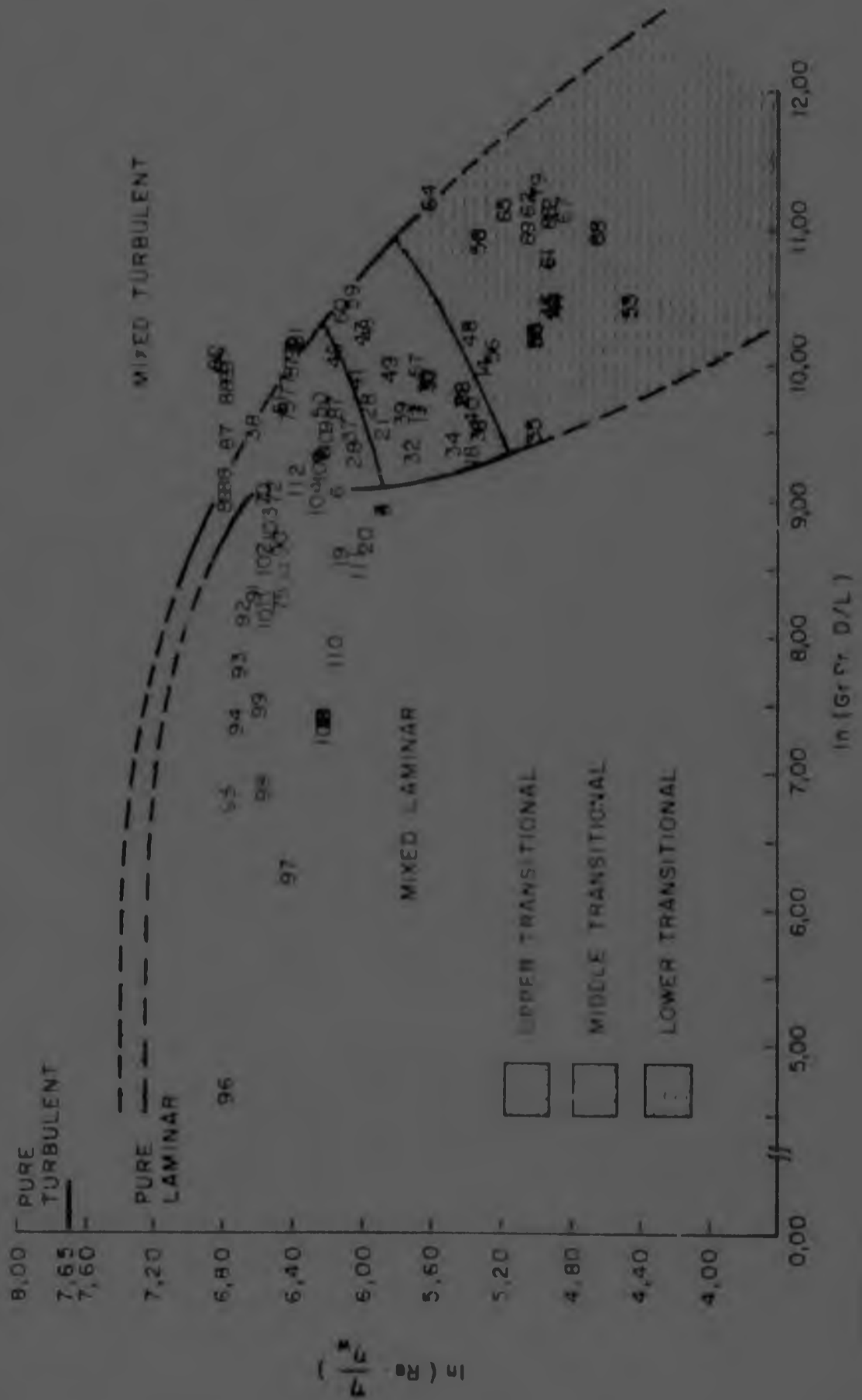


FIGURE 25 Mixed turbulent equation

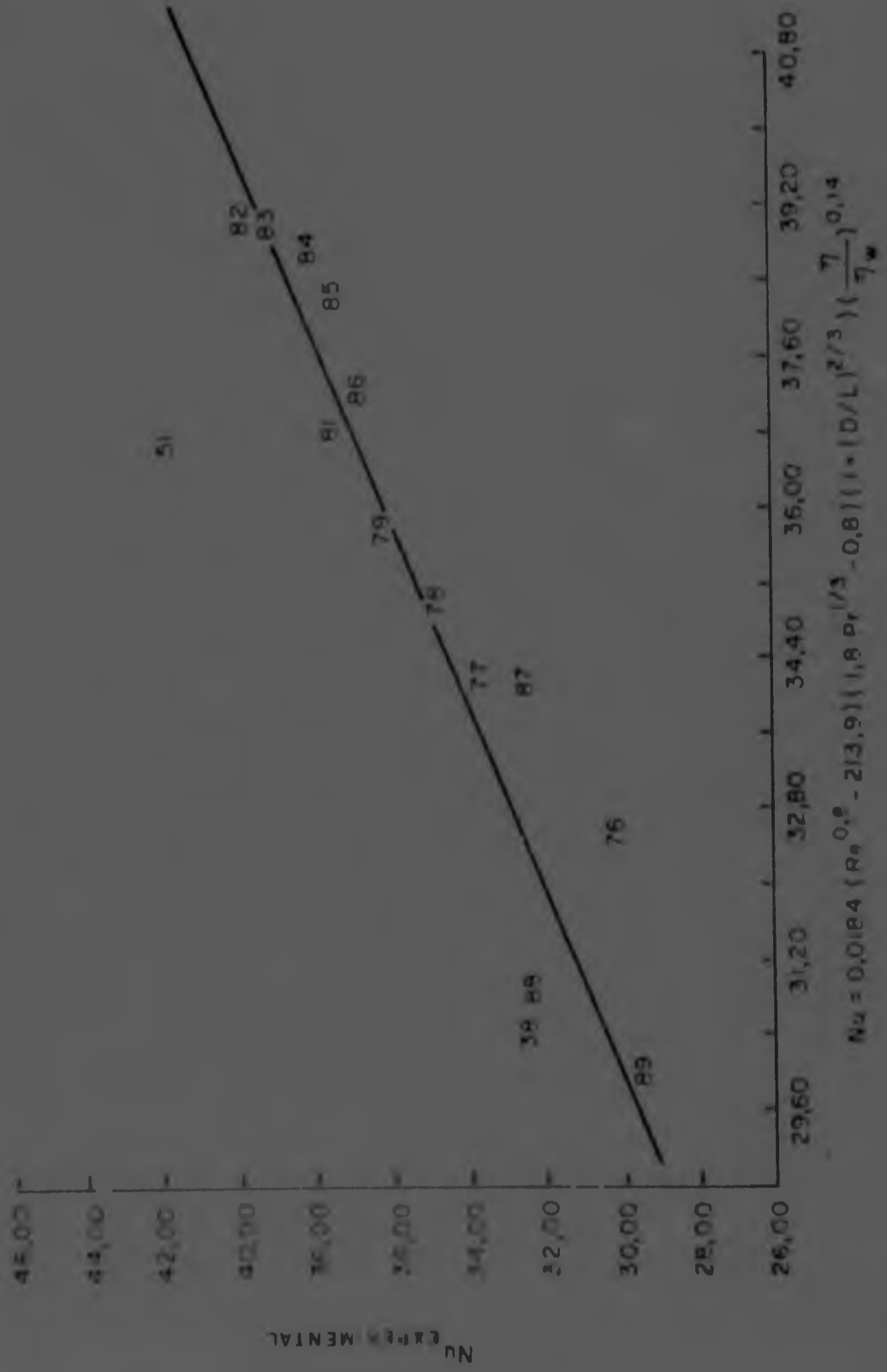


FIGURE 26 Upper transitional equation

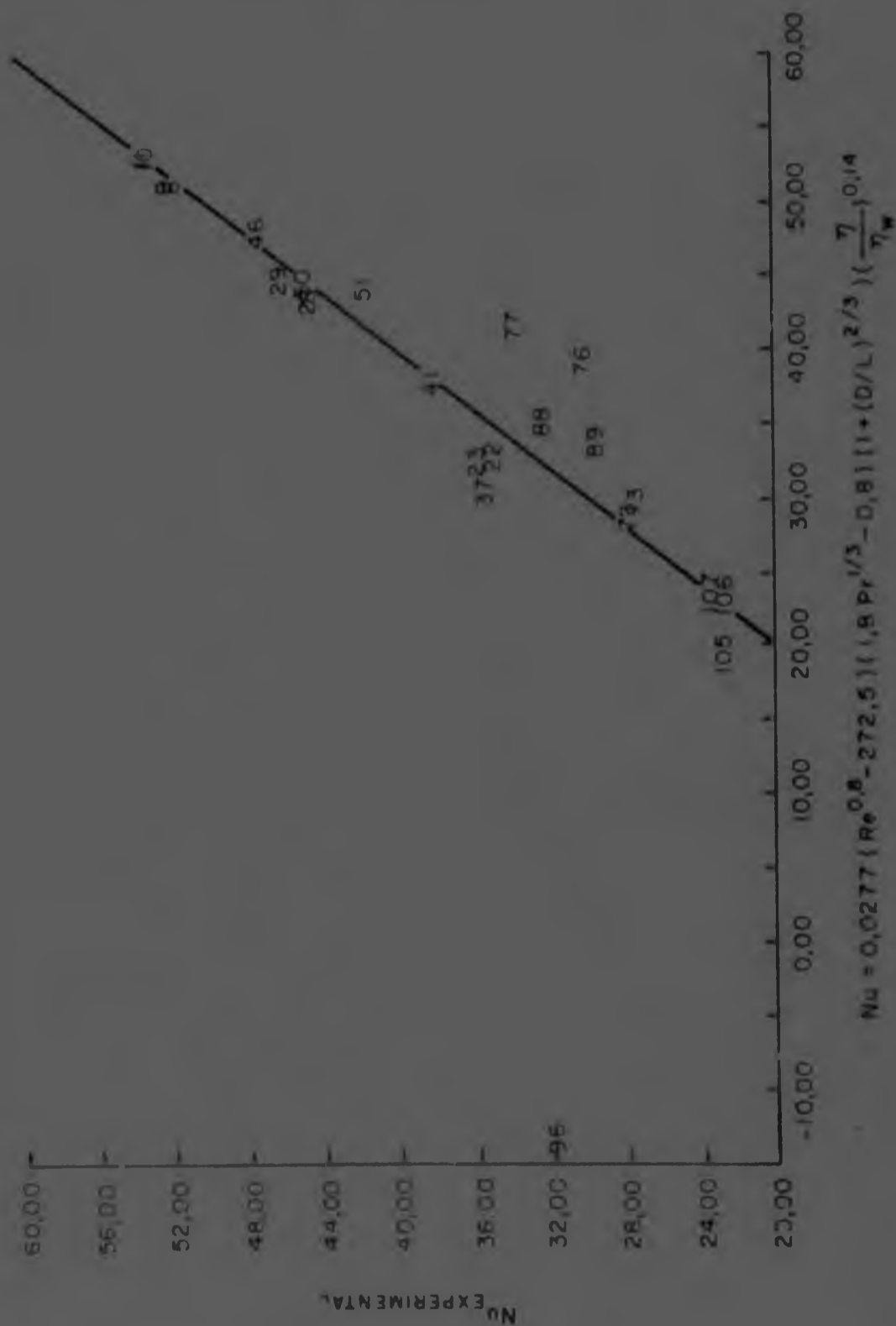


FIGURE 26 Upper transitional equation

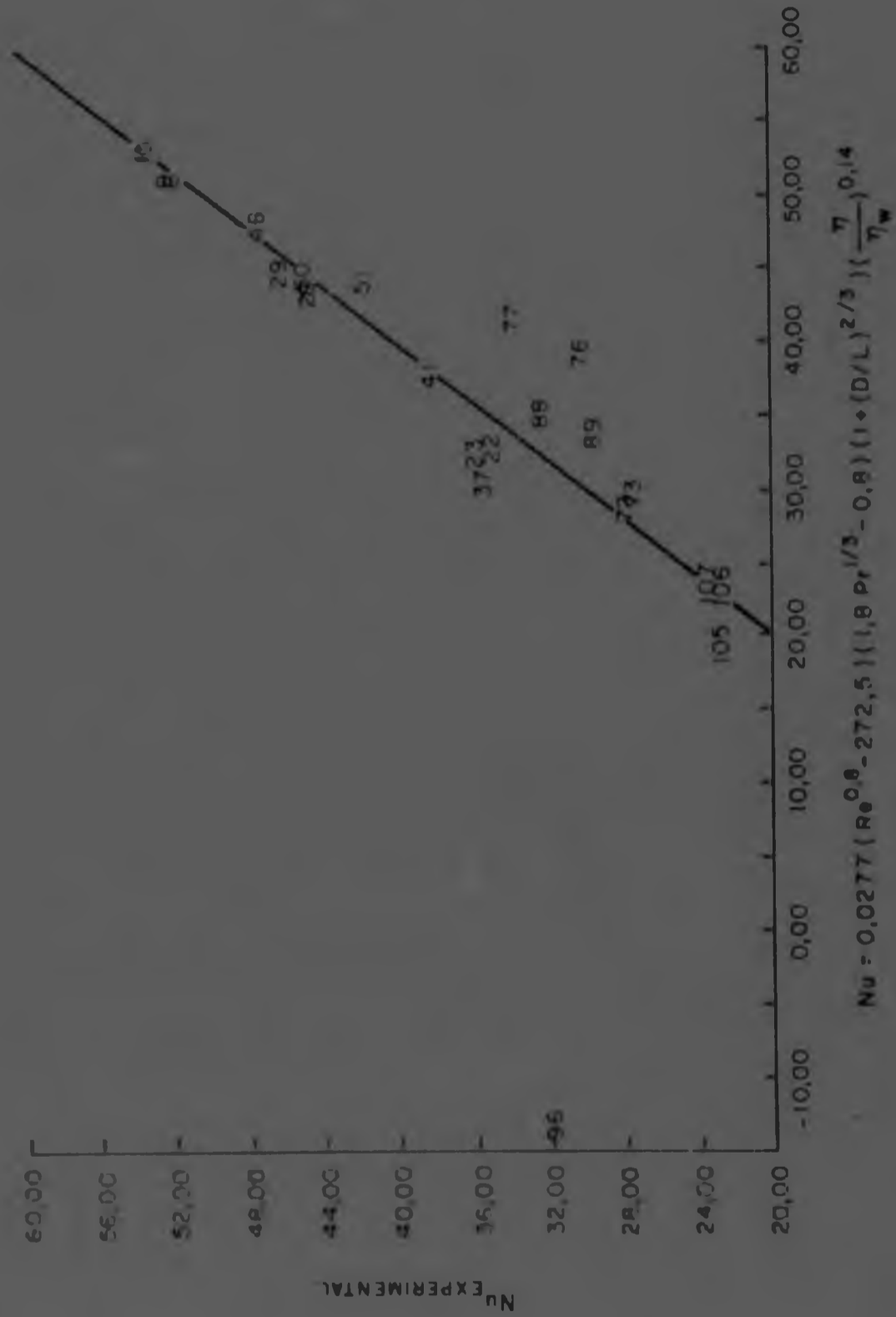


FIGURE 27 Middle transitional equation

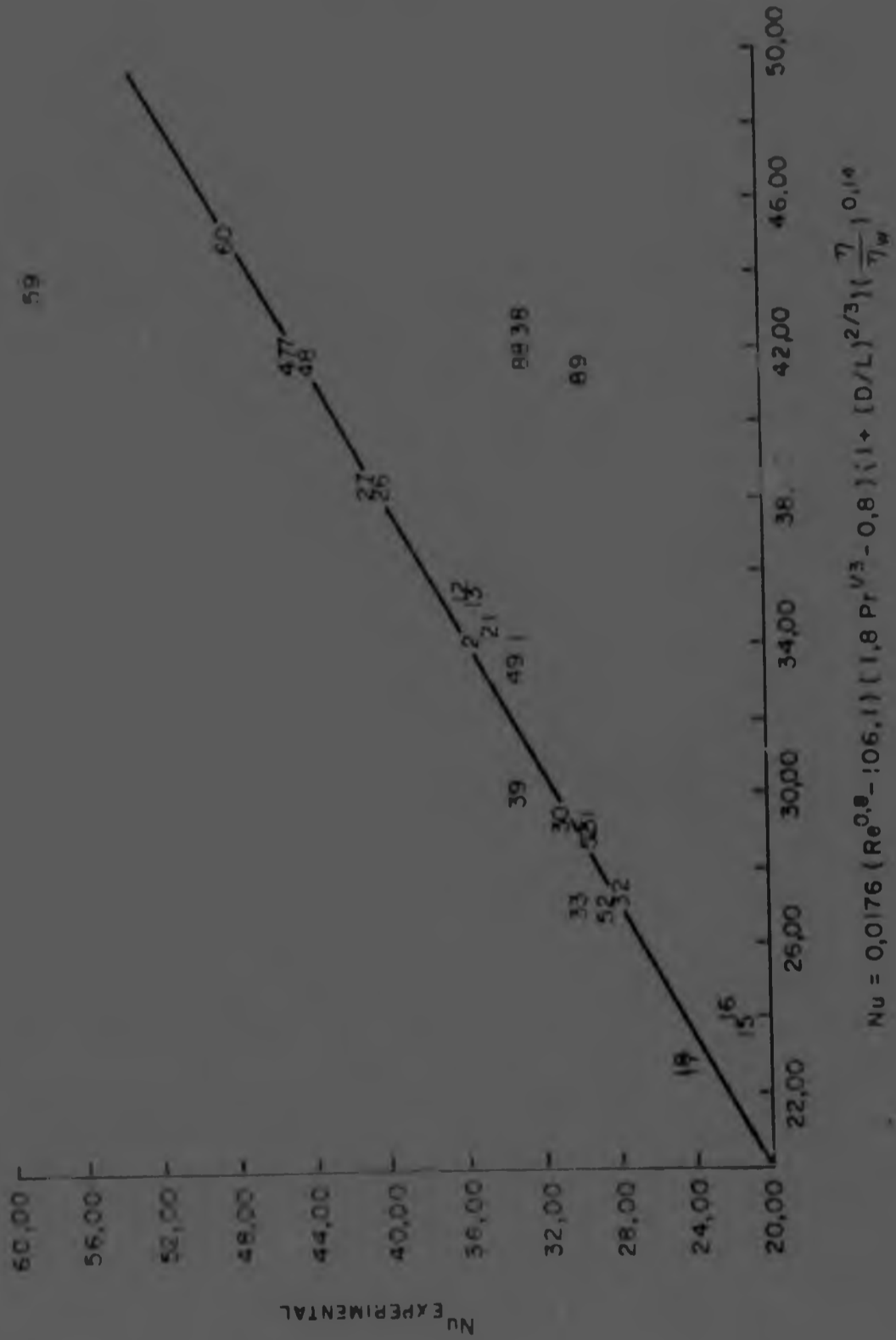


FIGURE 8 Mixed laminar equation

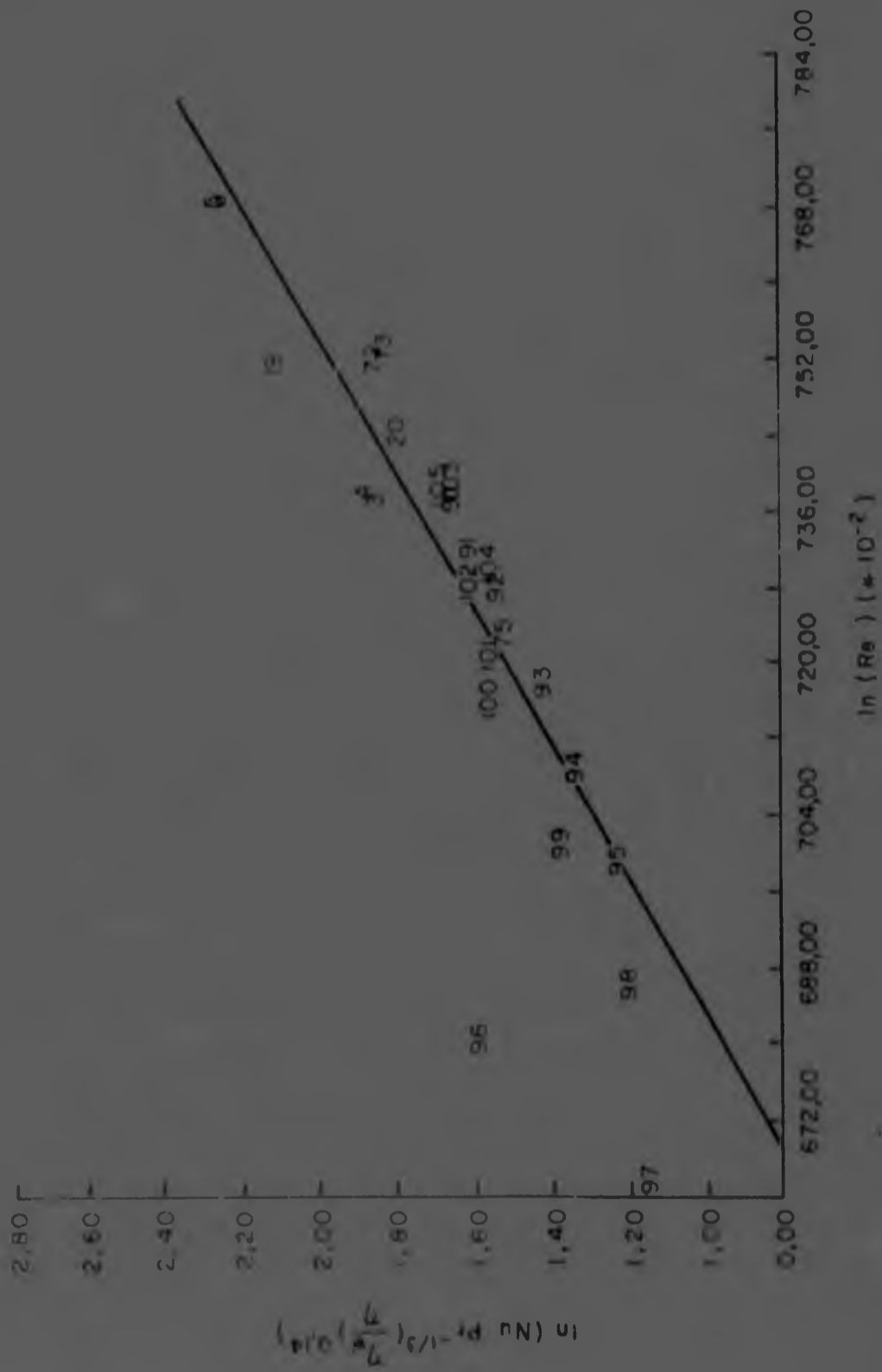




FIGURE 29 Lower transitional data as a function of the Stanton and Reynolds groups

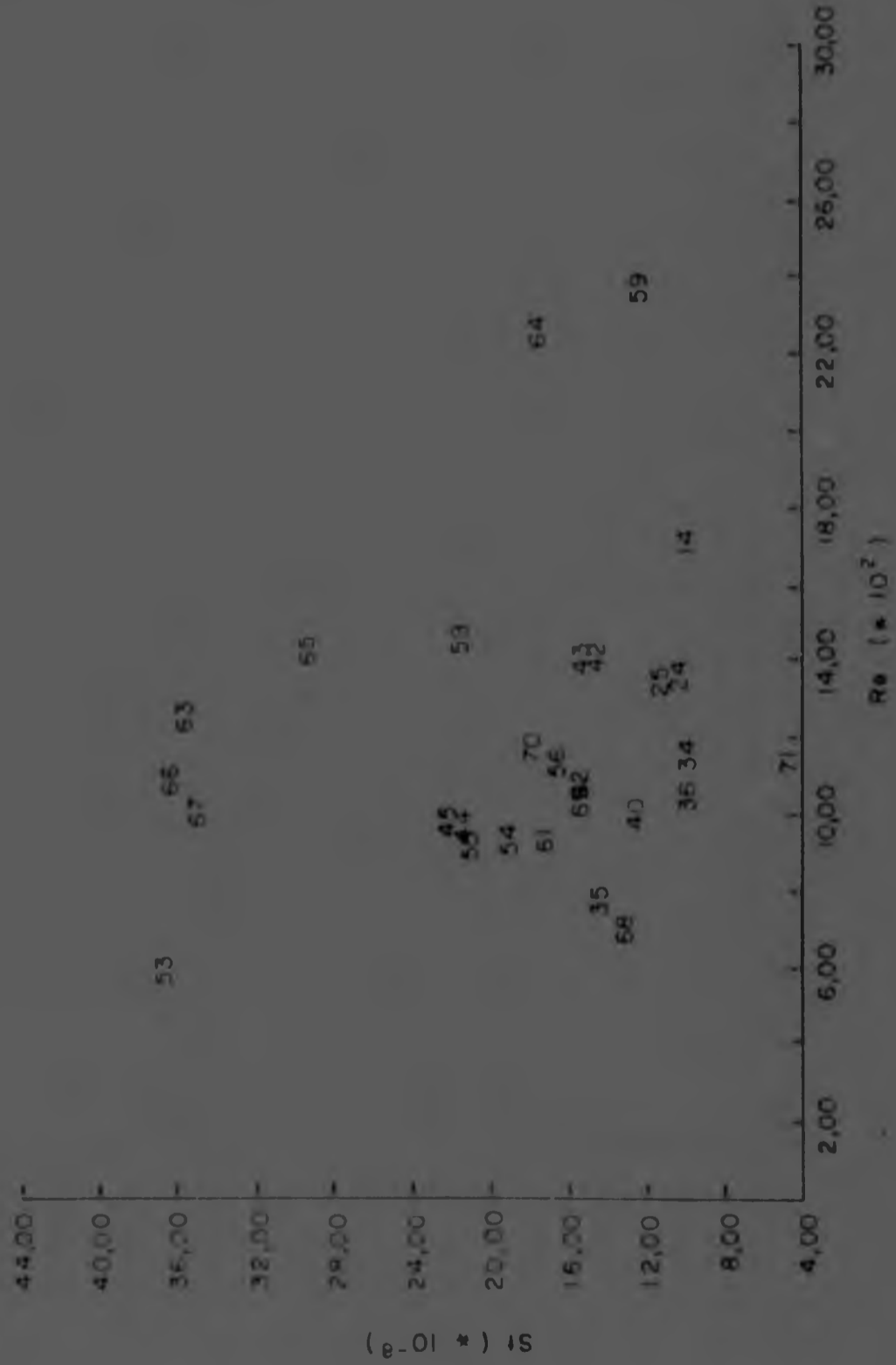


FIGURE 29 Lower transitional data as a function of the Stanton and Reynolds groups

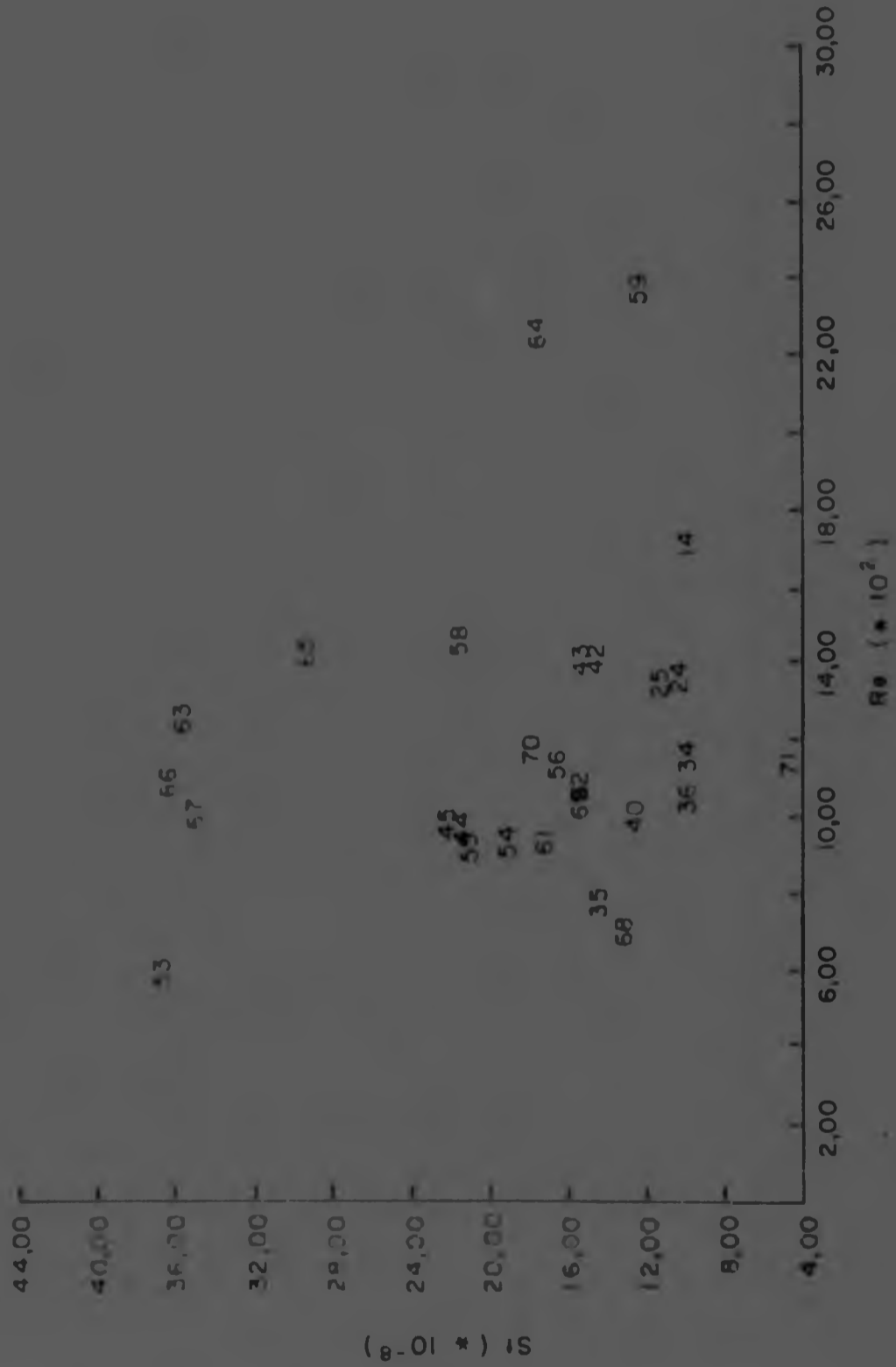


FIGURE 30 Lower transitional equation

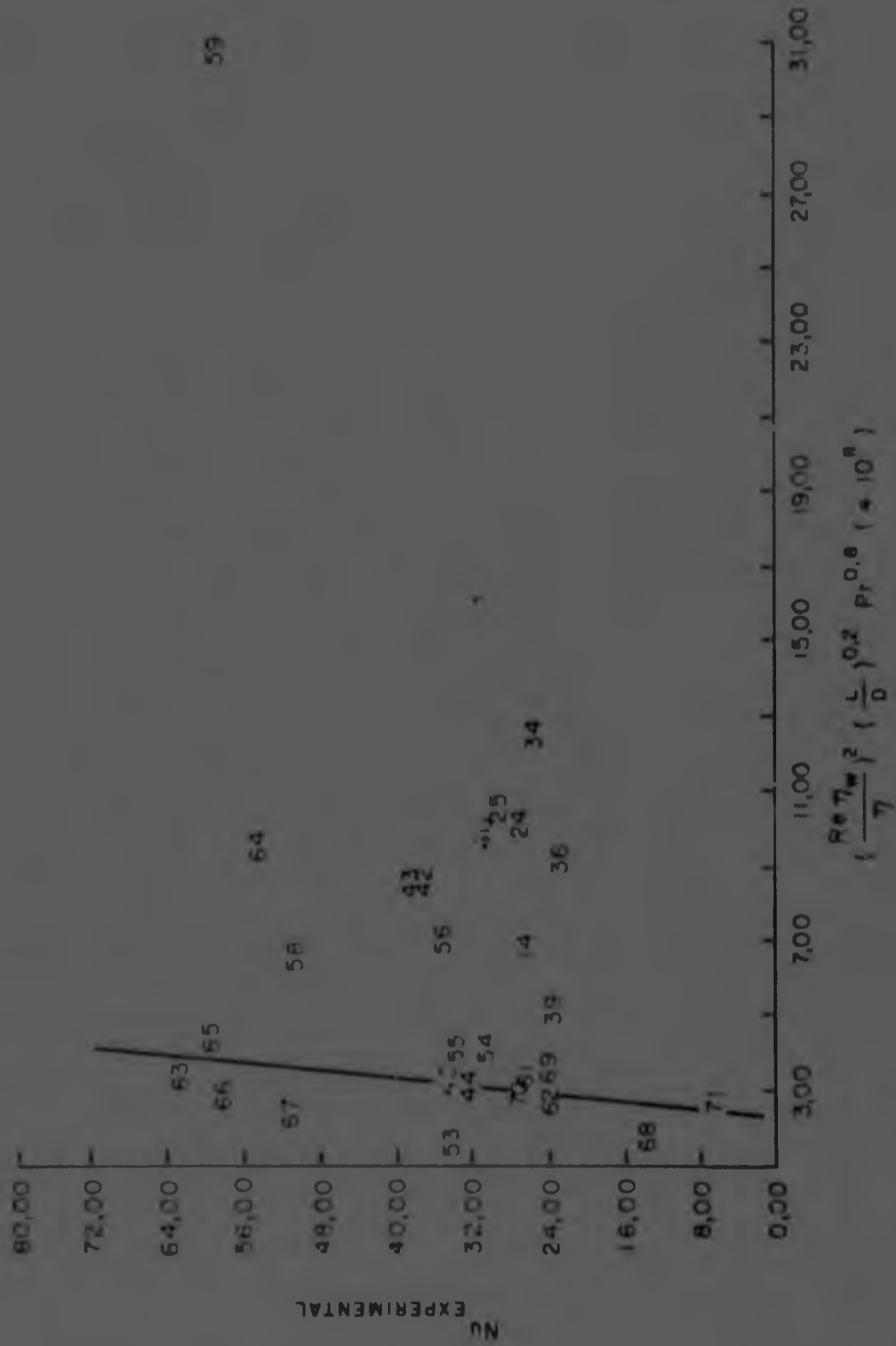
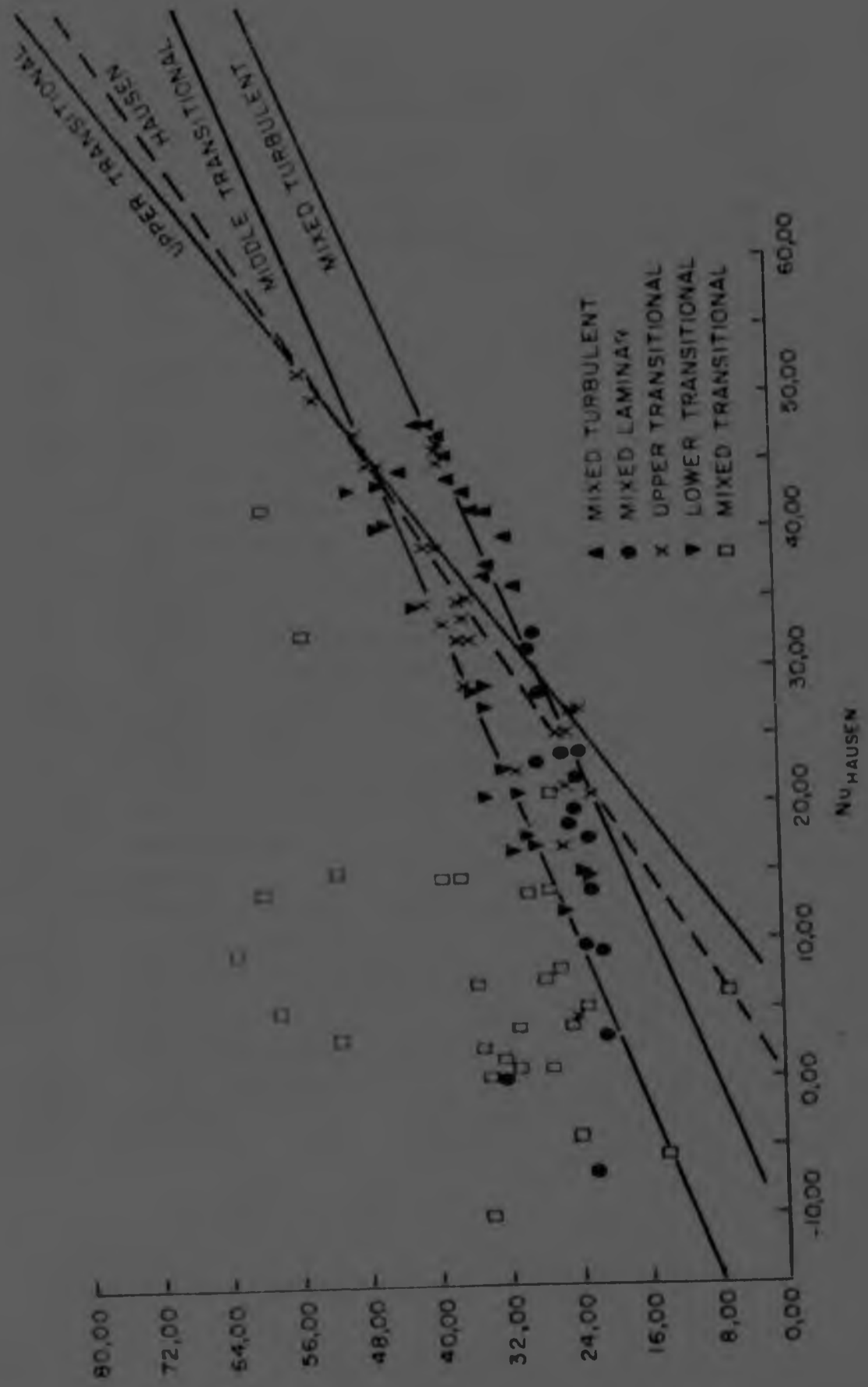


FIGURE 31 Hausen equation compared with derived equations



NOMENCLATURE

A	Surface area used in 6.7.2	$m^2$
$A_1$	Coefficient as defined in 6.7.1	$W^2/m^2$
$A_0$	Coefficient in Andrade equation	kg/ms
a	Coefficient in thermal conductivity equation	W/mK
$B_1$	Coefficient as defined in 6.7.1	WK
b	Coefficient in specific heat equation	J/kg
C	Specific heat	J/kgK
$C_0$	Specific heat at reference temperature	J/kgK
D	Inside diameter of tube	m
$E(x)$	The error in x	dimensions are that of x
$F(x)$	A general function used in 6.7	
g	Gravitational acceleration	$m/s^2$
h	Heat transfer coefficient	$W/m^2K$
$h_0$	Heat transfer coefficient in the annulus	$W/m^2K$
$\ell$	Characteristic length	m
dl	Incremental tube length	m
L	Tube length	m
m	Mass flow rate	kg/s
$m_1$	Mass flow of oil as calculated from calibration equation	kg/s
$m_2$	Mass flow of oil corrected for density	kg/s
$\bar{m}$	Arithmetic mean value	
$N_n$	Rotameter reading	%
p	Correction factor of Gregorig used in 3.4.1	
q	Heat flux	$W/m^2$
$\dot{Q}$	Heat transfer rate	W
R	Resistance	$\Omega$
$R_0$	Resistance of standard resistance thermometer	$\Omega$
$R_1$	Resistance as measured by bridge	$\Omega$
$R_2$	Resistance corrected to standard tables	$\Omega$
r	Radius	m
T	Absolute temperature	K
$T_c$	Critical temperature	K
w	Velocity	m/s

Greek symbols

$\beta$	Coefficient of volume expansivity	1/K
$\eta$	Dynamic viscosity	kg/ms
$\eta/\eta_w$	Ratio of bulk to wall viscosity	
$\theta$	Temperature	$^{\circ}\text{C}$
$\theta_0$	Datum temperature in thermal conductivity equation	$^{\circ}\text{C}$
$\theta_{wt}$	Temperature of the water in the cooling jacket	$^{\circ}\text{C}$
$\lambda$	Thermal conductivity	W/mK
$\lambda_0$	Thermal conductivity at datum temperature	W/mK
$\rho$	Density	$\text{kg/m}^3$
$\rho_f$	Density of rotameter float	$\text{kg/m}^3$
$\tau$	Response time of resistance thermometer	s

Subscripts used in general equations*(Specific instances are included previously)*

i	At inlet conditions
o	At outlet conditions
wi	At inside wall conditions
wo	At outside wall conditions
l	Pertaining to liquid
v	Pertaining to vapour
b	At bulk conditions
bi	At bulk inlet conditions
bo	At bulk outlet conditions

Dimensionless groups*(Properties are evaluated at the bulk temperature)*

Grashof number	$Gr = \beta g L^3 \rho^2 \Delta \theta / \eta^2$
	$Gr_D = \beta g D^3 \rho^2 \Delta \theta / \eta^2$
	$Gr_l = \beta g l^3 \rho^2 \Delta \theta / \eta^2$
Nusselt number	$Nu = \frac{h l}{k}$

Prandtl number	$Pr = \frac{C\eta}{\lambda}$
Reynolds number	$Re = \frac{\rho w D}{\eta} = \frac{4m}{\Pi \eta D}$
Stanton number	$St = \frac{h}{\rho w C} = \frac{Nu}{Re Pr}$

FORCED CONVECTIVE HEAT TRANSFER  
IN SINGLE PHASE FLOW OF A NEWTONIAN FLUID  
IN A CIRCULAR PIPE

A annotated summary of empirical correlations

DOUGLAS GORDON ROGER.

S Y N O P S I S

An extensive bibliography of empirical correlations for the Nusselt group for internal Newtonian pipe flow has been compiled to facilitate the design of heat transfer equipment. An index is provided for locating experimental heat transfer coefficients for particular fluids and flow conditions. The area of transitional flow is lacking in experimental data and more data must be collected in this region before reliable predictions of heat transfer coefficients may be made.

KEYWORDS Heat transfer coefficient, turbulent, laminar, transition, review  
internal flows, empirical correlations, fluid flow



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## INTRODUCTION

*Order and Simplification  
are the first step toward  
the mastery of a subject*

*Thomas Mann*

Reliable heat transfer coefficients for designing heat exchange equipment are often difficult to obtain and the design engineer may on occasion question the applicability of a chosen coefficient. This doubt may result in over-conservative design methods being used and more expensive units than necessary being designed.

This bibliography was compiled as an aid to the designer to find an accurate heat transfer coefficient for the conditions which he is designing and to clarify the state of experimental data and correlations as a first step to improving the design process.

The heat transfer coefficient for the transfer of heat to or from a non-porous wall to a fluid is denoted by the equation

$$q = h(T_w - T_f) \quad \text{where } h = \frac{-k_{\text{fluid}} \left. \frac{dT}{dx} \right|_{x=\text{wall}}}{T_w - T_f}$$

The magnitude of the heat transfer coefficient,  $h$ , has been determined to depend substantially on the velocity and it is therefore convenient to subdivide heat transfer coefficients according to the flow pattern. The dependence of  $h$  on other factors such as the Prandtl number, Schmidt number and Péclet number does not enable a convenient subdivision to be made.

### 1. FLOW REGIMES

Internal forced flow is characterized by the three regions, laminar, turbulent and transitional flow, the latter occurring in the intermediate region between laminar and turbulent flow. The laminar flow may be further subdivided into laminar to turbulent transitional and turbulent to laminar transitional (also called reverse transition) depending on the history of the flow. The three fundamental regions are traditionally characterized by the Reynolds number as

$$\begin{aligned} \text{laminar} & \quad Re < 2300 \\ \text{transitional} & \quad 2300 < Re < 10000 \\ \text{turbulent} & \quad Re > 10000 \end{aligned}$$

It should be noted that the geometry and the heat transfer rate affect the transition process.

## INTRODUCTION

It is well known that the heat transfer coefficient in a pipe is a function of the Reynolds number and the Prandtl number.

Figure 1

It is often found that the heat transfer coefficient in a pipe is a function of the Reynolds number and the Prandtl number. This is usually expressed in the form of a correlation equation. The form of the correlation equation is usually determined by the results of experimental work.

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$$h = \frac{k}{D} \left( \frac{Pr}{Pr_s} \right)^{1/4} \left( \frac{Re}{Re_s} \right)^m$$

The form of the correlation equation is usually determined by the results of experimental work. The form of the correlation equation is usually determined by the results of experimental work.

### 2.1 FLOW REGIME

Internal flow is characterized by the Reynolds number, laminar, turbulent and transitional flow. The latter occurring in the intermediate region between laminar and turbulent flow. The Reynolds number is a function of the velocity, diameter and viscosity of the fluid. The Reynolds number is a function of the velocity, diameter and viscosity of the fluid.

$$\begin{aligned} \text{Laminar } Re &= 2300 \\ \text{Transitional } Re &= 2300 \text{ to } 10000 \\ \text{Turbulent } Re &= 10000 \end{aligned}$$

The Reynolds number is a function of the velocity, diameter and viscosity of the fluid. The Reynolds number is a function of the velocity, diameter and viscosity of the fluid.

For internal pipe flow Metzger and Eckert (1964) recognised the effect of free convection on the flow pattern and incorporated the GrPr product to further subdivide the flow into the regions

Forced convection turbulent

Free convection turbulent

Forced convection laminar

Free convection laminar

Mixed convection turbulent

Mixed convection laminar

In this report only the three fundamental regions of laminar, turbulent and transitional flows and only experimental results and correlations are considered. For reviews on the theoretical models the texts of Shah and London (1978) for laminar flow models and Reynolds and Gebeci (1976) and Launder and Spalding (1972) for turbulent flow models are recommended. There is no specific text for transitional flow and this region is usually included in turbulent flow modelling.

## 1.2 LOCATING INFORMATION

Fluid and Equation indexes were prepared for locating original experimental data for particular fluids and flow conditions. For example, if a heat transfer coefficient is required for heating molasses in turbulent flow in a horizontal pipe, the Fluid index (Section 4.2) indicates that Friend and Metzner (1958) obtained experimental data. Cross-referring to 1958 in the Equation index (Section 4.3) for the article by Friend and Metzner will give the data and an accurate heat transfer coefficient.

Alternatively the Equation index may be used for evaluating a given correlation for the Nusselt number (or other heat transfer group). For example, if the equation of Malina and Sparrow (1964) was in question, the entry in the Equation index will indicate how the equation fits given experimental data.

The Equation index also contains entries for which there is no cross reference in the Fluid index. These entries are largely analytical solutions, or correlations based on other researchers' data.

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For literature other than English the VDI-Wärmeatlas Berechnungsblätter für den Wärmeübergang is recommended as an interesting summary.

## COEFFICIENT CORRELATIONS

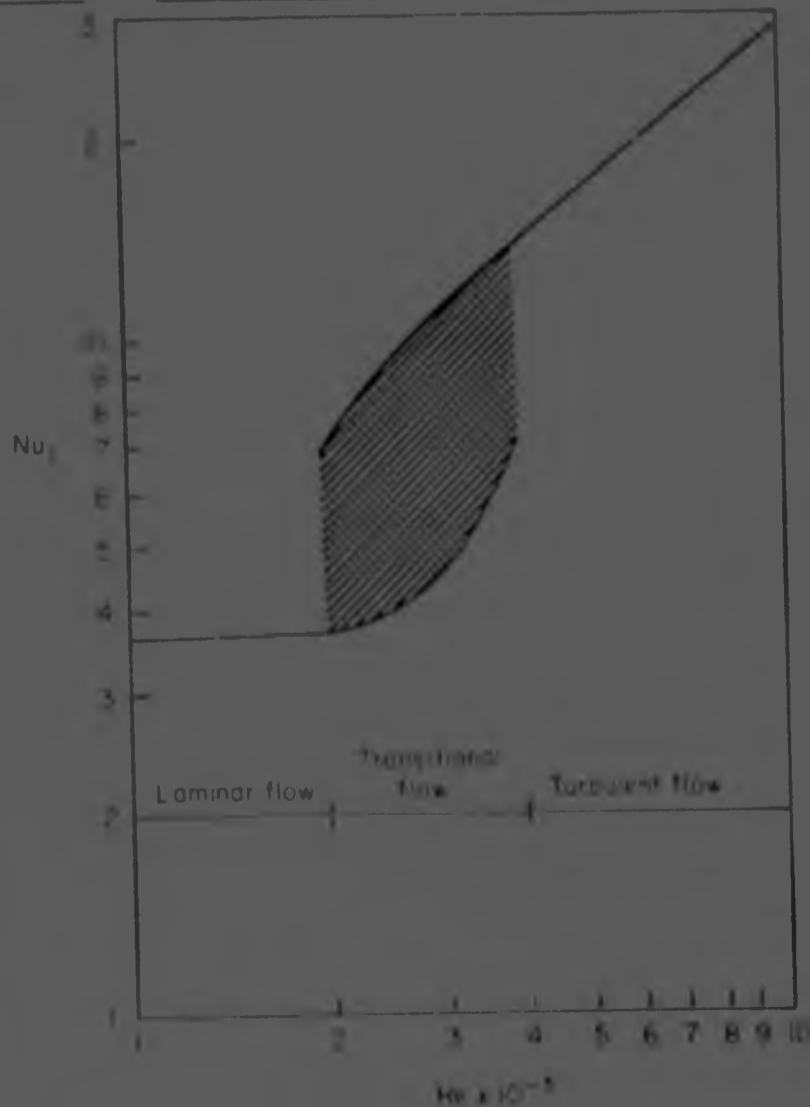
Osborne Reynolds (1874, 1884) was one of the first researchers to recognise and quantify the modes in which fluids flow in pipes. This he did as follows:

"... In the first place, it has shown that the property of viscosity or treachness, possessed more or less by all fluids, is the general influence conclusive to steadiness, while on the other hand, space and velocity are the counter influence..."

Reynolds therefore divided fluid flow into two regions which have since been termed *laminar* and *turbulent* flow, which in isothermal conditions are distinguished by the Reynolds group,  $Re$ .

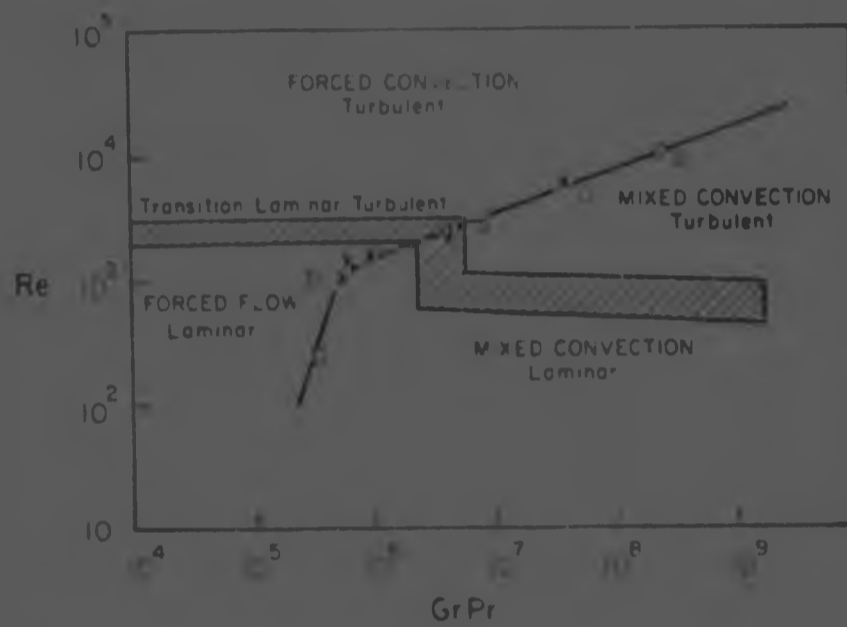
It was not until circa 1940 that an intermediate region of fluid flow which had a marked effect on the heat transfer, was discerned. This region was termed *transitional flow* and correlations of the form of Figure 1 were accepted, noteworthy is the lack of indication in the figure of how to select a Nusselt number in the transition region.

FIGURE 1 Nusselt numbers for transitional gas flow ( $Pr = 0.71$ )



Circa 1954 Eckert and Diaguila and later Metz and Eckert (1964) identified the effect of free convection on the extent and location of the three flow regions and presented results as in Figure 2

FIGURE 2 Regimes of free, forced, and mixed convection for flow through horizontal tube



(  $10^4 < GrPr < 10^9$  ) Metz and Eckert (1964)

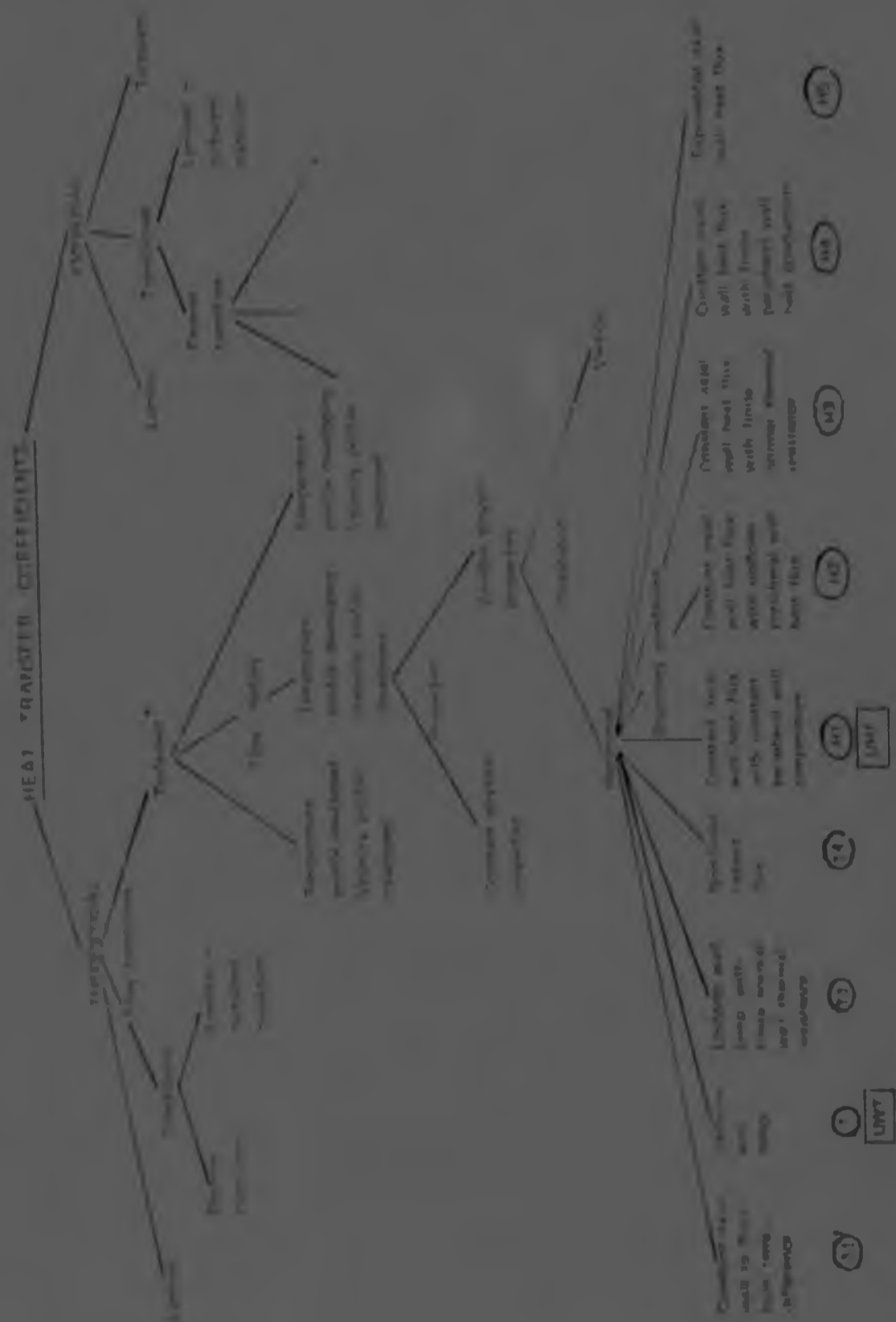
(The free convection limit was not established through lack of data)

However, correlation for determining the heat transfer in the transition region were still inadequate and circa 1970 Bankoff further subdivided the region into the separate cases of laminar to turbulent transition and turbulent to laminar transition or reverse transition.

With this subdivision of the flow conditions Figure 1 has been constructed. It is envisaged that this Figure will be enlarged in the future as a more complete understanding is reached.



FIGURE 3 Schematic representation of flow divider and boundary conditions for heat transfer in pipes



11 Boundary conditions (see text and Section 2.1.1)  
 12 Assumptions: Ra = 10^4, Pr = 10, Re = 10^4 (Churchill 1977), Diff. eq. (Eqn. 1.1.1)  
 13 All sublayers are fully developed in the sense of the fully developed flow conditions

21 TURBULENT FLOW HEAT TRANSFER CORRELATIONS

This section summarises the most important points that may be extracted from the extended survey in section 3.

Since the available literature on heat transfer extends over a relatively long time span, it is often difficult to visualise the progression of the science. To facilitate the visualisation of the state of experimental turbulent flow heat transfer coefficient correlations Table 1 has been constructed. From this it is relatively easy to grasp the chain of thought through the time span.

In the other sections on laminar and transition flow heat transfer correlations, similar tables have been drawn up to facilitate visualisation.

TABLE 1 A chronological summary of turbulent flow heat transfer correlation methods

UWT = const. wall temp  
UHF = uniform heat flux

BOUNDARY CONDITIONS	DATE	
	1905	Nusselt and Boussinesq from dimensional analysis suggest $Nu \propto (Re)^m (Pr)^n$
	1909	
UWT Pr = 0,7	1917	Nusselt suggests $Nu = 0,625 (RePr)^{1/4} L^{-1/4}$ for a developing velocity profile
	1916	Taylor propose $\frac{1}{C} = \frac{2}{C} \left[ 1 + \frac{U_1}{11} (10 - 7) \right]$
	1919	linking friction and heat transfer
	1922	McAdams and Frost take the viscosity at an average film temperature to align data
$L/D > 35$	1924	McAdams and Frost suggest $Nu = 0,625 (Re)^{1/4} (1 + a^d L)^{1/4}$ eliminating the effect of infinite tube length on the Nu
UWT Pr = 2,48 7,35	1924	Rice incorporates a temperature difference term, possibly to account for free convection
	1929	Keevil and McAdam note the effect of the wall to bulk temperature difference to give different velocity profile

21 TURBULENT FLOW HEAT TRANSFER CORRELATIONS

This section summarises the most important points that may be extracted from the extended survey in section 3

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	1909	
	1917	Nusselt suggests $Nu = 0,625 (RePr)^{1/4} L^m$ for a developing velocity profile
	1916 1919	Taylor propose $\frac{1}{C_1} = \frac{2}{C_2} \left[ 1 + \frac{U_f}{U} (Pr - 1) \right]$ linking friction and heat transfer
L/D = 35	1922	McAdams and Frost take the viscosity at an average film temperature to align data
	1924	McAdams and Frost suggest $Nu = 0,023 (Re)^{4/5} (1 + a^d/L)^{1/4}$ eliminating the effect of infinite tube length on the Nu.
UWT Pr = 2,48 7,35	1924	Rice incorporates a temperature difference term, possibly to account for free convection
	1929	Keevil and McAdams note the effect of the wall to bulk temperature difference to give different velocity profile

BOUNDARY CONDITIONS	DATE	
UWT L/D 59 - 224	1931*	Lawrence and Sherwood find that $L$ has only an effect for viscous liquids
	1931	Drew, Hogen and McAdams suggest using the Gr number ( $1 + a \text{RePr}^3 L$ )
UWT	1933	Colburn suggests using film temperature to align the data and suggests that for large Gr the group ( $1 + a \text{Gr}^{1/3}$ ) should be included.
UHF L/D 48 Pr 2 - 5	1945	Bernardo and Eian note that the St correlated results better in the lower turbulent region.
	1950	Deissler concludes that the effect of fluid properties across the tube can be eliminated by evaluating the properties at a temperature close to the average of the wall and bulk temperatures
	1951	Lyon notes that there is an expected minimum of the Nu as Pr $\rightarrow 0$ .
	1954	Deissler notes that increasing Pr eliminates the entrance effect and that the effect of variable viscosity can be eliminated by evaluating the viscosity at temperatures which are a function of the Pr
UWT L/D - 5 Pr 0.7	1954	Eckert and Dracilla note the effect of GrPr on determining the limits of the flow regions.
UHF	1955	Hartnett defines the thermal entrance length and notes that at high Re the Pr has little effect
UWT Pr 0.7	1961	Jackson, Spurlock and Purdy experimenting with developed velocity profiles note an effect of $L/D$ but not of free convection. In their experiments the boundary layer did not fill the tube
	1963	Petukhov and Popov suggest the equation
		$\text{Nu} = \frac{1.8 \text{RePr}}{1 + b \sqrt{1.8} (\text{Pr}^{1/3} - 1)}$

BOUNDARY CONDITIONS	DATE	
UHF $L/D = 30$ $Pr = 7 - 8$	1964	Allen and Eckert use the wall to fluid temperature difference to correlate the heat transfer.
UWT $L/D = 31$ $Pr = 0.71 - 5.52$	1965	Kotia <sup>2</sup> uses a turbulent $Re$ , $\frac{u_m d}{\sqrt{f}} \frac{d}{8}$ to correlate data.
UHF $L/D = 21$ $Pr = 0.7 - 14.3$	1967	Gowen and Smith show that the universal temperature profile is dependent on $Pr$ and $Re$ .
UWT $L/D = 80$ $Pr = 2 - 8$	1971	Herbert and Sterns for vertical tubes note that the $GrPr$ has little effect at high $Re$ but increases as $Re$ decreases.
	1972	Gross and Thomas note that inclusion of the $\frac{dP}{dx}$ term improves a theoretical model.
	1974	Mori, Sakakibara and Tanimoto find that for $Gz = 50$ the ratio of the wall thermal conductivity to that of the fluid and the wall thickness may be significant.

## 2.1.1 Conclusions

For heat transfer to a fluid with constant properties, a fully developed velocity profile and without free convection effects, the Nusselt number correlation is of the form  $Nu = f(Re), g(Pr)$ .

For free convection effects the term  $(1 + a Gr^b)$  may be included as a multiplier. [The effect of using the group as a multiplier is discussed by Brown and Thomas (1965)]

For a developing velocity profile the term  $(L/D)^n$  or inclusion of a friction term  $\sqrt{f/8}$  may be included as a multiplier.  $(L/D)^n$  may be criticised as predicting an infinite  $Nu$  for an infinite tube length and  $[1 + (L/D)^n]$  is sometimes used to eliminate this incongruity. The alternative use of the Graetz number is not to include entrance effects but to define the ratio of the rate of heat transfer by convection to the rate of heat transfer by conduction.

Variable fluid properties are accounted for by the ratio  $Nu/Nu_0$  as explained in section 2.4

The most often cited equation is that of Petukhov and Popov (1963), with the variable fluid property correction term of Hufschmidt, Burck and Riebold (1966) which is

$$Nu = \frac{1.07 + 12.7 \sqrt{1 - Pr_0}}{1 + 12.7 \sqrt{1 - Pr_0}} \left[ \frac{Pr}{Pr_0} \right]^{0.11}$$

Alternatively the equations recommended by McNeil and Eckert (1964) may be used

## 2.2 LAMINAR FLOW HEAT TRANSFER CORRELATIONS

As in section 2.1, Table 2 has been constructed to facilitate visualisation.

TABLE 2 A chronological summary of laminar flow heat transfer correlations

BOUNDARY CONDITIONS	DATE	
JWT	1885	Gratz and Nusselt formulate an analytical solution for fully developed velocity and temperature profiles
UHF	1910	
UWT	1928	Leveque extends the result to developing velocity profile giving $Nu = a(RePr)^{1/3} L^{1/3}$
UHF	1930	Dittus and Boelter include the term $T_{LM}$ to include variable physical properties.
UWT $Pr = 2 - 8$	1930	Colburn and Hougen find that $Nu = aPr^{1/3} Gr^{1/4}$ with properties based on a film temperature, independent of the Re
UHF $L/D = 150$	1931	Kirkland and McCabe propose the form $Nu = \frac{b}{RePr d/L} + \frac{c}{(RePr d/L)^n}$
	1933	Colburn includes the term $1 + a Gr^{1/4}$ to account for free convection and base properties on a film temperature
$L/D = 90$	1936	Sieder and Tate suggest the property correction $Nu = a(RePr)^{1/3} \left[ \frac{\mu}{\mu_0} \right]^{0.14}$

BOUNDARY CONDITIONS	DATE	
UWT Pr = 43 L/D = 20 - 602	1942	Martinelli et al criticise the use of $1 + aGr^{1/3}$ as a multiplier as this would indicate an increasing effect of free convection with increasing Re, contrary to their observed results for vertical tubes (See Brown and Thoma 1965)
UWT Pr = 60 - 1000 L/D = 40 - 193	1943	Kern and Othmer suggest $\frac{1 + aGr^{1/3}}{\log Re}$
	1959	Stephan suggests $Nu = 1 + (L/d) Pe$ for constant properties and $Nu = f(L/d, Re, Pr)$ for variable properties
UHF Pr = 0.7 - 8 L/D = 72	1961	Ede suggests $Nu = a + bGr^{1/3}$
	1962	Olive criticises Ede as producing a term in $D^4$ and suggests $Gr^{1/3}$
L/D = 36 - 72	1965	Brown and Thoma find that for horizontal tubes the free convection increases with increasing Re
UHF L/D = 12; Pr = 0.7	1966	Mori et al find that free convection effect starts at $ReRe = 10^4$ and that the critical $Gr$ depends on the intensity of the secondary flow
UHF Pr = 0.7 L/D = 80	1966	McConaughy and Eckert notice that the free convection effect increases as the ratio of $Gr$ to $Re$ increases
	1967	Iqbal and Stanislawicz analytically find that the tube inclination has little effect on the Nu
UHF Pr = 2 - 8 L/D = 700 L/D = 28	1968	Shannon and Depew notice that for $\frac{(GrPr)^{1/4}}{Nu}$ the natural convection is negligible
	1971	Depew and August suggest using the group $G, Gr^n Pr^n$

Conclusion

The above analysis includes all the important physical properties and in the subsequent sections the theory is extended to include the case of a tube

$$\text{with wall temperature } T_w = T_0 + \Delta T \cos \theta$$

$$\text{and the uniform heat flux } q_w = \frac{2k \Delta T}{r_0} \cos \theta$$

The boundary conditions for the tube  $(r=r_0, z=0)$  are to be included as a function. The above analysis includes the problem of uniform tube lengths giving infinite heat transfer coefficient.

The boundary conditions for the tube  $(r=r_0, z=L)$  are to be included as a function. The above analysis includes the problem of uniform tube lengths giving infinite heat transfer coefficient.

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$$N_{Gr} = \frac{g \beta \Delta T r_0^3}{\nu^2} \left[ \frac{r_0}{L} \right]^{1/4}$$

The above analysis includes the problem of uniform tube lengths giving infinite heat transfer coefficient.



2.2.1 Conclusions

For a fully developed parabolic velocity profile, constant physical properties and no buoyancy effects, the Nusselt number depends upon geometry and:

$$\begin{aligned} \text{constant wall temperature} \quad N_{D, \text{conv}} &= 3.66 \\ \text{and for uniform heat flux} \quad N_{D, \text{conv}} &= \frac{48}{15} = 3.20 \end{aligned}$$

For a developing velocity profile the term  $(1 - 1.1 \times 10^{-4} Pe^2)$  must be included and included. This factor has been developed for profiles of constant heat transfer coefficient that result in uniformity.

The convective film coefficient for fully developed flow, however, the  $(1 + 0.047 Pe^{0.16})$  and  $\frac{1 + 0.047 Pe^{0.16}}{1 + 0.047 Pe^{0.16}}$  are included as multipliers. In these instances the Graetz number alone is included as a multiplier, and in the case of fully developed flow the Graetz number is expressed as a function of the Nusselt number alone.

The extent of validity of the correlation may be expressed as a function of the Prandtl number  $Pr$  as  $Pr^{0.4} / Pr_0^{0.4}$ .

Applicable physical properties are included by using the  $\frac{\mu_0}{\mu}$  form of correction as given in equation 2.4.

An explicit equation is recommended for all fluids and conditions, however, the user should refer to the *Design and Analysis Handbook* (1977).

$$N_{D, \text{conv}} = 1.74 Re^{0.4} Pr^{0.4} \left[ \frac{\mu_0}{\mu} \right]^{0.14} \left[ \frac{1 + 0.047 Pe^{0.16}}{1 + 0.047 Pe_0^{0.16}} \right]^{0.14}$$

may be related to other parameters.

TRANSITIONAL FLOW HEAT TRANSFER CORRELATIONS

As in the previous sections Table 3 has been constructed to facilitate visualisation.

TABLE 3 A chronological summary of transitional flow heat transfer correlations

BOUNDARY CONDITIONS	DATE	
P = 35 - 140 L/D = 234	1933	Colburn constructs empirical correlations using the ratio $\frac{L}{D} \frac{h_{conv}}{k_{wall}}$ as a distinguishing criterion
	1936	Sieder and Tate in an analogous form on $\frac{L}{D} \frac{h_{conv}}{k_{wall}}$
	1940	Nusselt and Sieder note that there is a transition region which may extend from $Re = 2100$ to $Re = 10000$
	1942	Nusselt and Sieder experimentally determine the relation: $\frac{h}{C_p G} = 0.0067 Pr^{-0.8} \left[ \frac{L}{D} \right]^{0.14} St$
UWT L/D = 77 - 231	1966	Prater and Gill (re-examine form) the relation: $St = St_{10000} \left[ \frac{St}{St_{10000}} \right]^{-0.4} \left[ \frac{Re}{10000} \right] \left[ \frac{10000}{2100} \right]^{1.16}$ and found no effect of $L/d$
	1970	McElroy, Goun and Perkins note that the acceleration parameter may be used for predicting laminarisation
	1970	Bankston links the relaminarisation to the reverse transition in external flows
	1970	Colson and Perkins find evidence for predicting the transition with $Re$
	1970	Dyster and Epke suggest that more than one exponent for the $Re$ is necessary to correlate the heat transfer
1975	Estimation form: $Nu_{eff} = C Re^n$ where C and n are functions of $L/D$	
1977	Churchill proposes a comprehensive correlating equation for the total flow region, however, the equation is very cumbersome and is made up of various individual equations	

### 2.3.1 Conclusions

The extent of the transition region is not clearly definable and depends on the conditions of the system.

The most promising correlations for the heat transfer coefficient appear as a function of the Stanton number, rather than the Nusselt number.

The effects of free convection, variable physical properties and  $L/D$  ratio on the heat transfer have not been determined extensively.

The only available data are

for air	Pechenegov 1975
and for liquids	Norris and Sims 1942

and these are therefore recommended.

### 2.4 CORRECTION METHODS FOR VARIABLE PHYSICAL PROPERTIES

It is most often convenient to interpret experimental results as a deviation from some specific datum. In heat transfer to a fluid it is convenient to represent the datum as the limiting case of zero heat flux, under which restriction physical properties will be effectively constant due to the uniform temperature fields.

The heat transfer under finite heat flux is then some function of the limiting case of zero heat flux. This is expressed as

$$\frac{Nu}{Nu_0}$$

where  $Nu_0$  is the limiting case of zero heat flux. Table 1 has been constructed from correlations in the literature to summarise available correction methods.

TABLE 4 Correction factors for variable physical properties

FLUID	AUTHOR	YEAR	CORRECTION FACTOR		
G A S E S	H'in	1957	$Nu_c$	$CR_1^{0.8} \left[ \frac{T_c}{T_b} \right]^n$	Air
			$\left[ \frac{T_c}{T_b} \right]$	0,5 - 0,9    0,9 - 1,2    1,2 - 2,3	
			$n$	0,0218    0,0212    0,0223 0    -0,27    -0,58	
	Humble, Lowdermilk and Desmon	1951	$\frac{Nu_c}{Nu_b}$	$\left[ \frac{T_c}{T_b} \right]^n$	Air
			$n$	0 at $\frac{T_c}{T_b} = 1$ -0,55 at $\frac{T_c}{T_b} = 1,5$	
	Kays	1955	$\frac{Nu_{isothermal}}{Nu_{heated mass}}$		
			$n$	0,25 heating 0,08 cooling	
	Bialokoz and Saunders	1956	$\frac{Nu_c}{Nu_b}$	$\left[ \frac{T_c}{T_b} \right]^{0,5}$	Air
$\frac{T_c}{T_b}$			1,1 - 1,73		
Wright and Walters	1959	$\frac{Nu_c}{Nu_b}$	$\left[ \frac{T_c}{T_b} \right]^{0,675}$	Hydrogen	
		$\frac{T_c}{T_b}$	1 - 4		
McCarthy and Wolf	1960	$\frac{Nu_c}{Nu_b}$	$\left[ \frac{T_c}{T_b} \right]^{-0,3}$	Hydrogen	
		$\frac{T_c}{T_b}$	1,5 - 2,8		
Taylor and Kirchgessner	1960	$\frac{Nu_c}{Nu_b}$	1	Helium	
		$\frac{T_c}{T_b}$	1,6 - 3,9		
McCarthy and Wolf	1960	$\frac{Nu_c}{Nu_b}$	$\left[ \frac{T_c}{T_b} \right]^{-0,7}$	Hydrogen Helium	
		$\frac{T_c}{T_b}$	1,5 - 9,9		

FLUID	AUTHOR	YEAR	CORRECTION FACTOR
G A S E S	Wieland	1962	$Nu$ $Nu_c$ +
	Taylor	1963	$T_w$ $T_f$ < 2,8 1,5 - 5,6
	Petukhov and Popov	1963	$Nu$ $Nu_c$ $\left[ \frac{T_w}{T_f} \right]^{-0,47}$
	Kutateladze (through Petukhov)	1963	$\frac{Nu_w}{Nu_f}$ $\left[ \sqrt{\frac{T_w}{T_f} + 1} \right]$
	Kirillov and Malugin	1963	$Nu$ $Nu_c$ $\left[ \frac{T_w}{T_f} \right]^{-0,5}$ $T_w$ $T_f$ 1,1 - 2,3
	McEligot, Mays and Leppert	1965	$Nu$ $Nu_c$ $\left[ \frac{T_w}{T_f} \right]^{-0,4}$
	Lechuk, Elphimov and Fedotkin	1965	$T_w$ $T_f$ 1,1 - 2,5 1,1 - 2,7
	Perkins and Worsoe Schmidt	1965	$Nu$ $Nu_c$ $\left[ \frac{T_w}{T_f} \right]^{-0,7}$ $T_w$ $T_f$ 1,3 - 1,6
	Volkov and Ivanov	1966	$Nu$ $Nu_c$ $\left[ \frac{T_w}{T_f} \right]^{-0,31}$ $T_w$ $T_f$ 1,1 - 2,1
	Petukhov Kirillov and Maidonic	1966	$Nu$ $Nu_c$ $\left[ \frac{T_w}{T_f} \right]^{-0,31}$ $\pi$ $(0,9 \log \frac{T_w}{T_f} + 0,205)$ $T_w$ $T_f$ 1 - 6

FLUID	AUTHOR	YEAR	CORRECTION FACTOR
GASES	Kutzbender (through Grigull 1970)		$\frac{Nu}{Nu_0} = 1,27 - 0,27 \left[ \frac{T_w}{T} \right]$ cooling  $\left[ \frac{T}{T_w} \right]^{0,55}$ heating
	McEligot, Ormand and Perkins	1966	$\frac{Nu}{Nu_0} = \left[ \frac{T_w}{T} \right]^{-0,5}$
	Zucchetto, and Thorsen	1973	$\frac{Nu}{Nu_0} = \left[ \frac{T_w}{T} \right]^{0,25}$
LIQUIDS	Kaye and Furnas	1934	$Nu_{cooling} = f_{heating} \left[ \frac{Nu_{cooling}}{Nu_{heating}} \right]^n$  $f = 0,5$ liquids $f = 1,0$ gases
	Dittus and Boelter	1930	$Nu = a Re^0 Pr^n$  $n = 0,3$ cooling $n = 0,4$ heating
	Colburn	1933	$\frac{Nu}{Nu_0} = \left[ \frac{T_w}{T} \right]^{0,14}$
	Kraussold	1933	$Nu = a Re^{0,8} Pr^{0,054} P_r^n$  $n = 0,37$ heating $n = 0,3$ cooling
	Sieder and Tate	1936	$\frac{Nu}{Nu_0} = \left[ \frac{Nu_{wall}}{Nu_0} \right]^{0,14}$
	Michejev and Zukauskas (through Grigull 1970)	1952	$\frac{Nu}{Nu_0} = \left[ \frac{Nu_{wall}}{Nu_0} \right]^{0,14}$

FLUID	AUTHOR	Year	CORRECTION FACTOR
LIQUIDS	Jakovlev (through Gregorig 1970)	1955	$\frac{Nu}{Nu_0} = \left[ \frac{Pr}{Pr_0} \right]^{0.11}$
	Kutateladze (through Gregorig 1970)	1968	$\frac{Nu}{Nu_0} = \left[ \frac{Pr}{Pr_0} \right]^m$ $m = 0.25$ (cooling) $m = 0.01$ (heating)
	Malina and Sparrow	1964	$\frac{Nu}{Nu_0} = \left[ \frac{Pr}{Pr_0} \right]^{0.01}$
	Hufschmidt, Burck and Riebold	1966	$\frac{Nu}{Nu_0} = \left[ \frac{Pr}{Pr_0} \right]^{0.11}$
	Hackl and Groll	1969	$\frac{Nu}{Nu_0} = 0.645 \left[ \frac{Pr}{Pr_0} \right] + 0.355$
	Shannon and Dejev.	1969	$\frac{Nu}{Nu_0} = \left[ \frac{Pr}{Pr_0} \right]^m$ $m = f(n, Gz)$ in graphical form
	Gregorig	1970	$\frac{Nu}{Nu_0} = \frac{C_1 Pr_0^{0.25}}{Pr}$
	Kutateladze	1972	$\frac{Nu}{Nu_0} = \left[ \frac{Pr}{Pr_0} \right]^{0.11}$
LIQUIDS	Gregorig	1976	$\frac{Nu}{Nu_0} = \left[ \frac{Pr}{Pr_0} \right]^F$ $F = \frac{0.20(1 - xPr)^{0.20} + 0.0757}{(Pr_0)^{0.05} (Pr_0)^{0.02} (1 - xPr)^{0.01}}$ $x = \frac{Pr_0 - Pr}{Pr_0 - Pr_0}$

From this Table it may be concluded that the correlations for gases and liquids are of different form.

For gases the form

$$\frac{Nu}{Nu_0} = f_n(T/T_{ref})$$

appears to be the only form of correction

For liquids there is a large variety of correction forms however, it is physically most likely that use of the Prandtl group may be the most reliable form. The correction term of Gregorig (1976) is thus possibly the most favourable correction method.

## 2.5 CONCLUSIONS

Data and correlations for the heat transfer coefficient for laminar and turbulent flow of fluids with variable physical properties, developing velocity profiles and free convection effects are inadequate. It is therefore not a problem for the design engineer to obtain a reliable coefficient for most circumstances and for unusual circumstances reliable prediction methods are available.

The area of transition flow has not been extensively examined and methods for determining the extent of this region and obtaining heat transfer coefficients may be unreliable.

More data must be collected for the transition region before reliable predictions of heat transfer coefficients may be made.



### 3 EXTENDED CHRONOLOGICAL SURVEY

#### 3.1 TURBULENT FLOW CORRELATIONS

Although turbulence as such was not observed until circa 1884, Boussinesque in 1877 proposed a theory of *eddy diffusion* to account for the larger measured pressure gradient in pipe flow than that predicted by the theory of Hagen (1839) and Poiseuille (1841, 1846). He effectively introduced a mixing coefficient  $A$  for the Reynolds stress in turbulent flow by defining

$$-\rho \overline{u'v'} = \epsilon \frac{d\bar{u}}{dy}$$

This has the disadvantage that  $A$  is not a property of the fluid but is dependent on the mean velocity.

Later Boussinesque (1905) from dimensional analysis derived the functional form

$$h = \frac{k}{d} \phi \left[ \frac{U_m}{v} \right], \left[ \frac{U_m}{v} \right]$$

where  $\phi$  and  $\psi$  were functions to be derived from experimental results. He did not, however, verify this experimentally.

Around the same time Nusselt (1909) also using dimensional analysis, suggested that

$$h = \frac{k}{d} \left[ \frac{U_m}{v} \right]^n$$

and experimentally determined  $n = 0.786$  and  $t = 15.90$ . The group  $\frac{U_m}{v}$  was later to be termed the *Nusselt number*.

In 1917 Nusselt extended his previous results to include a developing velocity profile and determined that for air

$$Nu_x = 0.03622 \left[ \frac{U_m}{v} \right]^{0.8} \left[ \frac{\rho U_m C}{k} \right]^{0.4}$$

was found to hold. The  $x$  term allowed for the developing profile.

G. I. Taylor (1916-1919) extended Reynold's analogy to include two regions of flow inside the pipe, a laminar region where  $\epsilon$  (the "coefficient of turbulent exchange") is negligible and a turbulent region where the viscosity is negligible. From this he deduced the relation

$$\frac{C_p \rho U_m}{k} = \frac{1}{\epsilon} \left[ 1 + \frac{U_m}{v} (Pr - 1) \right]$$

where  $u_c$  is the velocity at the lamina-turbulent layer

Grober (1921) proposed the provisional equation for both liquids and gases

$$Nu = \frac{h_c d}{k} = \frac{u_c \rho C_p d}{k} \quad (79)$$

possibly as an extension of Nusselt's results. This was later criticised by McAdams and Frost (1924) as failing to allow satisfactorily for variations in diameter and velocity

McAdams and Frost (1922) noted that a *critical velocity* that defined the boundary between laminar and turbulent flow could be expressed as

$$u_c = \frac{16\mu}{\rho d}$$

and also noted that in certain cases the transition could be delayed far in excess of this velocity. Using the results from experiments with light oils and water they suggested a simplified form of the Boussinesque equation as

$$Nu = a Re^b$$

where the viscosity was taken at the average film temperature in an attempt to correlate the data satisfactorily.

Two years later McAdams and Frost (1924), after experimenting with the heating of water put forward the equation

$$Nu = 3 \left[ 1 + \frac{50d}{L} \right] Re^{0.8}$$

after noticing a pronounced effect of the  $L/d$  ratio

Rohsenow (1924), using collected data and basing physical properties on the film temperature proposed an equation of the form

$$Nu = a Pr^{0.6} Re^{0.6}$$

for use with gases and liquid in flow well above the critical value. The use of the  $Pr^{0.6}$  suggests the influence of free convection

In 1925 Prandtl developed his mixing length hypothesis

$$\tau = \rho l^2 \left| \frac{du}{dy} \right| \left| \frac{du}{dy} \right| \quad (l \text{ is the Prandtl mixing length}).$$

This theory has subsequently been used extensively and still finds numerous applications

Cox (1928), using semi-theoretical considerations derived the equation

$$Nu = aRe^{0.5} Pr^{0.33}$$

and based the physical properties on the film temperature as had McAdams and Frost (1922).

Morris and Whitman (1926) experimented extensively with three petroleum oils and using the form

$$Nu = a (Re)^{0.5} (Pr)$$

presented the results in graphical form, differentiating between heating and cooling. The Nu was defined as a point value rather than length average value and the use of average fluid properties rather than film properties gave more consistent results. Figures 4 and 5 are the plots obtained and it is interesting to note that there is more scatter in the data for cooling, suggesting an additional mechanism not allowed for in the equation used.

FIGURE 4 The heat transfer coefficient as a function of a dimensional Reynolds number for heating of oils (Morris and Whitman 1926)

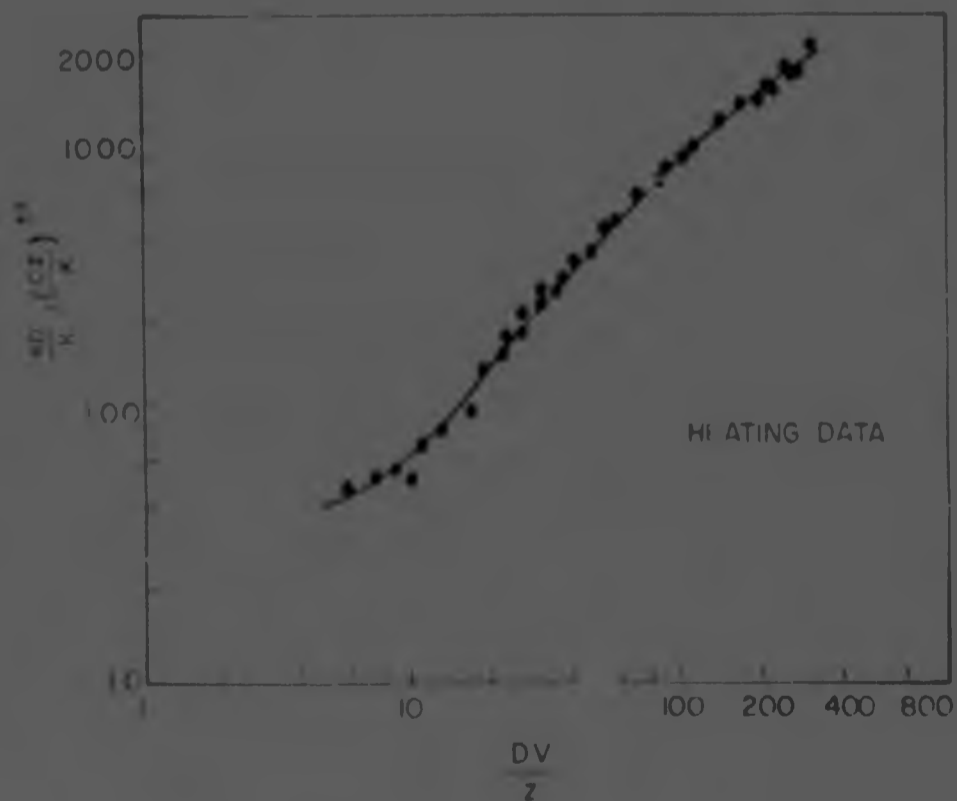
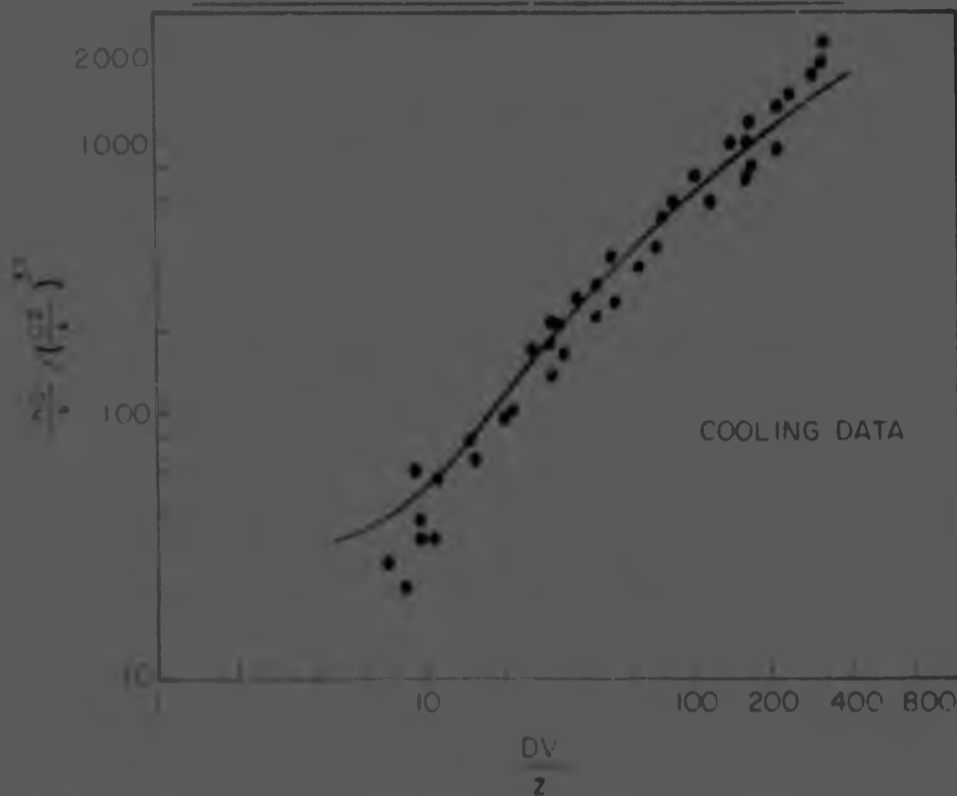


FIGURE 7 The heat transfer coefficient as a function of a dimensional Reynolds number for cooling of oils. (Morris and Whitman 1928)



Morris and Whitman (1928) further developed the results of Taylor (1916 - 1919) by observing that for smooth pipes

was proportional to  $Re^{0.75}$ . This led to the equation:

$$Nu = \frac{0.62}{2} \frac{1}{1 + Re^{-0.4} (Pr - 1)}$$

which was found to hold for small Pr only

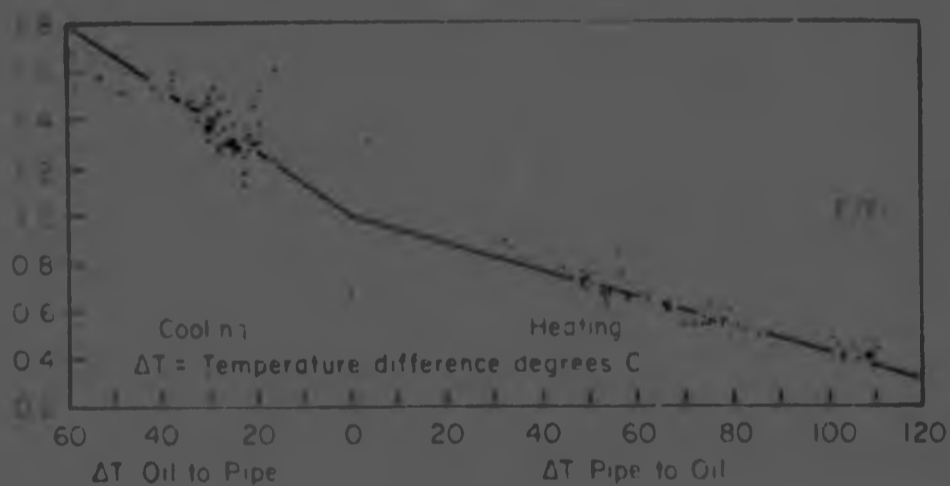
McAdams (1929) noted the effect of the direction of the heat transmission on the Nusselt number and on the friction factor. Physically the situation was represented as in Figure 7. In analyzing the experimental results of two oils they quantified the effect in the form of the wall temperature difference as in Figure 7.

FIGURE 6 Effect of heat transmission on velocity distribution in viscous motion (Keevil and McAdams 1929)



- Curve 1 - isothermal flow.  
 Curve 2 - heating of liquid or cooling of gas.  
 Curve 3 - cooling of liquid or heating of gas.

FIGURE 7 The effect of  $\Delta T$  on the friction factor in laminar flow (Keevil and McAdams 1929)



As with the results of Morris and Whitman (1928) the data for cooling were more scattered indicating an unaccounted for mechanism.

Eglik and Ferguson (1930), after an exhaustive survey of the published data, concluded that it was impossible to deduce any general rule by which heat transfer coefficients could be predicted under any given conditions.

They also proposed that in addition to a film and core region there was an intermediate layer, which led to the equation

$$\frac{au}{k_o} = A + B (Pr - 1) + C(Pr - 1)^2$$

where A, B and C are functions of the Re. Experimenting with water, they noted that there was agreement between the equation and the experimental data. They also postulated that the scatter in experimental results could be caused by a free convection effect.

Dittus and Boelter (1930) in a now classic paper experimentally determined correlations for the heating and cooling of oils. As with previous investigators they could not reconcile the heating and cooling data and suggested the equations

$$Nu = aRe^{0.8}Pr^{0.4} \quad \text{for heating, and}$$

$$Nu = bRe^{0.8}Pr^{0.3} \quad \text{for cooling}$$

The effect of an entrance region was noted but was not correlated due to insufficient data. An equation for laminar flow was also suggested in which a  $d/L$  term was included for entrance effects and the term  $T_{\infty}$  to account for free convection effects

Lawrence and Sherwood (1931) investigated the effect of the tube length using water. Stender\* in 1930 had noted that the use of the factor  $\left[ \frac{d}{L} \right]^2$  used as a multiplier in the Nusselt equation indicated zero heat transfer for infinite pipes, and had proposed the equation

$$Nu = a(RePr)^{0.75} + bRePr(d/L)$$

However, using experimental data on flow without a developing section Lawrence and Sherwood concluded that the  $d/L$  ratio had no effect and proposed the equation

$$Nu = aRe^{0.7}Pr^{0.4}$$

They noted, however, that for oils in the semi-turbulent region the  $d/L$  ratio appeared to have a pronounced effect but did not have sufficient data to correlate the results.

Drew, Hogan and McAdams (1931) reviewed the available equations and using previously published data concluded that no models were adequate to describe the true situation, and suggested

\*Stender, W. Veröfentlicht Siemens Konzern 9, 88 (1930) as referred to by Lawrence and Sherwood (1931).

the use of the group  $\frac{WCp}{kL}$  (Graetz number) to correlate the data. It is very interesting to note that the  $Gz = aRePr^{0.4}L$  which fits in with previous correlations

Nusselt (1931), using the data of Burbach\* and of Eagle and Ferguson (1930) substantiated his postulate that the equation was of the form

$$Nu = a(Re)^n (Pr)^{0.4} \left(\frac{L}{D}\right)^{0.0652}$$

Sherwood and Petric (1932) experimented extensively with the heating of several liquids of  $Pr$  from 1 to 20 and successfully correlated the results using an equation of the Dittus and Boelter (1930) form as suggested in 1931. They further noted that the results were inaccurate for lower  $Re$ .

Murphree (1932) criticised the Prandtl model as not showing the effect of large  $Pr$  effectively and, assuming the model of eddy currents whose value was zero at the wall and increased to a constant value in the bulk of the fluid, derived the formula

$$Nu = \frac{1}{\frac{1}{1 + \frac{1}{2} \left[ \frac{1}{Pr} \right]^{0.25}} + \frac{1}{Pr}} \left[ \frac{1}{2} \left( \frac{1}{Pr} \right)^{0.25} + 1 \right]$$

where  $\left[ \frac{1}{2} \left( \frac{1}{Pr} \right)^{0.25} + 1 \right]$  is a complicated function. When applied to experimental results the model was found to be reasonable.

Colburn (1933) introduced the  $j_H$  factor which he defined as

$$j_H = \frac{h}{C_p G} Pr^{1/3}$$

and postulated a direct link between this and the friction factor  $f = \frac{h}{C_p G} Pr^{1/3}$

Using others' experimental results and basing the fluid properties on a film temperature defined as

$$j_H = 0.023 Re^{-0.2} Pr^{1/3} \left(\frac{L}{D}\right)^{-0.14} \quad \text{for turbulent flow and}$$

$$j_H = 0.36 Re^{-1} Pr^{1/3} \left(\frac{L}{D}\right)^{-0.14} \quad \text{for laminar flow,}$$

\*Burbach, Th and Hermann, R. "Strömungswiderstand und Wärmeübergang in Rohren", Leipzig 1930, p 45, as referred to by Nusselt (1931).

the use of the group  $\frac{WCp}{k_1}$  (Gratz number) to correlate the data. It is very interesting to note that the  $Gz = aRePr^m L$  which ties in with previous correlations.

Nusselt (1931), using the data of Burbach\* and of Eagle and Ferguson (1930) substantiated his postulate that the equation was of the form

$$Nu = aRe^{0.764} Pr^{0.355} \left[ \frac{L}{D} \right]^{0.0852}$$

Sherwood and Petrie (1932) experimented extensively with the heating of several liquids of Pr from 1 to 20 and successfully correlated the results using an equation of the Dittus and Boelter (1930) form as suggested in 1931. They further noted that the results were inaccurate for lower Re.

Murphree (1932) criticised the Prandtl model as not showing the effect of large Pr effectively and, assuming the model of eddy currents whose value was zero at the wall and increased to a constant value in the bulk of the fluid, derived the formula

$$Nu = \frac{1}{1 + \frac{1}{Pr^{1/4}}} \left[ \frac{1}{1 + \frac{1}{Pr^{1/4}}} \right]^{1/4} \left[ \frac{L}{D} \right]^{1/4} Pr^{1/4}$$

where  $\phi$  is a complicated function. When applied to experimental results the model was found to be reasonable.

Colburn (1933) introduced the  $j_H$  factor which he defined as

$$j_H = \frac{h}{G} Pr^{1/4}$$

and postulated a direct link between this and the friction factor  $f' = \frac{R}{\rho u^2}$

Using others' experimental results and basing the fluid properties on a film temperature defined as

$$T_f = T_{avg} + \frac{1}{2}(T_{wall} - T_{avg}) \quad \text{for turbulent flow and}$$

$$T_f = T_{avg} + \frac{1}{4}(T_{wall} - T_{avg}) \quad \text{for laminar flow,}$$

\*Burbach, Th and Hermann, R. "Strömungswiderstand und Wärmeübergang in Röhren", Leipzig 1930, p 45, as referred to by Nusselt (1931).



he derived the equations

$$j_H = a + b \left[ \frac{dG}{\mu} \right]^{0.32} \quad \text{for turbulent flow and}$$

$$j_H = a \left[ \frac{dG}{\mu} \right]^{-2/3} \left[ \frac{L}{d} \right]^{-1/4} \quad \text{for laminar flow}$$

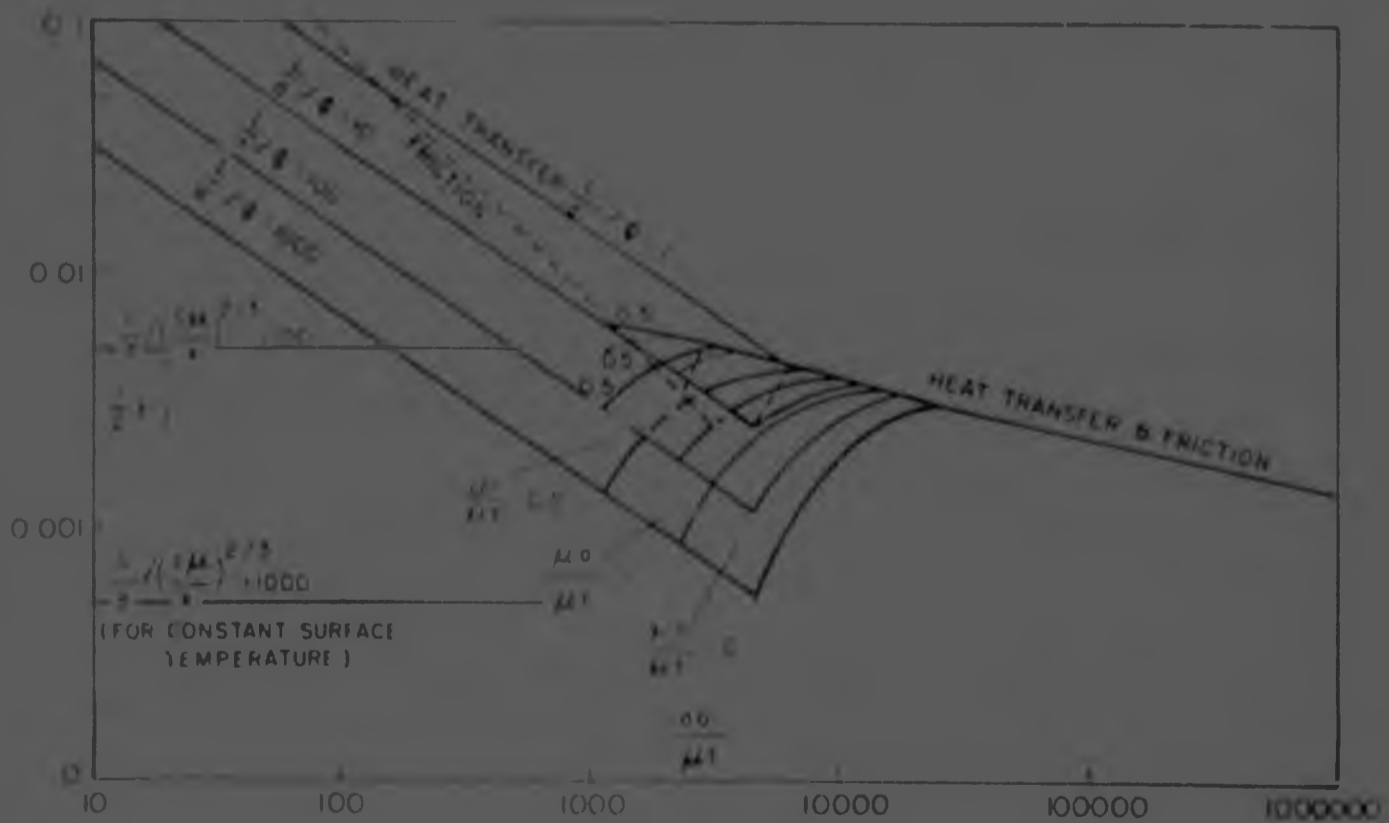
He suggested the use of the parameter

$$\left[ \frac{j_H}{\mu} \right]^{1/4}$$

to bring the heating and cooling data into line and noted that for large Gr, inclusion of the term  $(1 + 0.015 Gr^{0.1})$  was necessary

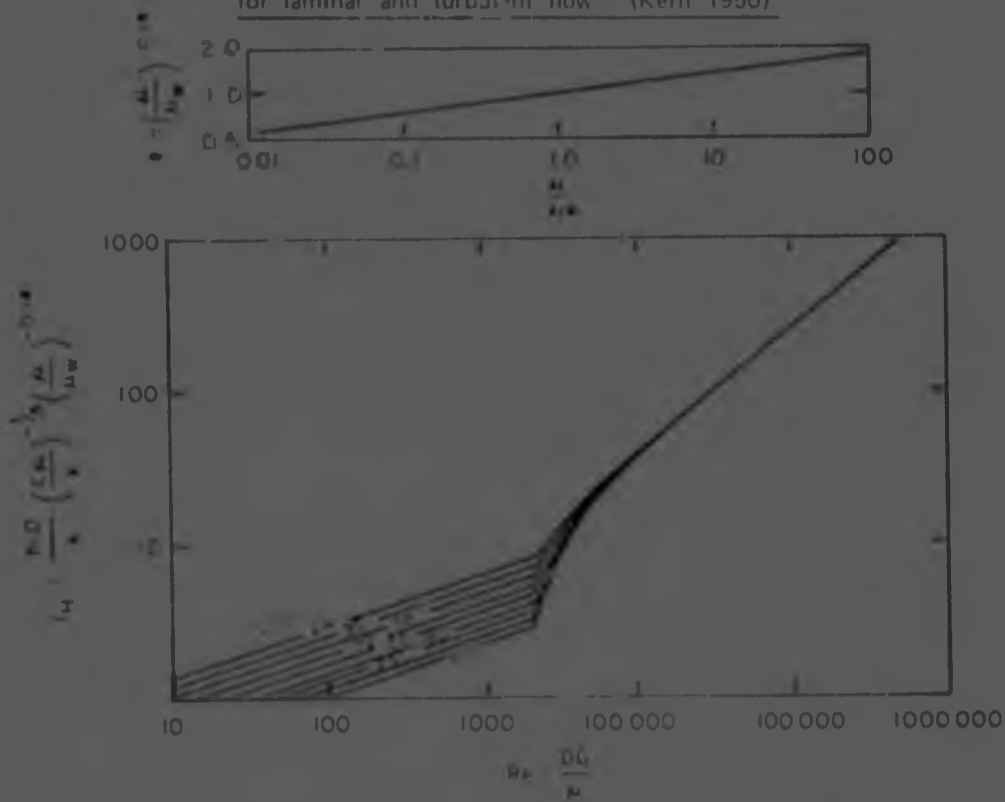
The  $j_H$  and  $f_f$  factors did not correlate in the laminar region but were in good agreement for large Re. In accordance with friction data Colburn presumed an analogous form for the heat transfer coefficient and hence constructed Figure 8, which extends from laminar to turbulent flow. The form of the curve in the transition region is, however, arbitrary and not based on sound experimental data.

**FIGURE 8** The functions  $j$  and  $j_{fr}$  as a function of the Reynolds number for laminar and turbulent flow. (Colburn 1933)



This work of Colburn's was a major advance and graphs of the form of Figure 8 are still used, as for example, in Kern's book "Process Heat Transfer" (Figure 9)

FIGURE 9 The functions  $f$  and  $j_H$  as a function of the Reynolds number for laminar and turbulent flow (Kern 1950)



Kraussold (1933) experimentally determined a correlation of the form

$$Nu = a Re^{0.8} Pr^{0.37} \left[ \frac{d}{L} \right]^{0.054}$$

for the heating of fluids and changed the exponent of the Pr to 0.3 for cooling. This indicated that there was an entrance effect, contrary to previous results.

Kaye and Furnas (1934) advanced the theory of a stationary film at the wall to explain the difference in heating and cooling data and suggested a correction factor

$$h_{cooling} = h_{heating} \left[ \frac{\text{heating}}{\text{cooling}} \right]^f$$

where  $f$  was 0.5 for liquids and 1.0 for gases.

Sieder and Tate (1936) in a now renowned paper, presented experimental results for the heating and cooling of oils, and based on the results, suggested the use of the factor

to bring the heating and cooling data into agreement. They suggested a value of 0.14 for the exponent of this correction factor and noting that the ratio  $\frac{h}{C_p G}$  was not necessary in turbulent flow produced a correlation as in Figure 6. This form is still widely accepted although the scatter of the original data is very large.

Von Kármán (1939) introduced the concept of a buffer layer between the laminar sublayer and the core in pipe flow. Using this theory and the Reynolds analogy he obtained, using Nikuradse's velocity profiles, (1932), the relation

$$\frac{1}{C} = \frac{2}{C_s} + 2.3 \left[ \frac{2}{C_s} \right]^{0.75} \left\{ Pr - 1 + \ln \left[ 1 + 3.6 (Pr - 1) \right] \right\}$$

which for  $Pr = 1$  reduces to the Reynolds analogy and for  $Pr = 1$  small reduces to a form similar to that of Taylor (1916 - 1919). Comparison of this equation with the results of Dittus and Boelter (1930) was good for  $Pr < 25$ . This effect of the  $Pr$  may be expected when using a Reynolds analogy.

Boelter, Martinelli and Jonassen (1941) extended Von Kármán's model by including variable viscosity in the laminar sublayer and constant viscosity outside. This constant viscosity was an average viscosity, not a viscosity at an average temperature. They derived an equation by adding the three resistances

$$\begin{aligned} \frac{q}{A} &= -k \frac{dT}{dy} && \text{laminar layer} \\ \frac{q}{A C_p} &= - \left[ \frac{1}{Pr} \right] \frac{dT}{dy} && \text{buffer layer} \\ \frac{q}{A C_p} &= - \frac{dT}{dy} && \text{turbulent core} \end{aligned}$$

which were in agreement when compared with the data of Morris and Whitman (1928).

Hausen (1943) used the results of Sieder and Tate (1936) and others with special attention to the effect of  $\mu$  and arrived at the function

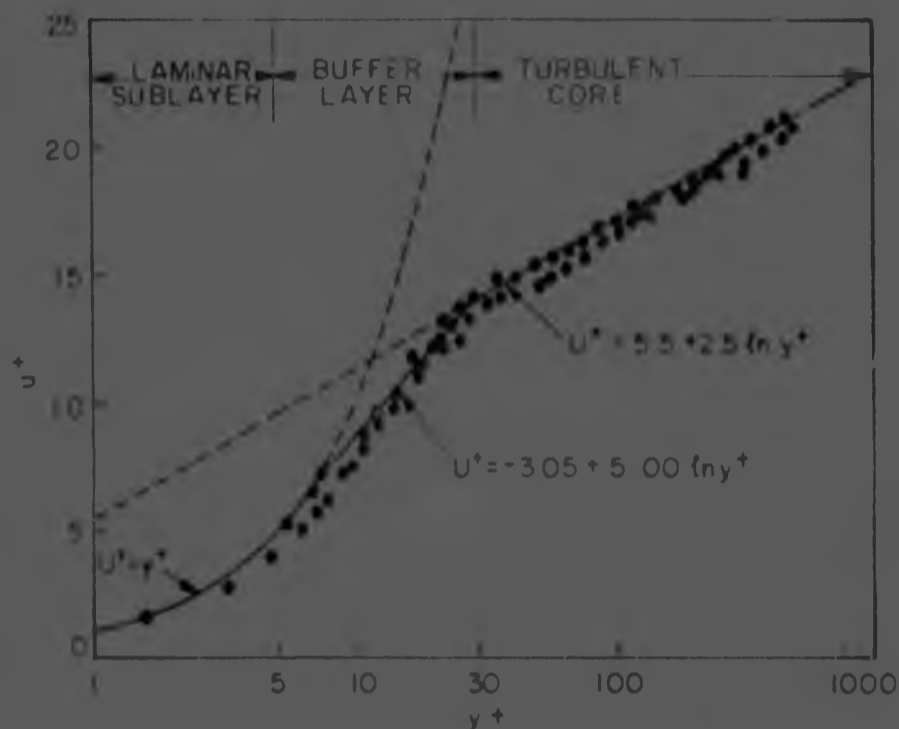
$$Nu = a(Rc^{0.3} - 125)Pr^{0.14} \left[ 1 + (d/L)^{0.3} \right] \left[ \frac{\mu}{\mu_s} \right]^{0.14}$$

This equation does not show the anomaly of zero heat transfer for infinite tube length.

Bernardi and Egan (1945) experimented with ethylene glycol and found that the Dittus and Boelter (1930) form of the equation correlated the results satisfactorily. They also noted that the Stanton number  $\frac{h}{C_p G}$  correlated the results better in the lower turbulent region.

Martinelli (1947) extended the model proposed by Von Kármán to include low Prandtl numbers by considering molecular conduction in the turbulent core. He also constructed a generalised velocity distribution using the results of Nikuradse (1932) and Reichardt (1943) as given in Figure 10.

FIGURE 10 Generalised velocity distribution for turbulent flow in tubes (Martinelli 1947)



Using dimensional analysis, Deissler (1950) determined that the eddy conductivity could be expressed as  $k = n^2 u y$

where  $n^2$  is experimentally determined. From this he developed velocity profiles that agreed with experimental results. Subsequently he extended this analysis (1950) to include variations in physical properties and concluded that the effect of fluid properties across the tube could be eliminated by evaluating the properties at a temperature close to the average of the wall and bulk temperatures.

Two years later Deissler and Ewing (1952) extended this work to include Prandtl number and found that the predicted results agreed with experimental results for air.

Danckwerts (1951) while considering liquid film coefficients in gas absorption proposed the *surface renewal* model. The idea of a stagnant film near the wall was abandoned and the idea of eddies continually exposing fresh sections of the surface was considered.

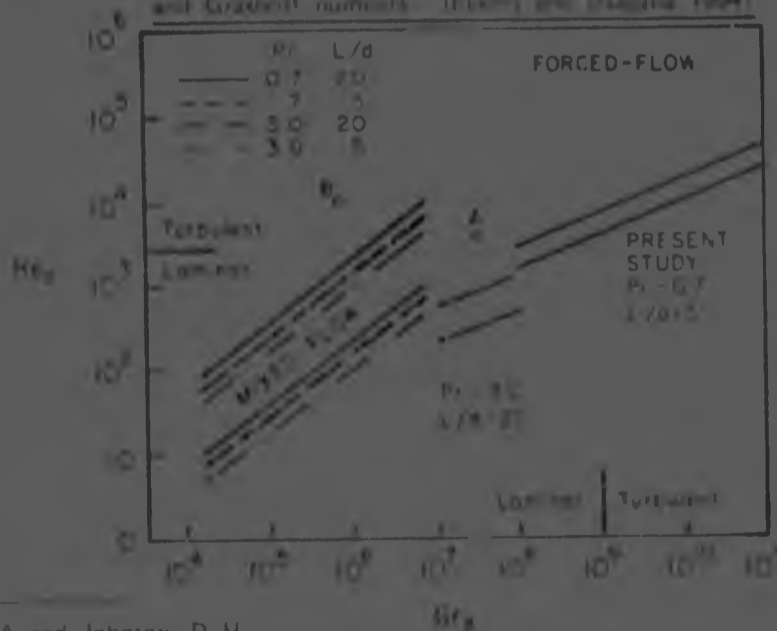
Lyons (1951), using the assumptions of fully developed velocity and temperature profiles, uniform heat flux and  $Pr_1$ , arrived at an integral form for the Nusselt number which was shown to be in reasonable agreement with experimental results. He further noted from this integral form that there was an expected minimum for the  $Nu$  as  $Pr \rightarrow 0$  in turbulent flow, which was around 7. This was later used by Churchill (1977) in his method of using asymptotes to obtain interpolatory equations.

Lin, Moulton and Putnam (1953) assumed an eddy viscosity in the region close to the wall and derived an equation that fitted experimental velocity profiles very well. This then eliminated the concept of a laminar sublayer as in the surface renewal model.

Deissler (1954) pointed out that the inadequacy of his previous results was due to the expression used for the eddy diffusivity close to the wall. A new expression, modified to account for the effect of kinematic viscosity in reducing turbulence close to the wall, gave improved results and indicated that except at low  $Re$  the entrance effect decreased with increasing  $Pr$ . His analysis also indicated that the effect of variable viscosity could be eliminated by evaluating the viscosity at a temperature which was a function of the  $Pr$ .

Eckert and Diaguila (1954) experimented with air flowing in vertical tubes and found that with properties based on a film temperature Hausen's equation underestimated the heat transfer coefficient at high  $Re$ . Using the results of Martinelli and Boelter and Watzinger and Johnson (1939)\* they divided the fluid flow into the three regions of forced, mixed and free flow using the criterion  $Re_s = a/(PrG_s)^{0.33}$  and plotted the results as in Figure 11.

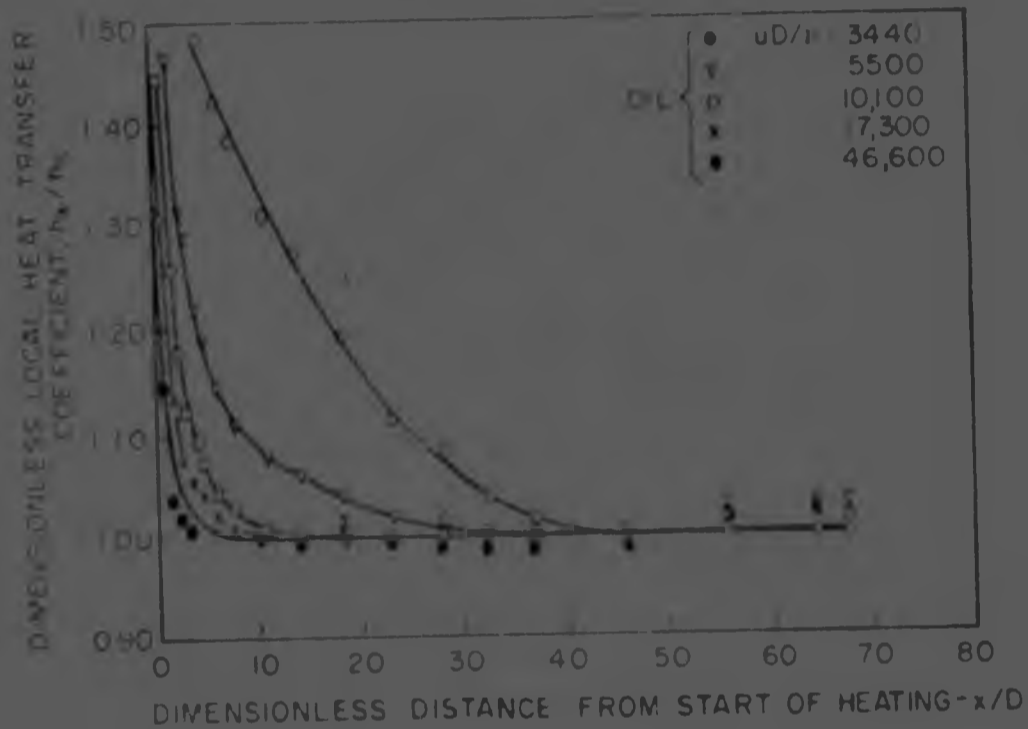
FIGURE 11 Subdivision of the flow regions as a function of the Reynolds and Grashof numbers (Eckert and Diaguila 1954)



\* Watzinger, A and Johnson, D H  
*Forsch. Ing. Arch.* 10: 182 (1939)  
 as referred to by Eckert and Diaguila (1954)

Hartnett (1955) experimented with oil and water with a fully developed velocity profile to determine the *thermal entrance length*, defined as the length needed for the heat transfer coefficient to reach a constant value. He concluded that at high  $Re$  the  $Pr$  had little effect on the entry length. Figure 12 is a typical result.

FIGURE 12 Thermal Entry Length results for oil flow in the transition region (Hartnett 1955)



Deissler (1955) analytically derived expressions for the thermal entrance length for uniform heat flux and uniform wall temperature and obtained results which agreed well with experimental results.

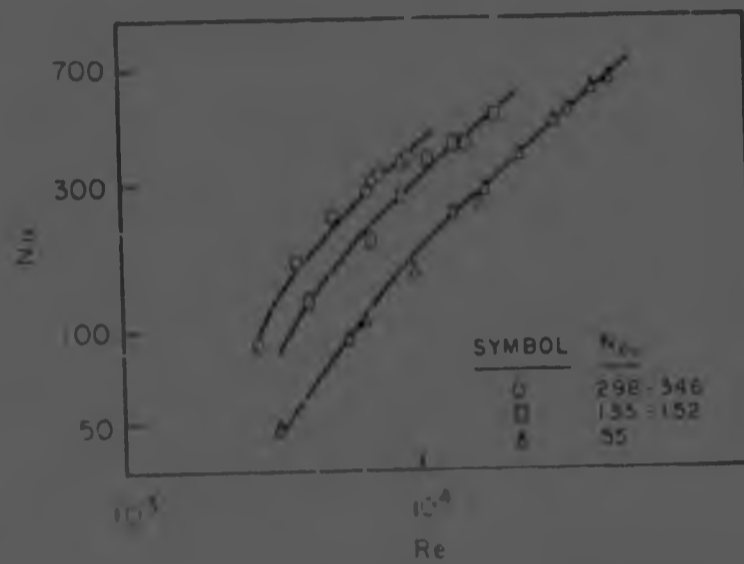
Sparrow, Hallma and Siegel (1958) analytically derived an expression for the thermal entrance length assuming a uniform temperature and a fully developed velocity profile. A similar approach to that of Grantz was used and yielded results that agreed with those of Deissler (1955). They noted that the effect of increasing  $Pr$  was to decrease the thermal entrance length, which is expected from the physics.

Friend and Metzner (1958) modified and simplified Reichardt's (1943) analysis for the case of moderate  $Pr$  to the form

$$St = \frac{1}{1.2 + (Pr - 1)b(Pr)\sqrt{1/2}}$$

where  $(Pr)$  was experimentally determined to be  $11.8 Pr^{-3}$ . Interesting to note is the consistency of the results through the transition region in Figure 13

**FIGURE 13** The Nusselt number as a function of the Reynolds number for heating of molasses (Friend and Metzner 1958.)



Azer and Chao (1960) modified Prandtl's mixing length hypothesis to include a continuous change of momentum and energy during the flight of an eddy. This translates to

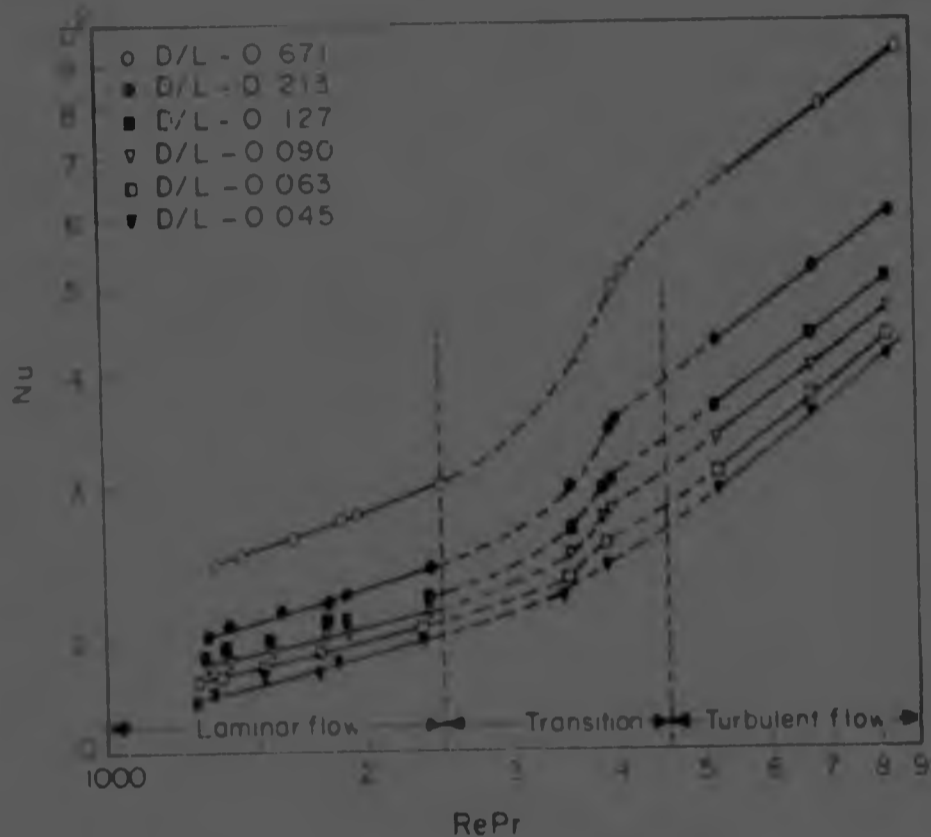
$$\frac{L}{\lambda} = \text{variable}$$

Results from this model agreed with experimental results

Jackson, Spurlock, and Purdy (1961) experimented with air with a developing velocity profile and found an effect of tube length but not of free convection. The tubes used in the experiments were not long enough to allow the boundary layer to completely fill the tube. Their results for the lower turbulent-transitional region were consistent. See Figure 14

FIGURE 14

Nusselt number as a function of the Reynolds Prandtl group for air (Jackson, Spurlock and Pomeroy, 1961)



Petukhov and Popov (1963) used a numerical procedure to solve the heat and momentum equations assuming variable physical properties, and found a more generalised form of the Lyon (1951) equation. From experimental results for constant physical properties they derived the interpolation formula

$$Nu = \frac{f_1 \cdot 8 \cdot RePr}{1.07 + 12.7 \sqrt{f_1/8} (F^{0.1} - 1)}$$

where  $f_1$  is similar to the form of Friend and Metzner (1958). For variable physical properties Petukhov and Popov (1963) suggested the use of

$$\frac{Nu_s}{Nu_t} = \left[ \frac{T_s}{T_t} \right]^0$$

for the temperature loads. They also mentioned the factor

$$\frac{Nu_t}{Nu_s} = \left[ \frac{\sqrt{1 + \frac{1}{Pr_s}}}{\sqrt{1 + \frac{1}{Pr_t}}} \right]^0$$

to correct equation (10b) to the temperature literature



Dipprey and Sabersky (1963) looked at the effect of surface roughness and proposed the *cavity vortex* theory. This assumes that the wall consists of small cavities and that the flow is laminar and about these cavities consists of one or more standing vortices.

Metals and Eckert (1964) summarised the correlations for horizontal and vertical flow and presented the results in graphical form. Available experimental data did not permit the establishment of a limit between free and mixed convection in horizontal tubes. Their results are reproduced in Figures 15 and 16.

FIGURE 15 The extent of the flow regions for flow in horizontal tubes  
(Metals and Eckert 1964)

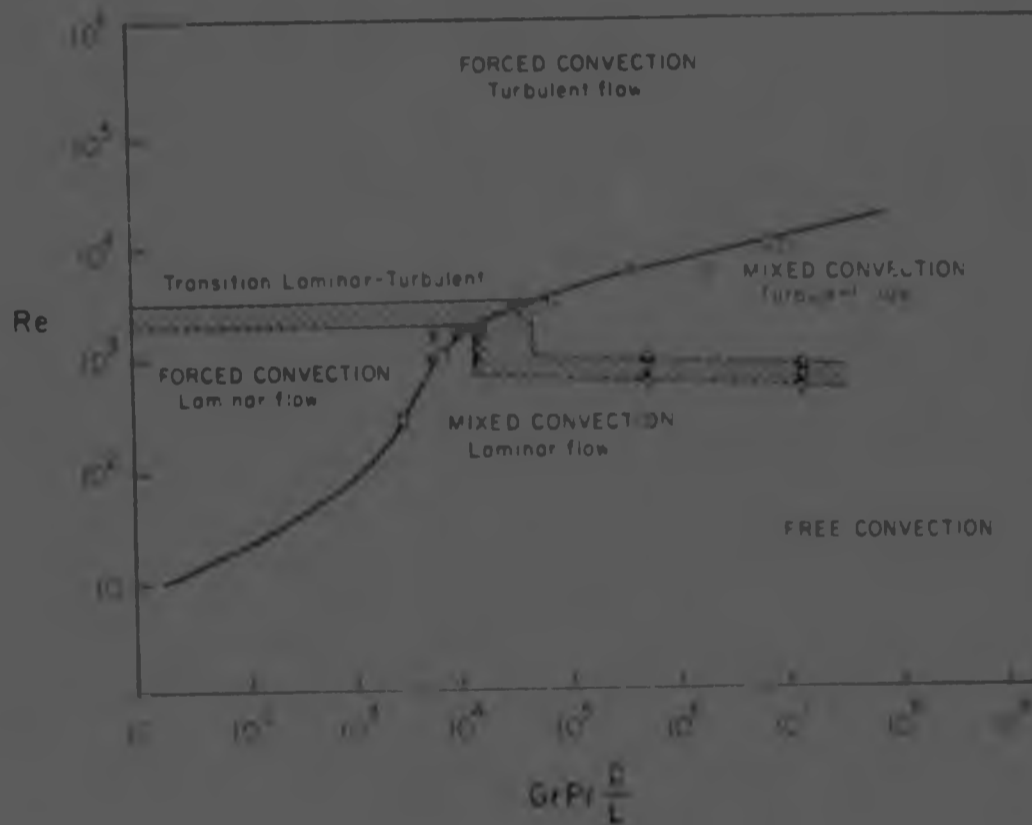
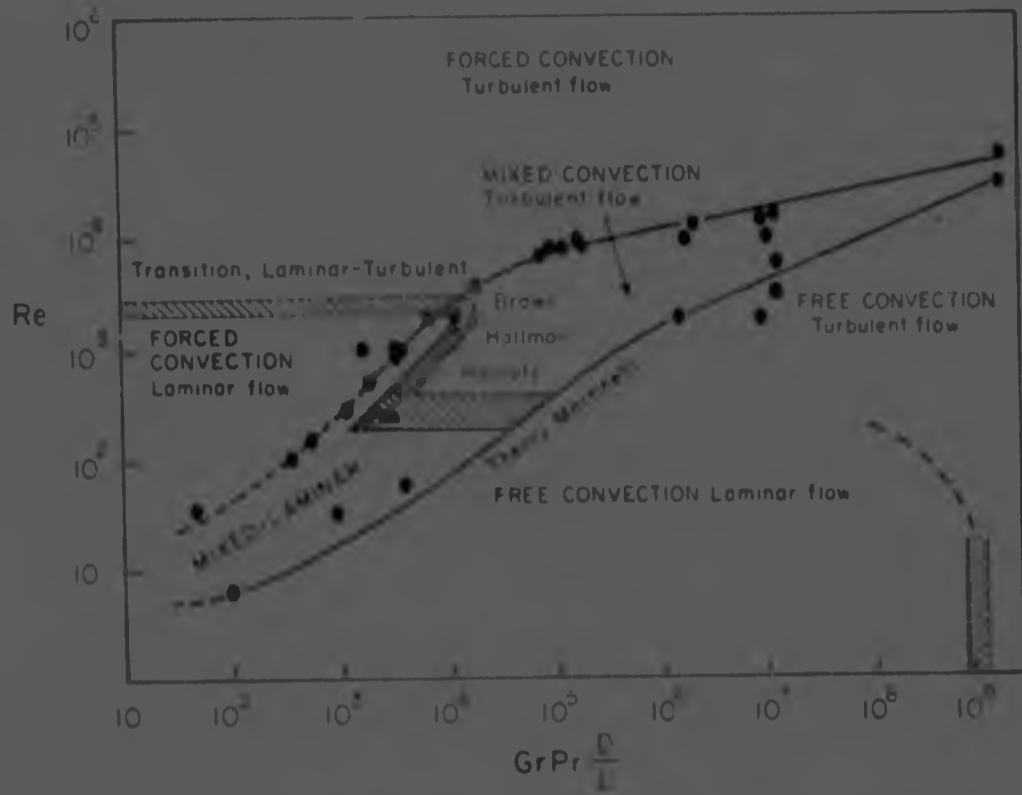
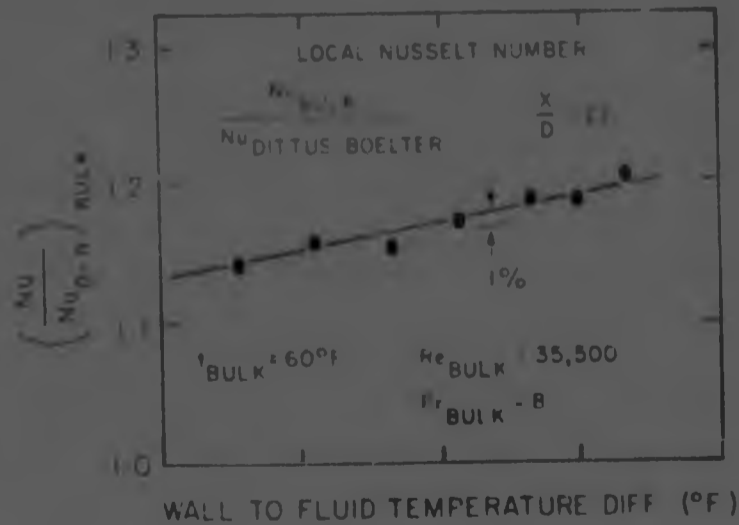


FIGURE 16 The extent of the flow regions for flow in vertical tubes  
(Metals and Eckert 1964)



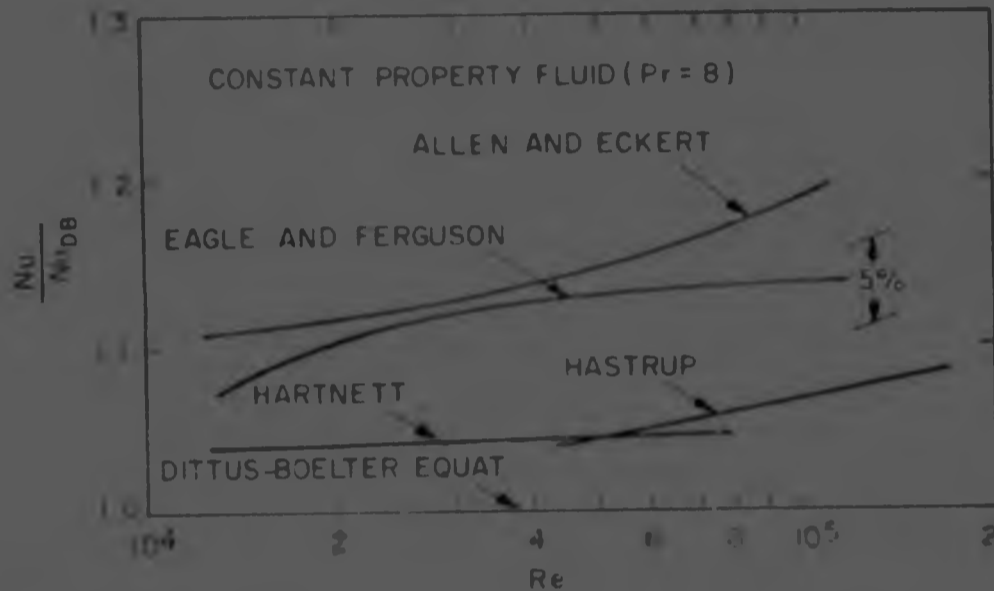
Allen and Eckert (1964) experimented with water and used the wall to bulk fluid temperature difference, to show the effect of variable physical properties, to correlate the results. Their results, compared with the equations of Dittus and Boelter (1930) are shown in Figure 17. They noted that the entrance effects were very similar to those of Hartnett (1955).

FIGURE 17 Comparison of results of Allen and Eckert (1964) with Dittus and Boelter (1930)  
(Allen and Eckert 1964)



They also noted that the correction factor of Sieder and Tate (1936) was not satisfactory but did not suggest an alternative. Their data are compared with those of others in Figure 18

**FIGURE 18** Comparison of heat transfer coefficients with those of other investigators. Thermally developed, constant property case. Pr = 8. Uniform wall heat flux. (Allen and Eckert 1964)



Malina and Sparrow (1964) extended the work of Allen and Eckert (1964) to include oils and suggested the correction factor

$$\left[ \frac{\mu_s}{\mu_w} \right]^{0.05}$$

to be used for variable physical properties

Subbotin, Ibragimov and Nomofilov (1965), after experiments over a wide range of Re and Pr, suggested that

$$Nu = 7.24 + \frac{0.025 Re}{\log Re}$$

to be used for constant properties

Kolar (1965) derived the semi-theoretical relation

$$Nu = a \left[ \frac{u d}{\nu} \right]^b Pr^c$$

where the group was called the turbulent Re,

Hufschmidt, Burck and Riichold (1966) suggested, after experiments with water, that the group

$$\left( \frac{\mu_b}{\mu_s} \right)^{0.14}$$

be used for temperature dependent properties and used with the equation of Petukhov and Popov (1963).

McEligot, Ormand and Perkins (1966) found that at  $Re > 4000$  the Dittus and Boelter (1930) relation was satisfactory and suggested the use of

$$\frac{Nu_w}{Nu_s} = \left( \frac{T_w}{T_s} \right)^{0.4}$$

for variable properties. They used the ratio of  $T_w/T_s$  plotted against the modified Reynolds group

$$Re = \frac{4m T_b}{\mu_d T_s}$$

to distinguish between the flow regions.

Gowen and Smith (1967) experimentally investigated the effect of the  $Pr$  based on a film temperature on the temperature profile. Assuming equal eddy diffusivities for heat and momentum they derived a universal temperature profile of the form

$$t^* = A_1 \ln \left( \frac{1}{1 - t^*} \right) + B$$

where  $A_1$  was constant and  $B$  a function of the  $Pr$ . Their experiments showed  $t^*$  to have a dependence on the  $Re$ .

Polvakov (1968) after experimentation, concluded from the inadequacy of the Petukhov and Popov (1963) equation to correlate results at large  $Gr$ , that free convection must still be influencing heat transfer in the turbulent region.

In 1968 the Engineering Sciences Data Unit published the first of their summaries on heat transfer in pipe. These summaries are updated at regular intervals.

Lawn (1969), from a theoretical discussion, concluded that the turbulent  $Pr_t = \frac{\epsilon_H}{\nu}$  must be dependent on the  $Re$  which severely limits its usefulness in practical calculations and which questions its physical significance.

Hufschmidt, Burck and Ruibold (1966) suggested, after experiments with water, that the group

$$\left[ \frac{Nu}{Pr} \right]^{0.14}$$

be used for temperature dependent properties and used with the equation of Petukhov and Popov (1963).

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$$\frac{Nu_s}{Nu_b} = \left[ \frac{T_m}{T_b} \right]^{0.4}$$

for variable properties. They used the ratio of  $T_m$  to  $T_b$  plotted against the modified Reynolds group

$$Re = \frac{4d_1 T_m}{\mu d T_w}$$

to distinguish between the flow regions.

Gowen and Smith (1967) experimentally investigated the effect of the Pr based on a film temperature on the temperature profile. Assuming equal eddy diffusivities for heat and momentum they derived a universal temperature profile of the form

$$t^+ = A_s \ln y^+ + B_s$$

where  $A_s$  was constant and  $B_s$  a function of the Pr. Their experiments showed  $t^+$  to have a dependence on the Re

Polyakov (1968), after experimentation, concluded from the inadequacy of the Petukhov and Popov (1963) equation to correlate results at large Gr, that free convection must still be influencing heat transfer in the turbulent region.

In 1968 the Engineering Science Data Unit published the first of their summaries on heat transfer in pipes. These summaries are updated at regular intervals.

Lawn (1969), from a theoretical discussion, concluded that the turbulent  $Pr_t = \frac{\mu}{\rho k}$  must be dependent on the  $Gr$  which severely limits its usefulness in practical calculations and which questions its physical significance.

Hackl and Coll (1969) and Hausen (1969) suggested the correction factor

$$Nu/Nu_0 = 0.645 \left[ \frac{\mu_0}{\mu} \right] + 0.355$$

to be used in preference to the Sieder and Tate (1930) form. This correction was obtained after numerous experiments with oils.

McEligot, Smith and Bankston (1970), in an excellent paper, reviewed the theoretical models for turbulent flow, and using a finite difference scheme solved the flow equations assuming temperature dependent physical properties and found a substantial improvement on previous models. They concluded that mixing length models are better than eddy diffusivity models for predicting the heat transfer and suggested the Van Driest mixing length model as the best. A summary of the models is given in the Appendix 4.4.

Gregoric (1970), from theoretical considerations, arrived at a complicated correction factor for physical properties. This factor was of the form

$$K_b K_{Pr}$$

where each  $K_i$  is a function of the temperature and the Prandtl ratio.

Herbert and Sterns (1971) experimented with water flowing in vertical tubes and noted that at high Re the product GrPr had little effect, but had a strong influence at low Re.

Gross and Thoma (1972) introduced the term  $\frac{d\mu}{dx}$  into the eddy penetration model and found that this improved the correlation. This suggests that it may be necessary to include the  $\frac{d\mu}{dx}$  term in empirical correlations.

Lauder and Spalding (1972) summarised the models of turbulence in a very readable form and this is recommended for a study of the theoretical models.

Pennell, Sparrow and Eckert (1972) and Kudva and Sesonske (1972) found from experiments that in the region near the wall and in the region near the centre line the ratio of the turbulence intensity to the friction velocity is independent of the Re.

Zucchetto and Thorsen (1974) noted that for air the variable physical properties could be corrected by

$$\frac{Nu}{Nu_0} = \left[ \frac{\mu_0}{\mu} \right]^{0.17}$$

Mori, Sakakibara and Tanimoto (1974) analytically found that at  $Gz = 50$ , the ratio of the thermal conductivity of the wall to that of the fluid, and the thickness of the tube wall, may become significant factors on the heat transfer.

Sleicher and Rouse (1975) reviewed a few empirical equations and recommended Petukhov (1970) as the best. Using others' experimental data they developed an equation similar in form to that of Petukhov but in which the exponents of the  $Re$  and  $Pr$  were functions of the  $Pr$ .

Hrycak and Andrushkiw (1974) used Meissner's entropy principle to develop a method of predicting the critical  $Re$  and found the method to be fairly accurate.

Gregorig (1976) considered the case where the  $Pr$  number is a non-linear function of temperature, and using similarity considerations arrived at a correction factor

$$\frac{N_{Nu}}{N_{Nu_0}} = \left[ \frac{Pr}{Pr_0} \right]^P$$

where  $P$  is a complicated function of the  $Re$  and  $Pr$ .

Hanna and Sandall (1978) developed an analytical approximation for

$$\frac{9/\sqrt{2}}{[\sqrt{2}]}.$$

and found it to be in reasonable agreement with Petukhov's (1970) experimental results.

Polley (1979) using data on air, determined that the Engineering Sciences Data Unit correlations were significantly better than the Dittus and Boelter (1930) or Sieder and Tate (1936) equations.

No further papers have been traced since this

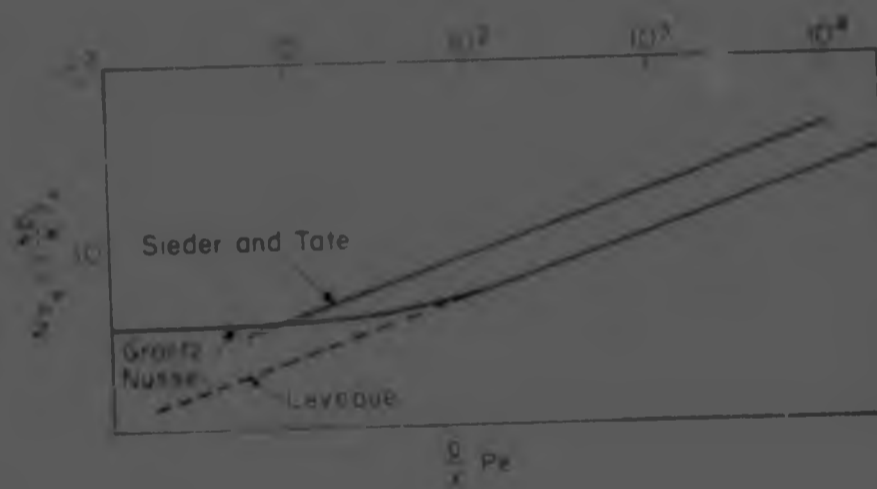
32 LAMINAR FLOW CORRELATIONS

Graetz in 1885 and later Nusselt in 1910 used the following assumptions to obtain a solution for laminar pipe flow

- i) fully developed parabolic velocity profile
- ii) constant wall temperature boundary condition
- iii) negligible axial conduction
- iv) constant fluid properties.

The resulting partial differential equation was solved to give a series solution for the Nusselt number as a function of the Peclet number which gives the result in Figure 19.

FIGURE 19 Local Nusselt number for laminar flow (HSU 1963)



The results of Sieder and Tate and Leveque's extension of the Graetz result, are included in the plot for comparison.

Leveque (1928) extended the result to include a developing velocity profile, giving

$$Nu_x = 1.607 \left[ \frac{d}{l} Re Pr \right]^{1/3} \quad (\text{Figure 1})$$

Dittus and Boelter (1930) using experimental data on the heating and cooling of oils added the term  $\mu/\mu_w$  to include variation in physical properties



Colburn and Houghton (1930) experimented with water in vertical flow and concluded that the heat transfer was independent of the velocity and proposed an expression of the form

$$Nu = 1.13 Pr^{1/4} Gr^{1/2}$$

with properties evaluated at the film temperature.

Park and McCabe (1931) used the use of a constant velocity profile since this was applicable to isothermal cases only. Their experimental results for water and air they derived the equation

$$Nu = 1.13 \frac{Pr^{1/4}}{Re Pr^{1/2}} \left[ \frac{Gr}{Pr} \right]^{1/2}$$

Colburn (1933) introduced the  $Gr$  factor and suggested using the term  $1 + 0.015 Gr^{1/2}$  to correct for buoyancy effects. Experiments were based on a film temperature defined as

$$T_f = \frac{T_w + T_\infty}{2}$$

Reynolds and Taylor (1934) experimented with air and recommended the use of Colburn's (1933) correction factor for the procedure. The equation

$$\frac{Nu}{Gr^{1/2}} = 1.13 Pr^{1/4}$$

was suggested as a correction for various physical properties and they proposed

$$Nu = 1.13 \left[ \frac{Pr}{Pr_s} \right]^{1/4} \left[ \frac{Gr}{Gr_s} \right]^{1/2}$$

Martinelli et al (1942) experimented with vertical flow and liquid air systems developed by Martinelli and Gruber

$$Nu = 1.76 Pr^{1/4} \left[ \frac{Gr}{Gr_s} \right]^{1/2} \left[ \frac{Pr}{Pr_s} \right]^{1/4}$$

and in reasonable agreement with the experimental results. This corrected the use of  $1 + 0.015 Gr^{1/2}$  as a correction as this would imply that the effect of free convection would increase with increasing  $Gr$  which was contrary to what was physically reported.

Heisen (1943) performed experiments to the effect of the  $Gr$  ratio and used Lewis and Prandtl's procedure to give the equation

$$Nu = 1.13 \frac{Pr^{1/4}}{1 + 0.015 Gr^{1/2}}$$

Kern and Othmer (1943) experimented with oils and from the results suggested the term

$$2,25 (1 + 0,01 Gr^{0,3}) \log Re$$

as a correction for free convection effect.

Kays (1955) numerically solved the energy equations using the velocity profiles of Langhaar and neglecting the radial velocity and axial conduction terms. Agreement with experimental results for gases was good and the correction factor

$$\frac{h}{h_{\text{isothermal}}} = \left[ \frac{1 + m}{1 - m} \right]^m$$

where  $m = 0,25$  for the heating and  $0,08$  for the cooling of gases; for temperature dependent properties, was advanced.

Stephan (1959) argued that for constant properties the function was of the form

$$Nu = f_1 \left[ \frac{L}{D} Pe \right]$$

and that for variable properties

$$Nu = f_2 \left[ \frac{L}{D} Re Pr \right]$$

From semi-theoretical considerations he arrived at the interesting form

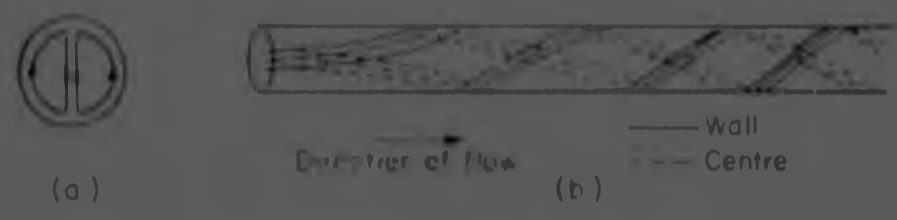
$$Nu = 3,06 + \frac{0,067 (Pr Re_c)^{1/4} L^{1/4}}{1 + 0,1 Pr (Re_c d/L)^{0,83}}$$

Edr (1961) experimented with air and water and found that the data could be correlated using the Grashoff number only in the form

$$Nu = 4,36(1 + 0,06 Gr^{0,3})$$

Oliver (1962) criticised the use of the term  $Gr^{0,3} L$  since the product is a term in  $D^4$  which may have a strong influence not physically expected and suggested the use of  $Gr Pr^{1/4} d$ . He noted that the effect of the tube length on the free convection effects should physically be different for horizontal and vertical tubes and postulated that for horizontal tubes the flow pattern would be as in Figure 20.

FIGURE 20 Probabl. flow pattern in horizontal tube with heated wall,  
 (a) end elevation and (b) horizontal elevation (Oliver 1962)



Brown and Thomas (1965) experimented with water in horizontal tubes and found that contrary to Martinelli's work on vertical tubes, it appeared that free convection effects increased with the flow rate. This was reconciled by postulating that at higher flow rates the temperature gradient at the wall would be higher than at low flow rates and hence the free convection effect would be greater.

Mori et al (1966) found from experiments with air at high heat flux that the free convection had a pronounced effect which started at about  $ReRa = 10^3$ . They also noted that the critical Reynolds appeared dependent on the intensity of the secondary flow and suggested

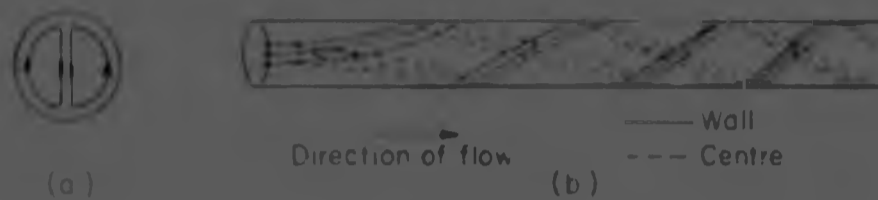
$$Re_{cr} = 128 (ReRa)^{1/4}$$

McComas and Eckert (1966) noticed that the effect of free convection in air flows increased as the ratio of  $Gr$  to  $Re$  increased and suggested using Oliver's (1962) equation

Iqbal and Staszko (1967) looked analytically at the effect of the tube inclination on combined free and forced convection and concluded that this had little effect on the  $Nu$  but did affect the friction factor. They noted that a temperature dependent density affected mainly the velocity field and not the temperature field.

Mori and Futagami (1967) extended the work of Mori et al (1966) and experimentally determined that the vortices in the secondary flow pattern moved closer to the wall with increasing  $ReRa$  (intensity of the secondary flow). The results are reproduced in Figure 21.

FIGURE 20 Probable flow pattern in horizontal tube with heated wall.  
 (a) end elevation (b) horizontal elevation. (Oliver 1962)



Brown and Thomas (1965) experimented with water in horizontal tubes and found that contrary to Martinelli's work on vertical tubes it appeared that free convection effects increased with the flow rate. This was reconciled by postulating that at higher flow rates the temperature gradient at the wall would be higher than at low flow rates and hence the free convection effect would be greater.

Oliver (1962) found from experiments with air at high heat flux that the free convection had a pronounced effect which started at about  $ReRa = 10^3$ . They also noted that the critical Reynolds number depended on the intensity of the secondary flow and suggested

$$Re_c = 128 (ReRa)^{0.2}$$

Miyama and Eckert (1966) noticed that the effect of free convection in air flows increased as the ratio of  $Gr$  to  $Re$  increased and suggested using Oliver's (1962) equation.

Iqbal and Stankiewicz (1967) looked analytically at the effect of the tube inclination on combined free and forced convection and concluded that this had little effect on the  $Nu$  but did affect the friction factor. They noted that a temperature dependent density affected mainly the velocity field and not the temperature field.

Mori and Futagami (1967) extended the work of Mori et al (1966) and experimentally determined that the vortices in the secondary flow pattern moved closer to the wall with increasing  $ReRa$  (intensity of the secondary flow). The results are reproduced in Figure 21.

FIGURE 21 Secondary flow pattern (Mori and Futamura 1967)  
 (a)  $ReRa = 2 \times 10^4$  (b)  $ReRa = 9 \times 10^4$  (c)  $ReRa = 1.6 \times 10^5$   
 Centres of vortices move closer to wall for increasing  $ReRa$



Shannon and Depew (1968) experimented with the heating of water at low Reynolds number and noticed a pronounced effect of the  $L/d$  ratio and from this concluded that for

$$\frac{(GrPr)^{1/4}}{Nu_0} < 2$$

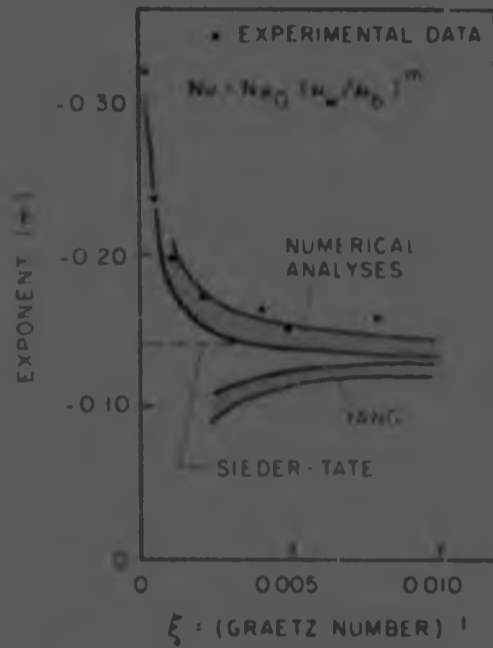
the natural convection influence was negligible

A year later Shannon and Depew (1969) concluded from a theoretical consideration that for an experimental dependence of  $\eta$  on  $T$  the temperature distribution in the fluid is a function of the viscosity ratio and the  $Gz$  only. From experiments with ethylene glycol they found the relation

$$\frac{T_{wall} - T_c}{Nu_0} = \left[ \frac{Gz}{m} \right]^m$$

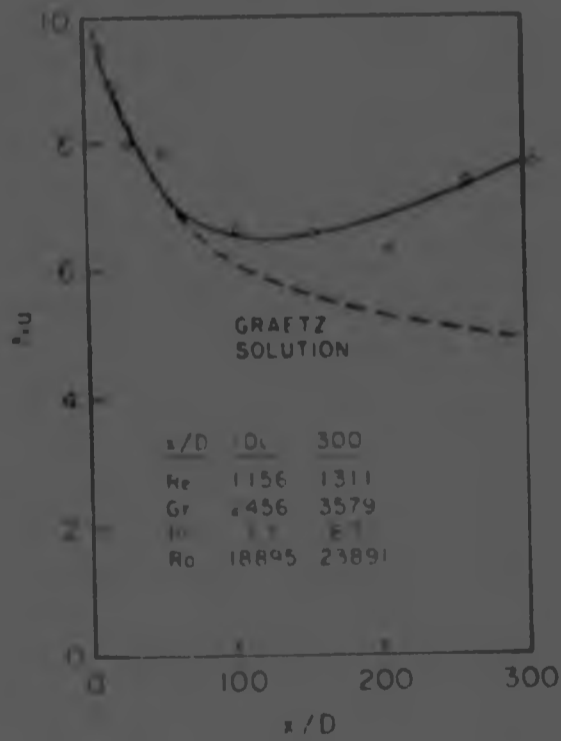
where  $m$  is specified in Figure 22

FIGURE 22 Exponent  $m$  versus  $\xi$  (Shannon and Dwyer (1969))



Kupper, Hauptmann and Iqbal (1969) experimented with uniform heat flux on water and although unable to correlate the data to show the effect of free convection, found a strong correlation with  $L/d$  ratio as in Figure 23

FIGURE 23 Variation of Nusselt number along tube (Kupper et al 1969)



Shah and London (1971) summarised the analytical solutions for laminar flow in a large variety of geometries. A summary of their findings is reproduced in the Appendix 4.5.

Porter (1971) summarised (without the knowledge of Shah and London) the laminar flow of Newtonian and Non-Newtonian fluids.

Depew and August (1971) experimentally determined that the effect of free convection was not correlated adequately by  $GrPr^{L/d}$  for  $L/d > 50$  and suggested using  $GzGr^nPr^m$  since  $L/d$  ceased to have significance in this range.

There appears to have been no further experimental investigation after this date.

### 3.3 TRANSITIONAL FLOW CORRELATIONS

Colburn (1933) and Sieder and Tate (1936) were the first to suggest heat transfer correlations in the transition region. These correlations derived from the analogous plots for fluid friction, give a discontinuity where the extended turbulent range meets the laminar range. Colburn used the ratio

$$\frac{h_{film}}{k_w}$$

to correlate the results and Sieder and Tate used

$$\frac{h_{wall}}{k_w}$$

Norris and Strid (1940) noted that there may be a distinct region where the flow "may be neither laminar nor turbulent, but of an intermediate nature". They noted that the transition region began at approximately  $Re = 2,100$  and extended as far as  $Re = 10,000$  for large values of  $L/d$ .

Norris and Sims (1942) experimented with the cooling of oils of high  $Pr$  in vertically downward flow in the semi-turbulent region of  $1,500 < Re < 11,000$ . The velocity profile was undeveloped and  $L/d$  was fixed in the experiment. Using physical properties based on an average inlet and outlet temperature they derived the equation

$$\frac{h}{C_p G} = 0.0067 Pr^{-0.8} \left[ \frac{L}{d} \right]^{0.14}$$

which correlated the results better than the methods of Colburn or of Sieder and Tate.

Senecal and Rothfus (1953) investigated laminar to turbulent transition in isothermally flowing air and noted that transition was limited to a narrow range of  $Re = 2,100$  to  $2,800$  and marked by extreme changes in the velocity profile. They found that friction data were independent of the

direction of approach to the transition region but did not extend their experiments to non-isothermal flow.

Ede (1961) experimented with air and water in the transition region but could not find any satisfactory method of correlating the data.

Petersen and Christiansen (1966) looked specifically at the transition region for Newtonian and Non-Newtonian fluids and empirically derived the relation

$$St = St_{turb} = 10,000 \left[ \frac{St_{critical, Re}}{Re - 10,000} \right]^{0.8(Re)} \left[ \frac{L}{10,000} \right]^3 \left[ \frac{10,000}{2,100} \right]^{0.05(Re)}$$

where  $\beta$  is the slope of the heat transfer correlation curve chosen for the turbulent region and

$$\beta(Re) = 1.635 \log \left[ \frac{1}{(Re - 710)} \frac{(Re - 1800)}{(Re - 1800)} \right]$$

They found that  $\beta(Re)$  effectively eliminated the effect of  $L/d$  ratio.

McEligot, Ormand and Perkins (1966) experimented in the transition region but the results could not be correlated.

Bankston (1970) experimented with hydrogen and helium undergoing turbulent to laminar transition at high heat loads and found that transition occurred at a bulk  $Re$  far in excess of the minimum turbulent  $Re$  for adiabatic flows. He also likened the process to the reverse transition of boundary layers in external flows.

McEligot, Coon and Perkins (1970) summarised the research on laminarisation in tubes and pointed out that the acceleration parameter used in external flows could be used for internal flows for predicting laminarisation.

Coon and Perkins (1970) looked specifically at the reverse transition of air and concluded that if

$$\frac{G^2 d T C_p}{\rho \mu} > 1.5 \times 10^5$$

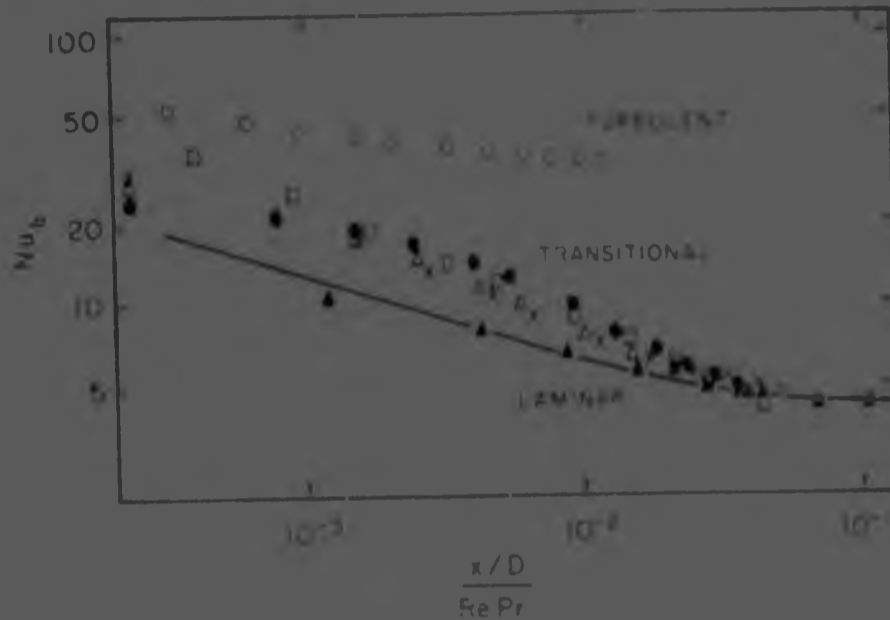
based on bulk properties which exceeded for an initially turbulent flow then the turbulent flow correlations did not predict the heat transfer coefficient acceptably. They further noted that downstream of the point

$$\left[ \frac{L}{D} \right] = 1.2 \times 10^5 \left( \frac{\rho \mu}{G^2 d T C_p} \right)^{0.5} \left[ \frac{1}{Re} \right]^{0.5}$$



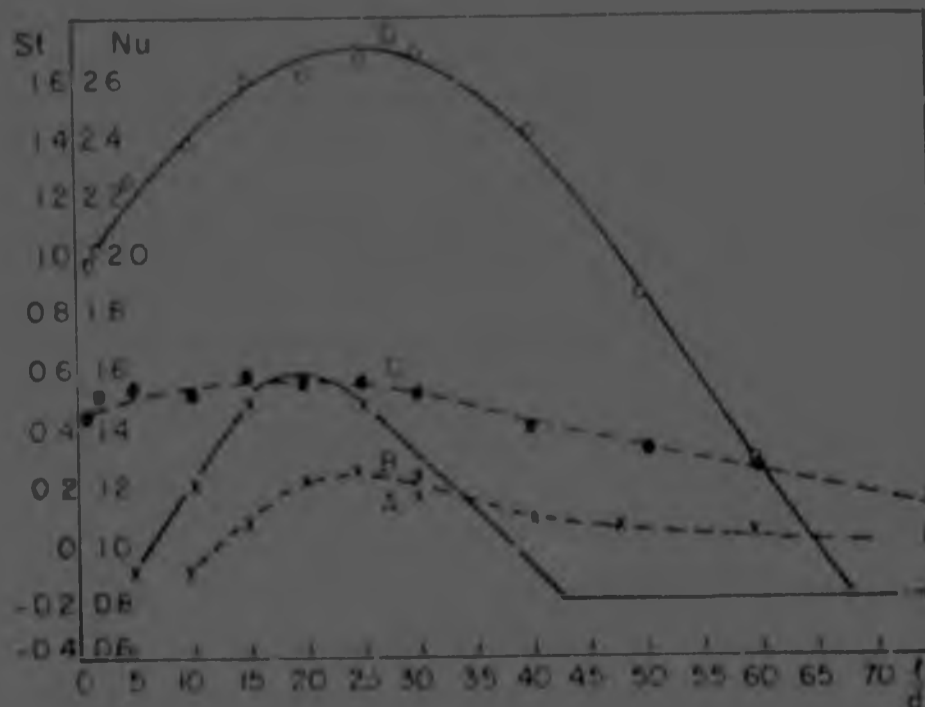
the laminar flow equations predicted the heat transfer coefficient. Their results in the transition region do not appear to be haphazard. See Figure 24

**FIGURE 24** Bulk Nusselt number data compared to Graetz Solution length laminar flow, turbulent flow and flow with reverse transition (Coon and Perkins 1970)



Dyban and Epik (1971) looked at reverse transition using air flow in nozzles and concluded that the local Nu could not be expressed in terms of the Re with only one exponent for the Re. However, they did not suggest a correlation. Figure 25 is a plot of the results obtained.

**FIGURE 25** Change in the Re exponent for transient flow regimes (solid line - local heat transfer, dashed lines - average heat transfer): A, B - conical convergent nozzle; C, D - Vitoshinsky nozzle (Dyban and Epik 1971)



Kuznetsova (1972) looked at the transitional flow of oil (without saying whether this was reverse transitional or laminar to turbulent) and from the results recommended that the equation

$$Nu = 0,013 Re^{0,8} Pr^{0,4} \left[ \frac{L}{d} \right]^{0,13}$$

be used. The results were well correlated by this. However, much of the original text was lost in translation.

Pechenegov (1975) experimented with air undergoing reverse transition and found that the results could be correlated using the equation

$$Nu_x = c R_x^n$$

where c and n are functions of  $L/d$ . This was in accordance with the expectations of Dybon and Epik (1971) that no single exponent could be used for the  $Re$ .

Churchill (1977) proposed a comprehensive correlating equation for the total flow region

$$Nu = \left[ Nu_{laminar} + \left[ Nu_{transitional}^c + \left[ \left( Nu_{Pr \rightarrow 0}^{Pr \rightarrow 0} \right)^2 + \left( Nu_{Pr \rightarrow \infty}^{Pr \rightarrow \infty} \right)^2 \right]^{1/2} \right]^{1/b} \right]^{1/a}$$

The individual expressions for the  $Nu$ 's were found from expressions in the literature. In particular, in the transition region, Churchill proposed using the asymptote

$$Nu_{transitional} = Nu_{laminar, Re = 2100} \left[ \frac{(Re - 2100) / 730}{(Re - 2100) / 730} \right]^c$$

and a value of the exponent  $c = -2$ . For the entire turbulent and transition region the function

$$Nu_x = \left[ \frac{1}{Nu_{laminar}^2} + \frac{1}{Nu_{turbulent}^2} \right]^{-1/2}$$

was proposed.

Since this time no literature has appeared.

4 APPENDIX4.1 REFERENCES

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42 FLUID INDEX

This index is for tracing original experimental data on the heat transfer to fluids in pipes. The heat transfer conditions are limited to the internal flow of Newtonian fluids in laminar, transitional or turbulent flow in horizontal or vertical circular pipes.

Cross-referencing with the EQUATION INDEX is intended and actual data may be obtained from that index or the original literature.

The work of Porter (1971) is recommended for reference for Non-Newtonian fluids in laminar flow and for Newtonian fluids in different geometries.

FLUID	RANGE		REFERENCE
	Reynolds	Orientation	
Air	Unknown	Horizontal	Nusselt (1917)
	300 - 100 000	Horizontal	Ede (1961)
	Laminar	Horizontal	Mori and Futagami (1967)
	100 - 900	Horizontal	McComas and Eckert (1966)
	100 - 13 000	Horizontal	Mori et al (1966)
	Laminar and turbulent	Horizontal	Jackson et al (1961)
	Turbulent		Subbotin et al (1965)
	4 500 - 140 000	Horizontal	Kolář (1965)
	Turbulent	Vertical	Gowen and Smith (1967)
	Turbulent	Horizontal	Pechenegov (1975)
	Turbulent	Horizontal	Zucchetto and Thorsen (1973)
	Turbulent		Rice (1924)
	Turbulent	Horizontal	Drissler and Eian (1952)
Turbulent	Vertical	Eckert and Diaguila (1954)	
Askarel	3 500 - 11 000	Vertical	Norris and Sims (1942)
Acetone	1 000 - 100 000	Horizontal	Sherwood and Petrie (1932)
Benzene	1 000 - 100 000	Horizontal	Sherwood and Petrie (1932)
n Butanol	1 000 - 100 000	Horizontal	Sherwood and Petrie (1932)
Butanol	5 000 - 300 000	Horizontal	Bernado and Eian (1945)
Corn syrup	Turbulent	Horizontal	Friend and Metzner (1958)
Ethanol	Laminar	Horizontal	Depew and August (1971)
	Laminar	Horizontal	Oliver (1962)
Ethylene glycol	6 - 300	Horizontal	Shannon and Depew (1969)
	5 000 - 300 000	Horizontal	Bernado and Eian (1945)
	Turbulent	Vertical	Gowen and Smith (1967)

FLUID	RANGE		REFERENCE
	Reynolds	Orientation	
Glycerol	Laminar	Horizontal	Sieder and Tate (1936)
	Laminar	Horizontal	Depew and August (1971)
	Laminar	Horizontal	Oliver (1962)
Helium	1 450 - 45 000	Vertical	McEligot et al (1966)
Kerosene	1 000 - 100 000	Horizontal	Sherwood and Petrie (1932)
Mercury	Turbulent		Subbotin et al (1965)
Molasses	Turbulent	Horizontal	Friend and Metzner (1958)
Nitrogen	1 450 - 4 500	Vertical	McEligot et al (1966)
Oil	Laminar	Horizontal	Sieder and Tate (1936)
	Laminar	Vertical	Martinelli et al (1942)
	Laminar and turbulent	Horizontal	Dittus and Boelter (1930)
	Laminar	Horizontal	Kern and Othmer (1943)
	Laminar	Vertical	Test (1968)
	2 300 - 48 000	Vertical	Hartnett (1955)
	3 500 - 11 000	Vertical	Norris and Sims (1942)
	Turbulent	Horizontal	Malina and Sparrow (1964)
	4 000 - 11 000	Horizontal	Hackl and Groll (1969)
	Turbulent		McAdams and Frost (1922)
Gas oil	1 862 - 43 000	Horizontal	Morris and Whitman (1928)
Straw oil	1 862 - 43 000	Horizontal	Morris and Whitman (1928)
Transformer Oil		Horizontal	Kuznetsova (1972)
Fuel oil		Horizontal	Kuznetsova (1972)
Light motor oil	1 862 - 43 000	Horizontal	Morris and Whitman (1928)
Velocite oil	Laminar and turbulent	Horizontal	Keevil and McAdams (1929)

FLUID	RANGE		REFERENCE
	Regime	Orientation	
Spindle oil	Laminar and turbulent	Horizontal	Keevil and McAdams (1929)
Light oil	Laminar	Horizontal	Kirkbride and McCabe (1931)
Rabboth oil	Laminar and turbulent	Horizontal	Keevil and McAdams (1929)
Heavy fuel oil	Laminar	Horizontal	Kirkbride and McCabe (1931)
Sodium	Turbulent		Subbotin et al (1965)
Sugar solution	Turbulent	Horizontal	Friend and Metzner (1958)
Water	1 862 - 43 000	Horizontal	Morris and Whitman (1928)
	Laminar	Vertical	Colburn and Hounen (1930)
	Laminar	Horizontal	Kirkbride and McCabe (1931)
	1 000 - 100 000	Horizontal	Sherwood and Petrie (1932)
	Laminar	Horizontal	Sieder and Tate (1936)
	Laminar	Vertical	Martinelli et al (1942)
	Laminar	Horizontal	Depew and August (1971)
	100 - 2 000	Horizontal	Kupper et al (1969)
	300 - 100 000	Horizontal	Ede (1961)
	2 300 - 48 000	Vertical	Hartnett (1955)
	Laminar	Horizontal	Oliver (1962)
	Laminar	Horizontal	Hown and Thomas (1965)
	120 - 2 300	Horizontal	Shannon and Depew (1968)
	5 800 - 71 000	Vertical	Hartnett and Sterns (1971)
	14 000 - 50 000		Walker and Bott (1973)
	5 000 - 300 000	Horizontal	Bernoldo and Eian (1945)
	14 000 - 500 000	Vertical	Dipprey and Sabersky (1963)
	13 000 - 111 000	Horizontal	Allan and Eckert (1964)
	Turbulent	Horizontal	Malina and Sparrow (1964)
	4 500 - 140 000	Horizontal	Kolac (1965)
Turbulent		Subbotin et al (1965)	
20 000 - 640 000	Horizontal	Hufschmidt et al (1966)	
Turbulent		McAdams and Frost (1922)	
Turbulent		Rice (1924)	



FLUID	RANGE		REFERENCE
	Reynolds	Orientation	
Water	Turbulent		McAdams and Frost (1924)
	Turbulent	Horizontal	Faulk and Ferguson (1930)
	Turbulent and laminar	Horizontal	Dittus and Boelter (1930)
	Turbulent	Horizontal	Lawrence and Sherwood (1931)
	Turbulent	Horizontal	Nusselt (1931)

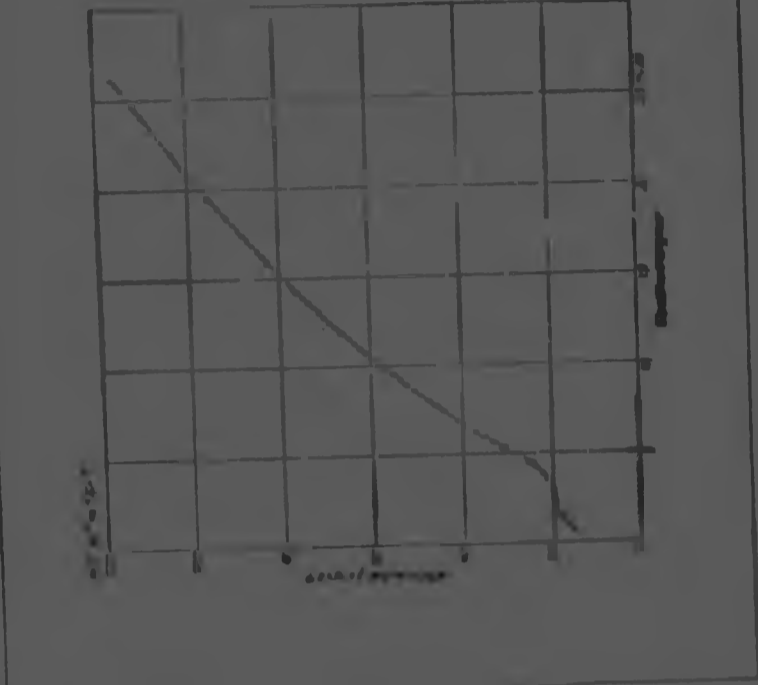
#### 4.3 EQUATION INDEX

This index is used in conjunction with the FLUID INDEX for tracing equations or direct experimental results. Likewise the heat transfer conditions are limited to the internal flow of Newtonian fluids in laminar, transitional or turbulent flow in horizontal or vertical circular pipes.

Both theoretically based and empirical equations are incorporated.

For a more complete review of the theoretical equations the works of *Shah and London (1978)* for laminar flow and *Launder and Spalding (1972)*, *Petukhov (1970)*, *McEligot et al (1970)* and *Reynolds and Cebecei (1976)* for turbulent flow are recommended.

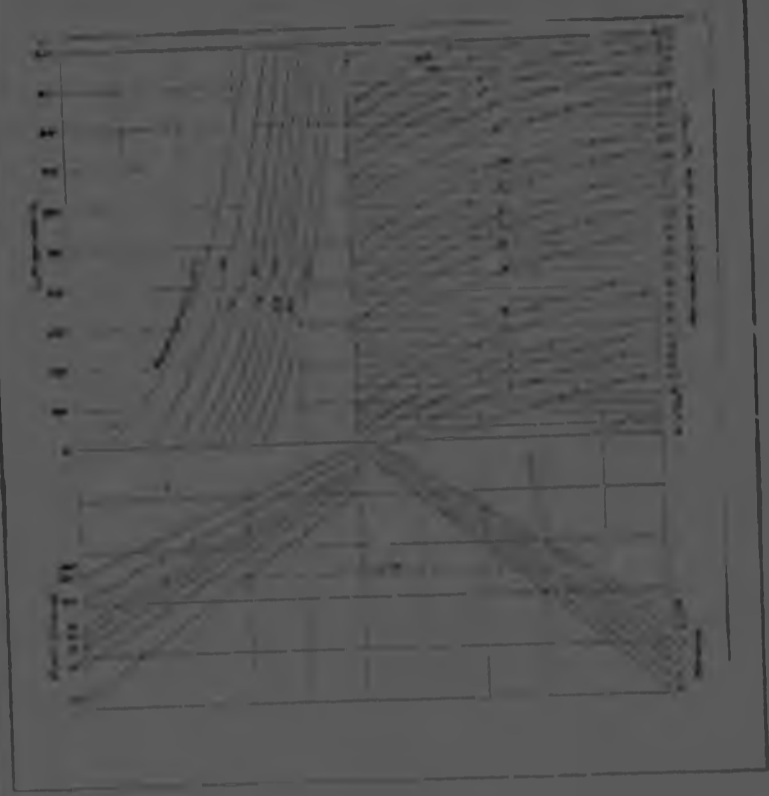
## HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air, CO<sub>2</sub></p> <p>METHOD: Steam heating in double pipe exchanger.</p>	<p>AUTHOR(S): Muehle, W</p>
<p>EQUATION PROPOSED:</p> $h = 15,000 \frac{k_{air}}{D^{0.25}} \left[ \frac{D \rho v}{\mu} \right]^{0.75} \left[ \frac{c_p \mu}{k} \right]^{0.14}$	<p>SOURCE: Z. VDI-Beibl. 53, Nr. 44, 1908 (1909)</p>
<p>REYNOLDS: Turbulent</p> <p>PRANDTL: ~0.7</p> <p>GRASHOF:</p>	
<p>AUXILIARY INFORMATION</p>	<p>HORIZONTAL/VERTICAL: Horizontal</p> <p>LENGTH/DIAMETER: Various</p>

# HEAT TRANSFER COEFFICIENTS IN TUBES

**AUTHOR(S):** Nusselt, Wilhelm Von

**SOURCE:** 7 VDI, Band 67, Nr 33, 685 (1917)



**FLUID(S):** Air

**METHOD:** Constant wall temperature (condensing steam)

**EQUATION PROPOSED:**

$$\frac{h_{cond}}{k_w} = 0.03622 \left[ \frac{d_w}{L} \right]^{0.684} \left[ \frac{GPr_w}{k_w} \right]^{0.186}$$

$$Nu_w = 0.03622 \left[ \frac{GPr_w}{k_w} \right]^{0.684} \left[ \frac{D_w Pr_w}{k_w} \right]^{0.186}$$

$$h_w = \frac{1}{\frac{1}{h_{gas}} + \frac{1}{h_{wall}}} \int_{T_w}^{T_m} \frac{dT}{T}$$

$T_m$  = gas temperature

$T_w$  = wall temperature

**REYNOLDS:** Unknown

**PRANDTL:** Unknown

**GRASHOF:** Unknown

**AUXILIARY INFORMATION**

**HORIZONTAL/VERTICAL:** Horizontal

**LENGTH/DIAMETER:** Variable

Paper in German and the layout is confusing.

# HEAT TRANSFER COEFFICIENTS IN TUBES

<b>FLUID(S):</b> Water etc. <b>METHOD:</b> Conduction method	<b>AUTHOR(S):</b> McAdams, W. M. From: T. H.
<b>EQUATION PROPOSED:</b> $h = 22.6 \left[ \frac{D_{out}}{L} \right]^{0.4} \left[ \frac{\mu}{k} \right]^{0.14}$ <p style="font-size: small;">             h = the viscosity of the fluid (the temperature referred to water at 60°F.)           </p>	
<b>SOURCE:</b> Ind. Eng. Chem., 34, 1109 (1922)	



<b>REYNOLDS:</b> Turbulent	<b>HORIZONTAL/VERTICAL:</b>
<b>PRANDTL:</b>	<b>LENGTH/DIAMETER:</b>
<b>GRASHOF:</b>	
<b>AUXILIARY INFORMATION</b>	

HEAT TRANSFER COEFFICIENTS IN TUBES

AUTHOR(S): BIRD, C. W.

SOURCE: Ind. Eng. Chem., 16, (6), 460 (1924)



FLUID(S): Air, Water

METHOD: Not given but assumed to be constant wall temperature.

EQUATION PROPOSED:

$$W_c = \left[ \frac{A k \Delta T}{K_p D} \right] \cdot \left[ \frac{c_p D}{k} \right]^{\frac{1}{4}} \cdot \left[ \frac{D v_c}{\mu} \right]^{\frac{1}{4}} \quad \text{(watts)}$$

For smooth pipes  $K_p = 60$ ,  $\frac{1}{4} = \frac{1}{4}$ ,  $\frac{1}{4} = \frac{1}{4}$

A = area of surface

$\Delta T$  = temperature difference

H R = mechanical resistance,  $(\text{watts}/\text{cm}^2)$ , then

$$W_c = \left[ \frac{A k \Delta T}{H R} \right] \cdot \left[ \frac{c_p D}{k} \right]^{\frac{1}{4}} \quad \text{(watts)}$$

REYNOLDS: Turbulent

PRANDTL: 2.48 - 7.35

GRASHOF: Unknown

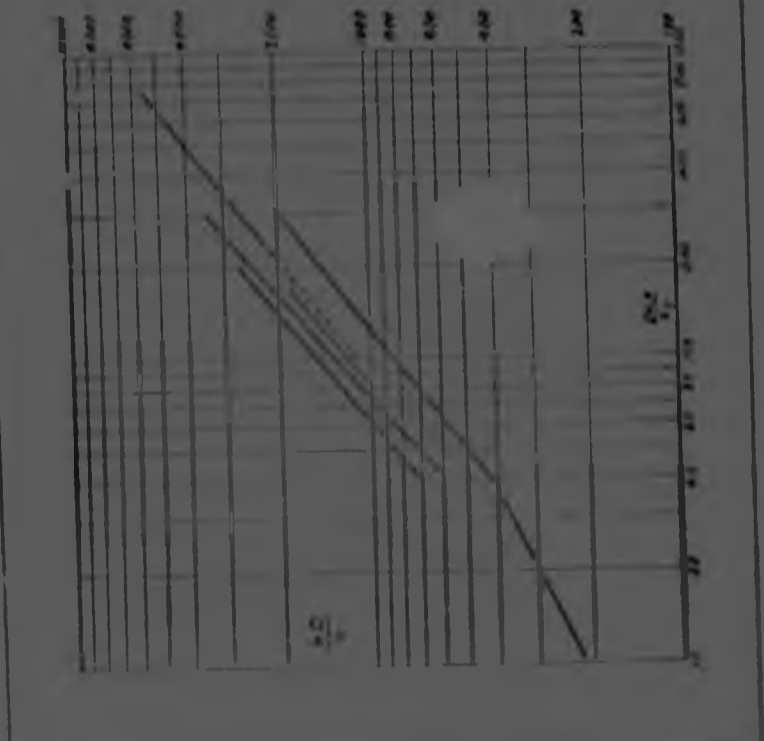
AUXILIARY INFORMATION

HORIZONTAL/VERTICAL: Not given

LENGTH/DIAMETER: Not given

Auxiliary information: Fluid variables taken at the average film temperature


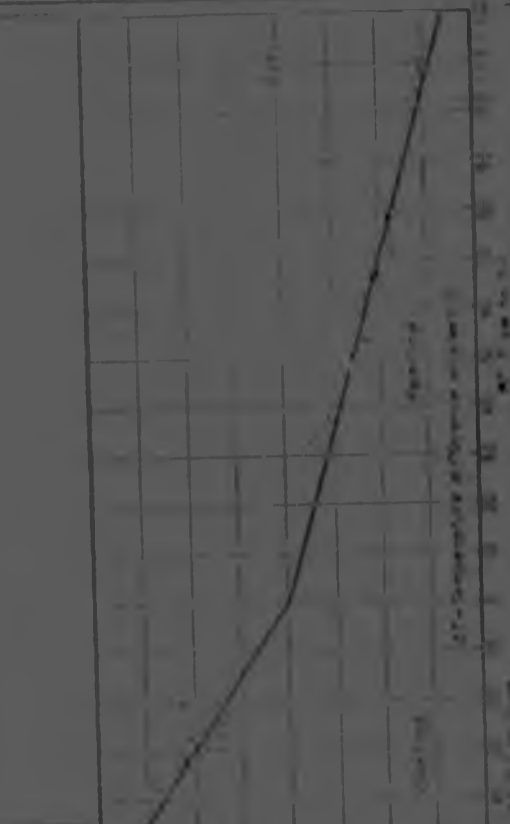
HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Water</p> <p><b>METHOD:</b> Using the data of Stanton, Webster and Bay and Savler.</p>	<p><b>AUTHOR(S):</b> McAdams, W H Frost, T H</p>
<p><b>EQUATION PROPOSED:</b></p> $h = \frac{138}{D^{0.3}} \left[ \frac{U}{L} \right]^{0.8} \left[ \frac{F}{L} \right] \left[ \frac{\text{BTU}}{\text{ft}^2 \text{ OF}} \right]$ <p>where <math>z</math> is the viscosity related to the viscosity at 68°F.  <math>k</math> is taken as constant 0.329 BTU/ft<sup>2</sup> OF</p> <p>The graph is a plot of the various data showing the spread of results</p>	<p><b>SOURCE:</b> Refrig Eng., 10, (9), 323 (1924)</p> 
<p><b>REYNOLDS:</b> Turbulent</p> <p><b>PRANDTL:</b></p> <p><b>GRA=ROF:</b></p> <p><b>AUXILIARY INFORMATION</b> L/D must be greater than .35</p>	<p><b>HORIZONTAL/VERTICAL:</b> Unknown</p> <p><b>LENGTH/DIAMETER:</b> &gt; 35</p>

### HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Gas oil, Steam oil, Light motor oil, Water.</p> <p><b>METHOD:</b> Steam heating, water cooling of double pipe heat exchanger.</p>	<p><b>AUTHOR(S):</b> Morris, F H Whitman, W G</p>	<p><b>SOURCE:</b> Ind Eng. Chem., 20, (3), 23A (1928).</p>
<p><b>EQUATION PROPOSED:</b> To non-dimensionalize:</p> $Nu = \frac{Nu_{max}}{12}$ $Re = 124.13 Re_{max}$ $Pr = 2.62 Pr_{max}$ <p>Use is made of the graphs rather than an equation. The parameters given are dimensional and must be adjusted as above.</p>	<p><b>REYNOLDS:</b> 1 852 - 43 445</p> <p><b>PRANDTL:</b> 2.83 - 750</p> <p><b>GRASHOF:</b></p>	<p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p> <p><b>LENGTH/DIAMETER:</b> 198</p> <p><b>AUXILIARY INFORMATION:</b> Physical properties at main bulk temperature, 12 inch entrance region.</p>

HEAT TRANSFER COEFFICIENTS IN TUBES


<p><b>FLUID(S):</b> Vegetable oil, Squalid oil, Rabbath oil</p> <p><b>METHOD:</b> Steam heating and water cooling of a jacketed pipe.</p>	<p><b>AUTHOR(S):</b> Krayl, C. S. Möhdams, W. H.</p> <p><b>SOURCE:</b> Chem. and Met. Eng. 36, (8) 464 (1920)</p>
<p><b>EQUATION PROPOSED:</b> Graphs are presented showing the variation of friction factor (laminar) with the difference between pipe and fluid average temperature.</p> 	
<p><b>REYNOLDS:</b> Viscous and turbulent</p> <p><b>PRANDTL:</b></p> <p><b>GRASSHOFF:</b></p>	<p><b>HORIZONTAL/VERTICAL:</b> Non-horizontal</p> <p><b>LENGTH/DIAMETER:</b> 105, 110</p>
<p><b>AUXILIARY INFORMATION:</b> Entrance region of 1000. Perfect at main bulk temperature.</p>	




# HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Water</p> <p><b>METHOD:</b> Electrical heating using alternating current</p>	<p><b>AUTHOR(S):</b> Exptl. A Ferguson, R M</p> <p><b>SOURCE:</b> Proc. Inst. Mech. Engrs. (London), 9301</p>	
<p><b>EQUATION PROPOSED:</b></p> $h_0 = h = \left[ \frac{0.07}{(1 + 0.05) - 1} \right]^2 \left[ \frac{-1}{2Pr} \frac{dPr}{dx} \right] \mu$ <p><math>h</math> = heat flux in BTU/hr<sup>2</sup> ft<sup>2</sup></p> <p><math>h_0</math> = coefficient at <math>h = 0</math> in BTU/hr<sup>2</sup> ft<sup>2</sup> and is referred to the tube wall temperature.</p> <p><math>V_0</math> = mass velocity</p> <p>The graph shows experimental results at a temperature of 60°F.</p>	<p><b>REYNOLDS:</b> Turbulent</p> <p><b>PRANDTL:</b> 2.99 - 10.5</p> <p><b>GRAZING:</b></p> <p><b>AUXILIARY INFORMATION:</b> <math>\alpha</math> and <math>\beta</math> are determined by experiment such that for water <math>\frac{\beta Pr}{\alpha + \beta(Pr - 1)} = \frac{Pr}{2.45 + Pr}</math></p> <p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p> <p><b>LENGTH/DIAMETER:</b> 48 - 144</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Oil, water  <b>METHOD:</b> Results of others used.</p>	<p><b>AUTHOR(S):</b> Dittus, F W          Boelter, L M K.</p>
<p><b>EQUATION PROPOSED:</b></p> 	<p><b>SOURCE:</b> Univ. Calif. Publ. in Eng. 2, (13) 443 (1930).</p>
<p><b>REYNOLDS:</b> Turbulent &amp; Laminar  <b>PRANDTL:</b>  <b>GRASSE:</b>  <b>AUXILIARY INFORMATION:</b> <math>x</math> is the viscosity. Properties are taken at the mean stream temperature. <math>S</math> is the specific gravity.</p>	<p><b>Turbulent Flow</b></p> $h/k = 19.8 \left[ \frac{C_p \mu}{k} \right]^n \left[ \frac{C_p \mu}{k} \right]^m$ <p><math>n = 0.3</math> cooling  <math>n = 0.4</math> heating</p> $m = 20 \left[ \frac{k \sqrt{S}}{C_p \mu} \right]^{0.5} \left[ \frac{C_p \mu}{k} \right]^{0.3} Co \left[ \frac{h/k}{\mu} \right]^{0.2}$ <p><math>\left[ \frac{h}{k} = \frac{50}{r} \right]</math> (STU) <math>h/k^2</math> <math>0.1</math>          in upward flow</p> <div style="border: 1px solid black; padding: 5px; width: fit-content; margin: 10px auto;"> <p><math>Nu = 0.024 Re^{0.4} Pr^{0.4}</math> heating  <math>Nu = 0.026 Re^{0.4} Pr^{0.3}</math> cooling</p> </div>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Water</p> <p><b>METHOD:</b> Steam heating of water rising. Upward and downward flow.</p>	<p><b>AUTHOR(S):</b> Colburn, A P Huguenin, O A</p>
<p><b>EQUATION PROPOSED:</b></p> $h = 0.128 k_s \left[ \frac{C_p \mu_s}{k_s} \right]^{1/4} \left[ \frac{L^2 \Delta T}{x^3} \right]^{1/4}$ <p>(BTU/hr ft<sup>2</sup> °F)</p> <p><math>h_w = 0.421 \sqrt[3]{N} \sqrt[3]{T}</math> for upward flow (BTU/hr ft<sup>2</sup> °F)</p> <p><math>h_w = 0.377 \sqrt[3]{N} \sqrt[3]{T}</math> for downward flow</p> <p><math>h_w = 0.44 \sqrt[3]{N} \sqrt[3]{T}</math> for downward flow</p> <p><i>h</i> is the temperature drop across film</p>	<p><b>SOURCE:</b> Ind. Eng. Chem. <u>23</u>, 572 (1930)</p> 
<p><b>REYNOLDS:</b> Laminar</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p>	<p><b>HORIZONTAL/VERTICAL:</b> Vertical</p>
<p><b>AUXILIARY INFORMATION</b></p> <p><math>t_f</math> = average temperature of water film</p> <p><math>t_w</math> = temperature of main water stream</p>	<p><b>LENGTH/DIAMETER:</b></p>


## HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Gas</p> <p>METHOD:</p>	<p>AUTHOR(S): Steiner</p> <p>Wiss Veröffentlich Siemens-Kazern 9, BP (1930)</p>
<p>EQUATION PROPOSED:</p> $\frac{h d}{k} = C_1 \left[ \frac{\rho u C_p}{k} \right]^{0.75} C_2 \left[ \frac{d}{L} \right]$	<p>SOURCE: Lawrence &amp; Sherwood (1931)</p>
<p>REYNOLDS:</p> <p>PRANDTL:</p> <p>GRASHOF:</p>	<p>HORIZONTAL/VERTICAL:</p> <p>LENGTH/DIAMETER:</p>
<p>AUXILIARY INFORMATION</p>	


# HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Water</p> <p><b>METHOD:</b> Condensing steam in a surrounding jacket</p>	<p><b>AUTHOR(S):</b> Lawrence, A E Shawood, T K</p> <p><b>SOURCE:</b> Ind Eng Chem 23-301 (1931)</p>
<p><b>EQUATION PROPOSED:</b></p> $\frac{h_c D}{k} = 450 \left[ \frac{D \cdot \Delta T}{\mu} \right]^{0.7} \left[ \frac{C_p \mu}{k} \right]^{0.4}$ <p style="font-size: small;"> <math>\mu</math> is the viscosity in centipoises taken at arithmetic mean film temperature  <math>\Delta T</math> is the specific gravity         </p> <div style="border: 1px solid black; padding: 5px; width: fit-content; margin: 10px auto;"> <math>Nu = 0.055 Re^{0.7} Pr^{0.4}</math> </div>	
<p><b>REYNOLDS:</b> Turbulent</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p>	<p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p> <p><b>LENGTH/DIAMETER:</b> 59 - 224</p> <p><b>AUXILIARY INFORMATION:</b> The experiments were conducted particularly to look at tube length effect. Arithmetic mean temperature for properties.</p>

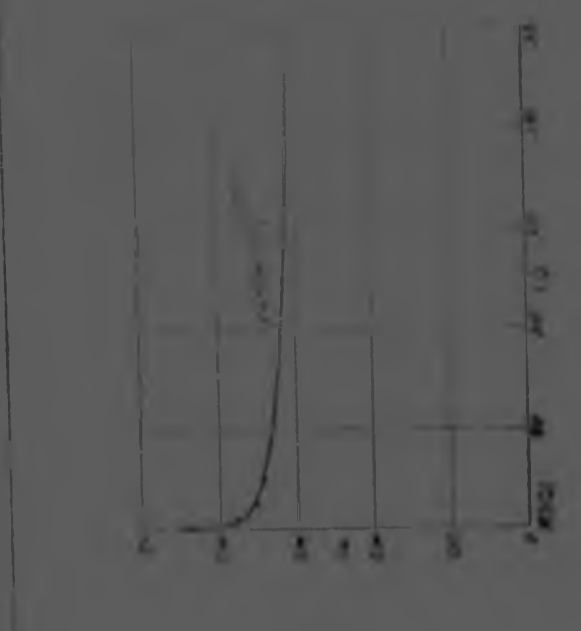
HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Water, light oil, heavy fuel oil</p> <p><b>METHOD:</b> Electric heating</p>	<p><b>AUTHOR(S):</b> Kishorale, C G McCabe, W L</p>
<p><b>EQUATION PROPOSED</b></p> $\frac{h_c D}{k} = 3.65 + \frac{0.0025}{(Re L/D)^{0.5}} + \frac{0.511}{(Re L/D)^{0.825}}$ <p>Proposed for tubes at atmospheric main fluid temperatures.</p>	<p><b>SOURCE:</b> Int. Eng. Chem. 22, 625 (1931)</p> 
<p><b>REYNOLDS:</b> Viscous flow</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p> <p><b>AUXILIARY INFORMATION:</b> 5 ft calming section</p>	<p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p> <p><b>LENGTH/DIAMETER:</b> 150</p>

HEAT TRANSFER COEFFICIENTS IN TUBES


<p><b>FLUID(S):</b> Water, keros oil, heavy fuel oil.</p> <p><b>METHOD:</b> Electric heating.</p>	<p><b>AUTHOR(S):</b> Kulkarni, C G Mishra, W L</p>
<p><b>EQUATION PROPOSED:</b></p> $\frac{h_c D}{k} = 3.65 + \frac{0.0035}{(Pr)^{1/4}} + \frac{0.613}{(Re L/D)^{1/4}} + 0.494$ <p>Properties taken at arithmetic mean fluid temperature.</p>	<p><b>SOURCE:</b> Ind Eng Chem., 53, 625 (1961)</p> 
<p><b>REYNOLDS:</b> Various flow</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p> <p><b>AUXILIARY INFORMATION</b> 5 ft cooling section</p>	<p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p> <p><b>LENGTH/DIAMETER:</b> 150</p>

HEAT TRANSFER COEFFICIENTS IN TUBES


<p>FLUID(S): Water</p> <p>METHOD: Liquid film of turbulent flow</p>	<p>AUTHOR(S): Schmidt, W.</p> <p>SOURCE: <i>Ergeb. Exptell. Physik</i>, Band 2, No. 11 (1935) (H.G.)</p>
<p>EQUATION PROPOSED:</p> $\frac{h}{k} = 0.023 \left[ \frac{v}{\nu} \right]^{0.8} \left[ \frac{c_p \mu}{k} \right]^{0.4} \left[ \frac{c_p \mu}{k} \right]^{0.14}$ <p>Nusselt suggested that the constants would apply to laminar flow as well.</p> <p>Using the data of Egan and Fanning his experiments would give 0.879 and 0.216.</p>	 <p>The effect of tube length</p>
<p>REYNOLDS: Turbulent</p> <p>PRANDTL: HORIZONTAL/VERTICAL: Horizontal</p> <p>GRASHOF: LENGTH/DIAMETER: 10 - 400</p> <p>AUXILIARY INFORMATION: The article is in German.</p>	




HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Water, Acetone, Benzene, Kerosene, n-butyl alcohol, Jacketed pipe, steam and hot water heating.</p> <p><b>METHOD:</b> Jacketed pipe, steam and hot water heating.</p>	<p><b>AUTHOR(S):</b> Sherwood, T K Peters, J M</p>
<p><b>EQUATION PROPOSED:</b></p> $\frac{hD}{k} = 0.024 \left[ \frac{ReD}{L} \right]^{0.8} \left[ \frac{Cp\mu}{k} \right]^{0.4} \quad \text{for turbulent flow}$ <p>Properties taken at the arithmetic mean temperature.</p>	<p><b>SOURCE:</b> Ind Eng Chem, 28, 736 (1932)</p>  <p>The graph plots the dimensionless heat transfer coefficient <math>\frac{hD}{k}</math> on the y-axis against the dimensionless group <math>\left[ \frac{ReD}{L} \right]^{0.8} \left[ \frac{Cp\mu}{k} \right]^{0.4}</math> on the x-axis. The y-axis ranges from 0 to 100, and the x-axis ranges from 0 to 100. A single straight line is drawn through the data points, indicating a power-law relationship. The legend identifies the data series: Water (solid circles), Acetone (open circles), Benzene (solid squares), Kerosene (open squares), n-butyl alcohol (solid triangles), and Jacketed pipe (open triangles). The graph also includes a note: 'Properties taken at the arithmetic mean temperature'.</p>
<p><b>REYNOLDS:</b> 1 000 - 100 000</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p> <p><b>AUXILIARY INFORMATION:</b> 55 inch developing section</p>	<p><b>HORIZONTAL/VERTICAL:</b> HORIZONTAL</p> <p><b>LENGTH/DIAMETER:</b> 97</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Air fluids</p> <p><b>METHOD:</b> Combined data</p>	<p><b>AUTHOR(S):</b> Colburn, A. P.</p>
<p><b>EQUATION PROPOSED:</b></p> $h = \left[ \frac{\mu}{C_p G} \right]^{1/4} \left[ \frac{C_p G}{k} \right]^{2/3} = 0.0007 + 0.0066 \left[ \frac{G}{\mu} \right]^{0.97}$ <p>Properties at film temperature. Turbulent flow.</p> $h = \left[ \frac{\mu}{C_p G} \right]^{1/4} \left[ \frac{C_p G}{k} \right]^{2/3} = 1.5 \left[ \frac{G}{\mu} \right]^{1/3} \left[ \frac{k}{D} \right]^{1/4}$ <p>Properties at arithmetic mean. Laminar flow.</p> <p>Multiply by <math>(1 + 0.015 \left[ \frac{D}{L} \right]^{1/3})</math> for large <math>D/L</math>.</p>	<p><b>SOURCE:</b> Trans. AIChE, 29, 175 (1933)</p> 
<p><b>REYNOLDS:</b> All</p> <p><b>PRANDTL:</b></p> <p><b>GRATZKE:</b></p>	<p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p>
<p><b>AUXILIARY INFORMATION:</b> <math>t_s = t_w + \frac{1}{2}(t_w - t_s)</math> in turbulent flow, and <math>t_s = t_w + \frac{1}{4}(t_w - t_s)</math> in laminar flow.</p>	<p><b>LENGTH/DIAMETER:</b></p>

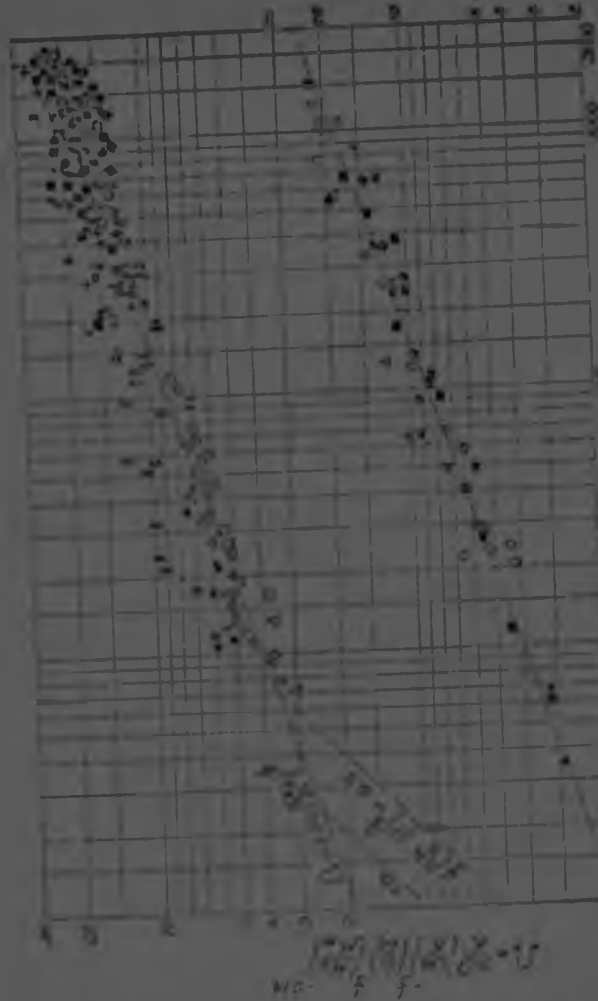
# HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> All fluids</p> <p><b>METHOD:</b> Combined data</p>	<p><b>AUTHOR(S):</b> Colburn, A P</p>	<p><b>SOURCE:</b> Trans. AICHE, 29, 176 (1933)</p> 
<p><b>EQUATION PROPOSED:</b></p> $h = \left[ \frac{h}{k} \right] \left[ \frac{C_p \mu}{k} \right]^{1/4} = 0.0031 + 0.005 \left[ \frac{h}{k} \right]^{0.32}$ <p>Properties at film temperature. Turbulent flow</p> $h = \left[ \frac{h}{k} \right] \left[ \frac{C_p \mu}{k} \right]^{1/4} = 1.5 \left[ \frac{G}{\mu} \right]^{0.75} \left[ \frac{k}{\mu} \right]^{1/4}$ <p>Properties at arithmetic mean. Laminar flow</p> <p>Multiply by <math>(1 + 0.015 Gr^{1/2})</math> for large Gr</p>	<p><b>REYNOLDS:</b> All</p> <p><b>PRANDTL:</b></p> <p><b>GRADE:</b> F</p> <p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p> <p><b>LENGTH/DIAMETER:</b></p>	<p><b>AUXILIARY INFORMATION</b> <math>t_f = t_w + \frac{1}{2}(t_w - t_c)</math> in turbulent flow, and <math>t_f = t_w + \frac{1}{4}(t_w - t_c)</math> in laminar flow.</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): METHOD:</p>	<p>AUTHOR(S): Kraussold, H. Forsch Ing, Wien 66, 3, 20 (1933)</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.032 Re^{0.8} Pr^{0.4} \left[ \frac{d}{L} \right]^{0.054} \text{ heating}$ $Nu = 0.032 Re^{0.8} Pr^{0.4} \left[ \frac{d}{L} \right]^{0.054} \text{ cooling}$	<p>SOURCE: Hauser, H. (1943)</p>
<p>REYNOLDS: PRANDTL: GRASHOF:</p>	<p>Turbulent</p>
<p>AUXILIARY INFORMATION</p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER:</p>

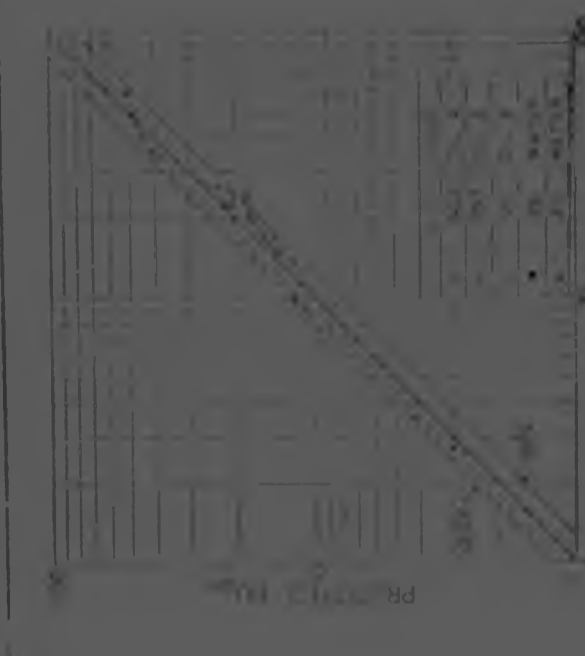
HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Oil, glycerol, water</p> <p><b>METHOD:</b> Water: heating and cooling</p>	<p><b>AUTHOR(S):</b> Sieder, E N Tate, G E</p> <p><b>SOURCE:</b> Ind Eng Chem, 28, 1429 (1936)</p>
<p><b>EQUATION PROPOSED:</b></p> $Nu = 1.86 Re^{1/4} Pr^{1/4} \left[ \frac{L}{D} \right]^{-1/4} \left[ \frac{\mu_s}{\mu_b} \right]^{1/4}$ <p>Multiply by 0.811 + 0.015 Gr<sup>1/4</sup> for Gr &gt; 25,000</p> <p>The equation was developed for laminar flow and is form with usual to apply in the turbulent regime as well</p>	
<p><b>REYNOLDS:</b> Viscous</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p>	<p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p> <p><b>LENGTH/DIAMETER:</b> 90</p>
<p><b>AUXILIARY INFORMATION</b> 2 ft calming section. Properties used, 100 at main bulk temperature.</p>	

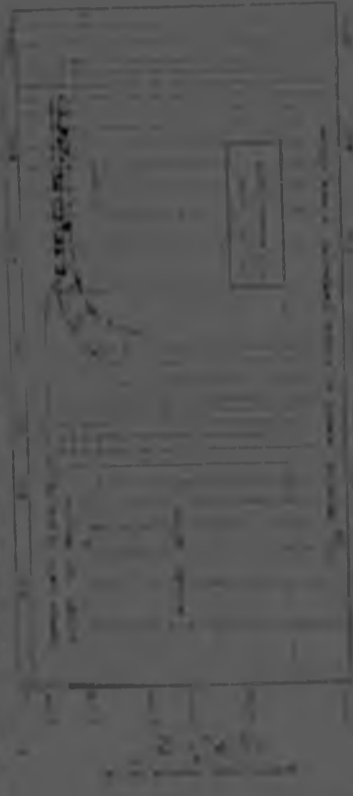
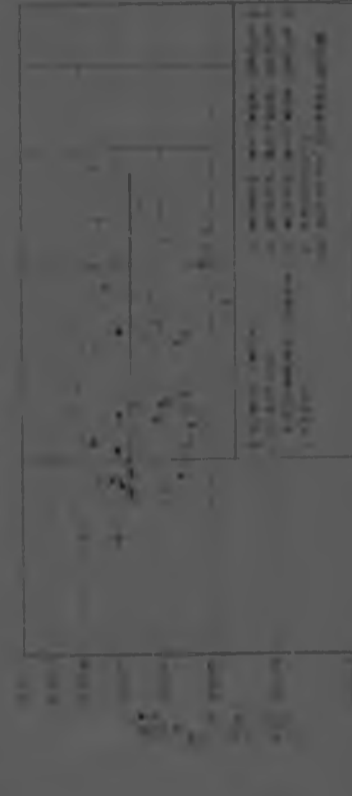
## HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Heating and cooling of fluids without change in state.</p> <p><b>METHOD:</b></p>	<p><b>AUTHOR(S):</b> Drew, T B Mottel, H C McAdams, W H</p>
<p><b>EQUATION PROPOSED:</b></p> <p>Turbulent flow</p> $\left[ \frac{h}{C_p G} \right] \left[ \frac{D_o}{k} \right]^n = 0.023 \left[ \frac{D_o}{L} \right]^{-0.2}$ <p><math>n = 0.6</math> heating <math>n = 0.7</math> cooling</p> <p>Stratified flow</p> $\left[ \frac{h}{k} \right] = 1.85 \left[ \frac{D_o}{L} \right]^{1/4} \left[ \frac{D_o}{L} \right]^{1/4} \left[ 1 + 0.015 G^{0.7} \right]$	<p><b>SOURCE:</b> Trans. AIChE 32, 271 (1936)</p>
<p><b>REYNOLDS:</b></p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p>	
<p><b>AUXILIARY INFORMATION:</b> Results are presented with no Latin for choice.</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES


<p>FLUID(S): Oil, water METHOD: Steam heating</p>	<p>AUTHOR(S): R. Artmann, R. C. Southwell, C. J. Albert, G. Gross, H. L. Wislizenus, E. D. Lansing, N. E. Bogner, L. M. K.</p>
<p>EQUATION PROPOSED:</p> $Nu_{gas} = 1.75 F_1 \sqrt{Gr_{gas}} + 0.0722 F_2 \left[ \frac{Gr_{gas} D}{L} \right]^{0.75}$ $F_1 = \left[ \frac{1 + Nu_{gas}}{Gr_{gas}} \right] \left[ \frac{2 \times \frac{1 + Nu_{gas}}{Gr_{gas}}}{2 - \frac{1 + Nu_{gas}}{Gr_{gas}}} \right]$ $F_2 = \frac{6}{8} \int_0^1 \left[ \frac{x}{L} \right]^{1/4} dx + \frac{Nu_{gas}}{F_1 Gr_{gas}} \left[ \frac{x}{L} \right]^{3/4} \Big _0^1$ <p>The use of <math>Nu_{gas}</math> exponent 0.75 in the Gr group is suggested as being superior to 0.75.</p> <p>REYNOLDS: Viscous flow PRANDTL: 49 GRASHOF: <math>10^5</math> to <math>10^6</math></p> <p>HORIZONTAL/VERTICAL: Vertical LENGTH/DIAMETER: 20, 176, 297, 692</p> <p>AUXILIARY INFORMATION: Arithmetic mean temperature is used for properties of Gr and <math>Nu_{gas}</math> in refer to wall temperature.</p>	<p>SOURCE: Trans. AIChE 493 (1942)</p> 

**HEAT TRANSFER COEFFICIENTS IN TUBES**

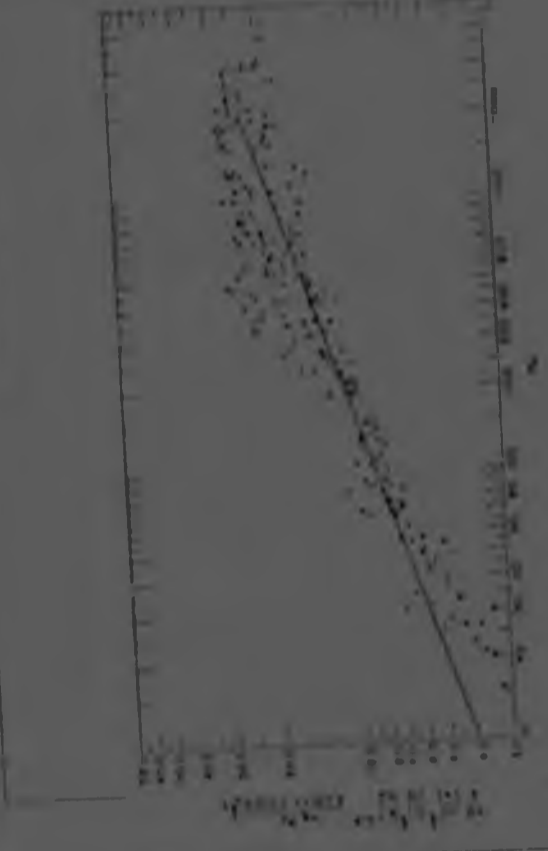
<p><b>FLUID(S):</b> OIL. Askerid (a chlorinated hydrocarbon)</p> <p><b>METHOD:</b> Water cooling in double pipe exchanger. Flow vertically downwards.</p>	<p><b>AUTHOR(S):</b> Harris, B H Sims, M W</p>
<p><b>EQUATION PROPOSED:</b></p> $\frac{h}{C_p G} = 0.0067 \left[ \frac{C_p G}{k} \right]^{-0.8} \left[ \frac{L}{D} \right]^{0.14}$ <p align="center">agreement of data is 88 ± 2%</p> $Nu = 0.0057 Re Pr^{0.74} \left[ \frac{L}{D} \right]^{0.14}$ <p align="center"><math>Re = 3\ 500 - 11\ 000</math>     <math>\frac{L}{D} = 0.20 - 0.76</math></p>	<p><b>SOURCE:</b> Trans. AICHE, 469 (1952)</p>  
<p><b>REYNOLDS:</b> Transitional</p> <p><b>PRANDTL:</b> 35 - 140</p> <p><b>GRASHOF:</b></p>	<p><b>HORIZONTAL/VERTICAL:</b> Vertical</p> <p><b>LENGTH/DIAMETER:</b> 234</p>
<p><b>AUXILIARY INFORMATION:</b> Properties based on average liquid temperature</p>	



HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S)      Oil</p> <p>METHOD     Stream (upward)</p>	<p>AUTHOR(S)    Kern, D O Othmer, L F</p>
<p>EQUATION PROPOSED:</p> $Nu = 7.86 \left[ \frac{Pe^0 \frac{d}{L}}{L} \right]^{0.14} \left[ \frac{d}{L} \right]^{0.14} \frac{2.25 \cdot 10^{-4} \cdot 0.010 Gr^{0.75}}{100 \cdot Re}$ <p>which is a corrected form of the Sieder and Tate equation.</p>	<p>SOURCE         Trans. AICHE 57: 1133</p> 
<p>REYNOLDS:    Laminar</p> <p>PRANDTL:     60 - 1000</p> <p>GRASHOF:     300 - 22.75 x 10<sup>4</sup> LENGTH/DIAMETER: 48-100-100</p> <p>AUXILIARY INFORMATION    3 ft entrance section    Properties at bulk average temperatures.</p>	<p>HORIZONTAL/VERTICAL: Horizontal</p>

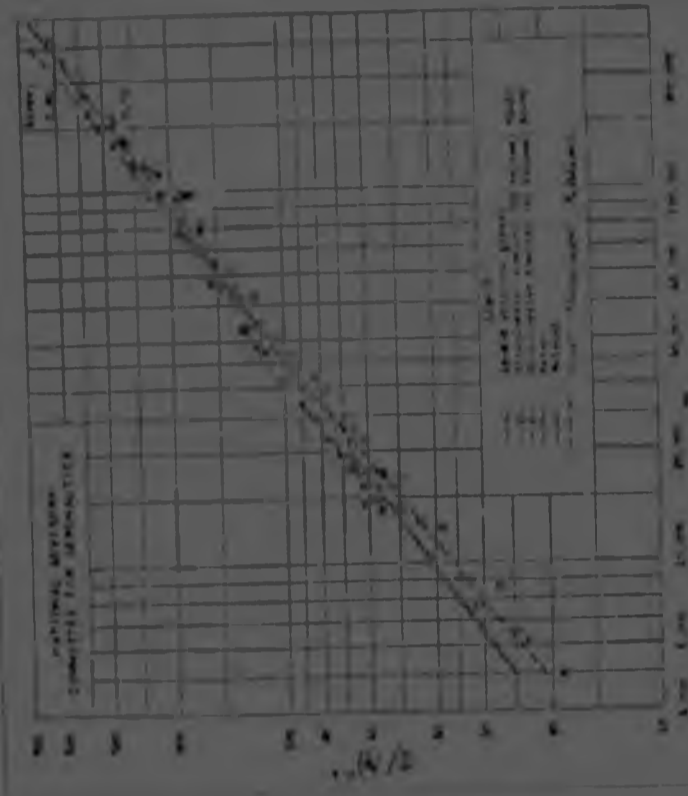
HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Oil METHOD: Steam-heating</p>	<p>AUTHOR(S): Kern, D O. Othman, D F</p>
<p>EQUATION PROPOSED:</p> $Nu = 1.88 \left[ \frac{h_{i0} f}{L} \right]^{1/4} \left[ \frac{D}{L} \right]^{0.74} \frac{2.25 (1 + 0.010 G_r^{1/4})}{\ln Re}$ <p>where is a corrected form of the Sieder and Tate equation.</p>	<p>SOURCE: Trans. AIChE 5:7 (1943)</p> 
<p>REYNOLDS: Laminar PRANDTL: 60 - 1 000 GRASHOF: 300 - 22,75 x 10<sup>4</sup> LENGTH/DIAMETER: 48 100 100 AUXILIARY INFORMATION: 3 ft entrance section. Properties at bulk average temperature.</p>	<p>HORIZONTAL/VERTICAL: Horizontal</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): All METHOD: From others' results.</p>	<p>AUTHOR(S): Hagen, H. SOURCE: Z. VDI Berichts Verfertigungstechnik, (4) 91 (1913)</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.116 \left[ Re^2 \lambda - 125 \right]^{0.125} \rho^{0.125} \left[ 1 + \left( \frac{d}{L} \right)^{0.4} \right] \left[ \frac{\mu}{\mu_s} \right]^{0.14}$ $Nu = 3.65 + \left[ \frac{0.0668 \left( \frac{Re \lambda}{L} \right)^{0.4}}{1 + 0.045 \left( \frac{Re \lambda}{L} \right)^{0.4}} \right]^{0.75}$ <p style="text-align: center;">in turbulent flow in laminar flow</p>	
<p>REYNOLDS: Turbulent and viscous PRANDTL: GRASHOF:</p>	<p>HORIZONTAL/VERTICAL: Horizontal LENGTH/DIAMETER:</p>
<p>AUXILIARY INFORMATION</p>	

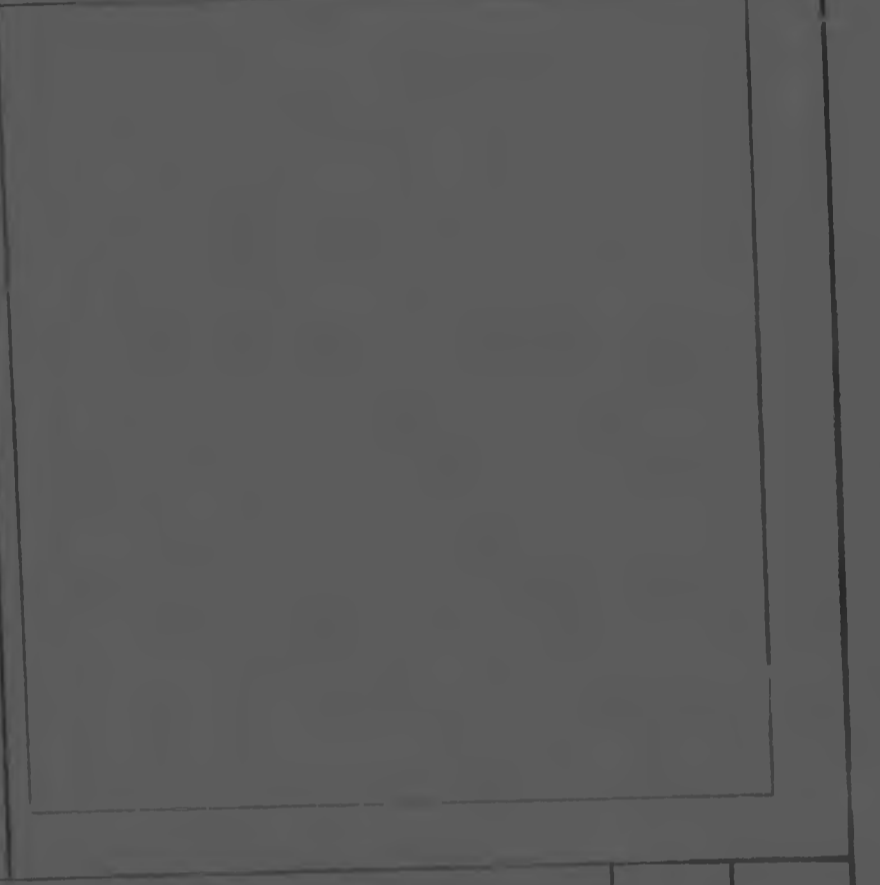
HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUIDS</b>    Engine cooling water, oil, steam</p> <p><b>METHOD</b>    Convective heat transfer</p>	<p><b>AUTHOR(S)</b>    Bernoulli, E. Eaton, C. S.</p>
<p><b>EQUATION PROPOSED</b></p> <p><math display="block">Nu = 0.048 Re^{0.8} Pr^{0.4}</math></p> <p>constant value of 1.125</p>	<p><b>SOURCE</b>    NACA Air Res. Rep. F-130, 1944</p>
<p><b>REYNOLDS</b>    5000 - 300 000    <b>HORIZONTAL/VERTICAL</b>    Horizontal</p> <p><b>PRANDTL</b>    2 - 5</p> <p><b>GRASHOF</b>    <b>LENGTH/DIAMETER</b>    48</p>	
<p><b>AUXILIARY INFORMATION</b>    Properties are taken at the mean bulk temperature</p>	

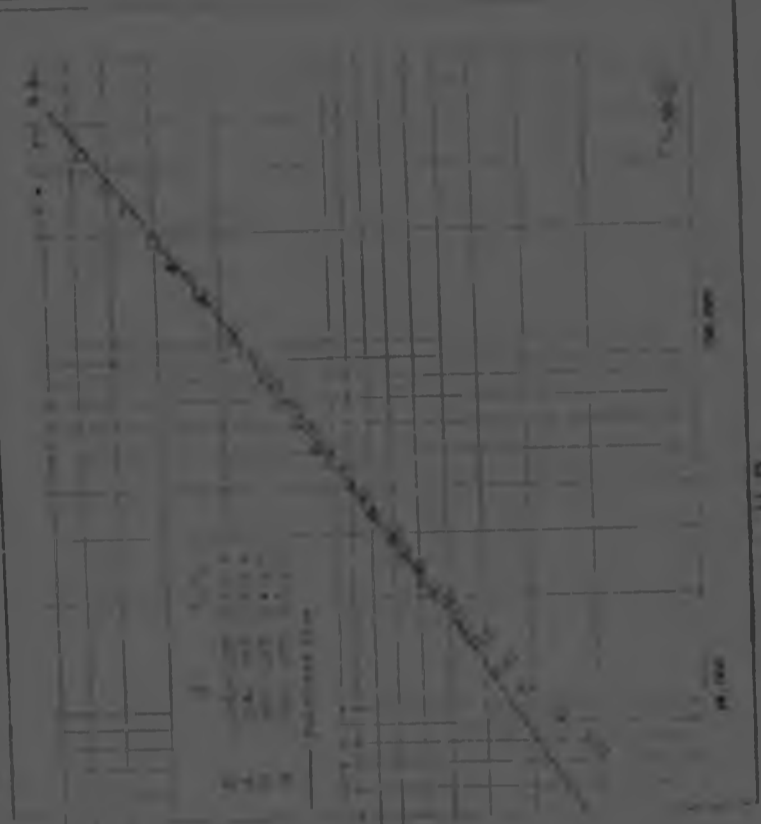
HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air</p> <p>METHOD:</p>	<p>AUTHOR(S): Ilin, L. N. Kontofortosova No. 1 (1951)</p>
<p>EQUATION PROPOSED</p> $Nu_c = C Re_c^m \left[ \frac{T_w}{T_c} \right]^n$ <p> <math>C = 0.5 - 0.9</math>    <math>m = 0.8 - 1.2</math>    <math>n = 2.3</math>  <math>C = 0.018</math>    <math>m = 0.0212</math>    <math>n = 0.0223</math>  <math>C = 0</math>    <math>m = -0.21</math>    <math>n = -0.58</math> </p>	<p>SOURCE: Tesis Penelitian (1970)</p>
<p>REYNOLDS: <math>7 - 60 \times 10^3</math>    HORIZONTAL/VERTICAL:</p> <p>PRANDTL:</p> <p>GRASHOF:</p>	
<p>AUXILIARY INFORMATION</p> <p><math>\frac{T_w}{T_c} = 0.56 - 2.3</math></p>	<p>LENGTH/DIAMETER: 59. 62</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S) Air</p> <p>METHOD</p>	<p>AUTHOR(S): Humble, L. V. Lofstrom, W. H. Deason, L. G.</p> <p>NACA Rept. 1020 (1951)</p> <p>SOURCE: Throughput (1970)</p>
<p>EQUATION PROPOSED:</p> $Nu_D = 0.023 Re_D^{0.8} Pr_D^{0.4} \frac{T_m}{T_b} \quad (S > 0.05)$ $\mu = 0 \text{ to } \frac{T_m}{T_b} < 1$ $\mu = -0.05 \text{ to } \frac{T_m}{T_b} > 1$	
<p>REYNOLDS: <math>Re = 300 \times 10^3</math></p> <p>PRANDTL:</p> <p>GRASHOF:</p>	<p>HORIZONTAL/VERTICAL:</p> <p>LENGTH/DIAMETER: 30 - 120</p> <p>AUXILIARY INFORMATION: <math>\frac{T_m}{T_b} = 0.46 - 3.5</math></p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air</p> <p>METHOD: Electrical analogy</p>	<p>AUTHOR(S): Dentler, R G Egan, C S</p>
<p>EQUATION PROPOSED</p> $U^* = \int_0^{\infty} \frac{dx}{(1 + 6x^2)^{1/2} + \frac{2.3U^*k}{k_f} \frac{x}{D}}$ $f^* = \int_0^{\infty} \frac{dx}{(1 + 6x^2)^{1/2} + \frac{2.3U^*k}{k_f} \frac{x}{D}}$	<p>SOURCE: NACA TN 2629 (1952)</p> 
<p>REYNOLDS: Turbulent</p> <p>PRANDTL: 0.73</p> <p>GRASHOF:</p>	<p>HORIZONTAL/VERTICAL: Horizontal</p> <p>LENGTH/DIAMETER: 87.0, 87</p>
<p>AUXILIARY INFORMATION</p> <p><math>t_{0,A} = 0.61(t_w - t_0) + t_0</math></p> <p><math>\rho</math> refers to wall</p>	<p><math>Re_{D,A} = \frac{\rho_0 \mu U_0 D}{\mu_0}</math></p>

# HEAT TRANSFER COEFFICIENTS IN TUBES

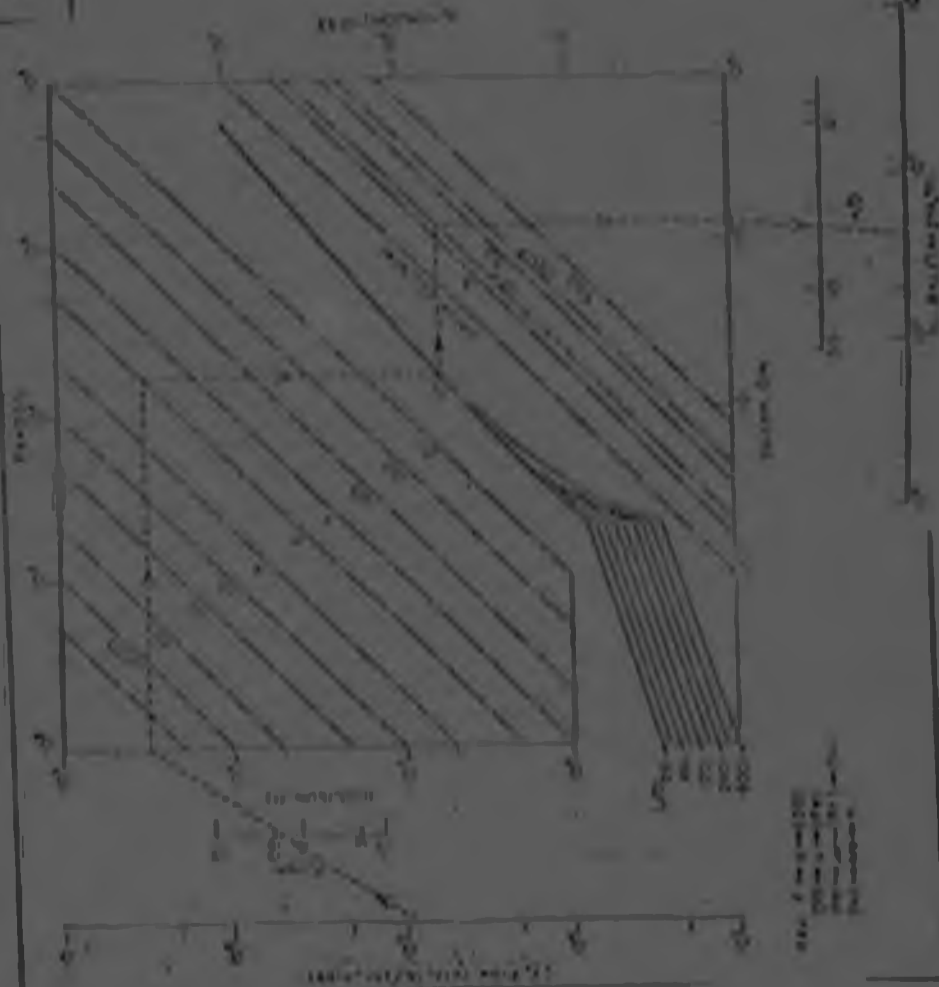
**AUTHOR(S):** TANG, Y S

**SOURCE:** Chem. Eng. 174, Dec. 1953

This is a nomogram of the equations given by McAdams in "Heat Transmission" (1942)

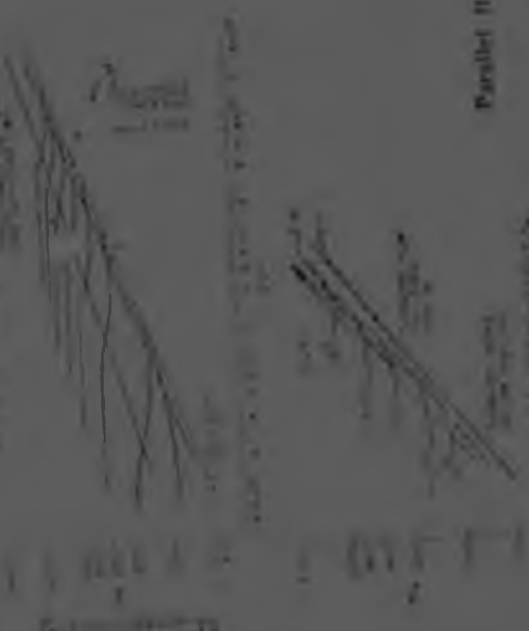

**FLUID(S):**

**METHOD:**



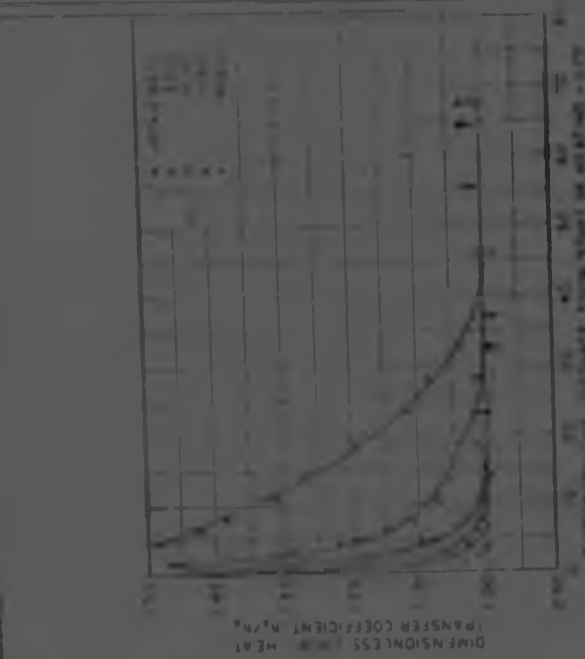


HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air</p> <p>METHOD: Steam heating</p>	<p>AUTHOR(S): Edvort, E B G Ducault, A J</p> <p>SOURCE: Trans ASME 803 (1954)</p>
<p>EQUATION PROPOSED:</p> <p>his equation is proposed and the results are presented for parallel flow</p>  <p>Parallel flow</p>	 <p>Counter flow</p>
<p>REYNOLDS: Turbulent</p> <p>PRANDTL: 0.7</p> <p>GRASSHOFF:</p>	<p>HORIZONTAL/VERTICAL: Vertical</p> <p>LENGTH/DIAMETER: 5</p>
<p>AUXILIARY INFORMATION: The equation of Martinielli and Boelter is examined.</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES


<p><b>FLUID(S):</b> OIL, water</p> <p><b>METHOD:</b> Electrical heating for the determination of thermal entry length</p>	<p><b>AUTHOR(SI):</b> Hartnett, J P</p>
<p><b>EQUATION PROPOSED:</b> No equation is proposed. Heat transfer coefficients are presented graphically as a function of the tube length</p>	<p><b>SOURCE:</b> Trans ASME 1211, (Nov 1955)</p>
<p><b>REYNOLDS:</b> 2200 17000</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p> <p><b>AUXILIARY INFORMATION:</b></p>	<p><b>REYNOLDS:</b> 2200 17000</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p> <p><b>AUXILIARY INFORMATION:</b></p>
<p><b>REYNOLDS:</b> 2200 17000</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p> <p><b>AUXILIARY INFORMATION:</b></p>	<p><b>REYNOLDS:</b> 2200 17000</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p> <p><b>AUXILIARY INFORMATION:</b></p>



HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air METHOD:</p>	<p>AUTHOR(S): Blaukoz, J. Sawinski, O. Combustion and Underhouse Engineering, New York (1956)</p>
<p>EQUATION PROPOSED</p> $Nu_c = 0.003 \rho_c^0.8 \mu_c^0.4 \left[ \frac{v_c}{\nu_c} \right]^{0.4}$	<p>SOURCE: Through Petukhov (1970)</p>
<p>REYNOLDS: <math>1.24 \times 10^5</math> PRANDTL: GRASHOF:</p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER: 29 - 72</p>
<p>AUXILIARY INFORMATION</p>	<p><math>\frac{T_w}{T_c} = 1.1 - 1.73</math></p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Cold water, molasses, sugar solution</p> <p><b>METHOD:</b> Steam jacket heating</p>	<p><b>AUTHOR(S):</b> Friend, W L Mettner, A B</p>
<p><b>EQUATION PROPOSED:</b></p> $St = \frac{h}{C_p G} = \frac{1}{2} \left( 1.76 + 11.8 \sqrt{\frac{1}{2}} (Pr - 1) \right)^{0.15}$ <p style="text-align: center;">+ is the second term</p> <p>The standard deviation from the experimental results is 3.4%</p> <p>The error of <math>\left[ \frac{h}{C_p G} \right]^{0.15}</math> is also specified</p>	<p><b>SOURCE:</b> AIChE J. 8, 703 (1952)</p> 
<p><b>REYNOLDS:</b> Turbulent</p> <p><b>PRANDTL:</b> 50 - 500</p> <p><b>GRASHPF:</b></p>	<p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p> <p><b>LENGTH/DIAMETER:</b> 158</p>
<p><b>AUXILIARY INFORMATION:</b> 10 ft calming section.</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S):</p> <p>METHOD:</p>	<p>AUTHOR(S): Petukhov, B S Kirillov</p> <p>Terminologica, 4, 63 (1951)</p>
<p>EQUATION PROPOSED</p> $Nu = \frac{0.023 Re^{0.8} Pr^{0.4}}{1 + 12.7 Re^{0.45} Pr^{0.16} \left[ \frac{1}{1 + 1.71 Re^{-0.1} Pr^{0.1}} \right]}$	<p>SOURCE: Sleicher, C A. Int Heat Mass, (1975)</p>
<p>REYNOLDS: 0.5 - 2 000</p> <p>PRANDTL: 0.7 - 1000</p> <p>GEOMETRY: HORIZONTAL/VERTICAL</p> <p>AUXILIARY INFORMATION: LENGTH/DIAMETER</p>	

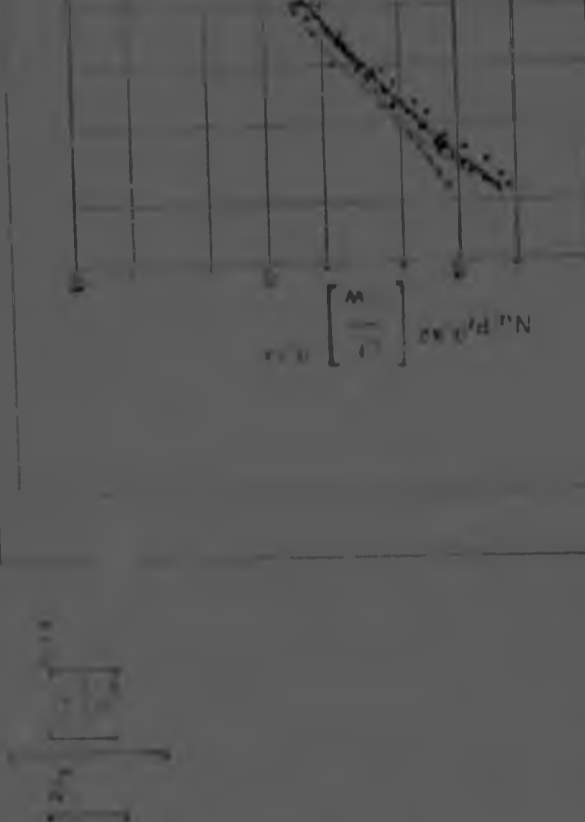
# HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air</p> <p>METHOD: Using the results of others</p> <p>EQUATION PROPOSED:</p> $Nu = 3.66 \left( 1 + \frac{0.0677(Pr)^{1/4}}{L} \right) \left( 1 + 0.49(Pr)^{1/4} \right)^{1/4}$	<p>AUTHOR(S): Stephan, K</p> <p>SOURCE: Chem. Ing. Tech. 31, 773 (1960)</p>
<p>REYNOLDS: Laminar</p> <p>PRANDTL: GRACE-OF:</p> <p>AUXILIARY INFORMATION: Face is in German</p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER:</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

FLUID(S):	AUTHOR(S):		Carter, J. Prestborg, B S
METHOD:	Summary		SOURCE:
EQUATION PROPOSED:			Chem. Eng. (6), (28) (7) (1959)
General Form			
Coefficient for Free Convection inside Tubes $h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4}$ $(\nu / \alpha)^{1/4}$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	
	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	
For $Pr > 1000$ For laminar flow, $Pr > 1000$ For $Pr > 1000$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	
For $Pr > 1000$ For $Pr > 1000$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	
For $Pr > 1000$ For $Pr > 1000$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	
For $Pr > 1000$ For $Pr > 1000$ For $Pr > 1000$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$	
REYNOLDS:	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$		
PRANDTL:	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$		
GRASHOF:	$h_c = 0.52 (k_f / D)^{1/4} (g \beta \Delta T / \nu)^{1/4} (\nu / \alpha)^{1/4}$		
AUXILIARY INFORMATION			
LENGTH/DIAMETER:			

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUIDS:</p>	<p>AUTHORS: Hansen, U.</p>
<p>METHOD:</p>	<p>System: Horizontal; 9; 70; 11000</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.077 (k/2.44)^{0.47} Pr^{0.47} \left[ \frac{1}{1 + 0.0125 Pr^{0.175}} \right]^{0.4} \left[ \frac{1}{1 + \frac{0.4}{Pr} \left( \frac{d}{L} \right)^{0.1}} \right]^{0.4}$ <p>for turbulent flow</p> $Nu = 3.65 + \frac{0.19 \left( \frac{Pr d}{L} \right)^{0.8}}{1 + 0.115 \left( \frac{Pr d}{L} \right)^{0.467}}$ <p>for laminar flow</p>	
<p>REYNOLDS:</p>	<p>Turbulent</p>
<p>PRANDTL:</p>	<p>HORIZONTAL/VERTICAL</p>
<p>GRASHOF:</p>	<p>LENGTH DIAMETER</p>
<p>AUXILIARY INFORMATION</p>	



HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): H<sub>2</sub>O vapor</p> <p>METHOD:</p>	<p>AUTHOR(S): Wright, F C Walters, H.</p> <p>WADC Tech Rep. 59 - 423, Aug (1959)</p>
<p>EQUATION PROPOSED:</p> $Nu_D = 0.023 Re_D^{0.8} Pr_D^{1/3} \left[ \frac{T_w}{T_b} \right]^{-0.5 Pr} \left( \frac{L}{D} \right)^{-0.15}$	<p>SOURCE: Through Petukhov (1970)</p>
<p>REYNOLDS: PRANDTL: GRAZHOFF:</p> <p>AUXILIARY INFORMATION</p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER:</p> <p><math>\frac{T_w}{T_b} = 1 + 4</math></p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): METHOD:</p>	<p>AUTHOR(S): Potukh, B S Naidu, L D Teylammogilica, G. T2 (1959)</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.21 \left[ RePr \frac{d}{L} \right]^{0.8} \left[ \frac{d}{L} \right]^{0.05} \left[ \frac{L}{d} \right]^{0.05}$	<p>SOURCE: Herbert (1971)</p>
<p>REYNOLDS: PRANDTL: GRASSE:</p>	<p>HORIZONTAL/VERTICAL: Vertical LENGTH/DIAMETER:</p>
<p>AUXILIARY INFORMATION</p>	

## HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Hydrogen, Helium</p> <p>METHOD:</p>	<p>AUTHOR(S): McCarthy, J. P. Woff, H.</p> <p>Rep. No. AF-60-12 Rocketdyne, Canoga Park, California (1960)</p>
<p>EQUATION PROPOSED:</p> $Nu_x = 0.045 Re_x^{0.4} \left[ \frac{L}{D} \right]^{-0.35} \left[ \frac{T_w}{T_b} \right]^{-0.7}$	<p>SOURCE: Through Population (1970)</p>
<p>REYNOLDS: <math>5 &lt; 1500 \times 10^3</math> HORIZONTAL/VERTICAL:</p> <p>PRANDTL:</p> <p>GRASHOF:</p>	
<p>AUXILIARY INFORMATION</p>	<p>LENGTH/DIAMETER: 26 - 87</p> <p><math>\frac{T_w}{T_b} = 1.5 - 9.9</math></p>

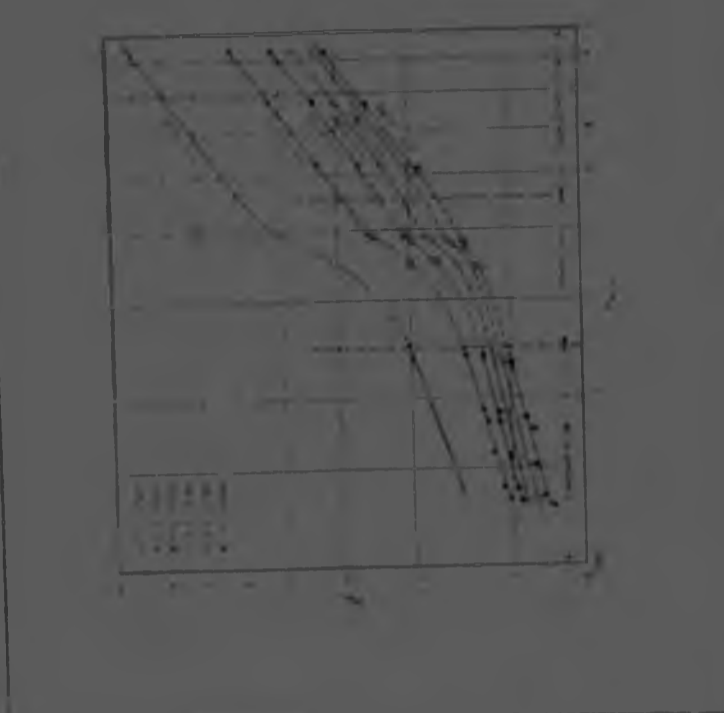
HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Hydrogen</p> <p>METHOD:</p>	<p>AUTHOR(S): McCarthy, J R Wulf, H.</p> <p>Am. Rocket Soc. J. 30, No. 4 (1960)</p>
<p>EQUATION PROPOSED:</p> $Nu_b = 0.023 Re_b^{0.8} Pr_b^{0.4} \left[ \frac{T_w}{T_b} \right]^{-0.14}$	<p>SOURCE: Through Petukhov (1970)</p>
<p>REYNOLDS: <math>7 \times 10^3</math> HORIZONTAL/VERTICAL:</p> <p>PRANDTL:</p> <p>GRASHOF:</p>	<p>LENGTH/DIAMETER: 43.67</p>
<p>AUXILIARY INFORMATION</p>	<p><math>\frac{T_w}{T_b} = 1.5 + 2.8</math></p>


HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Helium</p> <p>METHOD:</p>	<p>AUTHOR(S): Taylor, M F Kirchgesse, T A.</p> <p>Am Rocket Soc. J. 30, No 4 (1960)</p>
<p>EQUATION PROPOSED:</p> $Nu_s = 0.023 Re_s^{0.8} Pr_s^{0.4}$ $0.001 < \frac{L}{D} < 10$	<p>SOURCE: Through Petukhov (1970)</p>
<p>REYNOLDS: <math>3.2 \times 60 \times 10^3</math></p> <p>PRANDTL:</p> <p>GRASHOF:</p>	<p>HORIZONTAL/VERTICAL:</p> <p>LENGTH/DIAMETER: 50.92</p>
<p>AUXILIARY INFORMATION</p>	<p><math>\frac{T_s - T_f}{T_s} = 1.6 \times 3.9</math></p>

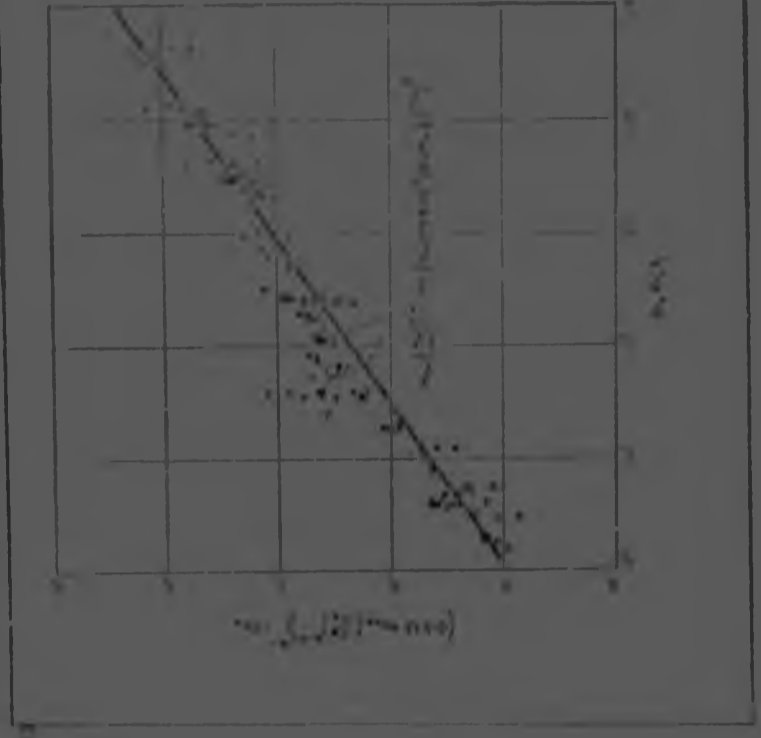
HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air METHOD: Steam heating</p>	<p>AUTHOR(S): Jackson, T W Sprack, J M Furdy, K R</p>
<p>SOURCE: AICHE J. L. 38 (1951)</p>	
	
<p>EQUATION PROPOSED:</p> $Nu = 2.67 Gr^{1/4} \left[ 1 + \frac{0.00872}{Gr^{1/2}} \left[ \frac{Gr Pr}{1 + Nu} \left( 1 + \frac{Pr}{0.1} \right) \right]^{1/4} \right]^{1/4}$ <p>for lamina flow <math>60 &lt; Gr &lt; 1300</math></p> $Nu = 0.023 Re^{0.8} Pr^{1/3} \left[ 1 + 3 \frac{d}{L} \right]^{1/4} \quad \text{for } Re > 7000$ <p>Graph is based on the log mean temperature difference</p>	
<p>REYNOLDS: Turbulent and laminar PRANDTL: 0.7 GRAEHOF: LENGTH/DIAMETER:</p>	<p>HORIZONTAL/VERTICAL: Horizontal AUXILIARY INFORMATION: Properties at bulk average temperatures.</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Water, air</p> <p><b>METHOD:</b> Electrical heating</p>	<p><b>AUTHOR(S):</b> Fife, A. J.</p>
<p><b>EQUATION PROPOSED:</b></p> $Nu = 0.023 Re^{0.8} Pr^{0.4} \quad \text{turbulent flow}$ $Nu = 4.36 (1 + 0.06 Gr^{0.5}) \quad \text{laminar flow}$	<p><b>SOURCE:</b> Int. J. Heat Mass A, 105 (1961)</p>
<p><b>REYNOLDS:</b> 300 - 100000</p> <p><b>PRANDTL:</b></p> <p><b>GR./SHOF:</b></p> <p><b>AUXILIARY INFORMATION:</b> Physical properties calculated at the bulk average temperature.</p>	<p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p> <p><b>LENGTH/DIAMETER:</b></p> 

HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Glycol, ethanol, water  <b>METHOD:</b> Water cooling and heating</p>	<p><b>AUTHOR(S):</b> Oliver, D R</p>
<p><b>EQUATION PROPOSED:</b></p> $Nu_{d,m} \left[ \frac{\mu}{\mu_s} \right]^{0.14} = 1.35 \left[ Gr_{d,m} + 5.6 \times 10^{-4} \left[ Gr_{d,m}^{0.75} \frac{L}{d} \right]^{0.10} \right]^{1/4}$ <p>for <math>Gr &gt; 2 Nu_{d,m}</math>          errors do not exceed <math>\pm 20\%</math></p>	<p><b>SOURCE:</b> Chem. Eng. 56, 12, 335 (1962)</p>
<p><b>REYNOLDS:</b> Laminar  <b>PRANDTL:</b>  <b>GRASHOF:</b></p> <p><b>HORIZONTAL/VERTICAL:</b> Horizontal  <b>LENGTH/DIAMETER:</b> 72</p> <p><b>AUXILIARY INFORMATION</b> 1.5 ft calming section. Mean fluid temperature used for properties.</p>	



## HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Helium, Hydrogen</p> <p>METHOD:</p>	<p>AUTHOR(S): Wieland, W F</p> <p>AIChE Symp. Nuc. Eng. Heat Transfer, Chicago, (1962)</p>
<p>EQUATION PROPOSED:</p> $Nu_L = 0.021 Re_L^{0.8} Pr_L^{0.4}$ <p><math>x_L</math> from the entry</p>	<p>SOURCE: Through Petukhov (1970)</p>
<p>REYNOLDS:</p> <p>PRANDTL:</p> <p>GRASHOF:</p>	<p>HORIZONTAL/VERTICAL:</p> <p>LENGTH/DIAMETER: 250</p>
<p>AUXILIARY INFORMATION</p>	<p><math>\frac{T_w}{T_b} &lt; 2.0</math></p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S):</p> <p>METHOD:</p>	<p>AUTHOR(S): Kurihara</p>
<p>EQUATION PROPOSED:</p> $\frac{Nu_c}{Nu_{opt}} = \left[ \frac{2}{\sqrt{\frac{Pr}{Pr_c} - 1}} \right]^4$ <p>Toshio Kurihara and R. P. Taylor (1963)</p>	<p>SOURCE: The Element of heat exchange (Hiroaki) 1962</p>
<p>REYNOLDS:</p> <p>PRANDTL:</p> <p>GRASHOF:</p>	<p>HORIZONTAL/VERTICAL:</p> <p>LENGTH/DIAMETER:</p>
<p>AUXILIARY INFORMATION</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Hydrogen, Helium</p> <p>METHOD:</p>	<p>AUTHOR(S): Taylor, M F                  ("Heat Transfer and Fluid Mechanics Institute") p 51, Stanford University Press, California (1963).</p>
<p>EQUATION PROPOSED:</p> $Nu_x = 0.021 Re_x^{0.4} Pr_x^{0.4}$	<p>SOURCE: Through Petukhov (1970)</p>
<p>REYNOLDS:</p> <p>PRANDTL:</p> <p>GRASHOF:</p> <p>AUXILIARY INFORMATION: <math>\frac{T_w}{T_b} = 1.5 - 5.6</math></p>	<p>HORIZONTAL/VERTICAL:</p> <p>LENGTH/DIAMETER: 77</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUIDS: METHOD:</p>	<p>AUTHOR(S): K. S. G. K. S. G. K. S. G. S. G. K. S. G. K. S. G. K. S. G.</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.023 Re^{0.8} Pr^{0.4} \left[ \frac{\mu}{\mu_s} \right]^{0.14}$	<p>SOURCE: K. S. G. K. S. G. K. S. G.</p>
<p>REYNOLDS: PRANDTL: GRASHOF: AUXILIARY INFORMATION</p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER:</p>

## HEAT TRANSFER COEFFICIENTS IN TUBES

FLUID(S): METHOD:	AUTHOR(S): Krasnothchikov, E. A. Sukomel, A. S. (1963)
EQUATION PROPOSED:  $Nu = 0.35 \left[ Re Pr \frac{d}{L} \right]^{0.6} \left[ Gr Pr \frac{d}{L} \right]^{0.1}$	SOURCE: Kuznetsov (1961)
REYNOLDS: PRANDTL: GRASHOF:	HORIZONTAL/VERTICAL: LENGTH/DIAMETER:
AUXILIARY INFORMATION	

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Nitrogen</p> <p>METHOD:</p>	<p>AUTHOR(S): Kudlov, U V Makogin, Yu, S</p> <p>High Temp. Heat Physics, 1, No. 2, (1963)</p>
<p>EQUATION PROPOSED:</p> $Nu_N = 0.021 Re_m^{0.6} Pr_m^{0.4} \left[ \frac{T_w}{T_b} \right]^{-0.8}$ <p style="text-align: center;">or <math>\frac{h}{D} &gt; 30</math></p>	<p>SOURCE: Through Metakhan (1970)</p>
<p>REYNOLDS: <math>7 \times 10^3</math> HORIZONTAL/VERTICAL:</p> <p>PRANDTL:</p> <p>GRASHOF: <math>1.78</math></p>	
<p>AUXILIARY INFORMATION <math>\frac{T_w}{T_b} = 1.1 \sim 2.3</math></p>	

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Water METHOD: Electrically heated.</p>	<p>AUTHOR(S): Dippney, D F Saborsky, R H</p>
<p>EQUATION PROPOSED:</p> $\frac{2C_h}{C_c} = N^{0.19}$ <p>where</p> $C_h = \frac{\pi_s}{U_m C_p (T_m - T_c)}$ $C_c = \frac{2k_s}{r U_m^2}$ <p><math>T_c</math> = mean temperature</p>	<p>SOURCE: Int. J. Heat Mass. 6: 329 (1963)</p>
<p>REYNOLDS: 14 000 - 500 000 PRANDTL: HORIZONTAL/VERTICAL: Vertical GRASHOF: LENGTH/DIAMETER: 00</p>	
<p>AUXILIARY INFORMATION: 45 D entrance region.</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES


<p>FLUID(S): Incompressible fluid</p> <p>METHOD: Numerically</p>	<p>AUTHOR(S): Petukhov, B S Ponomarev, V N</p>
<p>EQUATION PROPOSED:</p> $Nu = \frac{\lambda}{R} \frac{RePr}{1.07 + 12.7 \sqrt{\frac{1}{R} (Pr^{0.4} - 1)}}$ <p>not physical properties.</p> <p>For high temperature fluids</p> $\frac{Nu_s}{Nu_{s1}} = \left[ \frac{T_s}{T_{s1}} \right]^n \quad n = -0.82$	<p>SOURCE: High Temperature, 1, 69 (1963)</p> <p>The equation was proposed by Petukhov and Kirillov in 1958 Teploenergetika No. 4</p>
<p>REYNOLDS: Turbulent</p> <p>PRANDTL: HORIZONTAL/VERTICAL:</p> <p>GRASHOF: LENGTH/DIAMETER:</p> <p>AUXILIARY INFORMATION</p>	




# HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b></p> <p><b>METHOD:</b> Uniform wall temperature</p>	<p><b>AUTHOR(SI):</b> Metzger, B. University of Minnesota (1963)</p>
<p><b>EQUATION PROPOSED:</b></p> $Nu_s = 4.75 Re^{0.4} Pr^{0.41} \left[ \frac{d}{L} \right]^{0.38}$	<p><b>SOURCE:</b> Eckert and Metzger (1964)</p>
<p><b>REYNOLDS:</b> Turbulent</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p> <p><b>AUXILIARY INFORMATION</b></p>	<p><b>HORIZONTAL/VERTICAL:</b> Horizontal</p> <p><b>LENGTH/DIAMETER:</b></p>

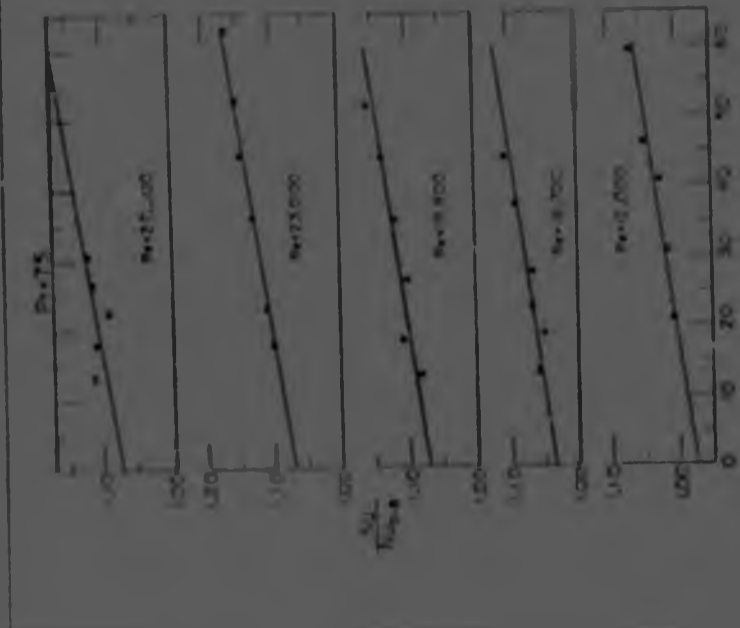
HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> <b>METHOD:</b></p>	<p><b>AUTHOR(S):</b> Metz, B Eckert, E R G</p>
<p><b>EQUATION(S) PROPOSED:</b>  Heat in conjunction with Metz (1963). Valid for uniform wall temperature and uniform mass flux</p>	<p><b>SOURCE:</b> J. of Heat Transfer, 295 (1964)</p>
<p><b>REYNOLDS:</b> <b>PRANDTL:</b> <b>GRASHOF:</b> <b>AUXILIARY INFORMATION</b></p>	<p><b>HORIZONTAL/VERTICAL:</b> Vertical <b>LENGTH/DIAMETER:</b></p> 

HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUIDS:</b> Water</p> <p><b>METHOD:</b> Uniform wall heat flux</p>	<p><b>AUTHOR(S):</b> Allen, R W Eckert, E R G</p>
<p><b>EQUATION PROPOSED:</b></p> $St = 0.023 Re^{-1/4} Pr^{-1/4}$ <p>Correlation is to be better than 1%.</p>	<p><b>SOURCE:</b> Trans. ASME, Vol. 80 (1958)</p>
<p><b>REYNOLDS:</b> Turbulent</p> <p><b>PRANDTL:</b> 7 - 8</p> <p><b>GRAZHOFF:</b> 30 D</p> <p><b>AUXILIARY INFORMATION:</b> Properties taken at bulk temperature 96 O entrance regions.</p>	 <p>Comparison of heat-transfer coefficients with those of other investigators. Summary developed, constant property case. Pr = 8. Uniform wall heat flux.</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Water, oil METHOD: Electrical heating</p>	<p>AUTHOR(S): Matina, J. A. Sparrow, E. M.</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.023 Re^{0.8} Pr^{1/3} \left[ \frac{\mu_b}{\mu_w} \right]^{0.14}$ <p>(i.e. in connection to the Dittus and Boelter equation.)</p>	<p>SOURCE: Chem. Eng. Sci. 19, 963 (1964)</p>
<p>REYNOLDS: Turbulent PRANDTL: 3 - 75 GRASHOF:</p>	
<p>AUXILIARY INFORMATION: 96 D developing section.</p>	<p>HORIZONTAL/VERTICAL: Horizontal LENGTH/DIAMETER: 30</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

FLUID(S): Water

METHOD: Water cooling and heating

AUTHOR(S): Brown, A R  
Thomas, M A

EQUATION PROPOSED

$$Nu = 1.75 \left[ C_p \mu \right]^{0.15} \left[ \frac{C_p G}{k} \right]^{0.74} \left[ \frac{U_m}{U_m} \right]^{0.34}$$

Correlated the majority of data to 8% and the majority of published data to 50%.

SOURCE: J. Mech. Eng. Sci., 7, (4), 440 (1965)



REYNOLDS: Laminar

HORIZONTAL/VERTICAL: Horizontal


PRANDTL:

GRASHOF:  $4 \times 10^4 - 480 \times 10^4$ ; LENGTH/DIAMETER: 72 - 108; 30

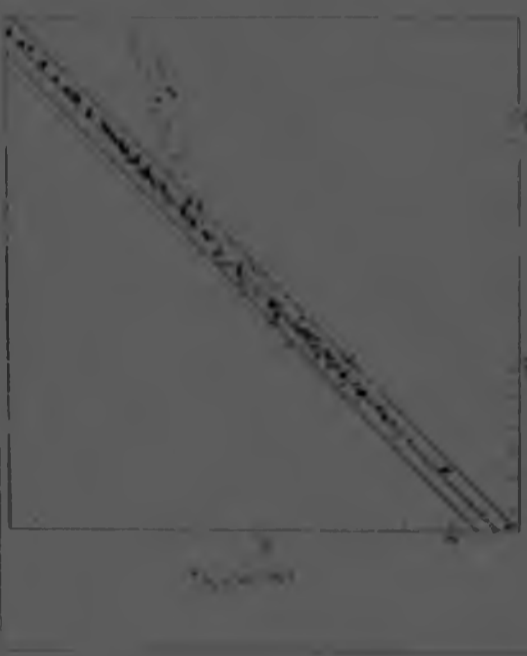
AUXILIARY INFORMATION: All properties are evaluated at the bulk average temperature. Calming section was included.

HEAT TRANSFER COEFFICIENTS IN TUBES


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<p><b>FLUID(S)</b>: Water</p> <p><b>METHOD</b>: Water cooling and heating</p>	<p><b>AUTHOR(S)</b>: Brown, A R Thomist, M A</p>
<p><b>EQUATION PROPOSED</b></p> $Nu = 5.75 \left[ Ge + 0.012 \left[ \frac{G}{G_0} \right]^{0.3} \right]^{1/2} \left[ \frac{\mu_b}{\mu_w} \right]^{0.14}$ <p>Correlates the majority of data by BS and the majority of published data to 50%.</p>	<p><b>SOURCE</b>: J. Mech. Eng. Sci., 7, (4), 440 (1965)</p> 
<p><b>REYNOLDS</b>: Laminar</p> <p><b>PRANDTL</b>: 4 x 10<sup>-4</sup> - 480 x 10<sup>4</sup></p> <p><b>CRASHOF</b>: 4 x 10<sup>4</sup> - 480 x 10<sup>4</sup></p> <p><b>AUXILIARY INFORMATION</b>: All properties are evaluated at the bulk average temperature. Calming section was included.</p>	<p><b>HORIZONTAL/VERTICAL</b>: Horizontal</p> <p><b>LENGTH/DIAMETER</b>: 72, 108, 36</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air, water          METHOD: Condensing steam, rough pipes.</p>	<p>AUTHOR(S): Klotz, V</p>
<p>EQUATION PROPOSED:</p> $Nu_d = 0.05614 Re_{d, \text{film}}^{0.8} Pr^{0.4}$ <p><math>Re_{d, \text{film}}</math> is the <math>Re_d</math> based on the film temperature:</p> $Re_d = \frac{\rho_{\text{film}} \sqrt{\frac{1}{8} \dot{m}}}{\mu_{\text{film}}}$	<p>SOURCE: Int. J. Heat Mass. 9, 639 (1968)</p>  <p>Plot is not of given equation but demonstrates the effect of using <math>Re_{d, \text{film}}</math></p>
<p>REYNOLDS: 4500 - 140000 HORIZONTAL/VERTICAL: Horizontal          PRANDTL: 0.71 - 5.52          GRASHOF: LENGTH/DIAMETER: 31          AUXILIARY INFORMATION <math>\tau_1 = \frac{\tau_2 + 1}{2}</math></p>	

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID:   METHOD:   EQUATION PROPOSED:   <math display="block">h = \frac{0.023 k \rho^{1/4} \mu^{1/4} Pr^{1/4}}{D} \left( \frac{L}{D} \right)^{-0.15} \left( \frac{G}{\mu} \right)^{0.8} \left( \frac{\mu}{\mu_s} \right)^{0.14}</math></p>	<p>AUTHOR(S): Subbotin, G. I.   (Institute of   Mechanics, Leningrad)</p>
<p>REYNOLDS: Turbulent   PRANDTL: 0.7 - 10   GRASHOF:   AUXILIARY INFORMATION:   HORIZONTAL/VERTICAL:   LENGTH/DIAMETER:</p>	<p>SOURCE: High Temp. J. 1957, 1, 100</p>
	



HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Nitrogen</p> <p>METHOD</p>	<p>AUTHOR(S): Perkins, H C Weyner-Schmitt, P</p> <p>Int. J Heat Mass Trans. B. 7, 4811 (1964)</p>
<p>EQUATION PROPOSED:</p> $Nu_D = 0.024 Re_D^{0.4} Pr_D^{0.4} \left[ \frac{T_w}{T_c} \right]^{-0.7}$ $\frac{L}{D} > 40$	<p>SOURCE: Through Database (1970)</p>
<p>REYNOLDS: <math>18 \times 10^3</math> HORIZONTAL/VERTICAL:</p> <p>PRANDTL:</p> <p>GRASHOF: LENGTH/DIAMETER: 100</p>	
<p>AUXILIARY INFORMATION</p> $\frac{T_w}{T_c} = 1.3 \text{ to } 7.5$	

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air, carbon dioxide, argon</p> <p>METHOD:</p>	<p>AUTHOR(S): Lelchuk, V. L. Elpimov, G. I. Farkov, Yu. P.</p> <p>Heat and Mass Transfer, <i>Mosk. VI</i>, (1965)</p>
<p>EQUATION PROPOSED:</p> $Nu_D = 0.021 Re_D^{0.8} Pr^{0.4} \left[ \frac{\mu_f}{\mu_w} \right]^{-0.14}$ $\text{at } \frac{l}{D} > 40$	<p>SOURCE: Through Putukov (1970)</p>
<p>REYNOLDS: <math>14 - 600 \times 10^3</math></p> <p>PRANDTL:</p> <p>GRASHOF:</p>	<p>HORIZONTAL/VERTICAL:</p> <p>LENGTH/DIAMETER: 77 - 206</p>
<p>AUXILIARY INFORMATION</p>	<p><math>\frac{T_w}{T_f} = 1.1 - 2.7</math></p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Nitrogen METHOD:</p>	<p>AUTHOR(S): Petukhov, R S Kirilina, V V Maklennik, V N Int. Heat Trans. Conf. 3, Chicago, p 29 (1966)</p>
<p>EQUATION PROPOSED:</p> $Nu_D = 0.023 Re_D^{0.8} Pr_D^{0.4} \left[ \frac{T_w}{T_b} \right]^{-0.11}$ $\text{for } \frac{L}{D} > 60$ $h = - \left[ 0.9 \log \left[ \frac{T_w}{T_b} \right] + 0.705 \right]$	<p>SOURCE: Through Petukhov (1970)</p>
<p>REYNOLDS: <math>13 &lt; Re &lt; 10^5</math> HORIZONTAL/VERTICAL: PRANDTL: GRASHOF: LENGTH/DIAMETER 80 - 100</p>	
<p>AUXILIARY INFORMATION <math>\frac{T_w}{T_b} = 1 - 6</math></p>	

## HEAT TRANSFER COEFFICIENTS IN TUBES

FLUID(S): Air, medium nitrogen METHOD:	AUTHOR(S): McEligot, D M McGee, P M Lepore, G J. Heat Transfer, 87, No. 1, (1965)
EQUATION PROPOSED:  $Nu_b = 0.023 Re_b^{0.8} Pr_b^{0.4} \left[ \frac{T_w}{T_b} \right]^{-0.5}$	SOURCE: Through Petukhov (1970)
REYNOLDS: PRANDTL: GRASHOF:	HORIZONTAL/VERTICAL: LENGTH/DIAMETER: 160
AUXILIARY INFORMATION	$\frac{T_w}{T_b} = 1.1 \rightarrow 2.5$

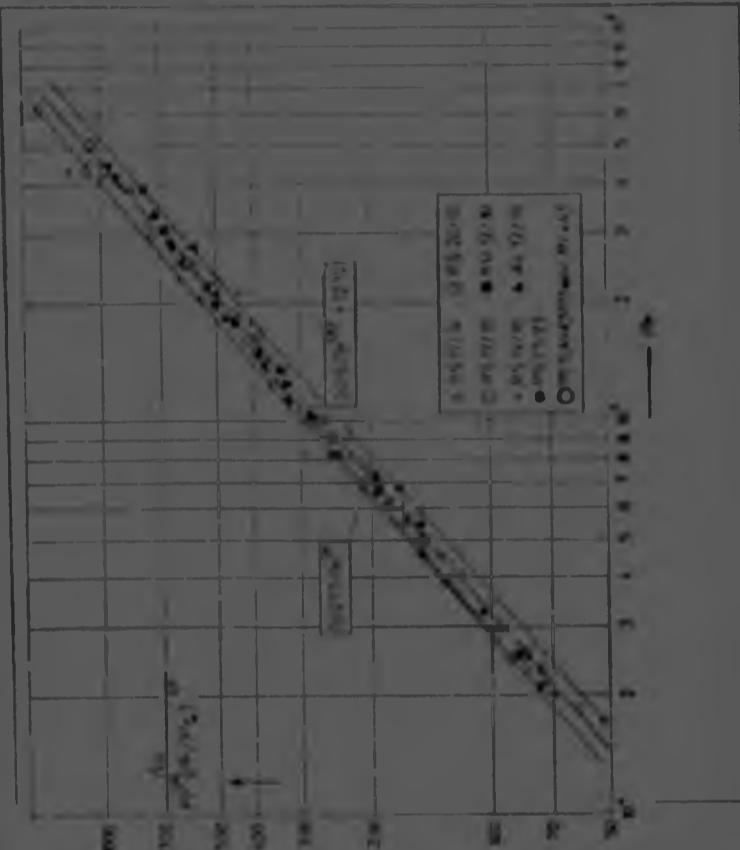
## HEAT TRANSFER COEFFICIENTS IN TUBES

FLUID(S): Air METHOD:	AUTHOR(S): Volkov, P M Lemov, A V Heat Transfer and Hydrodynamics in elements of power equipment Tr. TskTI 23, (1966)
EQUATION PROPOSED:  $Nu_x = 0.0103 Re_x^{0.8} Pr_x^{0.4} \left[ \frac{T_w}{T_b} \right]^{-0.11}$ $\frac{x}{D} > 100$	SOURCE: Through Petukhov (1970)
REYNOLDS: $14 + 400 \times 10^3$ PRANDTL: GRASHOF:	HORIZONTAL/VERTICAL: LENGTH/DIAMETER: 48 + 370
AUXILIARY INFORMATION	$\frac{T_w}{T_b} = 1.1 - 2.1$

## HEAT TRANSFER COEFFICIENTS IN TUBES

<b>FLUID(S)</b> Air <b>METHOD</b> Electrical heating	<b>AUTHOR(S)</b> Mori, Y. Furugami, K. Tokuda, S. Nakamura, M.
<b>EQUATION PROPOSED:</b> $Nu_s = 0.61 (RePr)^{1/4} \left[ 1 + \frac{1.8}{(RePr)^{1/4}} \right]$ <p>For laminar flow</p> <p>In turbulent flow at <math>Pr = 0.72</math></p> <p><math>Nu = 0.0201 Re^{0.8}</math> in agreement with Colburn</p>	<b>SOURCE:</b> Int. J. Heat Mass. 9, 453 (1968)
<b>REYNOLDS:</b> 100 - 12000 <b>PRANDTL:</b> <b>GRASHOF:</b> <b>AUXILIARY INFORMATION</b>	<b>HORIZONTAL/VERTICAL:</b> Horizontal <b>LENGTH/DIAMETER:</b> 1.77 7 m entrance section.

HEAT TRANSFER COEFFICIENTS IN TUBES


<p>FLUID(S): METHOD: Constant heat flux.</p>	<p>AUTHOR(S): Yabovlev, V V Atomnaya Energiya B, (3) 250 (1960) and Kernenergie, 3, 1028 (1960)</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.0277 Re^{0.8} Pr^{0.38} \left[ \frac{Pr}{Pr_{in}} \right]^{0.11} = 0.0277 Re^{0.8}$	<p>SOURCE: Hüfchmidt et al (1966)</p> 
<p>REYNOLDS: Turbulent PRANDTL: GRASHOF:</p>	<p>HORIZONTAL/VERTICAL: Horizontal LENGTH/DIAMETER:</p>
<p>AUXILIARY INFORMATION</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES

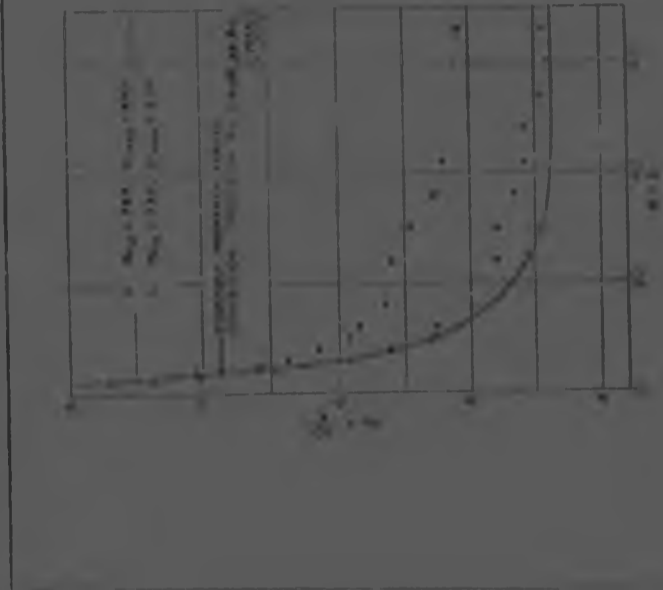
<p>FLUID(S): Water</p> <p>METHOD: Electrical heating. Results of others.</p>	<p>AUTHOR(S): Hufschmidt, W. Barrak, E. Rietold, W.</p>
<p>EQUATION PROPOSED:</p> <p>Using Prandtl and Sieder equations with a Pr correction factor</p> $Nu = \frac{\sqrt{Pr} \cdot Pr_0}{1.07 + 13.1 \sqrt{Pr} \left[ \frac{Pr_0}{Pr} - 1 \right]^{0.14}}$ <p>The use of the Pr<sub>0</sub> factor was first suggested by Yehoude (1966)</p>	<p>SOURCE: Int. J. Heat Mass. Tr. 9: 609 (1966)</p>
<p>REYNOLDS: 70000 - 600000</p> <p>PRANDTL: 2 - 5.5</p> <p>GRASHOF: LENGTH/DIAMETER</p> <p>AUXILIARY INFORMATION: A good review of different contributions is given.</p>	<p>HORIZONTAL/VERTICAL: Horizontal</p>



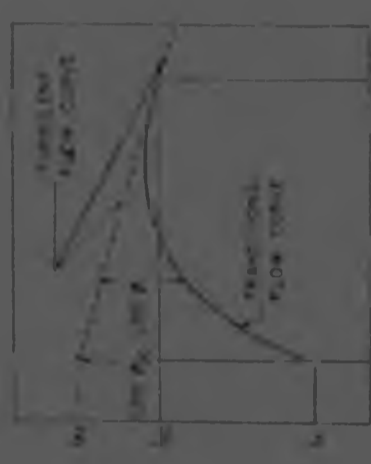
HEAT TRANSFER COEFFICIENTS IN TIRES

<p>FLUID(S): Air, nitrogen, helium.</p> <p>METHOD: Electrical heating.</p>	<p>AUTHOR(S): McElroy, D. M. Ormsand, L. W. Parkins, H. E.</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.21 Re^{0.8} Pr^{0.4} \mu^{0.14} (\mu_0/\mu)^{0.14} \quad \text{for } Re > 15,000$ <p>where <math>q'' = \frac{\text{heat flux}}{\text{periphery} \times \text{mass flow rate}}</math> (heat flux in Btu/ft<sup>2</sup> per centimeter)</p> <p><math>\mu_0 = 0.0004</math> (centipoise)</p> $0 < \mu/\mu_0 < 0.0004$ $Nu = Re_{\text{equivalent}}^{0.8} \left[ \frac{T_{\text{film}} - T_{\text{f}}}{T_{\text{f}} - T_{\text{w}}} \right]^{-0.5}$	<p>SOURCE: Trans. ASME, 82, 239 (1960)</p> 
<p>REYNOLDS: 1450 - 45000</p> <p>PRANDTL: 0.7 - 1.0</p> <p>GRASHOF: 10 - 1000</p> <p>AUXILIARY INFORMATION: Dittus and Boelter equation fits their data to <math>\pm 5\%</math>.</p>	<p>HORIZONTAL/VERTICAL: Vertical</p> <p>LENGTH/DIAMETER: <math>\geq 30</math></p>

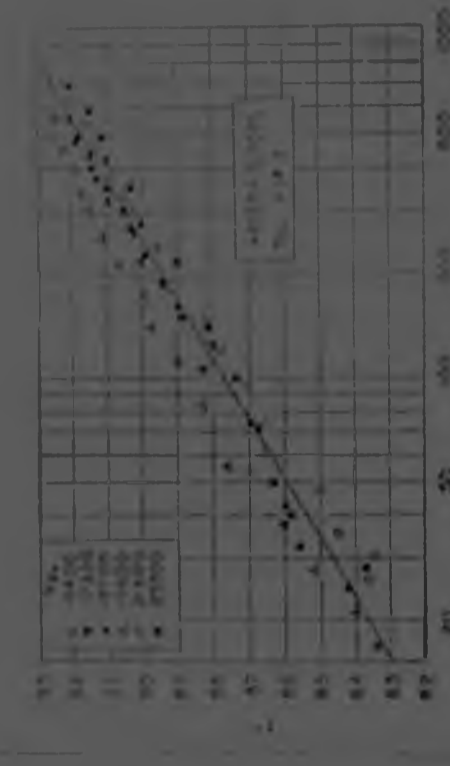
HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air METHOD: Electrical heating</p>	<p>AUTHOR(S): McCombs, S T Eckert, E R G</p>
<p>EQUATION PROPOSED</p> $Nu = A \left[ C_1 + B \left[ \frac{L}{d} \right]^{0.4} \right]^{1/2}$ <p>A and B are constants.</p>	<p>SOURCE: Trans. ASME C, 88, 441 (1966)</p>
<p>REYNOLDS: 100 - 1000 PRANDTL: — GRASHOF: 1 - 1000 AUXILIARY INFORMATION: Properties evaluated at local average temperature</p>	<p>HORIZONTAL/VERTICAL: Horizontal LENGTH/DIAMETER: — 80</p> 

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): METHOD: Other data</p>	<p>AUTHOR(S): Peterson, A W Christiansen, E B</p>
<p>EQUATION PROPOSED:</p> $St = St_{10000}^{0.75} \left[ \frac{Nu_{10000}}{Nu_{10000} - 10000} \right]^{0.25} \left[ \frac{Re}{10000} \right]^{\beta} \left[ \frac{10000}{2170} \right]^{\beta}$ <p>β is the slope of the y-factor vs Re curve in the turbulent region on a log-log plot</p> $\beta(Re) = 1.675 \log \left[ \frac{1}{1.131} \frac{Re - 710}{(Re - 1000)} \right]$	<p>SOURCE: AIChE J. 12, 2, 221 (1966)</p>  <p>β is the ratio of the gradient of the transition region of the actual heat transfer rate to the heat transfer rate that would be obtained if fully turbulent flow heat transfer mechanism existed.</p>
<p>REYNOLDS: Transitional PRANDTL: &gt; 2 GRASHOF:</p> <p>AUXILIARY INFORMATION: Data for air are not in good agreement.</p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER: 118 - 197</p>

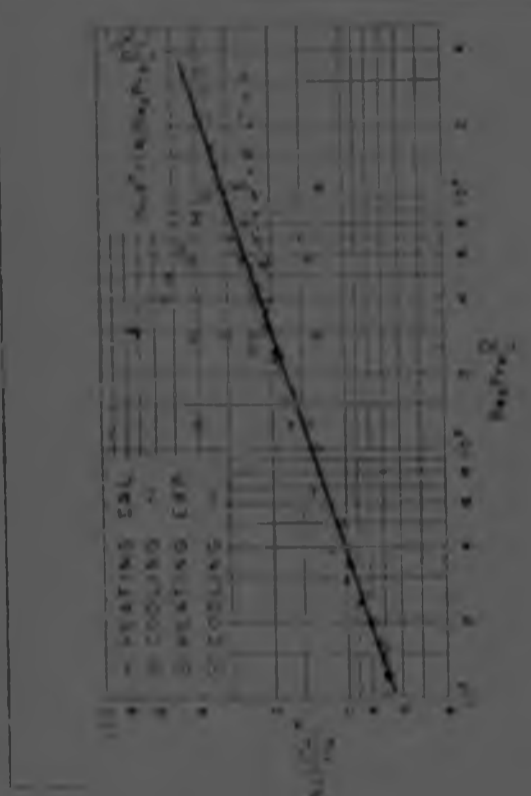
HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID: 51 - 58: polystyrene styrene glycol                  METHOD: Electrical heating</p>	<p>Author(s): Gowen R A                  Smith J A</p>
<p>EQUATION PROPOSED</p> $h_c = \frac{\sqrt{\frac{1}{2}}}{Nu \cdot \left[ \sqrt{Pr} + 0.4 \right]}$ $h_c = 5.17 \left[ \frac{0.55 + 1}{Pr} \right] + 0.55 + 5Pr$ <p><math>h_c</math> is the average for the universal temperature                  regime</p> $Nu = \frac{\sqrt{\frac{1}{2}}}{St}$	<p>SOURCE: Chem Eng Sci 37, 1151 (1962)</p> 
<p>REYNOLDS: 10 000 - 50 000                  PRAJDTL: 0.7 - 14.3                  GRASHOF:                  AUXILIARY INFORMATION: Film viscosity used for Pr.</p> $T_f = \frac{T_b + T_m}{2}$	<p>HORIZONTAL/VERTICAL: Vertical                  LENGTH/DIAMETER: 31</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

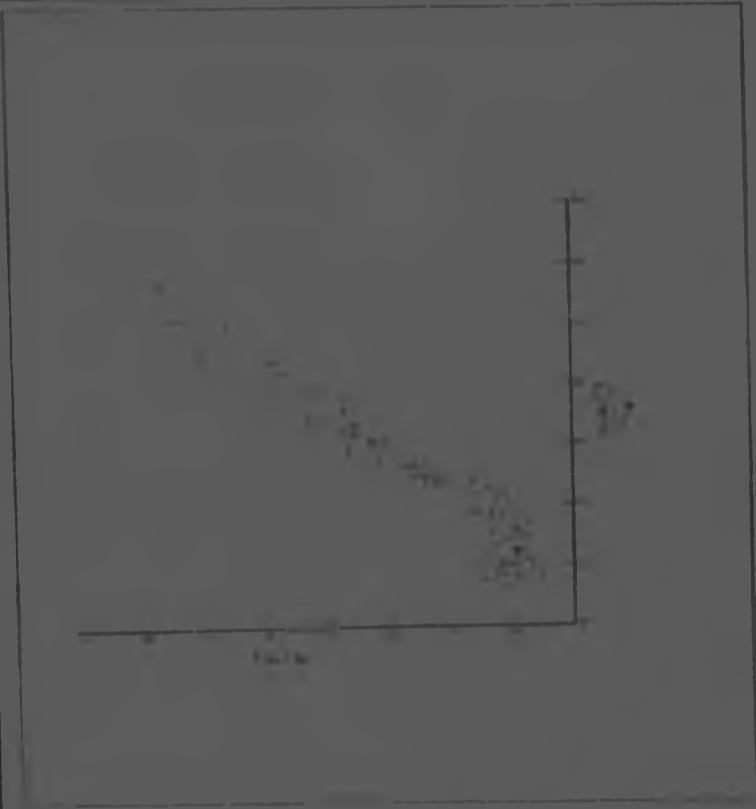
<p><b>FLUID IS:</b> Air <b>METHOD:</b> Uniform heating</p>	<p><b>AUTHOR(S):</b> Mori, Y Fujisawa, K</p>
<p><b>EQUATION PROPOSED:</b></p> $\frac{Nu}{Nu_0} = 1 + (0.026 - 0.00037 Pr + 0.3334 Pr^2) \left[ \frac{Re Pr}{1600} \right]^2$ <p>using the scales of their previous work.</p>	<p><b>SOURCE:</b> Int. J. Heat Mass <u>30</u>, 1801 (1967)</p>
<p><b>REYNOLDS:</b> Laminar <b>PRANDTL:</b> <b>GRASHOF:</b></p>	
<p><b>AUXILIARY INFORMATION:</b></p>	<p><b>HORIZONTAL VERTICAL:</b> Horizontal <b>LENGTH/DIAMETER:</b> LENGTH/DIAMETER</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): OIL          METHOD: Water in double heating.</p>	<p>AUTHOR(S): Tait, F L</p>
<p>EQUATION PROPOSED:</p> $Nu_b \left[ \frac{D}{L} \right]^m = C_1 \left[ \frac{\rho_b \mu_b}{\rho_f \mu_f} \right]^n$ <p><math>m = 0.05</math> heating  <math>m = 1/3</math> cooling  <math>C_1 = 1.4</math>  <math>n = 1/3</math></p>	<p>SOURCE: Trans ASME C. 385 (1908)</p> 
<p>REYNOLDS: Laminar          PRANDTL:          GRASHOF:          AUXILIARY INFORMATION: Properties based on bulk average temperature.</p>	<p>HORIZONTAL/VERTICAL: Vertical          LENGTH/DIAMETER: 240</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Water</p> <p>METHOD: Electrical Heating</p>	<p>AUTHOR(S): Shannon, R. E. DeJarni, C. A.</p>
<p>EQUATION PROPOSED: In practical form. It is suggested that actual coefficient be multiplied by <math>(\mu/\mu_s)^{0.14}</math> &amp; <math>Pr_s &lt; 2</math>, where <math>\mu_s</math> is the solution from Sieder et al. (1968)</p>	<p>SOURCE: Trans. ASME C 353 (1968)</p>
<p>REYNOLDS: 120 - 2300</p> <p>PRANDTL: 2.5 - 10<sup>6</sup></p> <p>AUXILIARY INFORMATION: Average properties were used.</p>	<p>HORIZONTAL/VERTICAL: Horizontal</p> <p>LENGTH/DIAMETER: 700</p>

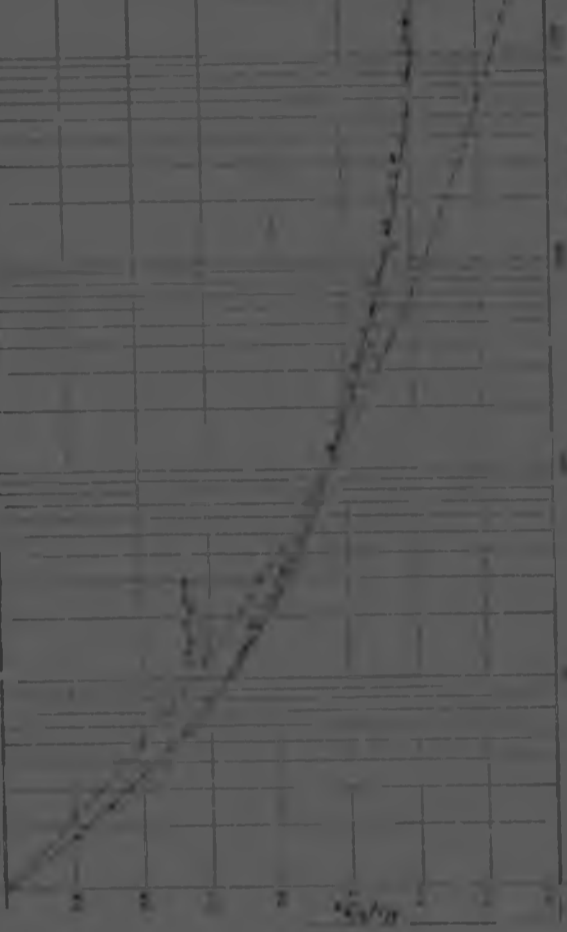


HEAT TRANSFER COEFFICIENTS IN TUBES

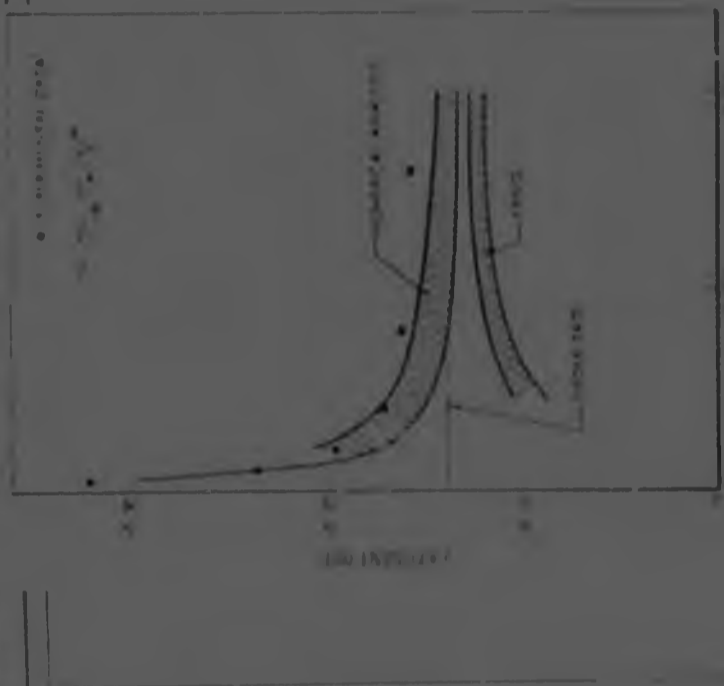
<p>FLUID(S) METHOD</p>	<p>AUTHOR(S) : Eversett, M.</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.023 Re^{0.8} Pr^{0.4} \left[ \frac{\mu}{\mu_w} \right]^{0.14}$ $x = 0.495 - 0.0225 \ln Re$ $x = \left[ \frac{Re}{870,000} \right]^{0.84} \quad Re < 60,000$ $x = 0.11 \quad Re > 62,500$ <p>These are modified forms of the ESDU (1967)</p>	<p>SOURCE : Chem. Eng., CE159, Sept (1969)</p>
<p>REYNOLDS: <math>&gt; 4 \times 10^3</math> PRANDTL: GRAEFHOF: AUXILIARY INFORMATION</p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER: Correlation is to within 10.2%</p>



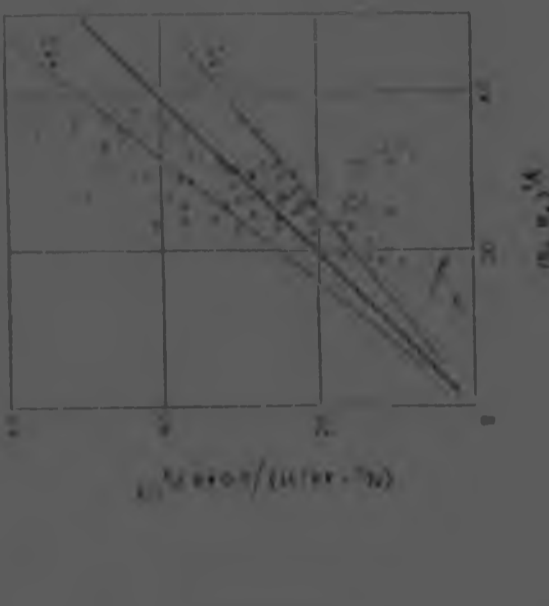
HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Oil</p> <p>METHOD:</p>	<p>AUTHOR(S): Ma<sup>1</sup>, A Ormi, W</p>
<p>EQUATION PROPOSED: Suggested a new correction factor for viscosity and presented this graphically.</p> <p>Haider, H (Met 3, (B) 356 (1968) and 480 L) proved that the equation</p> $\frac{h_{oil}}{h_{air}} = 0.845 \left[ \frac{\mu_{air}}{\mu_{oil}} \right]^{-0.2} + 0.255$ <p>covered the data.</p>	<p>SOURCE: Maschinenbauk 2(4) 141 (1968)</p> 
<p>REYNOLDS: 4000 - 11000</p> <p>PRANDTL:</p> <p>GRASSHOFF:</p> <p>AUXILIARY INFORMATION</p>	<p>HORIZONTAL/VERTICAL: Horizontal</p> <p>LENGTH/DIAMETER:</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Ethylene glycol</p> <p>METHOD: Electrical heating</p>	<p>AUTHOR(S): Shapiro, R L Dapkin, C A</p>
<p>EQUATION PROPOSED:</p> $Nu = Nu_0 \left[ \frac{L}{D} \right]^m$	<p>SOURCE: Trans. ASME C (1), 75 (1953)</p>
<p>REYNOLDS: 5 - 300</p> <p>PRANDTL: 26 - 500</p> <p>GRASHOF: - 2800</p>	
<p>AUXILIARY INFORMATION</p> <p>HORIZONTAL/VERTICAL: Horizontal</p> <p>LENGTH/DIAMETER: -</p>	<p>Fig. 7. Experiments in series I.</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Water METHOD: Electrical heating</p>	<p>AUTHOR(S): Kutzner, A. Hauptmann, E. S. Lohdal, M.</p>
<p>EQUATION PROPOSED:</p> $Nu = \frac{48}{11} + 0.048 Re^{1/4} (Pr/0.6)^{1/4}$ <p>covers 5% of the data at 10%</p>	<p>SOURCE: Solar Energy, 12, 439 (1968)</p>
<p>REYNOLDS: 100 - 2000 PRANDTL: 4 - 9 GRASHOF: 300 - 30000 ALXILIARY INFCRMATION: 1 ft entrance section. Bulk temperatures used for properties</p>	

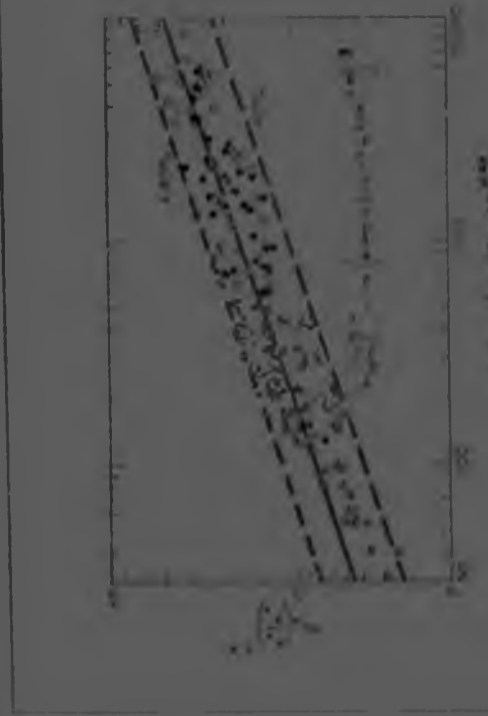
## HEAT TRANSFER COEFFICIENTS 4 TUBES

FLUID(S): METHOD:	AUTHOR(S): Karamanchikov Protomyshev Item Type: 4 (3) (1963) 9 (6) (1961)
EQUATION PROPOSED: $Nu_{\text{tr}} = Nu_{\text{tr}} \left[ \frac{\mu}{\mu_s} \right]^m \left[ \frac{C_p}{C_{p_s}} \right]^n$	SOURCE: Refukov. Design and Theory Series book. 1974
REYNOLDS: PRANDTL: GRASHOF: AUXILIARY INFORMATION	HORIZONTAL/VERTICAL: LENGTH/DIAMETER:

HEAT TRANSFER COEFFICIENTS IN TUBES

<p><b>FLUID(S):</b> Temperature dependent physical properties</p> <p><b>METHOD:</b> Using others' results and theoretical study</p>	<p><b>AUTHOR(S):</b> Grigoric, H</p>	
<p><b>EQUATION PROPOSED:</b></p> $K_a = \frac{K_w K_{pl}}{K_s}$ $K_w = \left[ \frac{k_w}{k_s} \right]^{0.15} \Delta T_w^{-0.25}$ $K_{pl} = \left[ \frac{Pr_w}{Pr_s} \right]^{0.25} \text{GRIGORIC EXP. 1}$ $K_s = \left[ \frac{r_{in}}{r_{out}} \right]^{0.34} \text{GRIGORIC EXP. 2}$ $X_b = \frac{r_{in} - r_{out}}{r_{in} + r_{out}} - 0$ $A = 0.44 \exp \left[ -0.185 \left( \frac{L}{D} \right)^{1.15} \right]$		<p><b>SOURCE:</b> Wärme und Stoffübertragung, 3, 26 (1970)</p>
<p><b>REYNOLDS:</b> HORIZONTAL/VERTICAL</p> <p><b>PRANDTL:</b> LENGTH/DIAMETER:</p> <p><b>GRASHOF:</b> AUXILIARY INFORMATION <math>w =</math> wall temperature</p> <p><math>\bar{t}_m =</math> mean temperature, <math>\bar{t}_s =</math> at <math>T_w = (T_{in} + T_{out})/2</math></p>		

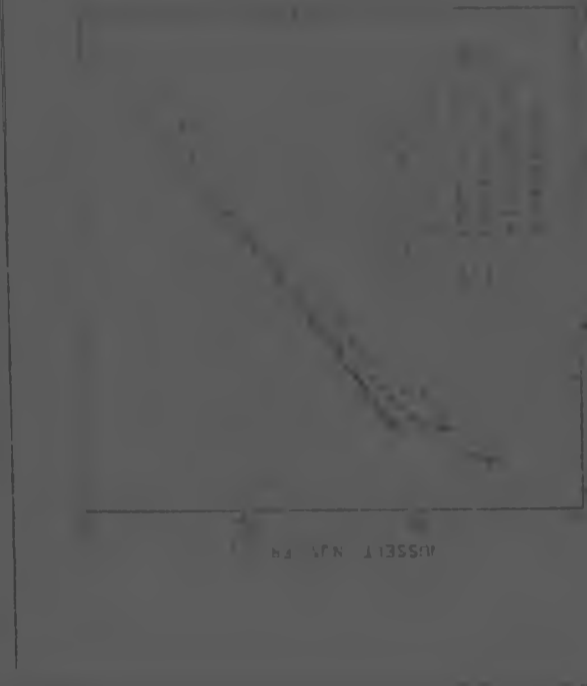
HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Water, ethanol, glycerol, water          METHOD: Cooling with refrigerant</p>	<p>AUTHOR(S): Dwyer, C A          August, S F</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.75 \left[ D_1 + 0.12 \left[ (Gr)^{1/3} Pr^{0.25} \right]^{0.88} \right]^{1/3} \left[ \frac{\mu_c}{\mu_w} \right]^{0.14}$ <p>covers most data in entire 40%</p>	<p>SOURCE: Trans. ASME C, 380 (1971)</p> 
<p>REYNOLDS: Laminar          PRANDTL:           GRASHOF:           AUXILIARY INFORMATION: Properties at bulk average temperatures</p>	<p>HORIZONTAL/VERTICAL: Horizontal          LENGTH/DIAMETER: 28.4</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

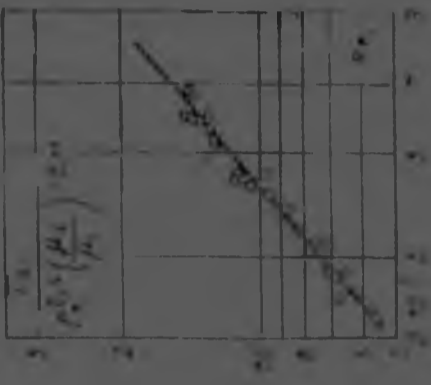
<p><b>FLUID(S):</b> Water</p> <p><b>METHOD:</b> Steam heating</p>	<p><b>AUTHOR(S):</b> Herbert, L.S. Sterne, U.J.</p>
<p><b>EQUATION PROPOSED:</b></p> <p><math>Nu = 0.56 Re^{0.4} Pr^{0.4}</math> Downward flow <math>Re &lt; 116000</math></p> <p><math>Nu = 0.5 \times 10^{-3} (G \mu)^{1/4}</math> Upward flow  <math>Re = 4500 - 15000</math>  <math>G = 3 \times 10^6 - 30 \times 10^6</math></p> <p><math>Nu = 0.525 Re^{0.4} Pr^{0.4} (G \mu)^{0.025}</math> (a)</p>	<p><b>SOURCE:</b> Chem. Eng. J. 4, 46 (1972)</p>
<p><b>REYNOLDS:</b> 5000 - 11000</p> <p><b>PRANDTL:</b></p> <p><b>GRASHOF:</b></p> <p><b>AUXILIARY INFORMATION:</b> Agreement with position (a) was not good</p>	<p><b>HORIZONTAL/VERTICAL:</b> Vertical</p> <p><b>LENGTH/DIAMETER:</b> 80</p> <p>Experimental results compared with equation (a)</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

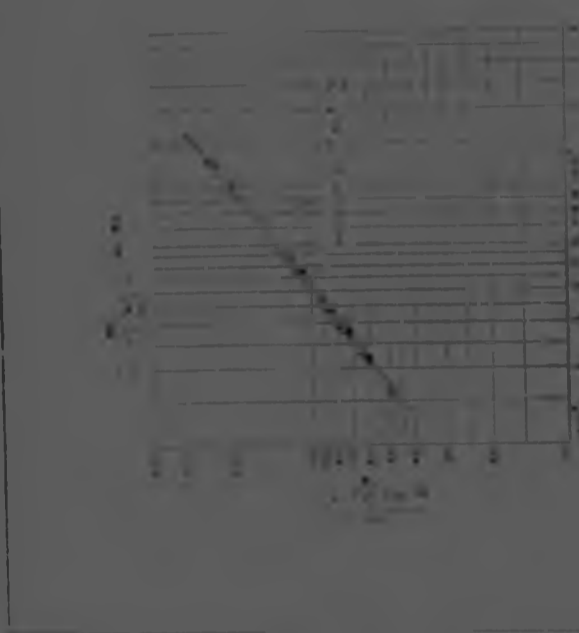
<p>FLUIDS: Data of Lewis (1961)</p> <p>METHOD: Solution of the form <math>\frac{d\theta}{dx} = -\frac{h\theta}{x}</math> in the fully developed region</p>	<p>AUTHOR(S): Lewis, R. D. Thomey, J. C.</p>
<p>EQUATION PROPOSED</p> $Nu_x = \frac{T_b - T_c}{T_c - T_c} \left[ \frac{U_c}{U_c} \right] \frac{1}{2} Re_x^{1/2} \left[ \frac{1 + \sqrt{1 + \left( \frac{16}{N} \right)}}{2} \right]$ $h = \left[ \frac{16}{N} \right] \left[ \frac{U_c \sqrt{1 + \left( \frac{16}{N} \right)}}{2} \right]$ <p><math>U_c</math> = entry velocity at the first entrance of conical</p>	<p>SOURCE: Trans. ASME E, 84M (1962)</p>  <p>Fig. 1. Comparison of predicted for fully developed flow in tubes with experimental data, after Lewis.</p>
<p>REYNOLDS: Turbulent - Transitional</p> <p>PRANDTL:</p> <p>GRASHOF:</p> <p>AUXILIARY INFORMATION</p> <p>HORIZONTAL/VERTICAL:</p> <p>LENGTH/DIAMETER:</p>	



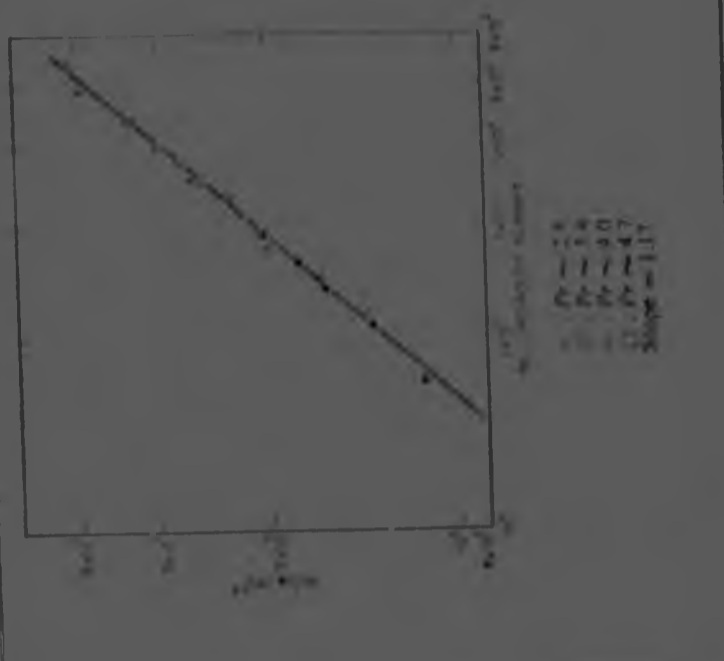
HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Transformer oil, fuel oil</p> <p>METHOD: Electrical heating</p>	<p>AUTHOR(S): Kuznetsov, V. V.</p> <p>SOURCE: Teploenergetika, 19 (6) 84 (1972)</p>
<p>EQUATION PROPOSED:</p> $Nu = 1.23 \left[ Pr \frac{d}{L} \right]^{0.4} \left[ \frac{\mu}{\mu_w} \right]^{0.14}$ <p>Re = 400 - 1500 Pr = 1.70 - 6.03</p> $Nu = 0.013 Re^{0.8} Pr^{0.4} \left[ \frac{\mu}{\mu_w} \right]^{0.14}$ <p>Re = 2000 - 6000 Pr = 1.70 - 2.00</p>	
<p>REYNOLDS: HORIZONTAL/VERTICAL: Horizontal</p> <p>PRANDTL: LENGTH/DIAMETER:</p> <p>GRASHOF: Properties at bulk temperature</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air</p> <p>METHOD: Uniform surface temperature</p>	<p>AUTHOR(S): Zucchetto, J Thomson, B S</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.023 Re^{0.8} Pr^{0.4} \left[ \frac{T_s}{T_b} \right]^{0.14}$	<p>SOURCE: Trans ASME C, 134 (1972)</p>
<p>REYNOLDS: Turbulent</p> <p>PRANDTL: _____</p> <p>GRASHOF: _____</p> <p>AUXILIARY INFORMATION: Properties evaluated at <math>T_b = T_s + 0.75 \Delta T</math></p>	<p>HORIZONTAL/VERTICAL: Horizontal</p> <p>LENGTH/DIAMETER: _____</p> 

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Water</p> <p>METHOD:</p>	<p>AUTHOR(S): Walker, R A Rort, T R</p>																				
<p>EQUATION PROPOSED</p> $Nu = 0.014 \sqrt{\frac{L}{D}} Re^{0.714} Pr^{0.41}$ <p>Proposed at average temperature</p>	<p>SOURCE: Chem. Eng. 151 (1973)</p>																				
<p>REYNOLDS: 14000 - 50000</p> <p>PRANDTL: 2.0 - 5.5</p> <p>GRASHOF:</p>	<p>HORIZONTAL/VERTICAL:</p> <p>LENGTH/DIAMETER:</p>																				
<p>AUXILIARY INFORMATION</p> <p>The authors were looking at the effect of wave grain roughness.</p>	 <table border="1" data-bbox="1254 850 1391 1015"> <tr> <td>log(Re)</td> <td>4.15</td> <td>4.25</td> <td>4.35</td> <td>4.45</td> <td>4.55</td> <td>4.65</td> <td>4.75</td> <td>4.85</td> <td>4.95</td> </tr> <tr> <td>log(Nu)</td> <td>1.15</td> <td>1.25</td> <td>1.35</td> <td>1.45</td> <td>1.55</td> <td>1.65</td> <td>1.75</td> <td>1.85</td> <td>1.95</td> </tr> </table> <p>Slope = 1.17</p>	log(Re)	4.15	4.25	4.35	4.45	4.55	4.65	4.75	4.85	4.95	log(Nu)	1.15	1.25	1.35	1.45	1.55	1.65	1.75	1.85	1.95
log(Re)	4.15	4.25	4.35	4.45	4.55	4.65	4.75	4.85	4.95												
log(Nu)	1.15	1.25	1.35	1.45	1.55	1.65	1.75	1.85	1.95												

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): METHOD:</p>	<p>AUTHOR(S): Hauser, H</p>
<p>EQUATION PROPOSED:</p> $Nu = 0.023 [Re^{0.8} - 230] \left[ 1.8 Pr^{0.3} - 0.6 \right] \left[ 1 + \left[ \frac{d}{L} \right]^{0.4} \right] \left[ \frac{\mu}{\mu_w} \right]^{0.14}$	<p>SOURCE: Wärme und Stoffübertragung, 1, 222 (1974)</p>
<p>REYNOLDS: Turbulent PRANDTL: GRASHOF: AUXILIARY INFORMATION: Altered form of the equation presented in 1959.</p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER:</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): METHOD: Ormsby results</p>	<p>AUTHOR(S): Gnielinski, V</p>
<p>EQUATION PROPOSED</p> $Nu = 0.0234 \left[ Re^{0.8} - 100 \right] Pr^{0.4} \left[ 1 + \left( \frac{d}{L} \right)^{1.1} \right] \left[ \frac{T_w}{T_b} \right]^{0.4}$ <p>0.6 &lt; Pr &lt; 500 2300 &lt; Re &lt; 10<sup>6</sup></p> $Nu = 0.013 \left[ Re^{0.8} - 200 \right] Pr^{0.4} \left[ 1 + \left( \frac{d}{L} \right)^{1.1} \right] \left[ \frac{Pr}{Pr_w} \right]^{0.11}$ <p>1.5 &lt; Pr &lt; 1000</p>	<p>SOURCE: Forsch. Ing.-Wes., 41, (1) 8 (1975)</p>
<p>REYNOLDS: Turbulent PRANDTL: GRASHOF: AUXILIARY INFORMATION</p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER:</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

FLUID(S):

METHOD: Using others' results.

EQUATION PROPOSED:

$$Nu_b = 5 + 0.015 Re_b^{0.8} Pr_b^{0.4} \dots (a)$$

$$0.1 < Pr_b < 10^6$$

$$10^4 < Re_b < 10^6$$

$$h = \frac{0.88 + 0.34}{1.8 + Pr_b}$$

$$h = 1.13 + 0.56 Pr_b^{0.07} Re_b^{0.4}$$

$$Nu_b = \frac{T_w - T_b}{T_w - T_m} > 2$$

$$Nu_b = 5 + 1.2 Re_b^{0.4} \left[ Pr_b + 0.29 \left[ \frac{T_w}{T_b} \right] \right] \dots (b)$$

$$n = -\log_{10} \left[ \frac{T_w - T_b}{T_w - T_m} \right] + 0.3$$

REYNOLDS: Turbulent

PRANDTL:

GRASHOF:

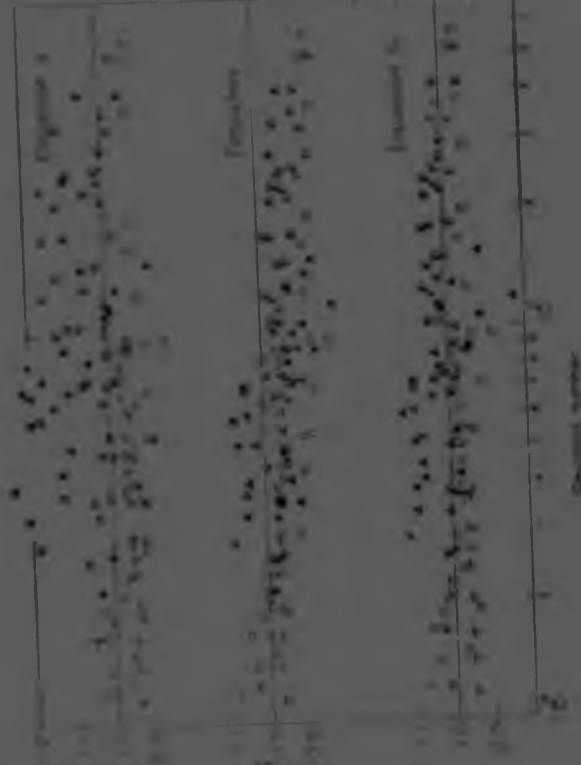
AUXILIARY INFORMATION

Petukhov, Adv. Heat Trans. 5, 503 (1970)


AUTHOR(S): Sleicher, C. A.

Reese, M. W.

SOURCE: Int. J. Heat Mass. TR. 677 (1975)



HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air          METHOD: Constant wall temperature</p>	<p>AUTHOR(S): Petukhov, Ya. Ya          SOURCE: Heat Trans. Sov. Res. 7 (6) 100 (1978)</p>
<p>EQUATION PROPOSED</p> $Nu_{D,i} = C Re_m^m Pr^k$ <p>where <math>C</math> and <math>k</math> are given in the graph          Conditions <math>k</math> is within 2%</p>	
 <p><math>C</math> and <math>k</math> in formula are functions of <math>Re/D</math>.</p>	
<p>REYNOLDS: Laminar/turbulent    HORIZONTAL/VERTICAL: Horizontal          PRANDTL:    LENGTH/DIAMETER: 77-154/231          GRASHOF:    AUXILIARY INFORMATION</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES

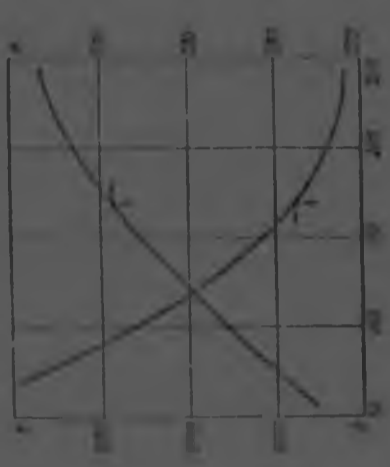
<p>FLUID(S): Air          METHOD: Constant wall temperature</p>	<p>AUTHOR(S): Pecherov, Yu. Ya          SOURCE: Heat Trans. Sov. Res. 7, 161 60 (1975)</p>
<p>EQUATION PROPOSED</p> $Nu_{D,air} = C Re^{0.4} Pr^{0.4}$ <p>where <math>C</math> and <math>n</math> are given in the graph</p> <p>Comparison of (8) with (7)</p>	
<p>REYNOLDS: Lower turbulent      HORIZONTAL, VERTICAL: Horizontal          PRANDTL:                              LENGTH/DIAMETER: 77, 154, 231          GRASHOF:                              AUXILIARY INFORMATION</p>	



c and n in formula as func. of Re/D.



HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): Air</p> <p>METHOD: Constant wall temperature</p>	<p>AUTHOR(S): Pechurinov, Yu Ya</p> <p>SOURCE: Heat Trans. Sov. Res., 7, (6) 60 (1976)</p>
<p>EQUATION PROPOSED</p> $Nu_{wall} = C Re_{max}^m$ <p>where C and m are given in the table.</p> <p>Conditions: <math>Re</math> within <math>Re_c</math></p>	 <p><math>\nu</math> and <math>\alpha</math> in formula as function of <math>x/D</math>.</p>
<p>REYNOLDS: Laminar-turbulent</p> <p>PRANDTL:</p> <p>GRASHOF: LENGTH/DIAMETER: 77: 154, 231</p> <p>AUXILIARY INFORMATION</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): METHOD: Others' results</p>	<p>AUTHOR(S): Graessner, R</p>
<p>EQUATION PROPOSED:</p> <p><math>\frac{L}{D} &gt; 70</math>, boiling of liquid only</p> $\frac{Nu_b}{Nu_s} = \left[ \frac{Pr_b}{Pr_s} \right]^c$ $c = \frac{0.76415 - \lambda_b Pr_b^{0.2} + 0.0157}{Pr_b^{0.33} Re_b^{0.1} (\lambda - \lambda_b)}$ $\lambda_b = \frac{Pr_s - Pr_b}{Pr_b - Pr_s} - \lambda_s$	<p>SOURCE: Wärme und Stoffübertragung, 9, 63 (1976)</p>
<p>REYNOLDS: PRANDTL: GRASHOF: AUXILIARY INFORMATION <math>\lambda_s = \frac{[T_b + T_w]}{2}</math></p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER:</p>

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUID(S): All</p> <p>METHOD: Generalized correlation.</p>	<p>AUTHOR(S): Churchill, S W</p>
<p>EQUATION PROPOSED:</p> $[Nu]^{1/4} = [Nu_L]^{0.4} \left[ \frac{(2300 - Pr)}{2300} \right]^{0.1} \left[ \frac{0.076 Re \sqrt{Pr}}{(1 + Pr)^{0.4}} \right]^{0.4}$ <p> <math>Nu_{L,c} = 0</math> - turbulent regime at <math>Re = 2100</math>  <math>Nu_{L,c} = 0</math> - in <math>Pr = 0</math> and <math>Re = 2100</math>  <math>Nu_L = 0</math> - laminar regime         </p>	<p>SOURCE: Ind. Eng. Chem. Fund. 16, (1) (1977) (1977)</p>
<p>REYNOLDS: HORIZONTAL/VERTICAL:</p> <p>PRANDTL: LENGTH/DIAMETER:</p> <p>GRASHOF:</p> <p>AUXILIARY INFORMATION</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES

<p>FLUIDS: Fluids with temperature dependent viscosity</p> <p>METHOD: Other: none</p>	<p>AUTHOR(S): HARRIS, J. T. SOMMER, R. C.</p>
<p>EQUATION PROPOSED:</p> $\frac{Nu}{\left(\frac{Re}{2}\right)^{0.4}} = \left[1 - \frac{1}{A}\right] + \frac{Nu_{\text{smooth}}}{\left(\frac{Re}{2}\right)^{0.4}}$ $Nu = \int_{x_0}^x \frac{2}{r} dx$ <p><math>T_w</math> - bulk condition</p> <p>For heating <math>A = 2.4</math>; For cooling <math>A = 4.05</math></p>	<p>SOURCE: Trans. ASME E, 100, 224 (1978)</p>
<p>REYNOLDS: Turbulent</p> <p>FRANZTL:</p> <p>GRASHOF:</p>	<p>HORIZONTAL/VERTICAL: LENGTH/DIAMETER:</p>
<p>AUXILIARY INFORMATION</p>	

HEAT TRANSFER COEFFICIENTS IN TUBES









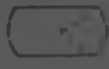












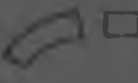




FLUID(S):	Folley, G T
METHOD:	
EQUATION PROPOSED	Chem. Eng. 233, (April 1979)
Suggests the use of the ESDU correlation (1967).	<p>Legend:          * Numerical          x High speed flow          - Other Numerical          --- ESDU</p>
REYNOLDS:	Turbulent
PRANDTL:	HORIZONTAL/VERTICAL:
GRAVITY:	LENGTH/DIAMETER:
AUXILIARY INFORMATION	









SUMMARY OF ANALYTICAL SOLUTIONS FOR TURBULENT FLOW.

FROM McELGOT, SMITH AND BANKSTON (1970)

"Title"/Reference	Basic Representation	n	y <sup>+</sup>	Range	PREDICTION FOR CONSTANT PROPERTIES, Re = 10 <sup>5</sup>	
					Nu/Nu <sub>DB</sub>	f/DKM
Reichardt "local"	$\frac{1}{\nu} = \frac{6}{y^+} \left[ 1 - \frac{y^+}{r_w^+} \tanh \frac{y^+}{r_w^+} \right] \left[ 1 + 2 \left( 1 - \frac{y^+}{r_w^+} \right)^2 \right]$	0.4	11.0	Wall q	0.945	0.989
Reichardt modified "wall"	Same except replaced y <sup>+</sup> by y <sub>w</sub> <sup>+</sup> in argument of tanh	Same	Same	Same	Same	Same
Reichardt "wall"	$\frac{1}{\nu} = \frac{6}{y_w^+} \left[ 1 - \frac{y^+}{y_w^+} \tanh \frac{y^+}{y_w^+} \right] \left[ 1 + 2 \left( 1 - \frac{y^+}{y_w^+} \right)^2 \right]$	Same	Same	Same	Same	Same
Three layer modified Martinelli	Smith, S B. MSE Report Univ Arizona, 1967.	0.4	-	-	0.964	0.993
Sparrow, Hallman and Siegel "local"	$\frac{1}{\nu} = n^2 u_y \left[ 1 - e^{-\frac{n^2 u_y}{\nu}} \right]$ n = 0.124 $\frac{1}{\nu} = x y^+ \left[ 1 - \frac{y^+}{r_w^+} \right] - 1$	0.36	26.0	y <sub>w</sub> <sup>+</sup> < y <sub>s</sub> <sup>+</sup> y <sub>w</sub> <sup>+</sup> > y <sub>s</sub> <sup>+</sup>	0.943	0.984
Sparrow, Hallman and Siegel "wall"	Viscous sublayer same (except range); Core $\frac{1}{\nu} = x y_w^+ \left[ 1 - \frac{y^+}{y_w^+} \right] - 1$	0.36	26.0	y <sub>w</sub> <sup>+</sup> < y <sub>s</sub> <sup>+</sup> y <sub>w</sub> <sup>+</sup> > y <sub>s</sub> <sup>+</sup>	Same	Same
Van Driest "local"	$\frac{1}{\nu} = xy \left[ 1 - \exp \left( -\frac{y^+}{y_w^+} \right) \right]$	0.4	26.0	Wall q	1.03	1.06
Van Driest "wall"	Same except replaced y <sub>w</sub> <sup>+</sup> by y <sub>w</sub> <sup>+</sup> in argument of exp	Same	Same	Same	Same	Same
Kendall et al "local"	$\frac{d^+}{dy^+} = \frac{xy^+}{y^+} \sqrt{\frac{y^+}{y_w^+}}$	0.42	11.83	Wall q	1.03	1.06
Kendall et al "wall"	$\frac{d^+}{dy^+} = \frac{xy_w^+}{y^+} \sqrt{\frac{y^+}{y_w^+}}$	Same	Same	Same	Same	Same
Kendall w Clauser "wall"	Wall region same form as Kendall et al. "wall", Core (or wake) $\epsilon^+ = 0.018 u_c^+$	0.44	11.83	$\epsilon^+ < u_c^+$ $u_c^+$	0.939	1.00

45 SUMMARY OF ANALYTICAL SOLUTIONS FOR LAMINAR FLOW  
IN VARIOUS GEOMETRIES [SHAH AND LONDON (1971)]

GEOMETRY	GEOMETRY
 Equilateral triangular duct with rounded corners	 Longitudinal flow between cylinders, square array
 Sine ducts	 Pascal's limaçon
 Circular sector ducts	 Curvilinear polygonal ducts
 Circular segment ducts	 Ovaloid ducts
 Flat sided circular duct	 Confocal elliptical ducts
 n sided cusped ducts	 Circular duct with rounded corner square cores
 Moon shaped ducts	 Circular duct with elliptical cores
 Cardioid duct	 Elliptical ducts with circular cores
 Eccentric annular ducts	 Internally finned tube
 Annular sector ducts	 Curved circular ducts
 Regular polygonal ducts with central circular cores	 Curved rectangular ducts
 Circular duct with central regular polygonal cores	 Curved elliptical ducts
 Longitudinal flow between cylinders, triangular array	 Curved concentric annular ducts

GEOMETRY	1Re	$Nu_{HI}$	$Nu_{HT}$
 Straight circular duct	11	48/11	3,657
 Parallel plates	11	140/17	7,541
 Rectangular ducts $2b/2a \sim 0 - 1$			
 Isosceles triangular ducts $2\phi \sim 0 - 180^\circ$			
 Right triangular duct $\phi \sim 0 - 180^\circ$			
 n-sided regular polygonal ducts $n \sim 3 - \infty$			
 Elliptical ducts $2b/2a \sim 0 - 1$			
 Concentric annular ducts $r_2/r_1 \sim 0 - 1$			



NOMENCLATURE

A	Transfer area
$a, b, c$ subscripted	Empirical coefficient
$C_p$	Specific heat (J/kgK)
$C_f$	Coefficient of friction $\frac{1}{d^2 u^2}$
$C_h$	Coefficient of heat transfer $\frac{q}{C_p u \theta}$
$d, D$	Inside diameter of tube (m)
$f$	(Fanning) friction factor $\frac{2R}{\rho u^2}$
$g$	Gravitational acceleration (m/s <sup>2</sup> )
$G$	Mass velocity $\frac{4m}{\pi d^2}$ (kg/sm <sup>2</sup> )
$Gr$	Grashof number $\frac{g \rho^2 \Delta T d^3}{\mu^2}$
$Gz$	Graetz number $\frac{m C_p}{k L} = \frac{\pi}{4} \frac{Re_d Pr_d}{L}$
$h$	Heat transfer coefficient (W/m <sup>2</sup> K)
$j$	Colburn's factor $j_h$
$k$	Thermal conductivity (W/mK)

$L$	Characteristic length (m)
$L$	Tube length (m)
$\ell$	Prandtl's mixing length
$m$	Mass flow rate (kg/s) $\frac{\pi d^2}{4} \rho u$
$Nu$	Nusselt number $\frac{hd}{k}$
$p$	Pressure (N/m <sup>2</sup> )
$Pe$	Peclet number $\frac{\rho u d C_p}{k} = Re_d Pr$
$Pr$	Prandtl number $\frac{C_p \mu}{k}$
$q$	Heat flux (W/m <sup>2</sup> )
$Q$	Heat flow rate (W)
$Ra$	Raleigh number $Gr Pr$
$Re$	Reynolds number $\frac{\rho u d}{\mu}$
$r$	Radius (m)
$R$	Resistance to flow
$S$	Cross sectional area (m <sup>2</sup> )
$St$	Stanton number $\frac{h}{\rho u C_p} = \frac{Nu}{Re_d Pr}$
$T$	Temperature (K)
$t$	Time (s)
$u$	Velocity $x$ (m/s)
$U$	Overall heat transfer coefficient (W/m <sup>2</sup> K)

Velocity $y$	(m/s)
Velocity $z$	(m/s)
Direction along axis of pipe	(m)
Direction along radius	(m)
Ratio of viscosity to the viscosity of water at 68 °F	
Thermal diffusivity	
$\frac{k}{\rho C_p}$	(m <sup>2</sup> /s)
Coefficient of volume expansion	
$(\frac{1}{K})$	
Dynamic viscosity	
$[\frac{kg}{ms}]$	
Kinematic viscosity	
$\frac{\mu}{\rho}$	(m <sup>2</sup> /s)
Temperature	(K)
Density	(kg/m <sup>3</sup> )
Shear stress	(N/m <sup>2</sup> )
Eddy diffusivity	(m <sup>2</sup> /s)
Eddy diffusivity of heat	(m <sup>2</sup> /s)
Eddy diffusivity of momentum	(m <sup>2</sup> /s)
Layer thickness	(m)

Subscripts

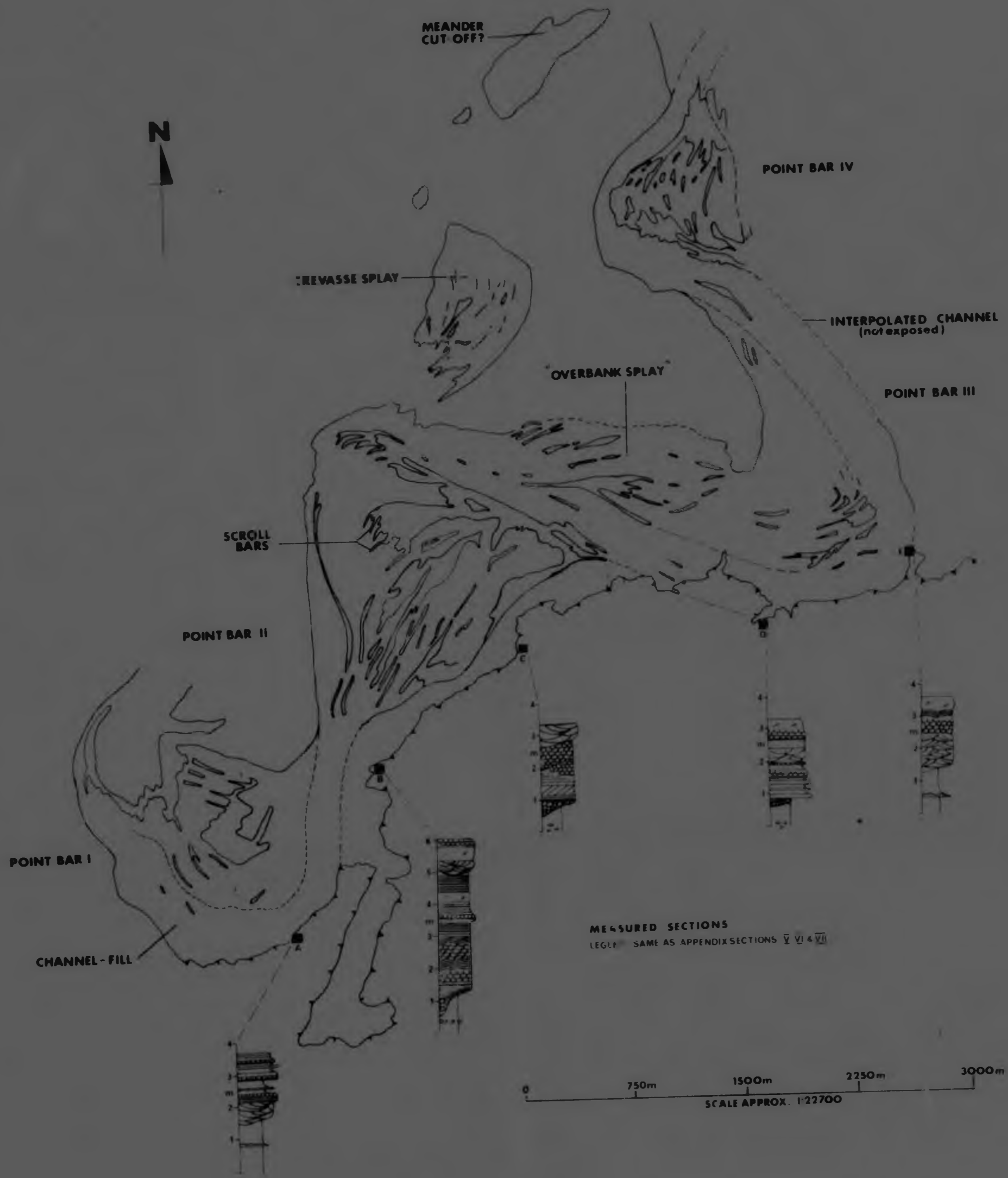
b	Bulk
am	Arithmetic mean
avg	Average
$C_c$	Centreline
w	Wall
i	Inlet, intermediate
ii	Outlet at zero heat flux;
$\alpha$	Constant properties
lm	Log mean
d	Based on diameter
x	Based on x distance
L	Based on tube length
f	Film
1,2, etc	Arbitrary basis
*	As infinite conditions are reached

11/11/11



APPENDIX III

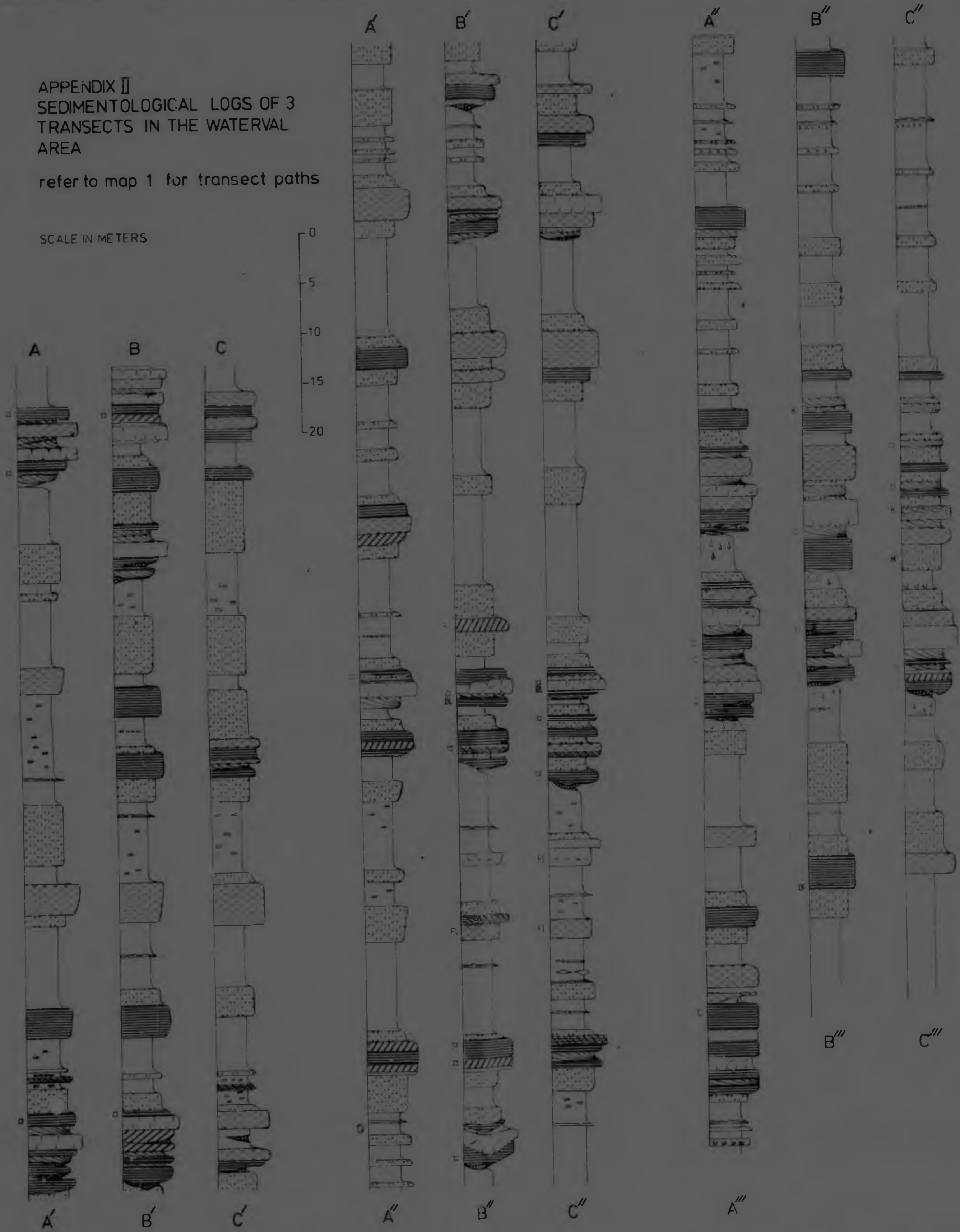
TYPE A BEDS : PLAN OF REIERSVLEI SANDSTONE



APPENDIX II  
 SEDIMENTOLOGICAL LOGS OF 3  
 TRANSECTS IN THE WATERVAL  
 AREA

refer to map 1 for transect paths

SCALE IN METERS



- TROUGH CROSS BEDDED SST.
- HORIZONTALLY BEDDED SST.
- MASSIVELY BEDDED SST.
- RIPPLE CROSS LAMINATED SST.
- MUDSTONE PEBBLY CONGLOMERATE







- CLIMBING RIPPLE LAMINATION
- PLANAR CROSS BEDDING
- INCIPIENT RIPPLE LAMINATION
- SLUMP STRUCTURES
- CONVOLUTE LAMINATION

- SEPTARIAN NODULE
- KIEFKER
- BLEACHING
- FLASER BEDDING
- PYRITES
- CALCAREOUS SILTSTONE
- BONE FRAGMENTS

- CALCAREOUS NODULE HORIZON
- PEBBLE SURFACE
- APPROXIMATE NODULAR LAYER
- CHERT LENS
- FOOT IMPRESSIONS



MAP 3: BIOZONE MAP OF THE WATERVAL - BERGVALLEI AREA

- LEGEND**
-  AULACEPHALODON / CISTECERPHALUS ASSEMBLAGE ZONE
  -  TROPIDOSTOMA / ENDOTHIIOON ASSEMBLAGE ZONE
  -  UPPER LOWER PRISTEROGNATHUS/DIICTODON ASSEMBLAGE ZONE
  -  DOLERITE
  -  SANDSTONE OUTCROP
  -  DYKE

0 750 1500 2250 3000 metres

Map 2

DETAILED BIOZONATION OF THE REGION SURROUNDING THE STUDY AREA

SCALE: 1:100 000







**MAP: THE GEOLOGY OF THE WATERVAL AREA**

**LEGEND**

- MAJOR SANDSTONE 1
- MAJOR SANDSTONE 2
- EXTENSIVE SANDSTONE OUTCROPS
- MINOR SANDSTONE OUTCROPS
- DOLERITE
- MUDROCK AND ALLUVIUM
- LINE OF TRANSECT
- DYKE

0 750 1500 2250 3000 metres

DEPARTMENT VAN MYNWESE  
DEPARTMENT OF MINES

GEOLOGIESE OPNAME  
GEOLOGICAL SURVEY

AFDELING/DIVISION

**MAP I: THE GEOLOGY OF THE WATERVAL AREA**

OPGESTEL DEUR  
DRAWN BY

TEKNISE KONTROLE  
TECHNICAL CONTROL

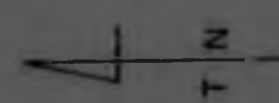
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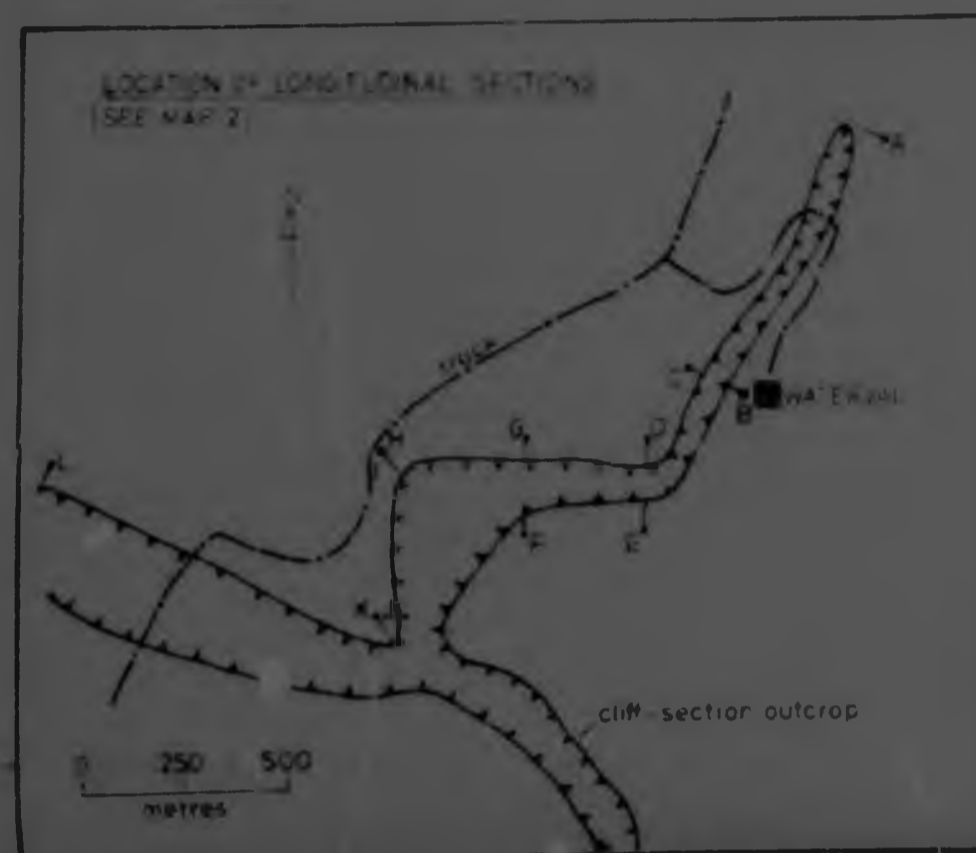
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LEGEND TO LONGITUDINAL SECTIONS

- SANDSTONE UNIT I
- SANDSTONE UNIT II
- SANDSTONE UNIT III
- ▨ EROSION AND REACTIVATION SURFACES



LEGEND TO COLUMNAR LOGS

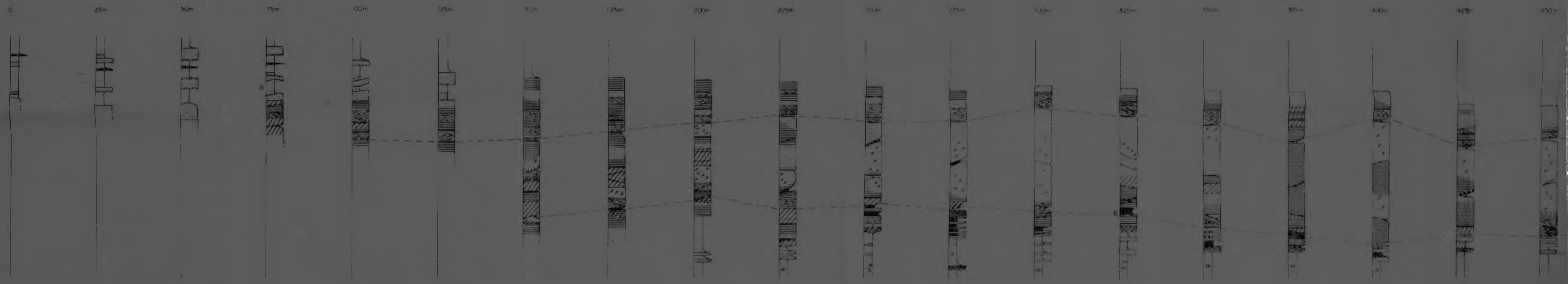
- |                               |  |
|-------------------------------|--|
| □ MUDSTONE                    | ▨ TROUGH CROSS-BEDDING                   |
| ▨ SILTSTONE                   | ▨ PLANAR CROSS-BEDDING                   |
| ▨ RIPPLE CROSS-LAMINATION     | ▨ MUDSTONE PEBBLE (CONGLOMERATE)         |
| ▨ CLIMBING RIPPLE LAMINATION  | ▨ CALCAREOUS NODULES AND SCORLICH STRIPE |
| ▨ INCIPENT RIPPLE LAMINATION  | ▨ PALEOSURFACE                           |
| ▨ WAVY OR PARALLEL LAMINATION | ▨ SOLE STRUCTURES                        |
| ▨ HORIZONTAL LAMINATION       | ▨ BONE FRAGMENTS                         |
| ▨ MASSIVE BEDDING             | ▨ SLIPPED MUDSTONE                       |
| ▨ FLAT BEDDING                | ▨ MANGROVES                              |

A

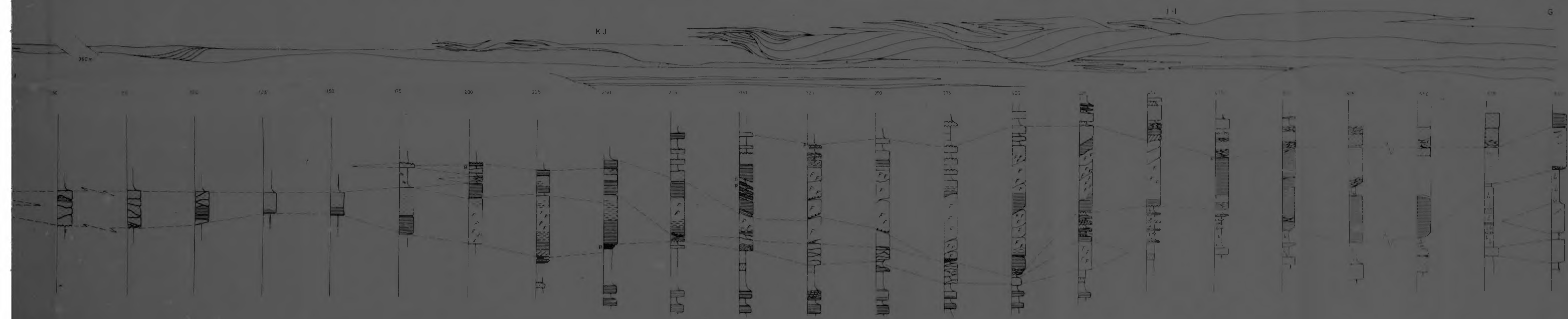
SCALE IN METRES IN 2000



HORIZONTAL SCALE IN METRES (1:500)







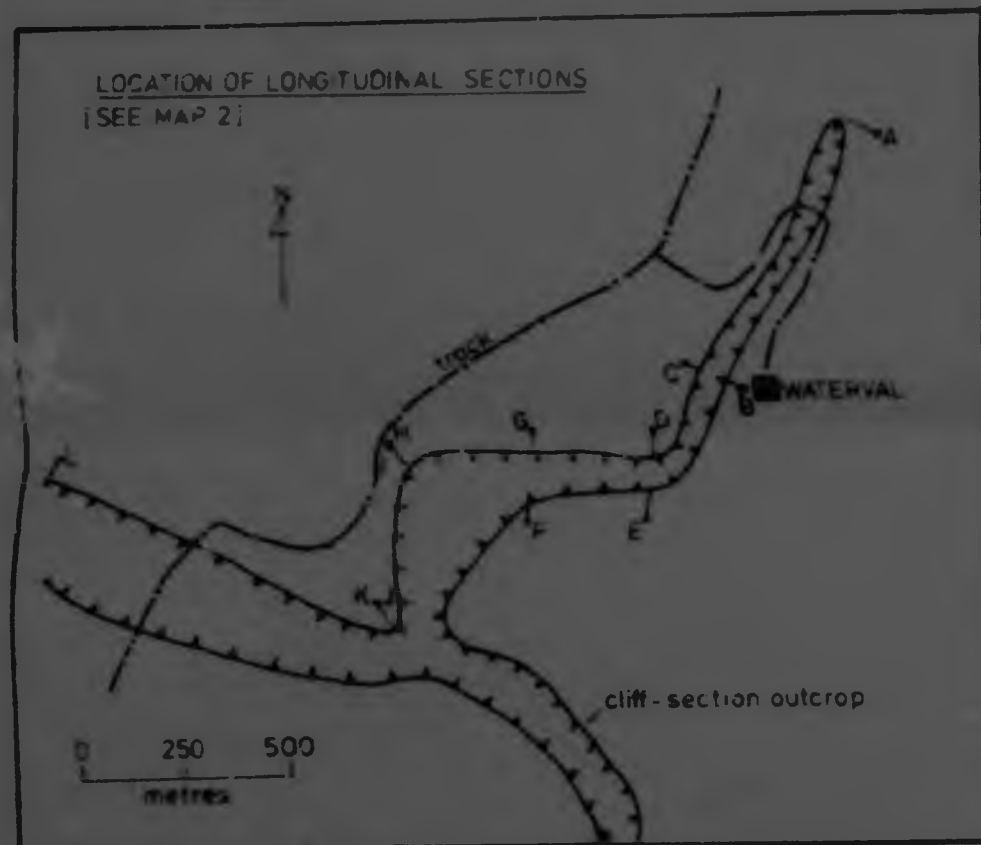
LONGITUDINAL AND COLUMNAR SECTIONS SHOWING



LONGITUDINAL AND COLUMNAR SECTIONS SHOWING SEDIMENTARY STRUCTURES OF A CLIFF EXPOSURE ON WATERVAL : SECTIONS G-H, I-J, K-L

LEGEND TO LONGITUDINAL SECTIONS

-  SANDSTONE UNIT I
-  SANDSTONE UNIT II
-  SANDSTONE UNIT III
-  EROSION AND REACTIVATION SURFACES

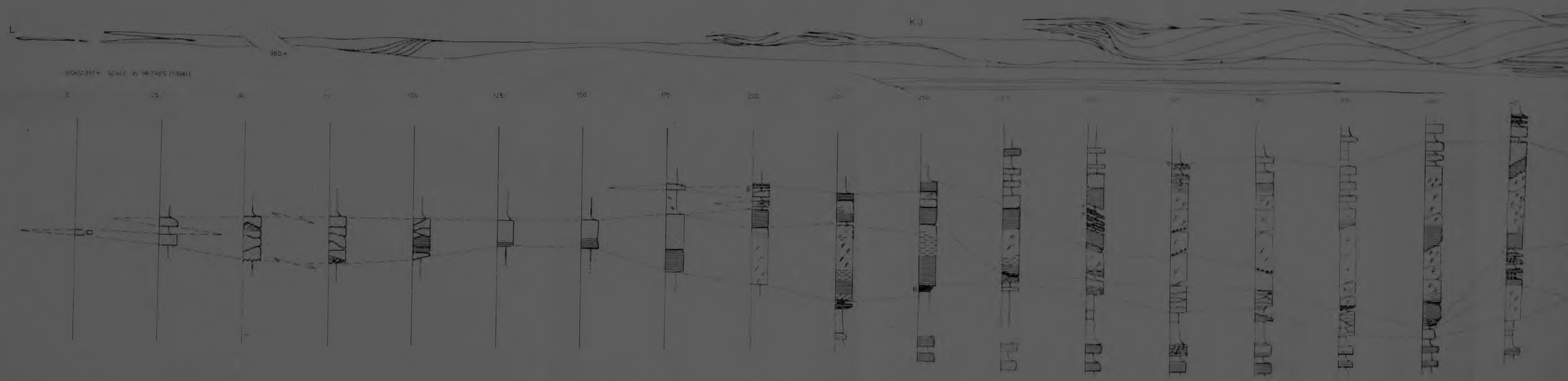


LEGEND TO COLUMNAR LOGS

- |   |   |
|---|---|
|  MUDSTONE                    |  TROUGH CROSS-BEDDING                  |
|  SILTSTONE                   |  PLANAR CROSS-BEDDING                  |
|  RIPPLE CROSS-LAMINATION     |  MUDSTONE PEBBLE CONGLOMERATE          |
|  CLIMBING RIPPLE LAMINATION  |  CALCAREOUS NODULES AND NODULAR LAYERS |
|  INCIDENT RIPPLE LAMINATION  |  PALAEO SURFACE                        |
|  WAVY OR PARALLEL LAMINATION |  SOLE STRUCTURES                       |
|  HORIZONTAL LAMINATION       |  BONE FRAGMENTS                        |
|  MASSIVE BEDDING             |  SLUMPED MUDSTONE                      |
|  FLAT BEDDING                |  PLANOLITES                            |





SCALE IN METRES BY 2000

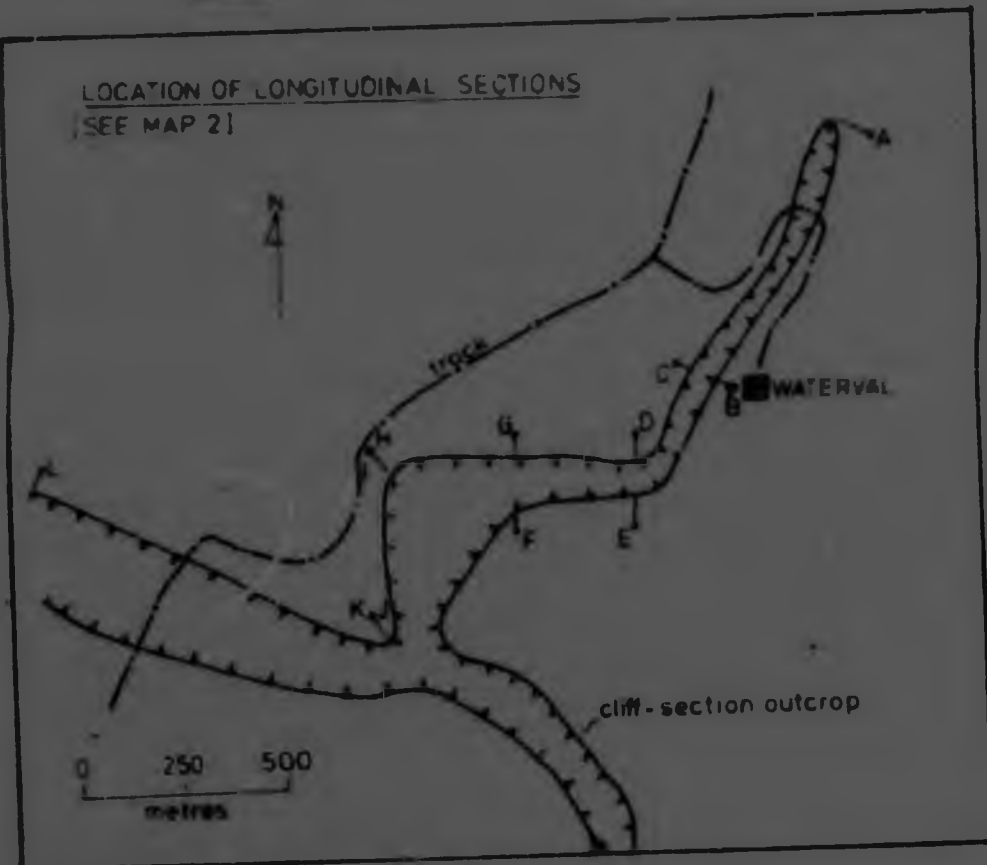
SCALE IN METRES BY 100





















LONGITUDINAL AND COLUMNAR SECTIONS SHOWING SEDIMENTARY STRUCTURES OF A CLIFF EXPOSURE ON WATERFALL : SECTIONS G-H, I-J, K-L

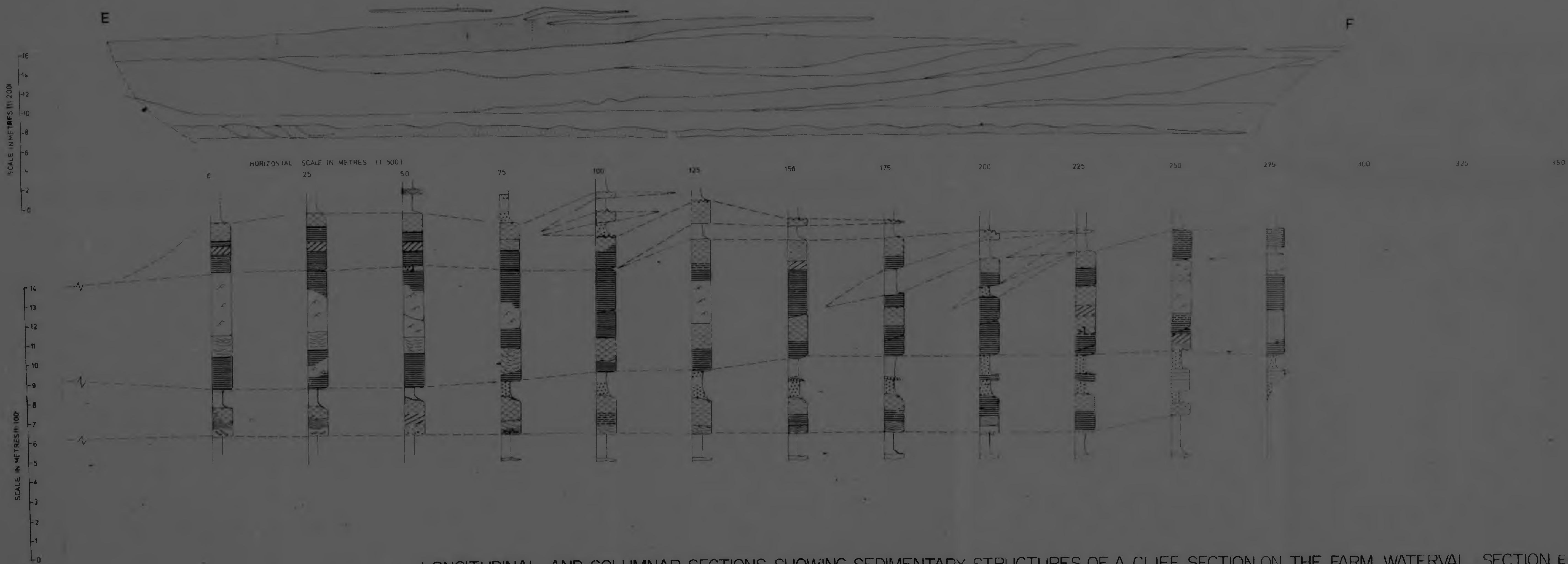
LEGEND TO LONGITUDINAL SECTIONS

-  SANDSTONE UNIT I
-  SANDSTONE UNIT II
-  SANDSTONE UNIT III
-  EROSION AND REACTIVATION SURFACES



LEGEND TO COLUMNAR LOGS

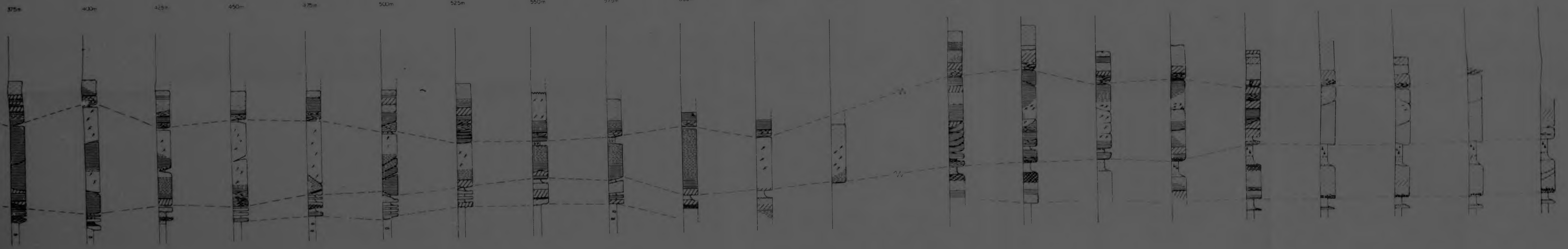
- |   |   |
|---|---|
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|  SILTSTONE                   |  PLANAR CROSS-BEDDING                  |
|  RIPPLE CROSS-LAMINATION     |  MUDSTONE PEBBLE CONGLOMERATE          |
|  CLIMBING RIPPLE LAMINATION  |  CALCAREOUS NODULES AND NODULAR LAYERS |
|  INCIPENT RIPPLE LAMINATION  |  PALAEOURFACE                          |
|  WAVY OR PARALLEL LAMINATION |  SCL STRUCTURES                        |
|  HORIZONTAL LAMINATION       |  BONE FRAGMENTS                        |
|  MASSIVE BEDDING             |  SLUMPED MUDSTONE                      |
|  FLAT BEDDING                |  PLANOLITES                            |



LONGITUDINAL AND COLUMNAR SECTIONS SHOWING SEDIMENTARY STRUCTURES OF A CLIFF SECTION ON THE FARM WATerval SECTION E-F

B C

D



PROFILE AND COLUMNAR SECTIONS SHOWING SEDIMENTARY STRUCTURES OF A CLIFF EXPOSURE ON THE FARM WATERVAL SECTION ABCD

**Author** Rogers D G

**Name of thesis** Experimental heat transfer coefficients for the cooling of oil in horizontal internal forced convective Transitional flow 1981

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